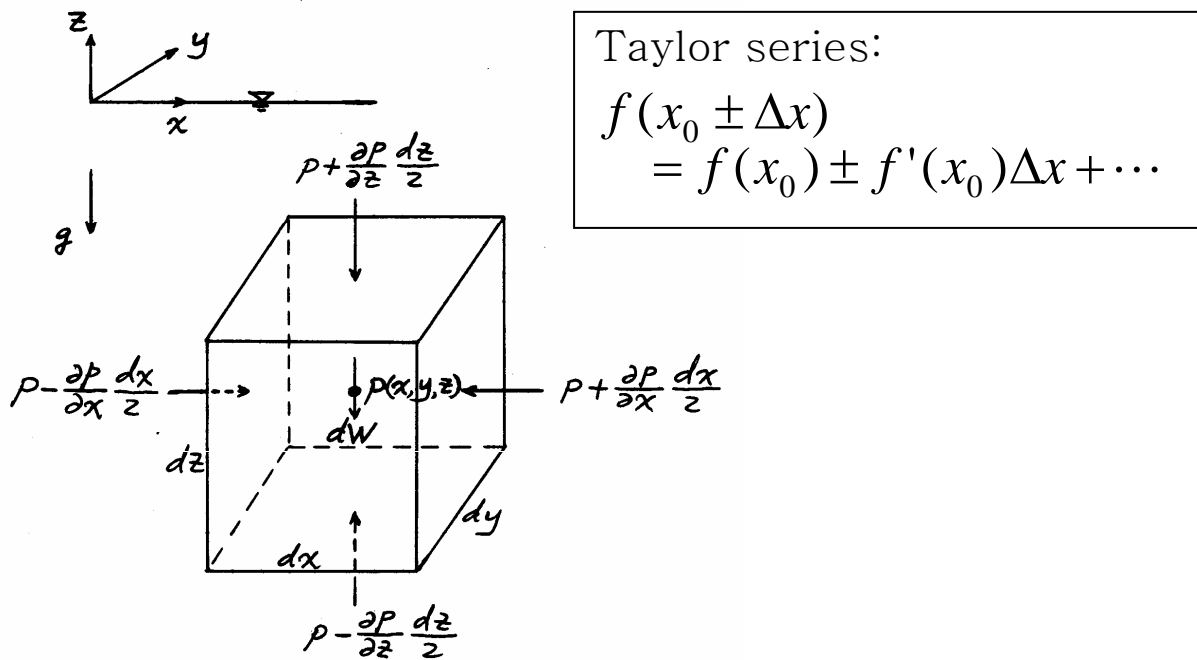


## Chap.2 FLUID STATICS

Fluid at rest  $\rightarrow$  No shear stress  $\rightarrow$  Pressure only

### 2.1 Pressure Variation with Elevation (relation between $p$ , $\rho$ , and $z$ )



Force balance:

$$\sum F_x = 0 = \left( p - \frac{\partial p}{\partial x} \frac{dx}{2} \right) dydz - \left( p + \frac{\partial p}{\partial x} \frac{dx}{2} \right) dydz$$

$$\sum F_y = 0 = \left( p - \frac{\partial p}{\partial y} \frac{dy}{2} \right) dx dz - \left( p + \frac{\partial p}{\partial y} \frac{dy}{2} \right) dx dz$$

$$\sum F_z = 0 = \left( p - \frac{\partial p}{\partial z} \frac{dz}{2} \right) dx dy - \left( p + \frac{\partial p}{\partial z} \frac{dz}{2} \right) dx dy - dW$$

$$dW = \rho g dx dy dz$$

$$\left. \begin{aligned}
 \sum F_x = 0 &\rightarrow \frac{\partial p}{\partial x} = 0 \rightarrow p = C_1(y, z) \\
 \sum F_y = 0 &\rightarrow \frac{\partial p}{\partial y} = 0 \rightarrow p = C_2(x, z)
 \end{aligned} \right\} p = p(z)$$

$$\begin{aligned}
 \sum F_z = 0 &\rightarrow \frac{\partial p}{\partial z} + \rho g = 0 \\
 &\rightarrow \frac{\partial p}{\partial z} = -\rho g \\
 &\rightarrow p = -\rho g z + C_3(x, y) \rightarrow C_3 = \text{const.}
 \end{aligned}$$

B.C.:  $p = 0$  at  $z = 0 \rightarrow C_3 = 0$

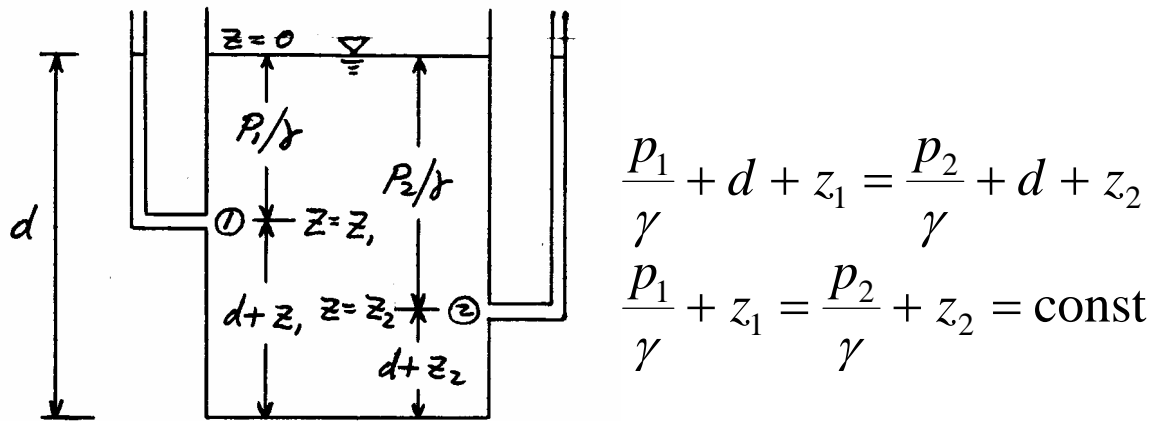
$$\boxed{p = -\rho g z}$$



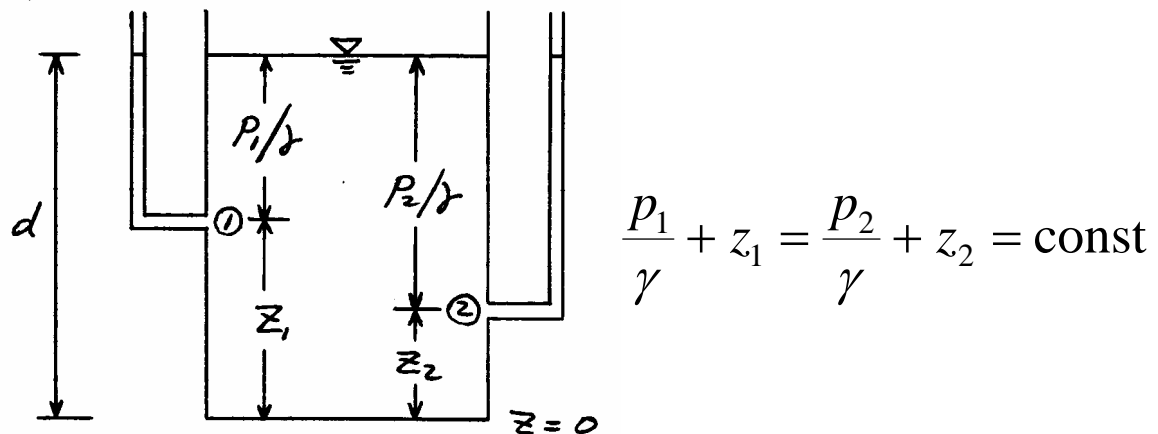
$$z = -h \quad p = -\rho g(-h) = \rho g h = \gamma h$$

Pressure varies linearly from 0 at surface to  $\gamma h$  at depth  $h$ .

Pressure head (壓力水頭):  $h = \frac{p}{\rho g} = \frac{p}{\gamma}$



$\updownarrow$   $z_1, z_2$  different



$$\frac{p}{\gamma} + z = \text{const}$$

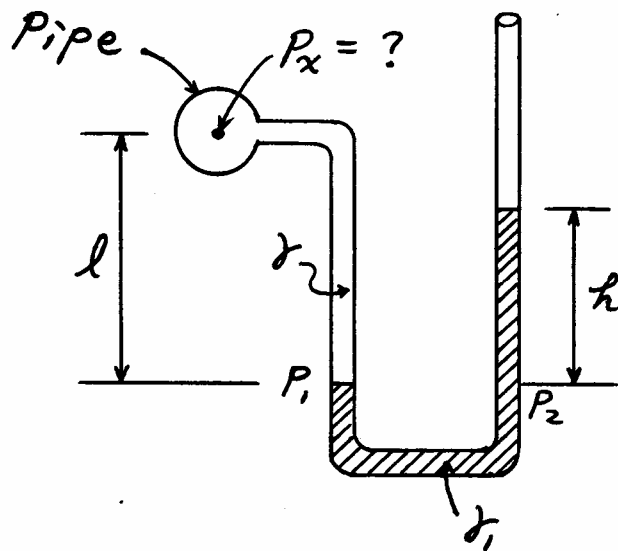
이 식은  $z$ 의 원점을 어디에 잡든지 성립.

$z = \text{elevation head (位置水頭)}$

## 2.2 Absolute and Gauge Pressure

- Absolute pressure (絕對壓力):  
진공상태를 기준으로(“0”으로) 하여 측정한 압력  
예: 대기압 = 101.3 kPa
- Gauge pressure (計器壓力):  
대기압을 기준으로(“0”으로) 하여 측정한 압력  
예: 타이어 압력 = 35 psi
- Gauge pressure = Absolute – Atmospheric
- Note: 공학적으로 사용되는 pressure는  
통상 gauge pressure를 의미함.

## 2.3 Manometer (液柱壓力計)



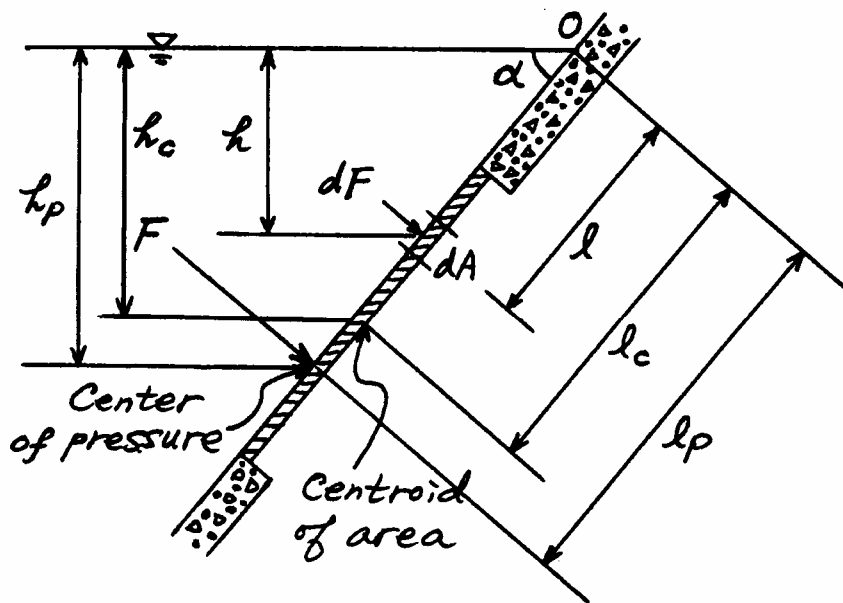
$$P_1 = P_2$$

$$\left. \begin{array}{l} P_1 = P_x + \gamma_2 l \\ P_2 = \gamma_1 h \end{array} \right\} \rightarrow P_x = \gamma_1 h - \gamma_2 l$$

Read text for other types of manometers.

## 2.4 Pressure Forces on Plane Surfaces

- { Magnitude ( $F$ )?
- { Direction (Normal to surface)
- { Point of action?



$$dF = p dA = \gamma h dA = \gamma l \sin \alpha dA$$

$$F = \int dF = \gamma \sin \alpha \int l dA = \gamma \sin \alpha l_c A = \gamma h_c A$$

where  $l_c = \frac{1}{A} \int l dA$  : Centroid of area

(depending on the shape  
of the area)

$$\boxed{F = \gamma h_c A}$$

Total force ( $F$ ) = pressure at centroid of  
area ( $\gamma h_c$ )  $\times$  Area ( $A$ )

- Point of action ( $l_p = ?$ )

Consider moment about  $O$ :

$$l_p F = \int l dF = \gamma \sin \alpha \underbrace{\int l^2 dA}_{I_o}$$

$$dF = \gamma l \sin \alpha dA$$

2nd moment about  $O$

$$= I_c + l_c^2 A$$

moment of inertia (Appendix 3)  
= 2nd moment about centroid  
of area

$$l_p F = \gamma \sin \alpha (I_c + l_c^2 A)$$

$$l_p \gamma \sin \alpha l_c A = \gamma \sin \alpha (I_c + l_c^2 A)$$

$$\boxed{l_p = \frac{I_c}{l_c A} + l_c}$$

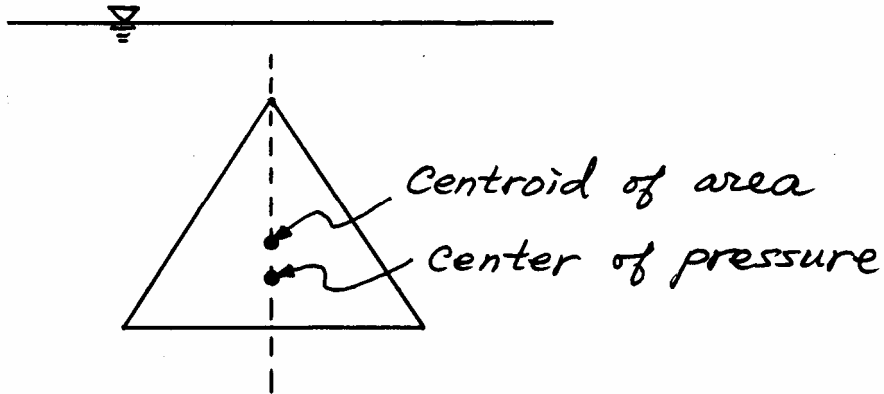
$$\therefore l_c, I_c, A \rightarrow F, l_p$$

$$I_o = \int l^2 dA$$

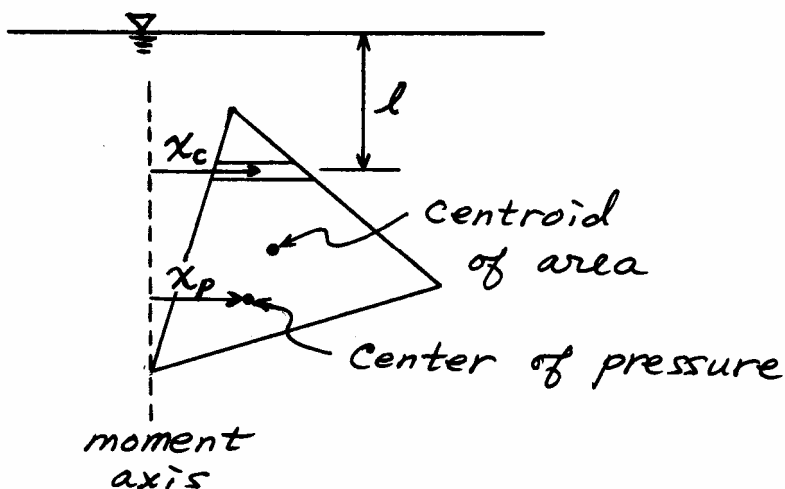
$$I_c = \int (l_c - l)^2 dA = l_c^2 \underbrace{\int dA}_{=A} - 2l_c \underbrace{\int l dA}_{=l_c A} + \underbrace{\int l^2 dA}_{=I_o}$$

$$\therefore I_o = I_c + l_c^2 A$$

Symmetric:



Non-symmetric:



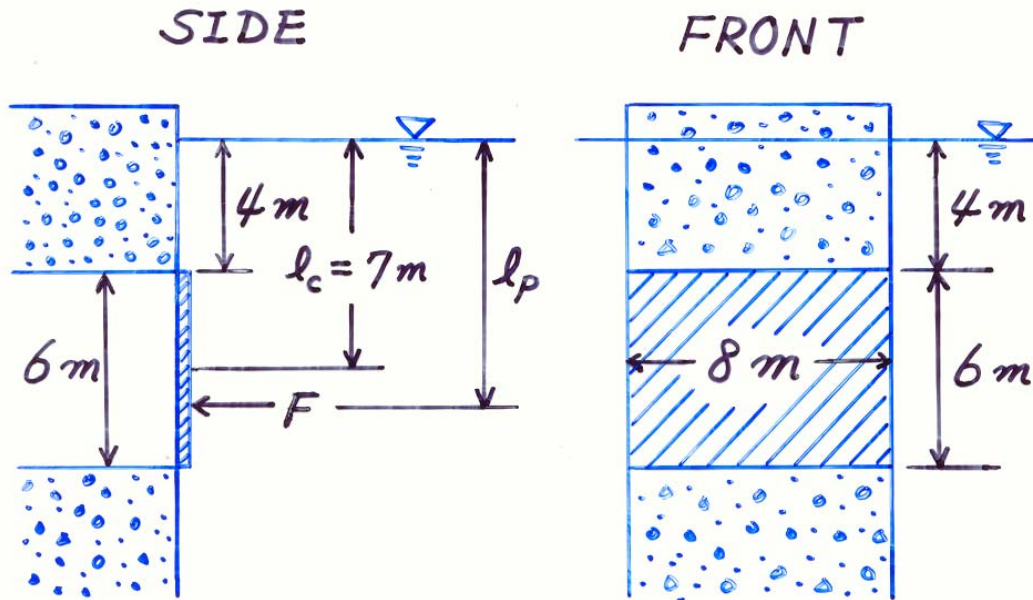
$F$  and  $l_p$  can be calculated as previous.

Lateral position?

$$x_p = \frac{\gamma \sin \alpha \int x_c l dA}{F} \quad (\text{Read text and IP2.9})$$



## Example



Vertical gate:  $\alpha = 90^\circ \rightarrow \sin \alpha = 1.0$

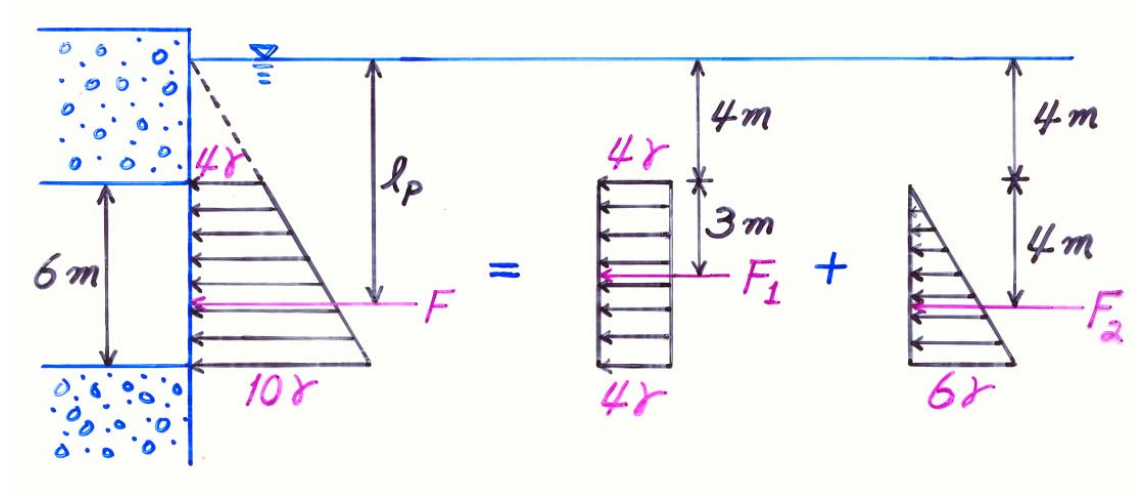
1) Direct integration

$$F = \gamma \sin \alpha \int l dA = \gamma \int_4^{10} l \cdot 8 dl = 8\gamma \frac{l^2}{2} \Big|_4^{10} = 336\gamma$$

$$\begin{aligned} l_p F &= \gamma \sin \alpha \int l^2 dA = \gamma \int_4^{10} l^2 \cdot 8 dl = 8\gamma \frac{l^3}{3} \Big|_4^{10} \\ &= \frac{8}{3} (1000 - 64)\gamma = \frac{8 \times 936}{3} \gamma \end{aligned}$$

$$l_p = \frac{\frac{8 \times 936}{3} \gamma}{336\gamma} = 7.43 \text{ m}$$

## 2) Pressure prism approach



$$\left. \begin{aligned} F_1 &= 4\gamma \times (6 \times 8) = 192\gamma \\ F_2 &= \frac{6\gamma \times 6}{2} \times 8 = 144\gamma \end{aligned} \right\} \rightarrow F = F_1 + F_2 = 336\gamma$$

$$l_p F = 7F_1 + 8F_2 \rightarrow l_p = \frac{7F_1 + 8F_2}{F} = 7.43 \text{ m}$$

## 3) Formula method

$$l_c = h_c = 4 + 3 = 7 \text{ m}$$

$$F = \gamma h_c A = \gamma \times 7 \times (6 \times 8) = 336\gamma$$

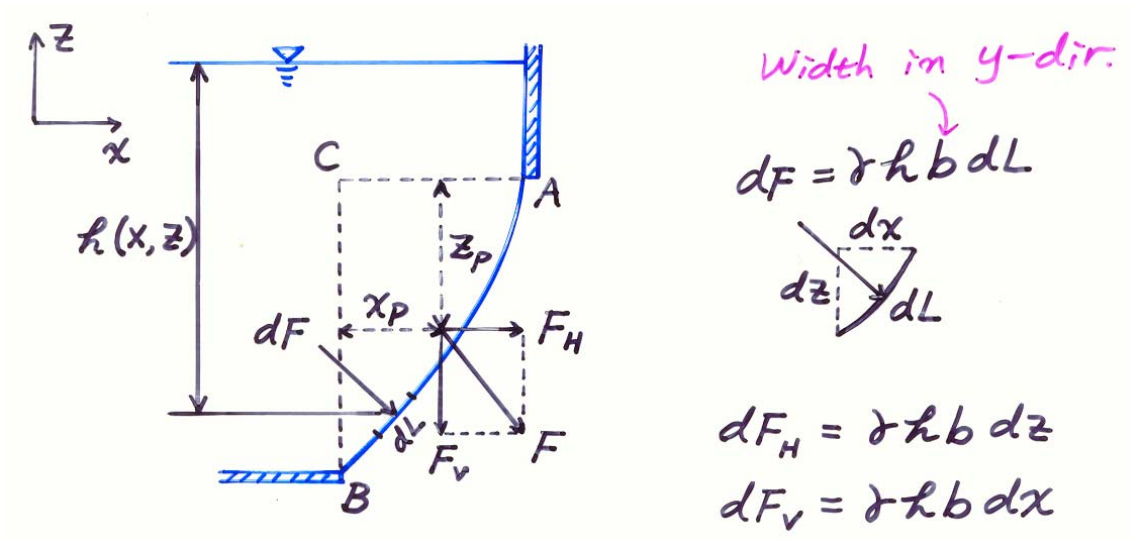
$$I_c = \frac{8 \times 6^3}{12} = 144$$

$$l_p = \frac{I_c}{l_c A} + l_c = \frac{144}{7 \times (6 \times 8)} + 7 = 7.43 \text{ m}$$

## 2.5 Pressure Forces on Curved Surfaces

Direction is not constant  $\rightarrow$  difficulty!

1) Direct integration method



▶ Curved surface =  $h(x, z)$   $\leftarrow$  No  $y$  variation

▶ Find  $F_H$ ,  $F_V$  and  $x_p$ ,  $z_p$  separately

▶ Vector sum of  $F_H$  and  $F_V \rightarrow F$

$$F_H = \int dF_H = \int \gamma h b dz, \quad F_V = \int dF_V = \int \gamma h b dx$$

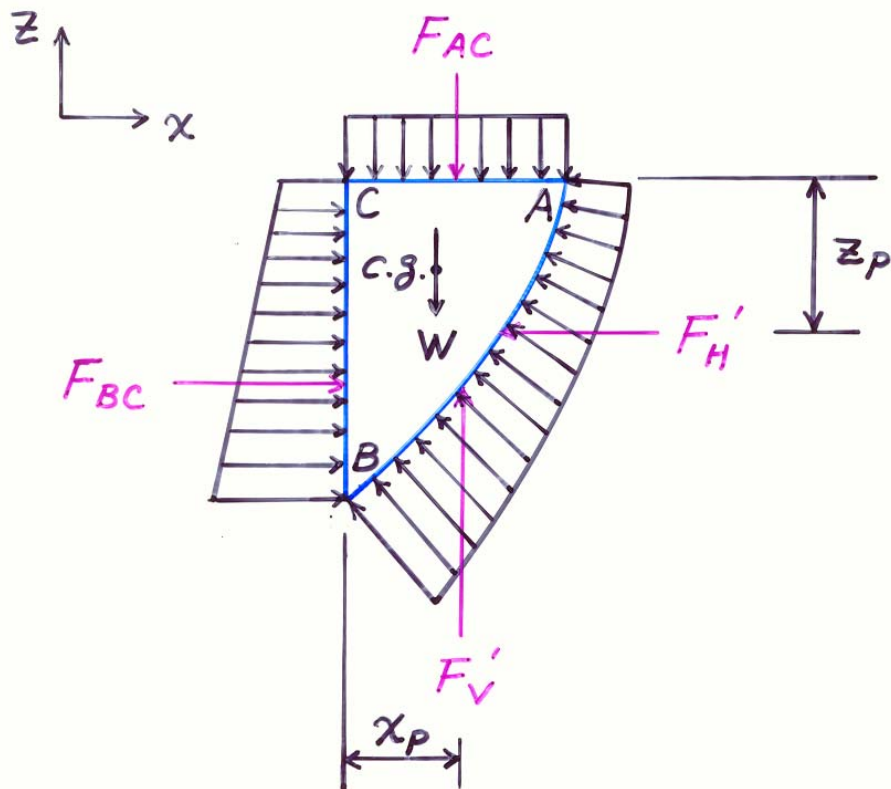
▶ Consider moment about  $C$ :

$$F_H z_p = \int z dF_H = \int z \gamma h b dz \rightarrow z_p$$

$$F_V x_p = \int x dF_V = \int x \gamma h b dx \rightarrow x_p$$

## 2) Basic mechanics method

- ▶ Isolate a fluid mass → free body
- ▶ Force balance on the free body



$$\sum F_x = F_{BC} - F_H' = 0 \rightarrow F_H' = F_{BC}$$

$$\sum F_z = F_V' - F_{AC} - W = 0 \rightarrow F_V' = F_{AC} + W$$

$F_{AC}, F_{BC}$  (크기 및 작용점) ← Plane surface

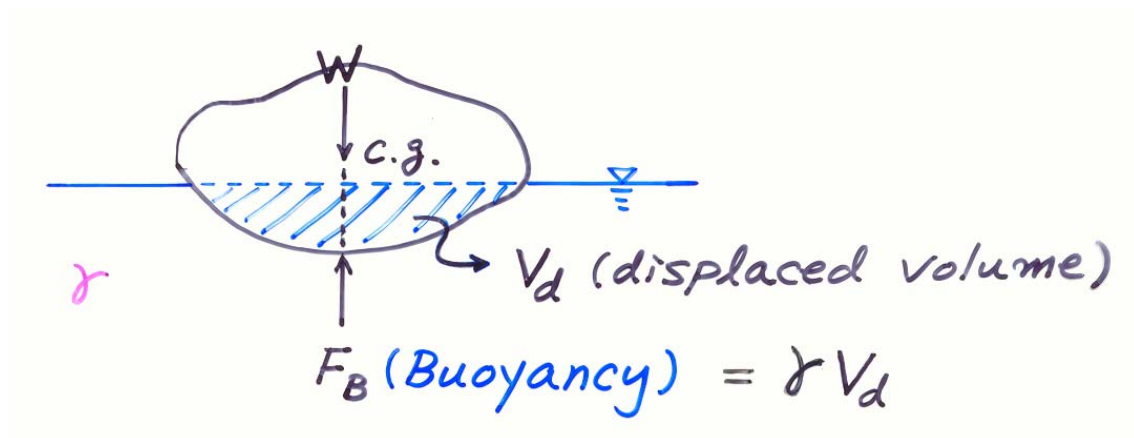
$W$  at c.g. of the free body

Horizontal moment about  $C \rightarrow z_p$

Vertical moment about  $C \rightarrow x_p$

## 2.6 Buoyancy and Stability of Floating Bodies

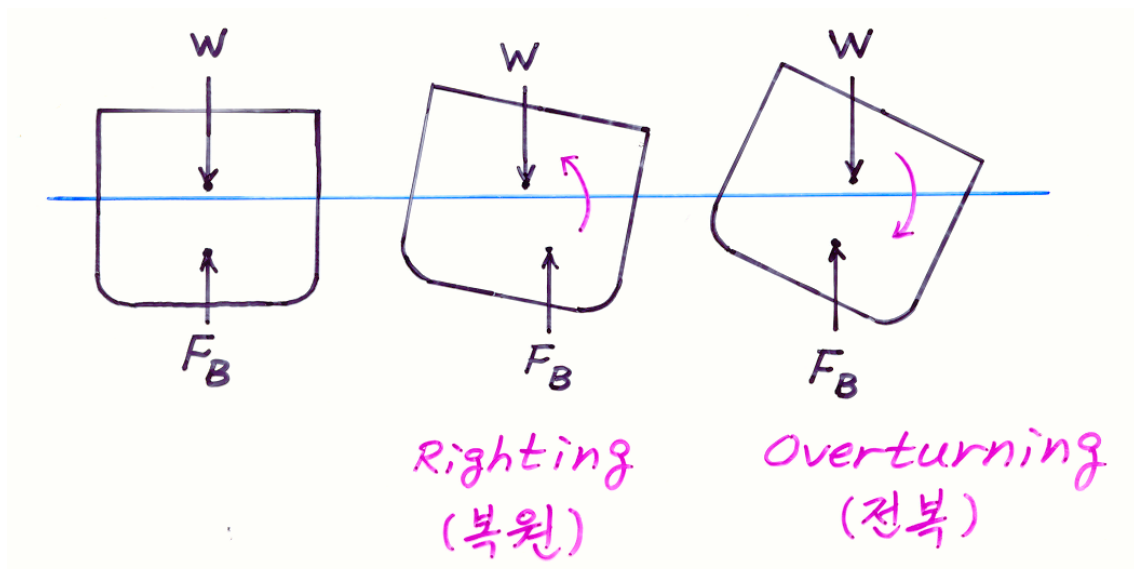
- Archimedes의 원리



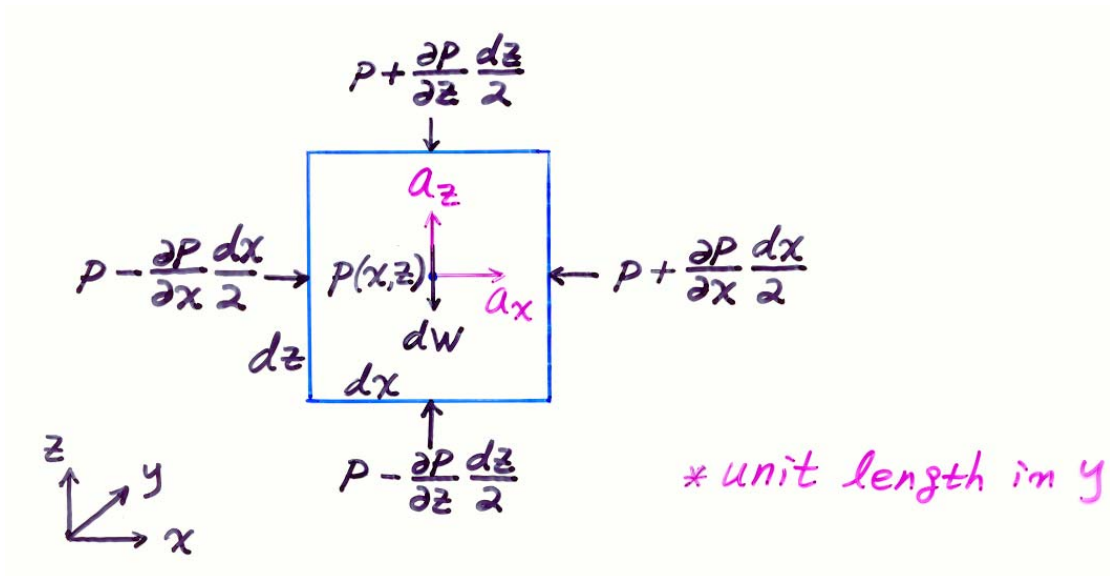
- Stability of floating bodies

$W$  at c.g. of the whole body

$F_B$  at c.g. of  $V_d$



## 2.7 Fluid Masses Subjected to Acceleration



► Newton's 2nd law:

$$F_x = \left( p - \frac{\partial p}{\partial x} \frac{dx}{2} \right) dz - \left( p + \frac{\partial p}{\partial x} \frac{dx}{2} \right) dz = \rho dx dz a_x$$

$$\boxed{-\frac{\partial p}{\partial x} = \rho a_x}$$

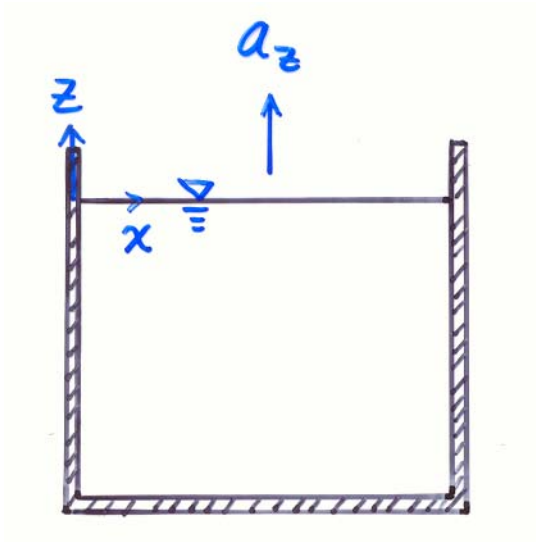
$$F_z = \left( p - \frac{\partial p}{\partial z} \frac{dz}{2} \right) dx - \left( p + \frac{\partial p}{\partial z} \frac{dz}{2} \right) dx - \rho g dx dz = \rho dx dz a_z$$

$$\boxed{-\frac{\partial p}{\partial z} = \rho(a_z + g)}$$

Line of constant pressure

$$dp = \frac{\partial p}{\partial x} dx + \frac{\partial p}{\partial z} dz = -\rho a_x dx - \rho(a_z + g) dz$$

$$dp = 0 \text{ (const. pressure)} \rightarrow \boxed{\frac{dz}{dx} = -\frac{a_x}{a_z + g}}$$



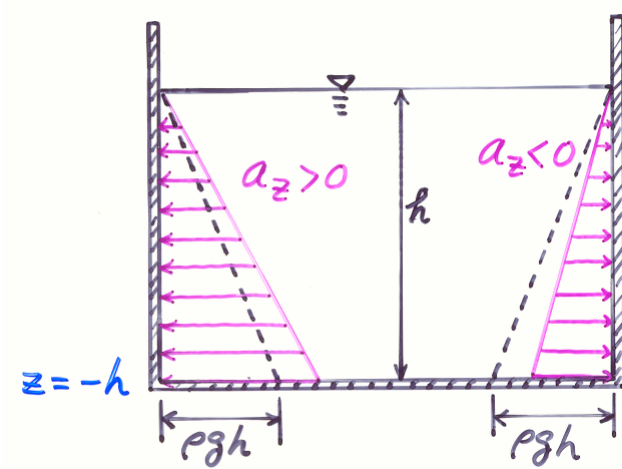
$$a_x = 0 \rightarrow \frac{\partial p}{\partial x} = 0$$

$$\frac{dp}{dz} = -\rho(a_z + g)$$

$$p = -\rho(a_z + g)z + C$$

$$\text{B.C. } p = 0 \text{ at } z = 0 \rightarrow C = 0$$

$$\therefore p = -\rho(a_z + g)z$$



If  $a_z = -g$

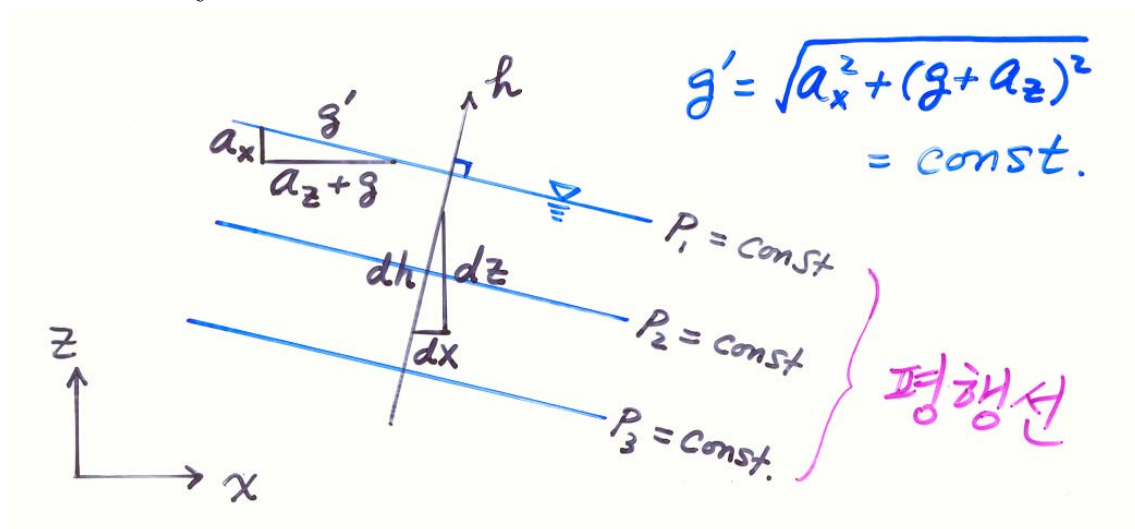
(자유낙하)

$p = 0$  everywhere

## Constant linear acceleration

Line of constant pressure:

$$\frac{dz}{dx} = -\frac{a_x}{a_z + g} = \text{const} \quad (\because a_x, a_z \text{ const.})$$



$$\frac{dx}{dh} = \frac{a_x}{g'} ; \quad \frac{dz}{dh} = \frac{a_z + g}{g'}$$

$$\begin{aligned} \frac{dp}{dh} &= \frac{\partial p}{\partial x} \frac{dx}{dh} + \frac{\partial p}{\partial z} \frac{dz}{dh} \\ &= -\rho a_x \frac{a_x}{g'} - \rho(a_z + g) \frac{a_z + g}{g'} \\ &= -\frac{\rho}{g'} (a_x^2 + (a_z + g)^2) \\ &= -\frac{\rho}{g'} (g')^2 \\ &= -\rho g' \end{aligned}$$

Pressure varies linearly with  $h$ , but is proportional to  $\rho g'$  instead of  $\rho g$ .