Lecture Note of Design Theories of Ship and Offshore Plant

Design Theories of Ship and Offshore Plant Part I. Ship Design

Ch. 6 Structural Design

Fall 2016

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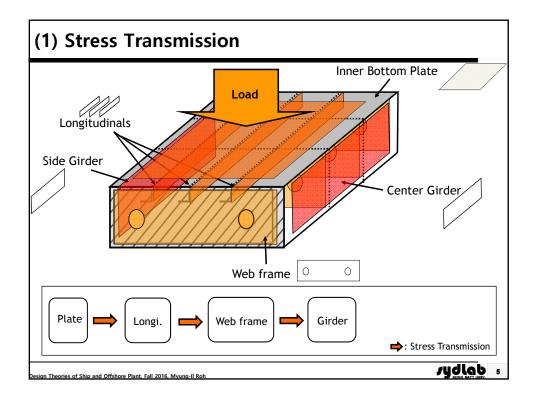
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6.1 Generals & Materials

- (1) Stress Transmission
- (2) Principal Dimensions
- (3) Criteria for the Selection of Plate Thickness, Grouping of Longitudinal Stiffener
- (4) Material Factors

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(2) Principal Dimensions

DNV Rules, Jan. 2004,Pt. 3 Ch. 1 Sec. 1 101

The following principal dimensions are used in accordance with DNV rule.

1) Rule length (L or L_s)

: Length of a ship used for rule scantling procedure

$$0.96 \cdot L_{WL} < L < 0.97 \cdot L_{WL}$$

- Distance on the summer load waterline (L_{WL}) from the fore side of the stem to the axis of the rudder stock
- Not to be taken less than 96%, and need not be taken greater than 97%, of the extreme length on the summer load waterline ($L_{W\!U}$)

Example of the calculation of rule length

L _{BP}	L _{WL}	0.96·L _{WL}	0.97·L _{WL}	L
250	261	250.56	253.17	250.56
250	258	247.68	250.26	250.00
250	255	244.80	247.35	247.35

2) Breadth

: Greatest moulded breadth in [m], measured at the summer load waterline

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(2) Principal Dimensions

DNV Rules, Jan. 2004,Pt. 3 Ch. 1 Sec. 1 101

3) Depth (D)

: <u>Moulded</u> depth defined as the vertical distance in [m] from baseline to moulded deck line at the uppermost continuous deck measured amidships

4) Draft (T)

: Mean moulded summer draft (scantling draft) in [m]

5) Block coefficient (C_R)

: To be calculated based on the rule length

$$C_{B} = \frac{\Delta}{1.025 \cdot L \cdot B \cdot T} \ \ \text{, (Δ: Moulded displacement in sea water on draft T)}$$

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(3) Criteria for the Selection of DNV Rules, Jan. 3 Plate Thickness, Grouping of Longitudinal Stiffener

DNV Rules, Jan. 2004,Pt. 3 Ch. 1 Sec. 1

1) Criteria for the selection of plate thickness

→ When selecting plate thickness, use the provided plate thickness.

(1) 0.5 mm interval

(2) Above 0.25 mm: 0.5 mm

(3) Below 0.25 mm: 0.0 mm

Ex) 15.75 mm → 16.0 mm 15.74 mm → 15.5 mm

2) Grouping of longitudinal stiffener

For the efficiency of productivity, each member is arranged by grouping longitudinal stiffeners. The grouping members should satisfy the following rule.

Average value but not to be taken less than 90% of the largest individual requirement (DNV).

Ex) The longitudinal stiffeners have design thickness of 100, 90, 80, 70, 60 mm. The average thickness is given by 80 mm×5. However, the average value is less than 100mm×90% = 90 mm of the largest individual requirement, 100 mm.

Therefore, the average value should be taken 90 mm×5.

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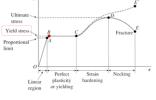
(4) Material Factors

1) DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec.2

 $^{2)}\,\text{James}$ M. Gere, Mechanics of Materials 7th Edition, Thomson, Chap.1, pp.15~26

• The material factor f_1 is included in the various formulae for scantlings and in expressions giving allowable stresses.¹⁾

Material Designation	Yield Stress (N/mm²)	$rac{\sigma}{\sigma_{\scriptscriptstyle NV-NS}}$	Material Factor (f_1)
NV-NS	235	235/235 = 1.00	1.00
NV-27	265	265/235 = 1.13	1.08
NV-32	315	315/235 = 1.34	1.28
NV-36	355	355/235 = 1.51	1.39
NV-40	390	390/235 = 1.65	1.47



* A: 'A' grade 'Normal Strength Steel'

* AH: 'A' grade 'High Tensile Steel'

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6.2 Global Hull Girder Strength (Longitudinal Strength)

- (1) Generals
- (2) Still Water Bending Moment (Ms)
- (3) Vertical Wave Bending Moment (Mw)
- (4) Section Modulus

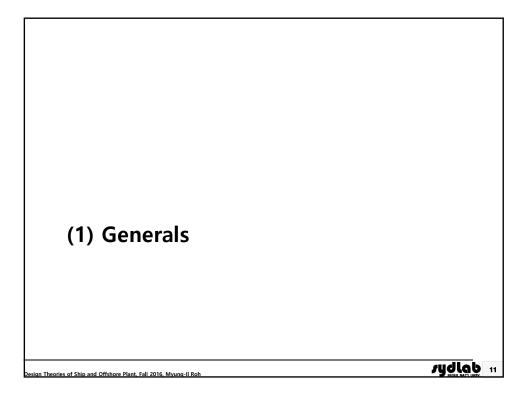
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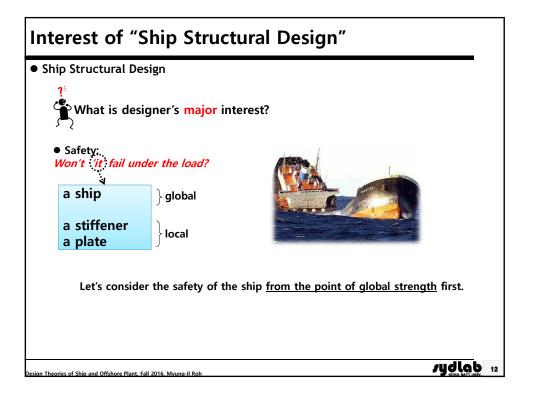
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^{*} NV-NS: Normal Strength Steel (Mild Steel)

^{*} NV-XX: High Tensile Steel

^{*} High tensile steel: A type of alloy steel that provides better mechanical properties or greater resistance to corrosion than carbon steel. They have a carbon content between 0.05-0.25% to retain formability and weldability, including up to 2.0% manganese, and other elements are added for strengthening purposes.





What are dominant forces acting on a ship in view of the longi. strength? w(x):weight of light ship, weight of cargo, and consumables hydrostatic force (buoyancy) on the submerged hull hydrodynamic force induced by the wave What is the direction of the dominant forces? The forces act in vertical (lateral) direction along the ship's length.

Longitudinal Strength

: Overall strength of ship's hull which resists the bending moment, shear force, and torsional moment acting on a hull girder.

Longitudinal strength loads

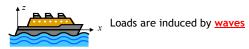
: Load concerning the overall strength of the ship's hull, such as the bending moment, shear force, and torsional moment acting on a hull girder

Static longitudinal loads



Loads are caused by <u>differences between weight and buoyancy</u> in longitudinal direction in the still water condition

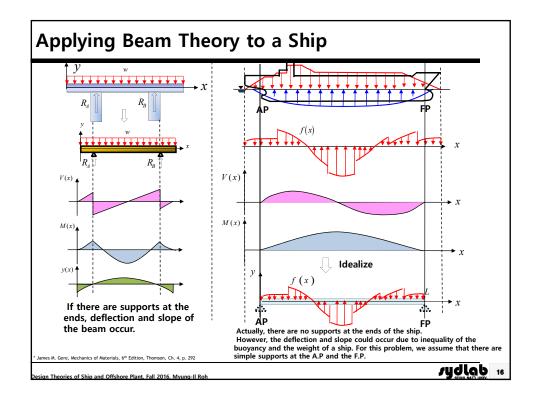
Hydrodynamic longitudinal loads

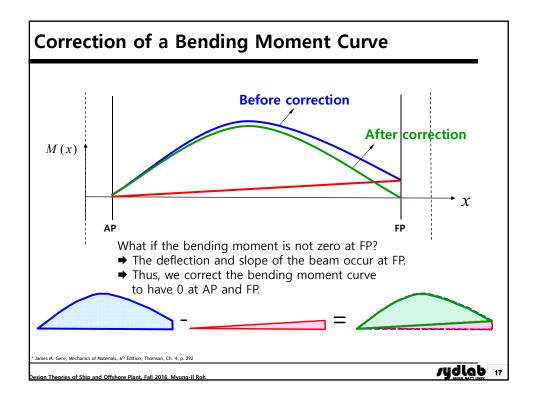


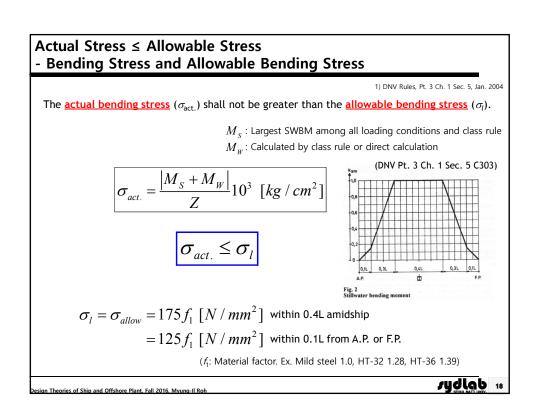
1) Okumoto, Y., Design of Ship Hull Structures, Springers, 2009, P.17

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How can we idealize a ship as a structural member? Structural member according to the types of loads Axially loaded bar: structural member which supports forces directed along the axis of the bar Bar in torsion: structural member which supports torques (or couples) having their moment about the longitudinal axis Beam: structural members subjected to lateral loads, that is, forces or moments perpendicular to the axis of the bar Since a ship has a slender shape and subject to lateral loads, it will behave like a beam from the point view of structural member.



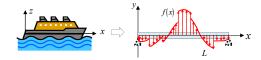




Criteria of Structural Design (1/2)

• Ship Structural Design

a ship



The <u>actual bending stress</u> ($\sigma_{act.}$) shall not be greater than the <u>allowable bending stress</u> (σ_l).

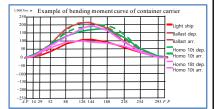
$$\sigma_{act.} \leq \sigma_l$$
 , $\sigma_{act.} = rac{M}{Z}$ =

 $\ensuremath{M_{\mathrm{S}}}\xspace$: Largest SWBM among all loading conditions and class rule ${\cal M}_{\it W} :$ calculated by class rule or direct calculation

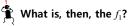
 σ_l : allowable stress

For instance, allowable bending stresses by DNV rule are given as follows:

$$\sigma_l = 175 f_1 \ [N/mm^2]$$
 within 0.4L amidship
$$= 125 f_1 \ [N/mm^2]$$
 within 0.1L from A.P. or F.P.



Actual bending moments at aft and forward area are smaller than that at the midship.



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Criteria of Structural Design (2/2)

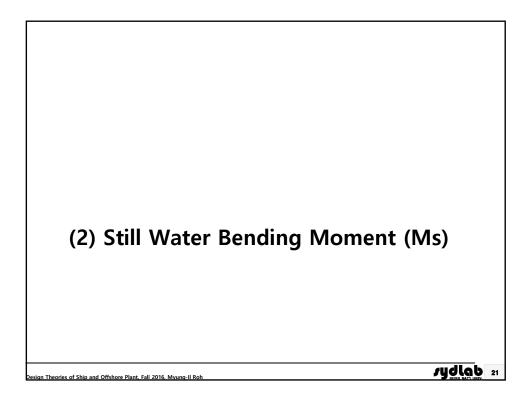
$$\sigma_{act.} \leq \sigma_l$$

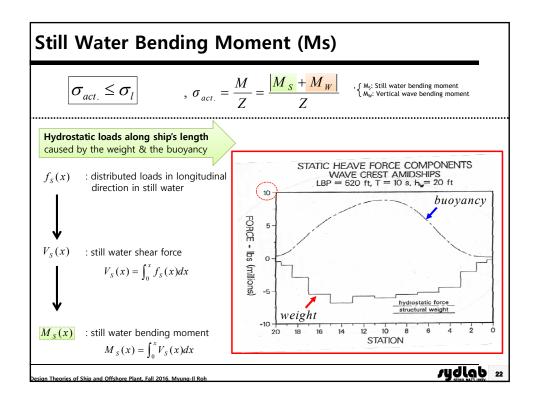
$$\sigma_{act.} \leq \sigma_l$$

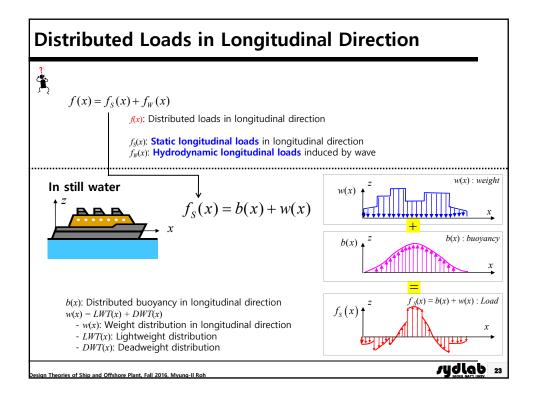
$$\sigma_{act.} = \frac{M}{Z} = \frac{|M_S + M_W|}{Z}$$

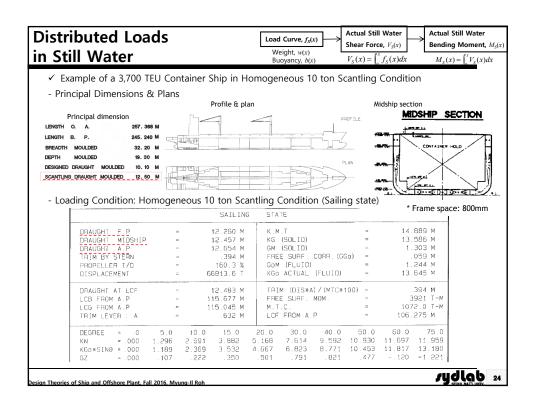
- (1) Still Water Bending Moment (Ms)
- (2) Vertical Wave Bending Moment (Mw)
- (3) Section Modulus (Z)

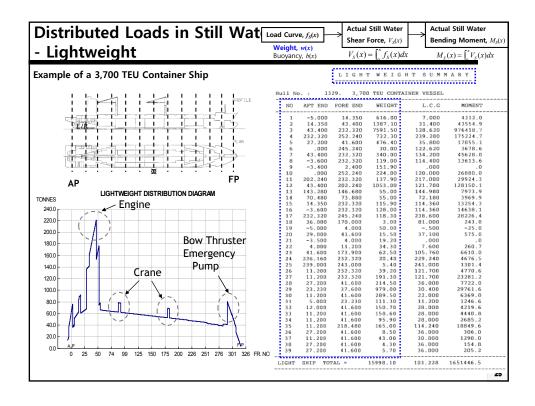
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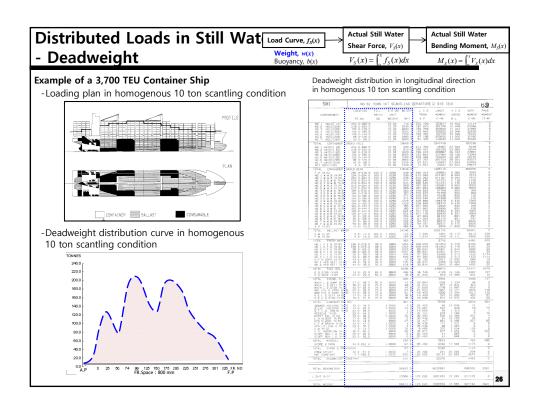


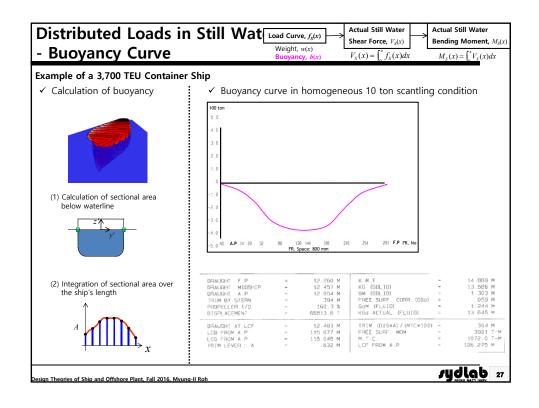


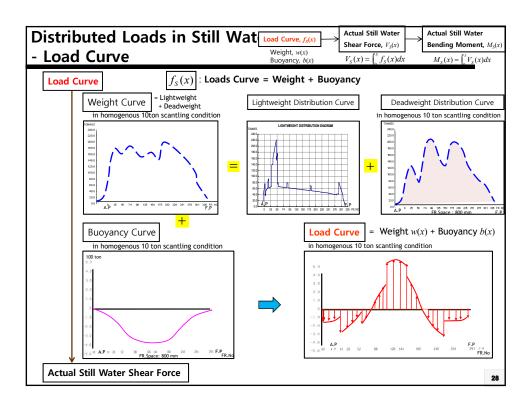


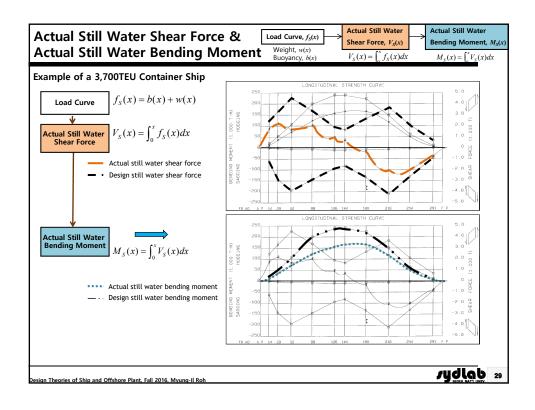












Rule Still Water Bending Moment by the Classification Rule

Recently, actual still water bending moment based on the loading conditions is used for still water bending moment, because the rule still water bending moment is only for the tanker.

• The design still water bending moments amidships are not to be taken less than

(DNV Pt. 3 Ch. 1 Sec. 5 A105)

$$M_{S} = M_{SO} [kNm]$$

$$M_{SO} = \underline{-0.065}C_{WU}L^2B(C_B + 0.7)$$
 [kNm] in sagging

 $= C_{WU} L^2 B (0.1225 - 0.015 C_{\scriptscriptstyle B}) \hspace{0.5cm} \text{[kNm] in hogging}$

 C_{WU} : Wave coefficient for unrestricted service

The still water bending moment shall not be less than the large of: the <u>largest actual still</u> water bending moment based on the <u>loading conditions</u> and the <u>rule still water bending moment</u>.

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Rule Still Water Shear Force by the Classification Rule

ullet The <u>design values of still water shear forces</u> along the length of the ship are normally not to be taken less than

(Dnv Pt.3 Ch.1 Sec. 5 B107)

$$Q_{S} = k_{sq} Q_{SO}(kN)$$

$$Q_{SO} = 5 \frac{M_{SO}}{L}(kN)$$

 k_{sq} = 0 at A.P. and F.P. = 1.0 between 0.15L and 0.3L from A.P. = 0.8 between 0.4L and 0.6L from A.P.

= 1.0 between 0.7L and 0.85L from A.P.

 $M_{SO} = -0.065C_{WU}L^2B(C_B + 0.7)$ [kNm] in sagging = $C_{WU}L^2B(0.1225 - 0.015C_B)$ [kNm] in hogging

 C_{WU} : wave coefficient for unrestricted service

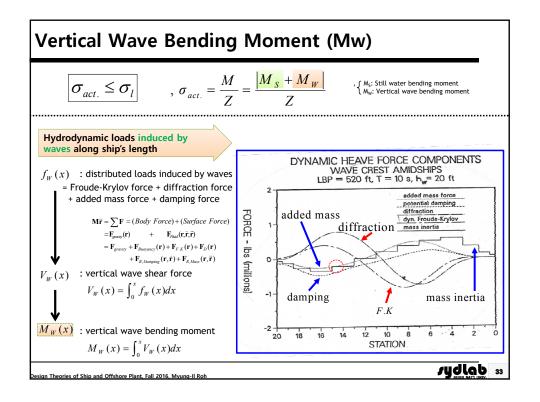
The still water shear force shall not be less than the large of: the <u>largest actual still water</u> <u>shear forces based on loading conditions</u> and the <u>rule still water shear force</u>.

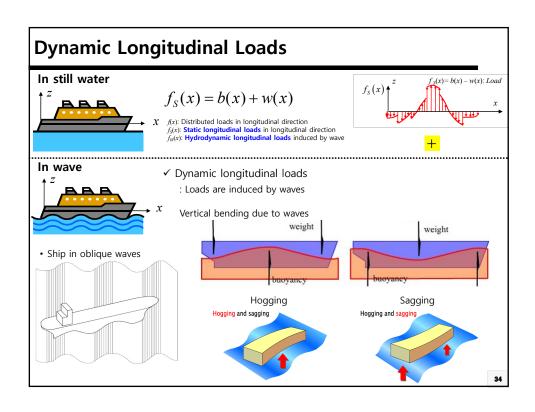
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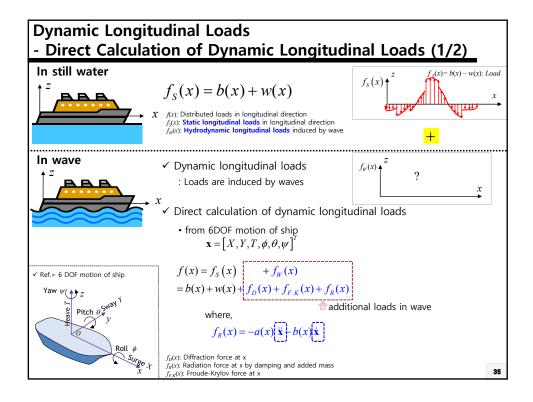
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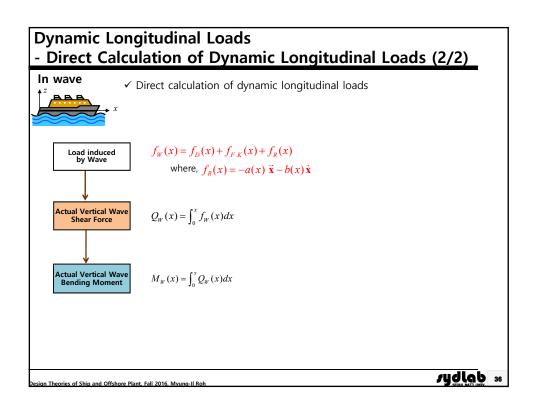
(3) Vertical Wave Bending Moment (Mw)

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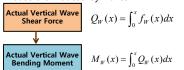




Rule Values of Vertical Wave Bending Moments

✓ Direct calculation of dynamic longitudinal loads





Recently, rule values of vertical wave moments are used,

because of the uncertainty of the direct calculation values of vertical wave bending moments.

The <u>rule vertical wave bending moments</u> amidships are given by:

$$\begin{split} M_W &= M_{WO} \quad \text{[kNm]} \\ M_{WO} &= -0.11 \alpha C_W L^2 B (C_B + 0.7) \quad \text{[kNm] in sagging} \\ &= 0.19 \alpha C_W L^2 B C_B \qquad \text{[kNm] in hogging} \\ &\alpha = 1.0 \text{ for seagoing condition} \end{split}$$

= 0.5 for harbor and sheltered water conditions (enclosed fiords, lakes, rivers) C_W : wave coefficient

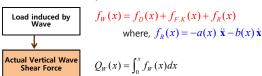
C_B: block coefficient, not be taken less than 0.6

<u>Direct calculation values of vertical wave bending moments can be used</u> for vertical <u>wave</u> bending moment instead of the rule values of vertical wave moments, if the value of the direct calculation is smaller than that of the rule value.

Rule Values of Vertical Wave Shear Forces

✓ Direct calculation of dynamic longitudinal loads

· Loads are induced by waves



The <u>rule values of vertical wave shear forces</u> along the length of the ship are given by: (DNV Pt.3 Ch.1 Sec.5 B203)

Positive shear force:
$$Q_{\mathit{WP}} = 0.3 \beta k_{\mathit{wqp}} C_{\mathit{W}} LB(C_{\mathit{B}} + 0.7)$$

Negative shear force: $Q_{\scriptscriptstyle WN} = -0.3 \beta k_{\scriptscriptstyle wgn} C_{\scriptscriptstyle W} LB(C_{\scriptscriptstyle B} + 0.7)$

 β : coefficient according to operating condition

 k_{wqp} , k_{wqn} : coefficients according to location in lengthwise C_W : wave coefficient

<u>Direct calculation values of vertical wave shear forces can be used</u> for vertical <u>wave shear</u> force instead of the rule values of vertical shear forces, if the value of the direct calculation is smaller than that of the rule value.

[Example] Rule Values of Still Water Bending Moments (Ms) and Vertical Wave Bending Moment (Mw)

Calculate L_{SCAND} and vertical wave bending moment at amidships (0.5L) of a ship in hogging condition for sea going condition.

Dimension: $L_{OA} = 332.0 \, m$, $L_{BP} = 317.2 \, m$, $L_{EXT} = 322.85 \, m$, $B = 43.2 \, m$, $T_s = 14.5 \, m$ Δ (Displacement (ton) at T_s) = 140,960 ton

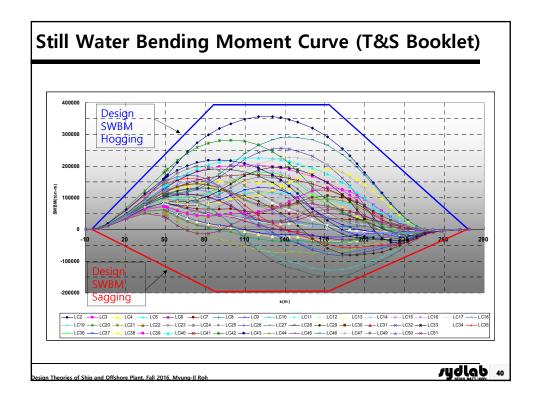
(Sol.)
$$L_s = 0.97 \times L_{EXT} = 0.97 \times 322.85 = 313.16$$
 $C_{B,SCANT} = \Delta / (1.025 \times L_s \times B \times T_s) = \frac{140,906}{1.025 \times 313.16 \times 43.2 \times 14.5} = 0.701$ $M_{SO} = -0.065C_{WU} L^2 B(C_B + 0.7), \text{ (in sagging)}$ $C_W = 10.75, \text{ if } 300 \le L \le 350 \text{ (wave coefficient)}$ $M_{WW} = -0.11 \alpha C_W L^2 B(C_B + 0.7), \text{ (in sagging)}$ $C_W = 1.0 \text{ between 0.4L and 0.65 L from A.P(=0.0) and F.P}$

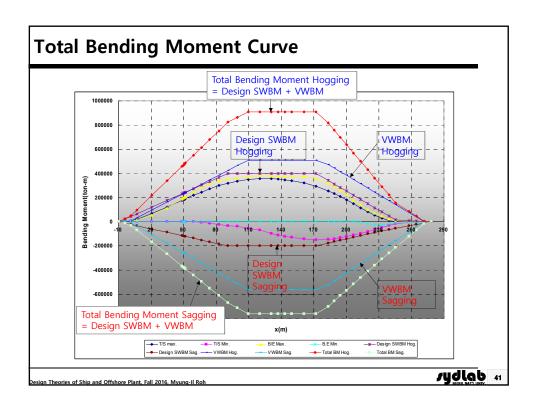
$$\begin{array}{c} \times 43.2 \times 14.5 \\ \hline \times 43.2 \times 14.5 \\ \hline \end{array} = 0.701 \\ = C_{WU}L^2B(0.1225 - 0.015C_B), \\ \underbrace{\left(M_w = M_{WO} \quad (kNm) \right]}_{M_{WO} = -0.11\alpha C_W L^2B(C_B + 0.7), \\ = 0.19\alpha C_W L^2BC_B, \\ \underbrace{\left(inhoggsing \right)}_{(inhoggsing)} \\ = 0.19\alpha C_W L^2BC_B, \\ \underbrace{\left(inhoggsing \right)}_{(inhoggsing)} \\ \end{array}$$

$$\begin{split} M_{WO} &= 0.19 \times \alpha \times C_W \times L^2 \times B \times C_{B,SCANT} \text{ (kNm)} \\ &= 0.19 \times 1.0 \times 10.75 \times 313.16^2 \times 43.2 \times 0.701 = 6,066,303 \text{ (kNm)} \\ \text{at } 0.5\text{L, } k_{nm} &= 1.0 \\ M_W &= 1.0 \times M_{WO} \end{split}$$

So, $M_W = 1.0 \times M_{WO} = 6,066,303 (kNm)$

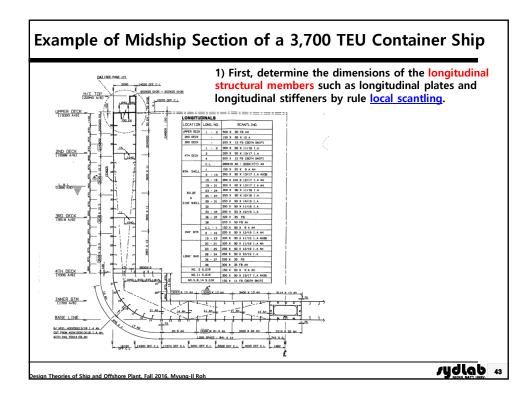
DSME, Ship Structural Design, 5-2 Load on Hull Structure, Example 4, 2005 gn Theories of Ship and Offshore Plant, Fall 2016, Myung-Il Rol





(4) Section Modulus

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Vertical Location of Neutral Axis about Baseline

2) Second, calculate the moment of sectional area about the base line.

$$\sum h_i \ A_i$$
 h_i : vertical center of structural member A_i : area of structural member

3) Vertical location of neutral axis from base line (\bar{h}) is, then, calculated by dividing the moment of area by the total sectional area.

$$\overline{h} = rac{\sum h_i \ A_i}{A}$$

 \overline{h} : vertical location of neutral axis A: total area

By definition, neutral axis pass through the centroid of the cross section.

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Midship Section Moment of Inertia about N.A

- The midship section moment of inertia about base line $({\it I}_{\it BL})$

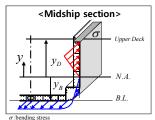
$$I_{B.L} = I_{N.A.} + A \ \overline{h}^2$$

- then calculate the midship section moment of inertia about neutral axis (I_{NA}) using I_{BL} .

$$I_{N.A.} = I_{B.L} - A \ \overline{h}^2$$

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Calculation of Section Modulus and Actual Stress at Deck and Bottom



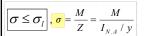
Section modulus

$$Z_D = \frac{I_{N.A.}}{y_D}, \quad Z_B = \frac{I_{N.A.}}{y_B}$$

Calculation of Actual Stress at Deck and Bottom

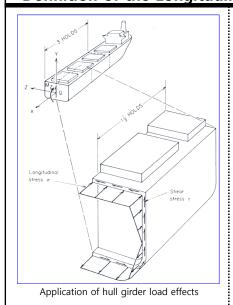
$$\sigma_{Deck} = \frac{M}{Z_D} = \frac{M}{I_{N.A} / y_D}$$

$$\sigma_{Bottom} = \frac{M}{Z_B} = \frac{M}{I_{N.A} / y_B}$$



ydlab #

Global Hull Girder Strength (Longitudinal Strength) - Definition of the Longitudinal Strength Members



X Example of Requirement for Longitudinal Structural Member

DNV Rules for Classification of Ships

Part 3 Chapter 1 HULL STRUCTUREALDESIGN SHIPS WITH LENGTH 100 METERS AND ABOVE

- C 300 Section modulus
 301 The requirements given in 302 and 303 will normally be satisfied when calculated for the midship section only, provided the following rules for tapering are complied with:
- a) Scantlings of all continuous longitudinal strength members shall be maintained within 0.4 L amidships.
- b) Scantlings outside 0.4 L amidships are gradually reduced to the local requirements at the ends, and the same material strength group is applied over the full length of the ship.

The section modulus at other positions along the length of the ship may have to be specially considered for ships with small block coefficient, high speed and large flare in the fore body or when considered necessary due to structural arrangement, see A106.

Hughes, Ship Structural Design, John Wiley & Sons, 1983

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The Minimum Required Midship Section Modulus and **Inertia Moment by DNV Rule**

DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec. 5

The midship section modulus about the transverse neutral axis shall not be less than: (Pt.3 Ch.1 Sec.5 C302)

$$Z_O = \frac{C_{WO}}{f_1} L^2 B(C_B + 0.7) \text{ [cm}^3]$$

 C_{WO} : wave coefficient

L	C_{WO}	
L < 300	$10.75 - [(300 - L)/100]^{3/2}$	
$300 \le L \le 350$		
L > 350	$10.75 - [(L - 350)/150]^{3/2}$	

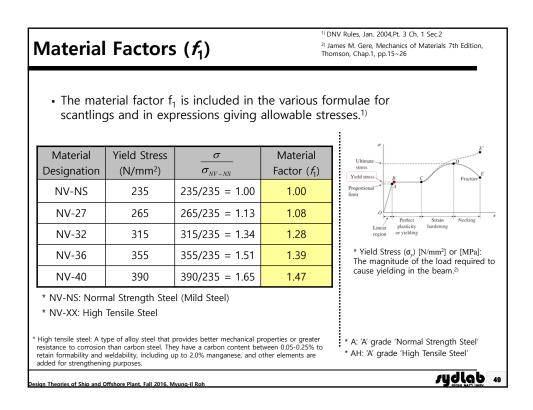
 C_B is in this case not to be taken less than 0.60.

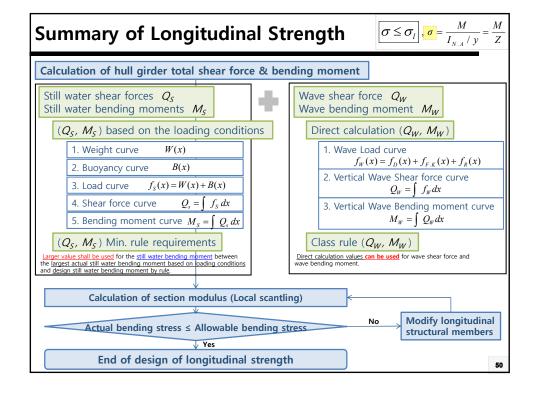
The midship section moment of inertia about the transverse neutral axis shall not (Pt.3 Ch.1 Sec.5 C400)

$$I_{ship} = 3C_W L^3 B(C_B + 0.7) \ [cm^4]$$

DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec. 5

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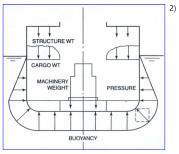
6.3 Local Strength (Local Scantling)

- (1) Procedure of Local Scantling
- (2) Local Strength & Allowable Stress
- (3) Design Loads
- (4) Scantling of Plates
- (5) Scantling of Stiffeners
- (6) Sectional Properties of Steel Sections

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Local Scantling

Ship structure members are designed to endure the loads acting on the ship structure such as hydrostatic and hydrodynamic loads¹⁾.



• For instance, the structural member is subjected to:

Hydrostatic pressure due to surrounding water Internal loading due to self weight and cargo weight Hydrodynamic load due to waves

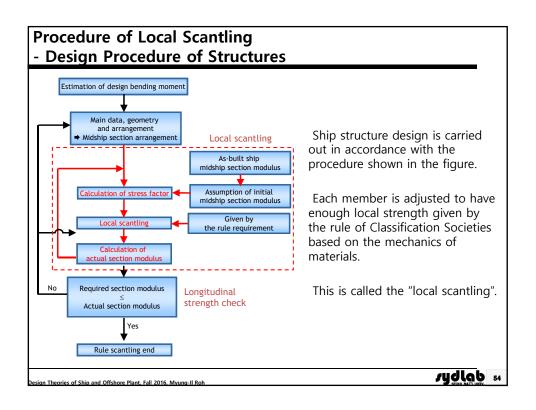
Inertia force of cargo or ballast due to ship motion

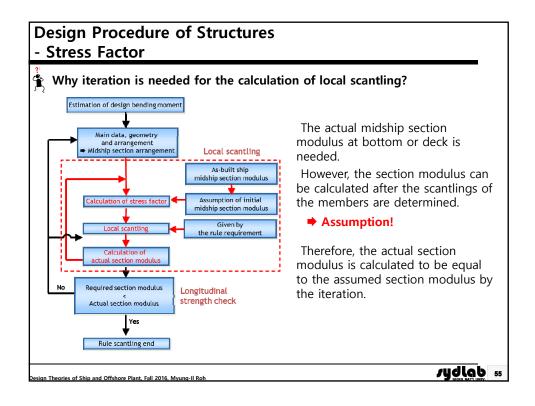
¹⁾ Okumoto, Y., Takeda, Y., Mano, M., Design of Ship Hull Structures - a Practical Guide for Engineers, Springer, pp. 17-32, 2009 ²⁾ Mansour, A., Liu, D., The Principles of Naval Architecture Series – Strength of Ships and Ocean Structures, The Society of Naval Architects and Marine Engineers, 2008

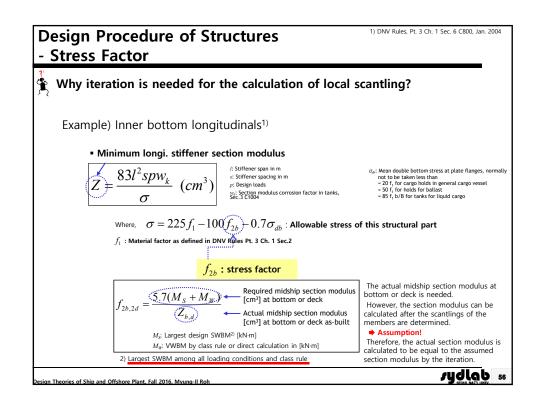
Theories of Ship and Offshore Plant, Fall 2016, Myung-Il Rol

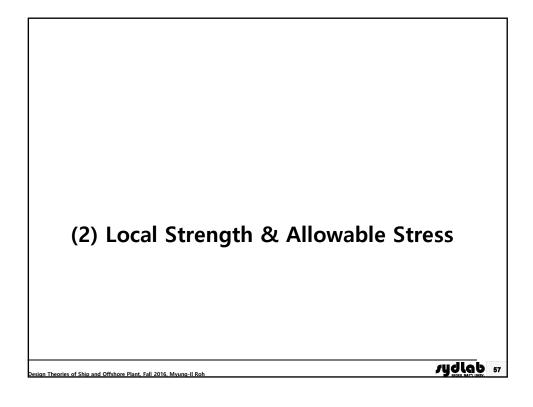
(1) Procedure of Local Scantling

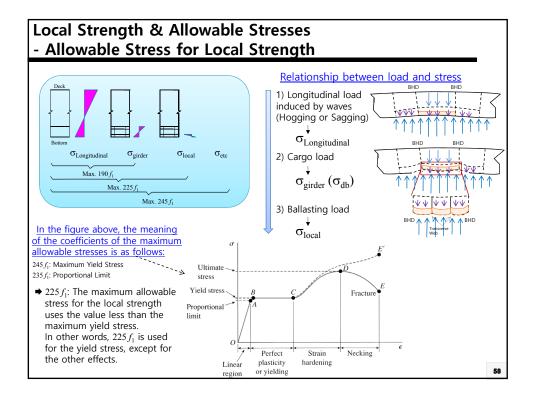
of Ship and Offshore Plant, Fall 2016, Myung-II Roh

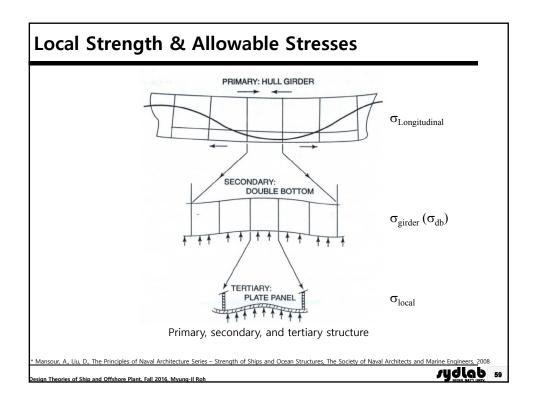


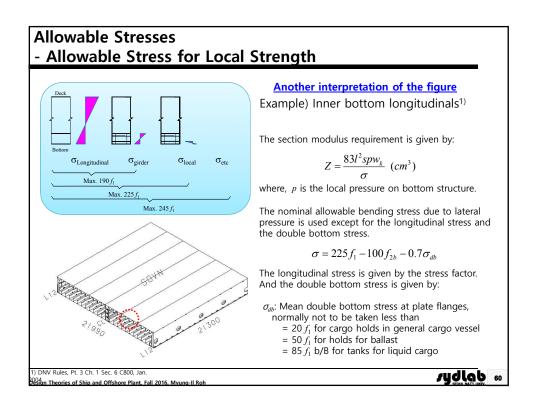


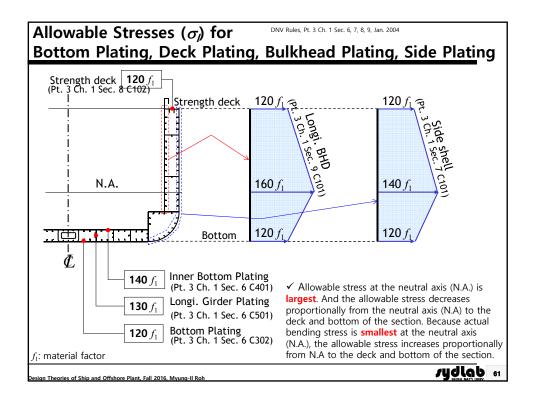


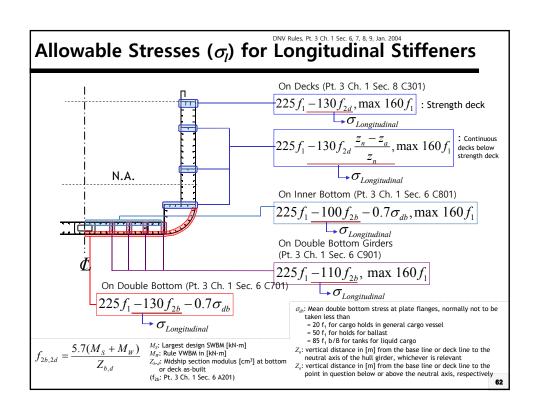


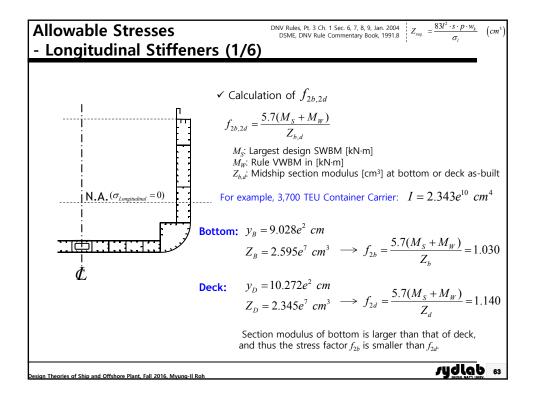


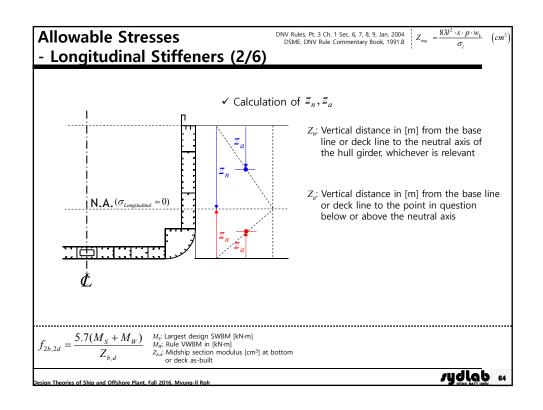


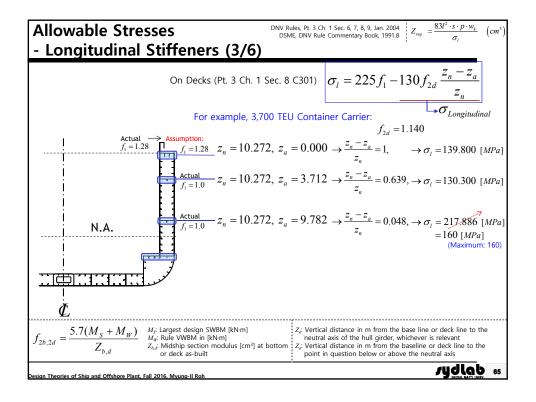


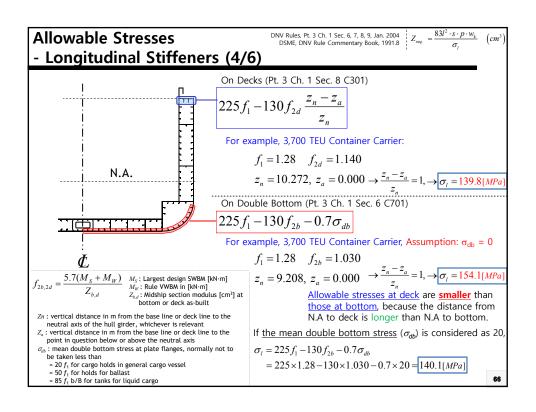


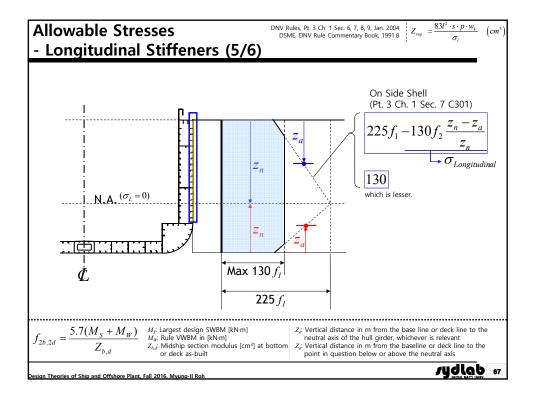


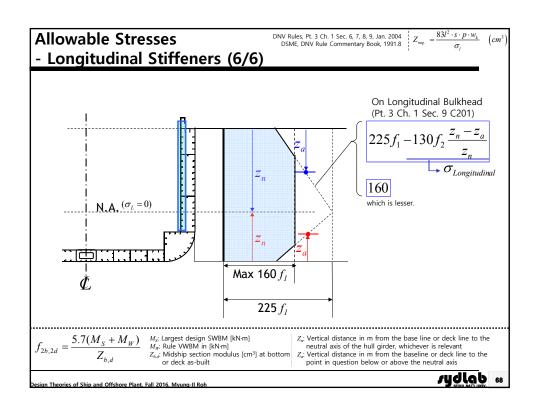






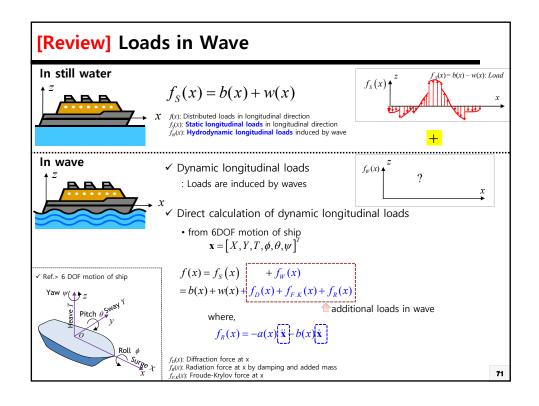


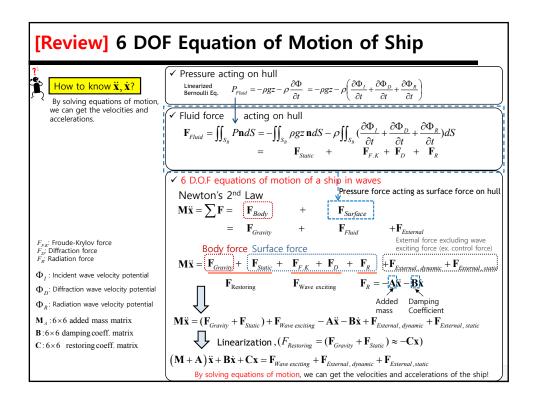




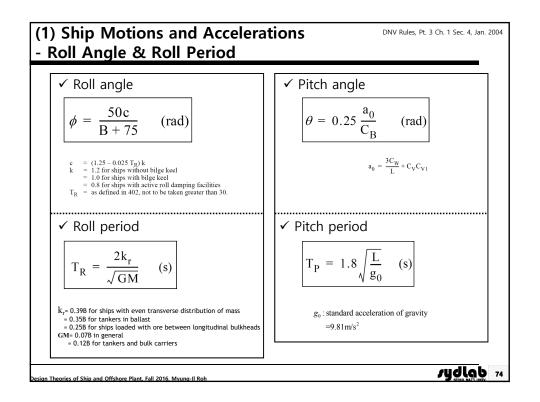
(3) Design Loads	
Design Theories of Ship and Offshore Plant, Fall 2016, Myung-Il Roh	/ydlab so

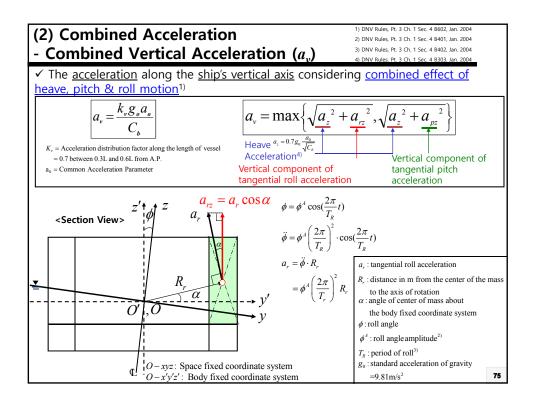
Contents ☑ Ship Motion and Acceleration ☑ Combined Acceleration ☑ Design Probability Level ☑ Load Point ☑ Pressure & Force ■ Sea Pressure ■ Liquid Tank Pressure

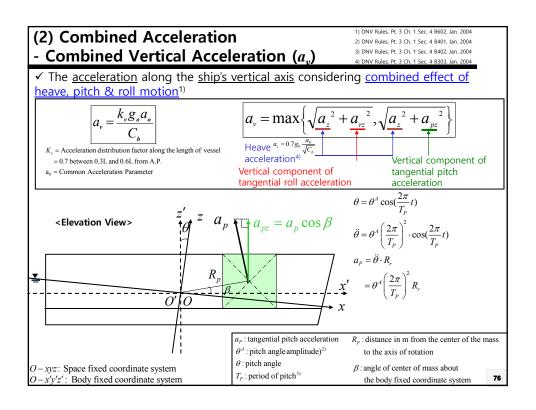


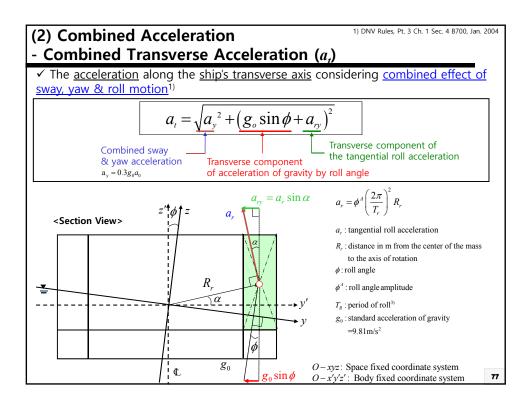


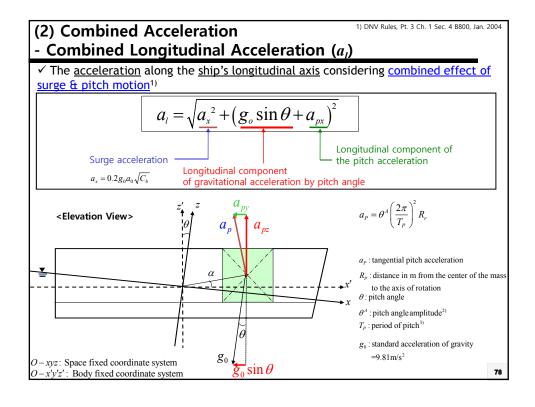
•	on and Acceleration rmula of DNV Rule	
		✓ Ref. 6 DOF motion of ship
Common Acceleration Parameter	$a_0 = \frac{3C_w}{L} + C_v C_{v1}$	Yaw w z Pitch θ
Surge Acceleration	$a_x = 0.2g_0 a_0 \sqrt{C_b}$	y y
Combined Sway/Yaw Acceleration	$a_y = 0.3g_0 a_0$	Roll \$ Surge }
Heave Acceleration	$a_z = 0.7g_0 \frac{a_0}{\sqrt{C_b}}$	Common Acceleration Parameter, a ₀
Tangential Roll Acceleration	$a_r = \phi \left(\frac{2\pi}{T_r}\right)^2 R_r$	$a_0 = \frac{3C_W}{L} + C_V C_{V1}$
Tangential Pitch Acceleration	$a_p = \theta \left(\frac{2\pi}{T_p}\right)^2 R_p$	$C_r = \frac{\sqrt{L}}{50}, \text{maximum } 0.2$ $C_{yz} = \frac{V}{\sqrt{L}}, \text{minimum } 0.8$ $C_{yz} = \frac{V}{\sqrt{L}}, \text{minimum } 0.8$ $C_{yz} = \frac{U}{1000} \frac{C_{yz}}{10.0792 \cdot L}$ $1000 \cdot L < 300 10.75 - \frac{1}{3}(300 - L)/100$
g_0 : standard accele =9.81 m/s ²	eration of gravity	$\begin{array}{c c} 300 \le L \le 350 \\ L > 350 & 10.75 \\ L > 350 & 10.75 - [(L - 350)/150]^{1/2} \end{array}$











(2) Combined Acceleration

- [Example] Vertical Acceleration

(Example) Calculate the vertical acceleration of a given ship at 0.5L (amidships) by DNV Rule.

[Dimension] L_s =315.79 m, V=15.5 knots, C_B =0.832

$$a_{v} = \frac{k_{v}g_{o}a_{o}}{C_{b}}$$

 $K_v =$ Acceleration distribution factor along the length of vessel

= 0.7 between 0.3L and 0.6L from A.P.

 $a_0 = Common Acceleration Parameter$

 $g_0 = Standard acceleration of gravity (=9.81 m/sec^2)$

(Sol.)
$$a_v = (k_v g_0 a_0) / C_B = (0.7 \times 9.81 \times 0.277) / 0.832$$

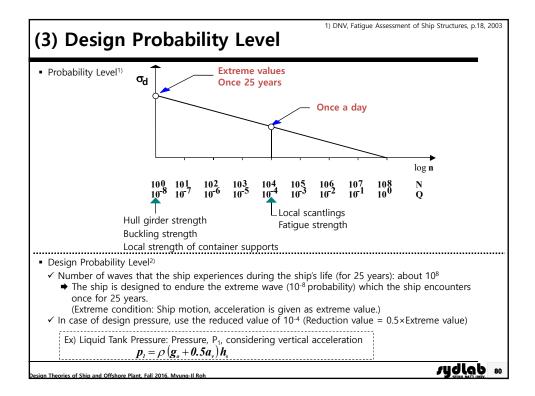
 $= 2.286 (m/sec^2)$
where, $k_v = 0.7$ at mid ship
$$a_0 = 3C_W / L + C_v C_{v1} = 3 \times 10.75 / 315.79 + 0.2 \times 0.872 = 0.277$$

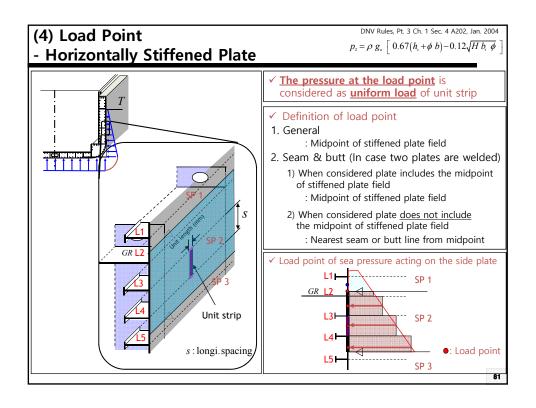
$$C_v = L^{0.5} / 50 = 315.79^{0.5} / 50 = 0.355 \text{ or Max. } 0.2$$

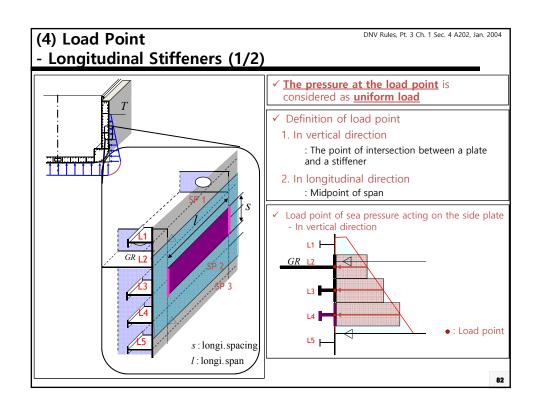
=0.2 $C_{v1} = V/L^{0.5} = 15.5/315.79^{0.5} = 0.872$ or Min. 0.8

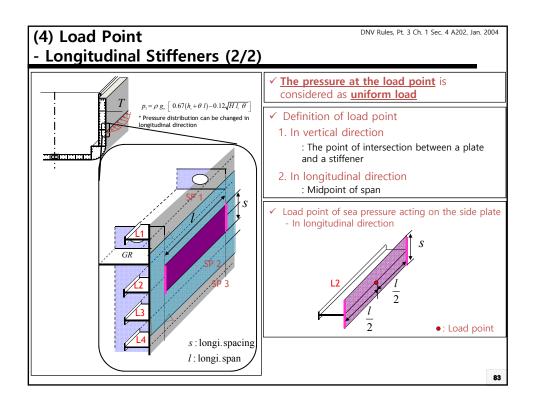
=0.872

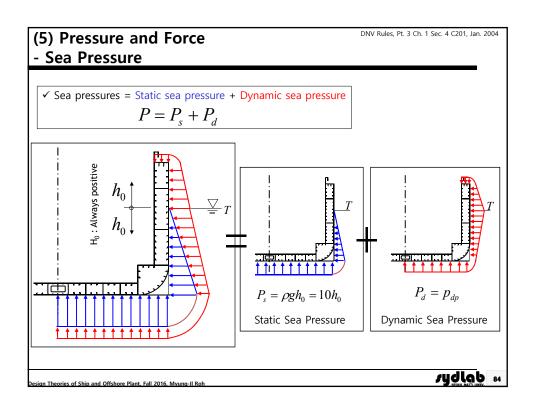
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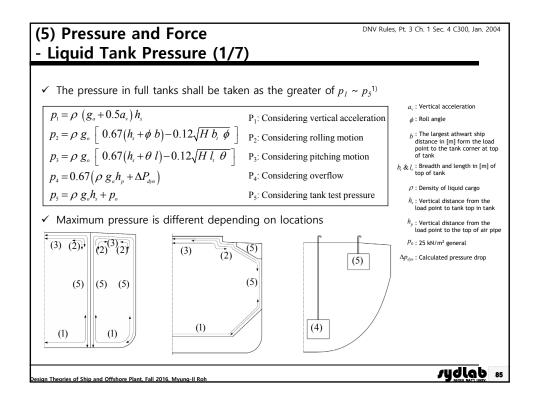


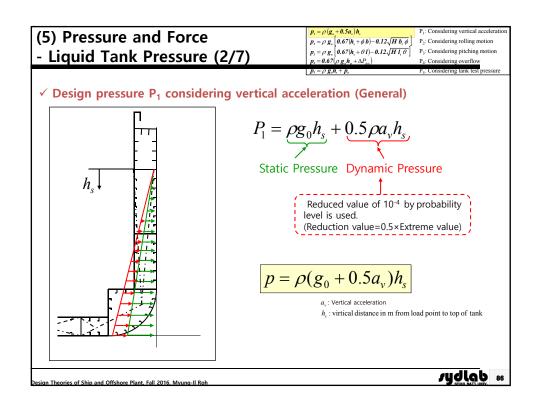


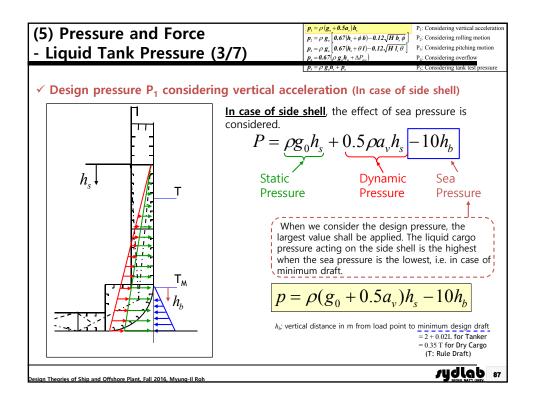


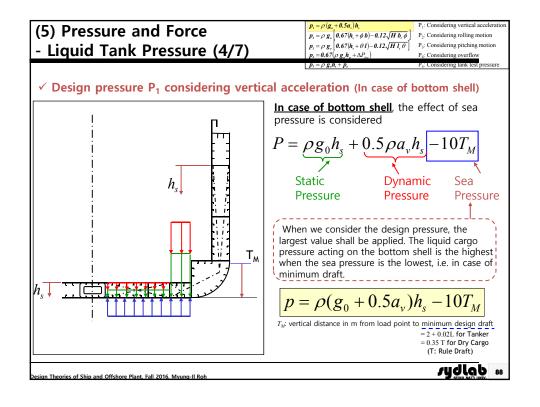












(5) Pressure and Force

DNV Rules, Pt. 3 Ch. 1 Sec. 4 B800, Jan. 2004

Example) Calculation of P₁ Pressure

(Example) When the tank is filled up, calculate the P₁ pressure of inner bottom and deck by using vertical acceleration (a_v =2.286 m/s²) and dimensions of tank which is given below.

[Dimension] Inner bottom height: 3.0 m, Deck height: 31.2m, $\rho = 1.025$ ton/m³

$$P_1 = \rho (g_0 + 0.5a_v) h_s$$

 $\rho = density (ton/m^3)$

a, = Vertical acceleration

 $g_0 = Standard acceleration of gravity (=9.81 m/sec^2)$

 h_s : virtical distance in m from load point to top of tank

(Sol.)
$$a_v = 2.286 \text{ m/s}^2$$

1 Inner Bottom

$$h_s = 31.2 - 3.0 = 28.8 \ m$$

$$P_1 = \rho \left(g_0 + 0.5 a_v \right) h_s$$

$$=1.025(9.81+0.5\times2.286)\times28.2$$

$$=316.6\,kN\,/\,m^2$$

② Deck

$$h_{\rm s} = 31.2 - 31.2 = 0 \ m$$

$$P_1 = \rho (g_0 + 0.5a_v) h_s$$

$$=1.025(9.81+0.5\times2.286)\times0$$

$$=0 kN/m^2$$

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(5) Pressure and Force

- Liquid Tank Pressure (5/7)

 \overline{b}

 $p_1 - p_1 g_2$, $0.3u_1 h_1$, $p_2 = \rho g_2$, $0.67(h_1 + \phi b) - 0.12\sqrt{H b_1 \phi}$ $\rho_3 = \rho g_2$, $0.67(h_1 + \theta l) - 0.12\sqrt{H l_1 \theta}$ $\rho_3 = 0.67(h_1 + \theta l) - 0.12\sqrt{H l_1 \theta}$ $\rho_3 = 0.67(h_1 + \theta l) - 0.12\sqrt{H l_1 \theta}$ $\rho_3 = 0.67(h_1 + \theta l) - 0.12\sqrt{H l_1 \theta}$

DSME, 선박구조설계 5-3 DNV Rules, Pt. 3 Ch. 1 Sec. 4, Jan. 200

✓ Design pressure P₂ considering the rolling motion

Air pipe

Load point

 h_2

 h_1

 h_{s}^{*}

When the ship is rolling, the higher static pressure is applied.

Assumption: $\phi \ll 1$

$$h_1 = h_s \cos \phi \approx h_s$$

$$h_2 = b \sin \phi \approx b \phi$$

$$\therefore h_s^* = h_1 + h_2$$
$$= \underline{(h_s + b\phi)}$$

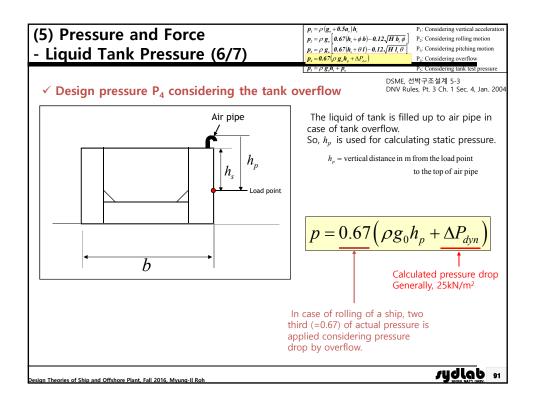
 $p_2 = \rho g_0 [0.67(h_s + \phi b) - 0.12\sqrt{H\phi b_r}]$

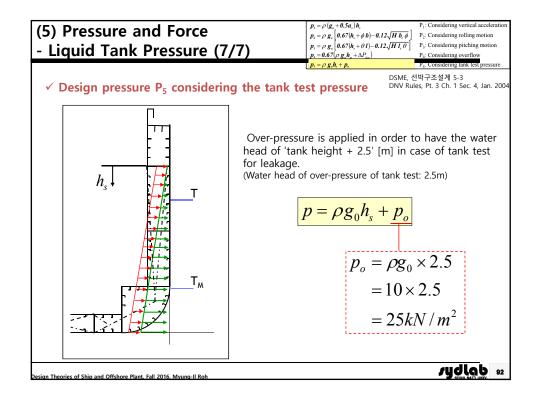
H: Height in m of the tank b_i : Breadth in m of top of tank

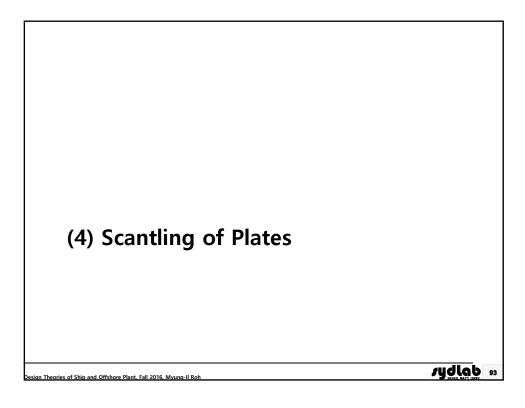
 h_{ς}

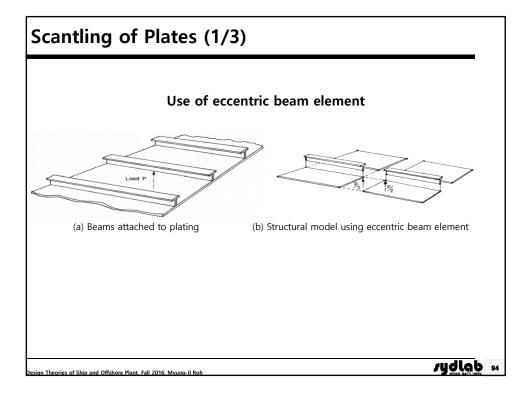
In case of rolling of a ship, two third (=0.67) of actual pressure is applied considering pressure drop by overflow

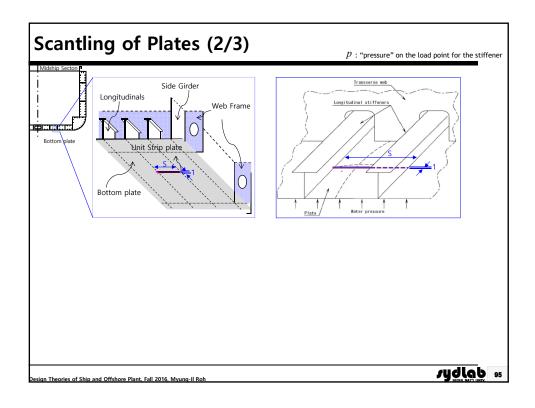
The filling ratio of the most tank is about 98%. That (about 2%) is considered.

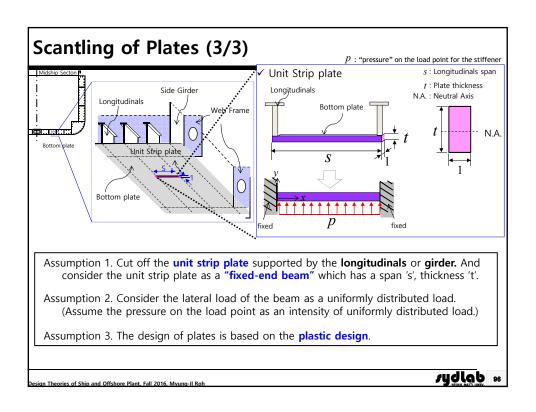


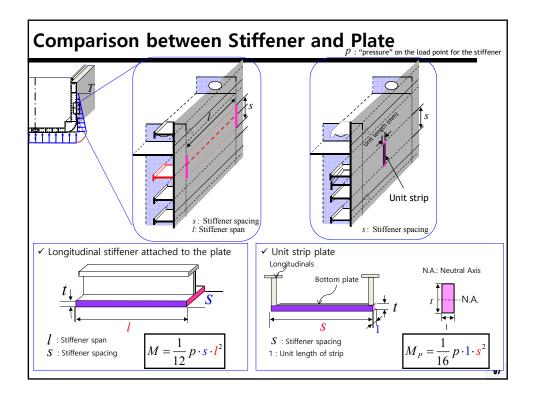


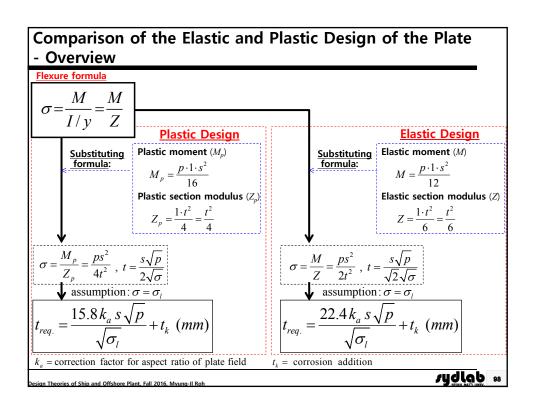






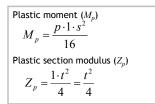


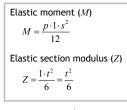


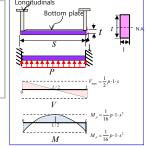


Comparison of the Elastic and Plastic Design

[Example] Thickness Requirements







1 A mild steel plate carries the uniform pressure of 100 kN/m² on a span length of 800 mm.

Compare the thickness requirement depending on the plastic design and elastic design.

$$t_{req.}_{plastic} = \frac{15.8k_a s \sqrt{p}}{\sqrt{\sigma_l}}$$

$$= \frac{15.8 \times 1 \times 0.8 \times \sqrt{100}}{\sqrt{235}} = 8.24 \text{ (mm)}$$

$$t_{req.}_{elastic} = \frac{22.4k_a s \sqrt{p}}{\sqrt{\sigma_l}}$$

$$= \frac{22.4 \times 1 \times 0.8 \times \sqrt{100}}{\sqrt{235}} = 11.69 \text{ (mm)}$$

$$t_{req.}_{relastic} = \frac{22.4 \kappa_a s \sqrt{p}}{\sqrt{\sigma_t}}$$
$$= \frac{22.4 \times 1 \times 0.8 \times \sqrt{100}}{\sqrt{235}} = [11.69 \text{ (mm)}]$$

The thickness requirement of the plate of plastic design is smaller than that of the elastic design at the same pressure and on the same span.

 k_a = correction factor for aspect ratio of plate field

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Comparison of the Elastic and Plastic Design

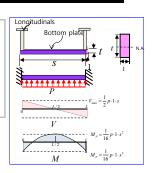
- [Example] Design Pressure

Plastic moment
$$(M_p)$$

$$M_p = \frac{p \cdot 1 \cdot s^2}{16}$$
Plastic section modulus (Z_p)

$$Z_p = \frac{1 \cdot t^2}{4} = \frac{t^2}{4}$$

Elastic moment (*M*)
$$M = \frac{p \cdot 1 \cdot s^2}{12}$$
Elastic section modulus (*Z*



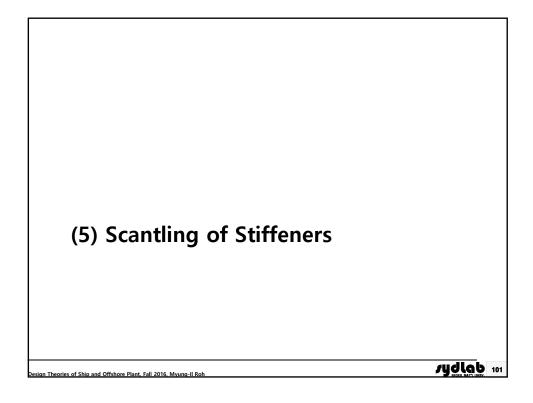
② A mild steel plate has a thickness of 10 mm on a span length of 800 mm.

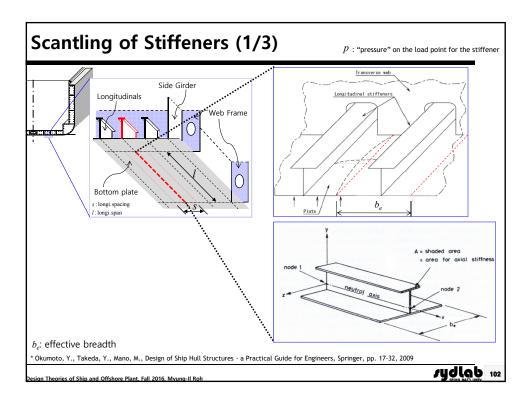
Compare the design pressure that the maximum stresses of the plate reaches the yield stress depending on the plastic design and elastic design.

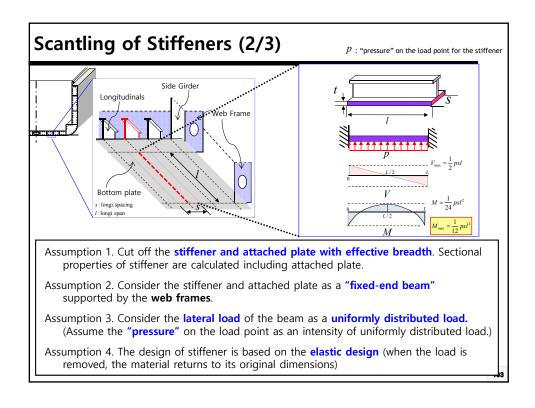
$$p_{plastic} = \frac{t^2 \sigma_l}{15.8^2 s^2}$$
$$= \frac{10^2 \times 235}{15.8^2 0.8^2} = 147 [kN/m^2]$$

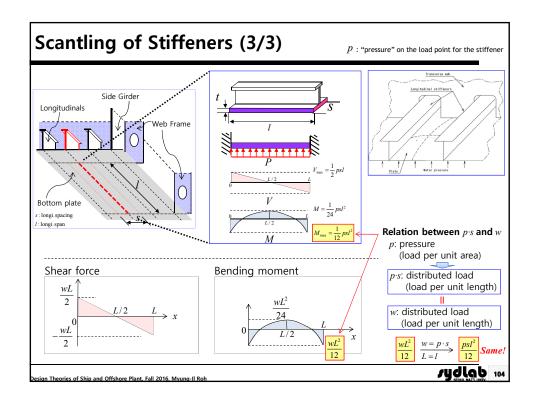
$$p_{elastic} = \frac{t^2 \sigma_l}{22.4^2 s^2}$$
$$= \frac{10^2 \times 235}{22.4^2 0.8^2} = 73 \left[\frac{kN}{m^2} \right]$$

The design pressure of plastic design that reaches the yield stress, is higher than that of the elastic design on the same span with the same thickness.



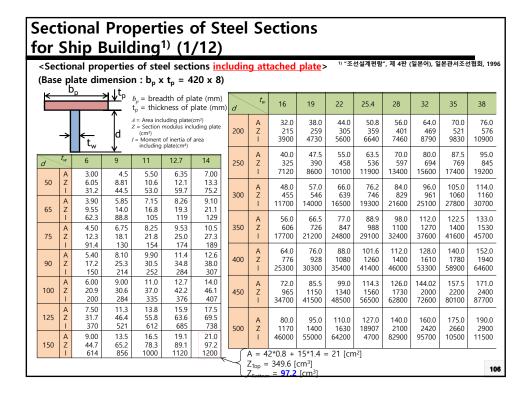


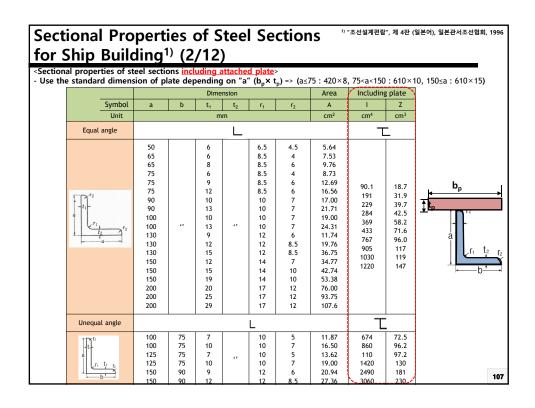


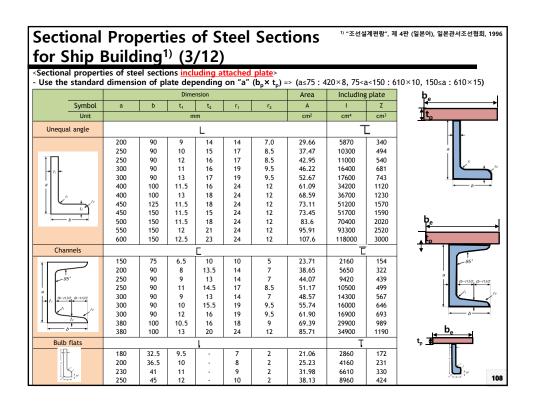


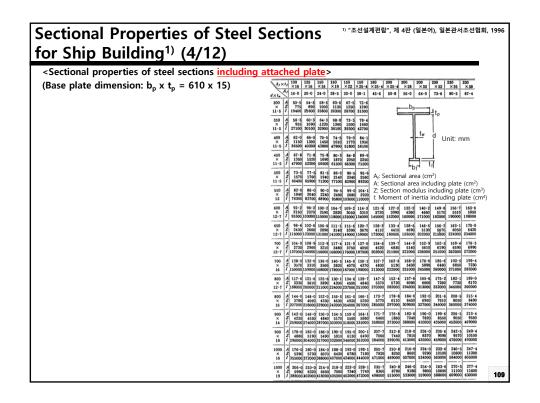
(6) Sectional Properties of Steel Sections

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Section shape	А	/	Z_e	Z_{P}
$\begin{array}{c} \Pi_{i} \\ \frac{\tilde{e}_{I}}{\tilde{e}_{I}} \\ \vdots \\ -r_{m} - r_{i} + l \\ -r_{I} \end{array}$	$\frac{1}{2}\pi(r_1^2 - r_1^2)$ $t/r_n b = 0 \Rightarrow b$ $A_{rn} = \pi r_n t$		$\epsilon_1 = r_1 - \epsilon_2$ $\epsilon_2 = \frac{4(r_1! + r_2r_1 + r_1!)}{3\pi(r_2 + r_1)}$ $\epsilon_{1rn} = \frac{2}{\pi}r_n = 0.6366 r_n$	$ \begin{split} & 2 \{ 2 (r_t^3 \sin^3 \theta_t \\ & - r_t^3 \sin^3 \theta_t) \\ & - (r_t^3 - r_t^3) \} / 3 \\ & \subset \zeta \zeta, \\ & r_1 \cos \theta_1 - r_t \cos \theta_t \end{split} $
e ₁ A B	$\frac{1}{2}r^{i}(2\alpha-\sin\!2\alpha)$	$I_A = r^4 \left[\frac{1}{16} (4\alpha - \sin \theta) \right]$ $I_B = \frac{r^4}{12} \left[3\alpha - 2 \sin 2\theta \right]$ $e_1 = r \left(1 - \frac{4 \sin^2 \alpha}{6\alpha - 3} \right)$ $e_2 = r \left(\frac{4 \sin^2 \alpha}{6\alpha - 3} \right)$	$\frac{2\alpha}{\sin 2\alpha}$	$\frac{2}{3}r^{3}(2\sin^{3}\alpha_{s}-\sin^{3}\alpha)$ $\subset \subset \mathcal{C},$ $\frac{2\alpha-\sin 2\alpha}{2\alpha_{s}-\sin 2\alpha_{s}}=4$
e, A B A B	2art	$I_A = r^2 I(\alpha + \sin\alpha\cos\alpha)$ $= 2 \frac{\sin^2\alpha}{\alpha}$ $I_B - r^2 I(\alpha - \sin\alpha\cos\alpha)$	$e_1 = r\left(1 - \frac{\sin\alpha}{\alpha}\right)$ $e_2 = r\left(\frac{\sin\alpha}{\alpha} - \cos\alpha\right)$	$ 2rt(r-t/2) \\ \times (2\sin\frac{\alpha}{2} - \sin\alpha) $
e A B B A B	ω	$I_{A} = \frac{1}{4} r^{4} (\alpha + \sin\alpha\cos\alpha)$ $-\frac{16 \sin^{4}\alpha}{9\alpha})$ $I_{B} = \frac{1}{4} r^{4} (\alpha - \sin\alpha\cos\alpha)$	$\epsilon_1 = r \left(1 - \frac{2 \sin \alpha}{3\alpha} \right)$ $\epsilon_2 = r \frac{2 \sin \alpha}{3\alpha}$	$\alpha > 0.955,$ $(2\alpha' - \sin 2\alpha' = \alpha)$ $2r^{3}(2\sin \alpha' - \sin \alpha)/3$ $\alpha < 0.996$ $\frac{2r^{3}}{3}\left[\sin \alpha - \sqrt{\frac{\alpha^{3}}{2\tan \alpha}}\right]$
15, # [7]	жab	$\frac{\pi}{4}a^3b = 0.7854a^3b$	$\frac{\pi}{4}a^{1}b = 0.7854 a^{1}b$	$\frac{4}{3}a^{i}b$

Section shape	А	1	Z_e	Z_{P}
16. bm	π(a;b1-a;b1) t/an, t/bn が 小さいとも An=π(an+bn)t	$\frac{\pi}{4}(a_1b_1-a_1b_1)$ $I_m = \frac{\pi}{4}a_m^3(a_m+3b_m)t$	$\frac{\pi}{4} \frac{a_1 b_2 - a_1 b_1}{a_1}$ $Z_m = \frac{\pi}{4} a_m (a_m + 3b_m) t$	$\frac{4}{3}(a_1!b_2-a_1!b_1)$
17. 半楕円	1 nað	$\left(\frac{\pi}{8} - \frac{8}{9\pi}\right) a^3 b$ $= 0 \cdot 1098 \ a^3 b$	$\begin{aligned} \epsilon_1 &= \left(1 - \frac{4}{3\pi}\right) a = 0.5756a \\ Z_1 &= 0.1908 \ a^2 b \\ \epsilon_2 &= \frac{4r}{3\pi} = 0.4244 \ a \\ Z_1 &= 0.2587 \ a^2 b \end{aligned}$	⇔0·35362 a²b
18, B t ₂ h A A h ₁ t ₃ t ₄	$2bt_1+h_1t_1$	$I_{A} = \frac{bh^{3} - (b - t_{1})h_{1}^{3}}{12}$ $I_{B} = \frac{2b^{3}t_{2} + h_{1}t_{3}^{3}}{12}$	$Z_{A} = \frac{bh^{3} - (b - t_{1})h_{1}^{3}}{6h}$ $Z_{A} = \frac{2b^{3}t_{2} + h_{1}t_{1}^{3}}{6b}$	$\frac{h_1^1 t_1}{4} + \frac{h t_2}{2} (h + h_1)$
19. e ₂ e ₁ t ₂ h A h ₁ t ₂ t	$2bt_1+h_1t_1$	$I_{A} = \frac{bh^{3} - (b - t_{1})h_{1}^{3}}{12}$ $I_{A} = \frac{2b^{3}t_{2} + h_{1}t_{1}^{3}}{3} - Ae_{2}^{3}$	$e_1 = b - e_2$ $e_2 = \frac{2b^2t_2 + h_1t_1^2}{4bt_2 + 2h_1t_1}$	18. と同じ

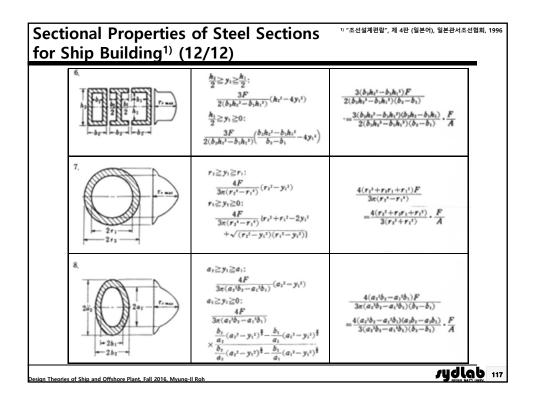
Section shape	Α	/	Z_e	Z_P
1 h	òh	$\frac{1}{12}\delta h^{i}$	$\frac{1}{6} bh^2$	$-\frac{1}{4} hh^z$
2. h, h,	$h_i^{z} - h_i^{z}$	$\frac{1}{12}(h_{7}{}^{*}-h_{7}{}^{*})$	$\frac{1}{6} \cdot \frac{h_2 - h_1}{h_2}$	$\frac{1}{4}(h_t{}^s{-}h_t{}^s)$
3. h	ħ²	$\frac{1}{12}h^{\epsilon}$	$\frac{\sqrt{2}}{12}h^2$	$\frac{\sqrt{2}}{6}h$
4. h ₂ h ₂	$h_2^2 - h_1^2$	$\frac{1}{12}(h_1{}^{i}-h_1{}^{i})$	$\frac{\sqrt{2}}{12} \frac{h_{1}^{4} - h_{1}^{4}}{h_{1}}$	$\frac{\sqrt{2}}{6}(h_2^3-h_1^3)$
5. 1 h	$-\frac{1}{2} \delta h$	$\frac{1}{36}bh^{3}$	$\epsilon_1 = \frac{2}{3}h, \ Z_1 = \frac{bh^2}{24}$ $\epsilon_i = \frac{1}{3}h, \ Z_2 = \frac{bh^2}{12}$	$\frac{2-\sqrt{2}}{6}bh^{\tau}$

Section shape	Α	/	Z_e	Z_P
6.	$\frac{1}{2}(b_1+b_2)h$	$\frac{h^2(b_1^2+4b_1b_2+b_2^2)}{36(b_1+b_2)}$	$e_1 = \frac{h(b_1 + 2b_1)}{3(b_1 + b_1)}$ $Z_1 = \frac{h^2(b_1^2 + 4b_1b_2 + b_1^2)}{12(b_1 + 2b_1)}$ $e_2 = \frac{h(2b_1 + b_1)}{3(b_1 + b_1)}$ $Z_1 = \frac{h^2(b_1^2 + 4b_1b_1 + b_1^2)}{12(2b_1 + b_1)}$	$\frac{Ah\ (b_1b_2+b_2b_3+b_3b_1}{3\ (b_1+b_2)(b_1+b_2)}$ $\subset \subset \mathcal{C},$ $b_1^2 = (b_1^2+b_2^2)/2$
7. 正 n 角形 A B	$\frac{1}{2}$ na r_1		$Z_{s} = \frac{A}{48 r_{1}} (12 r_{1}^{2} + a^{2})$ $Z_{s} = \frac{A}{24 r_{1}} (6 r_{1}^{2} - a^{2})$	$n: \bigoplus_{k=1}^{\infty} kk, Z_{PA} = \frac{a^2 r_1}{6}$ $+ \frac{2}{3} a r_1^2 \sum_{k=1}^{\frac{n}{2}-1} \sin \frac{2k\pi}{n}$
8.	$\frac{1}{4}\pi d^{2}$	$\frac{1}{64}\pi d^4$	$\frac{1}{32}\pi d^{3}$	$\frac{1}{6}d^3$
9. $d_m - t$	$\frac{\frac{1}{4}\pi(dz^{1}-dz^{1})}{t/d_{m}b^{(j)}} \stackrel{\text{def}}{\Rightarrow} b$ $A_{dm}=\pi d_{m} t$	$\frac{1}{64}\pi (d_1 - d_1)$ $I_{4m} = \frac{1}{8}\pi d_m^2 t$	$\frac{\pi}{32} \frac{d_2^4 - d_1^4}{d_2}$ $Z_{dm} = \frac{1}{4} \pi d_m^2 t$	$\frac{1}{6}(d_1^3-d_1^3)$
	1 xr2	$\left(\frac{\pi}{8} - \frac{8}{9\pi}\right)r^4$ $= 0.1098 r^4$	$e_1 = \left(1 - \frac{4}{3\pi}\right)r = 0.5756r$ $Z_1 = 0.1908 \ r^2$ $e_2 = \frac{4r}{3\pi} = 0.4244 \ r$ $Z_3 = 0.2587 \ r^3$; 0-37982 r³

Section shape	Α	/	Z_e	Z_{P}
20. H l2 l4	$\delta t_3 + h_1 t_1$	$I_{s} = \frac{h^{2}t_{1} + (b-t_{1})t_{1}^{3}}{3} - Ae_{1}^{3}$ $-Ae_{1}^{3}$ $I_{s} = \frac{b^{2}t_{2} + h_{1}t_{1}^{2}}{12}$	$e_1 = \frac{h^2 t_1 + (b - t_1) t_2^2}{2(bt_2 + h_2 t_1)}$ $e_2 = h - e_1$	$\begin{aligned} &t_1 \leq h, t_1 \neq b \not \supset \xi \geqslant \\ &\frac{bt_2}{2} \left(h - \frac{t_1}{t_1} b\right) \\ &+ \frac{h_1 t_1}{4} \left[h_1 + \left(\frac{t_1}{t_1}\right)^2 \right. \\ &\times \left(\frac{h}{h_1}\right) b\right] \\ &t_1 > h, t_1 \neq b \not \supset \xi \geqslant \\ &\frac{bt_1^2}{4} \left[1 - \left(\frac{h_1 t_1}{bt_1}\right)^2\right] \\ &+ \frac{h_1 h_1 t_1}{2} \end{aligned}$
h h e e	$(h+h_1)t$	$\frac{t}{3} \cdot (h^3 + h_1 t^2) - A \varepsilon_1^2$	$\varepsilon_1 = h - \varepsilon_1$ $\varepsilon_2 = \frac{h^2 + h_1 t}{2(h + h_1)}$	$\frac{t}{4} \big[(h\!-\!t)^{\scriptscriptstyle 2} \!+\! h^{\scriptscriptstyle 2} \big]$
22. B	$(h+h_1)t$	$I_{A} = \frac{(h+1)^{4}}{24} - \frac{h_{1}^{4} + 2t^{4}}{24}$ $-Ae_{2}^{3}$ $I_{B} = \frac{1}{12}(h^{4} - h_{1}^{4})$	$e_1 = \frac{h^2 + h_1 t}{\sqrt{2(h + h_1)}}$ $e_2 = \frac{h^2}{\sqrt{2(h + h_1)}}$	$\frac{t}{\sqrt{2}}[h(h-t)+t^2]$
23. 11 h h, e1 + e2	$bt_2+h_1t,$	$\frac{h^3t_1 + (b-t_1)t_2^3}{3} - Ae_2^3$	$e_1 = h - e_2$ $e_2 = \frac{h^2 t_1 + (b - t_1) t_2^2}{2(bt_2 + h_1 t_1)}$	20. と同じ

Section shape	А	/	Z_e	Z_{P}
24.	$b_0t_0+bt_2+h_1t_1$	$I = \frac{b_0 t_0^2}{3} + \frac{bh^3}{3} - \frac{(b)}{2h}$ $e_1 = t_0 + \frac{bh^3 - (b - t_1)}{2A}$ $e_2 = h - \frac{bh^2 - (b - t_1)}{2A}$	$h_1^2 - b_0 t_0^2$	$\begin{split} &t_0 \leq (bt_1 + h, t_1) / b_1 \sigma \succeq \S \\ &\frac{but_1}{2} (h_t + t_1) + \frac{bt_1h}{2} \\ &+ \frac{h_1^2t_1}{4} - \frac{1}{4t_1} \\ &\times (bt_1 - b_1t_2)^2 \\ &t_0 > (bt_1 + h, t_1) / b_1 \sigma \succeq \S \\ &\frac{b_1t_2^2}{4} - \frac{1}{4b_0} (bt_1 + h, t_1)^2 \\ &+ \frac{(h_1 + t_2)(h_1t_1 + bt_1)}{2} \\ &+ \frac{bt_1h}{2} \end{split}$
25. a = a = d = d = d = d = d = d = d = d =	t(a+b)	$\frac{td^{2}}{12}(3a+b)$	$\frac{td}{6}(3a+b)$	$\frac{adt}{2} + \frac{bdt}{4}$
26.	$at\left(1+\frac{\pi}{2}\right)+2 bt$ $=2.5708 at+2 bt$	$\frac{a^3t}{12} \left(1 + \frac{3}{4}\pi \right) + \frac{1}{2}a^3bt$ $= 0 \cdot 2797a^3t + 0 \cdot 5a^3bt$		$\frac{3}{4}a^2t + abt + \frac{t^3}{6}$

Section shape and distribution of shear force	$\tau_{r} = \frac{F}{z_{1}I} \int_{z_{1}}^{r_{1}} zy dy$	$\tau_{rmax} = \frac{\alpha F}{A}$
1. [9, 1]	$\frac{3}{2} \cdot \frac{F}{bh} \left\{ 1 - \left(\frac{2y_1}{h}\right)^2 \right\}$	$\frac{3}{2} \cdot \frac{F}{bh} = \frac{3}{2} \cdot \frac{F}{A}$
2	$\sqrt{2} \frac{F}{a^2} \left\{ 1 + \sqrt{2} \frac{y_1}{a} - 4 \left(\frac{y_1}{a} \right)^2 \right\}$	$\frac{9}{8}\sqrt{2}\frac{F}{\sigma^{l}}=1.591\frac{F}{A}$
3.	$\frac{4}{3} \cdot \frac{F}{\pi r^2} \Big\{ 1 - \Big(\frac{y_1}{r}\Big)^t \Big\}$	$\frac{4}{3} \cdot \frac{F}{\pi r^i} = \frac{4}{3} \cdot \frac{F}{A}$
1 2 1	$\frac{F}{\pi rt} \left\{ 1 - \left(\frac{y_1}{r}\right)^t \right\}$	$\frac{F}{\pi r t} = 2 \frac{F}{A}$
5.	$\frac{4}{3} \cdot \frac{F}{\pi ab} \left[1 - \left(\frac{y_1}{a} \right)^2 \right]$	$\frac{4}{3} \cdot \frac{F}{\pi ab} = \frac{4}{3} \cdot \frac{F}{A}$



6.4 Buckling Strength

- (1) Column Buckling
- (2) Buckling Strength of Stiffener
- (3) Buckling Strength of Plate
- (4) Buckling Strength by DNV Rule
- (5) Buckling Strength of Stiffener by DNV Rule
- (6) Buckling Strength of Plate by DNV Rule

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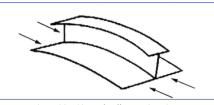
JUDIO 118

Buckling

 Definition: The phenomenon where lateral deflection may arise in the athwart direction* against the axial working load

*선측(船側)에서 선측으로 선체를 가로지르는

· This section covers buckling control for plate and longitudinal stiffener.



Flexural buckling of stiffeners plus plating

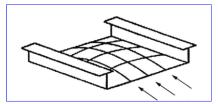


Plate alone buckles between stiffeners

* Mansour, A., Liu, D., The Principles of Naval Architecture Series - Strength of Ships and Ocean Structures, The Society of Naval Architects and Marine Engineers, 2008

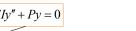
sydlab 119

(1) Column Buckling

James M. Gere, Mechanics of Materials 6th Edition, Thomson, pp. 748-762

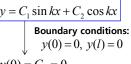
- The Equation of the Deflection Curve

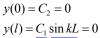
- Differential equation for column buckling:

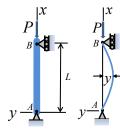


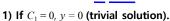
Using the notation $k^2 = \frac{P}{EI}$, $y'' + k^2 y = 0$

General solution of $y = C_1 \sin kx + C_2 \cos kx$ the equation:









- 2) If $\sin kl = 0$, ($\sin kl = 0$: buckling equation)
 - ① If kl = 0, y = 0 (trivial solution).
 - (1) If kl = 0, y = 0 (main solution.)

 (2) If $kl = n\pi$ (n=1, 2, 3) or $P = \left(\frac{n\pi}{l}\right)^2 EI$, it is nontrivial solution. E = modulus of elasticity $I = 2^{\text{nd}}$ moment of the section area and the solution.

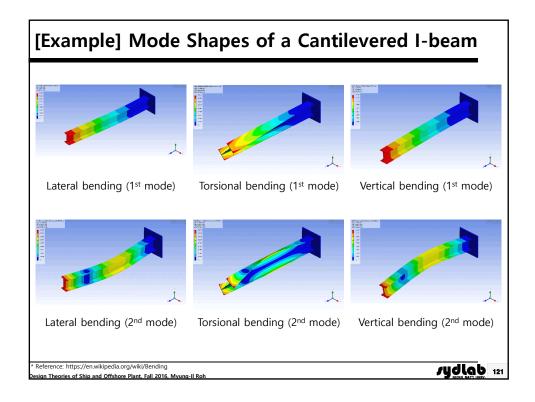
 $\therefore y = C_1 \sin kx = C_1 \sin \frac{n\pi x}{L}, n = 1, 2, 3...$

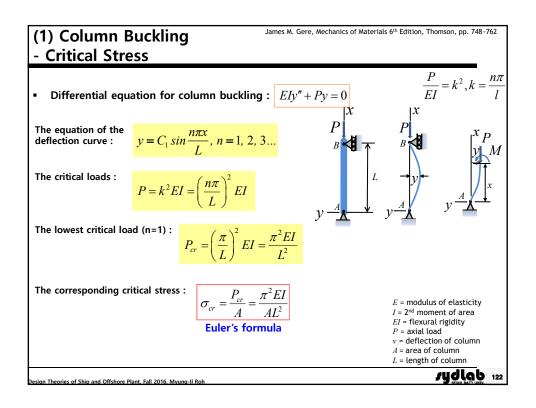
EI = flexural rigidity P = axial load

v = deflection of column

L = length of column

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(1) Column Buckling

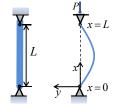
James M. Gere, Mechanics of Materials 6th Edition, Thomson, pp. 748~762

- Critical Load

Differential equation for column buckling : $y'' + \lambda y = 0$, y(0) = 0, y(L) = 0, where $\lambda = P/EI$

The equation of the deflection curve : $y_n(x) = C_1 \sin(n\pi x/L)$

The critical loads : $P_n = n^2 \pi^2 EI / L^2, n = 1, 2, 3...$



The lowest critical load (n=1): $P_{cr} = P_1 = \pi^2 EI/L^2$

E = modulus of elasticity I = 2^{nd} moment of area EI = flexural rigidity P = axial load y = deflection of column A = area of column L = length of column

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(1) Column Buckling

- Critical Buckling Stress

A critical buckling stress is often used instead of a buckling load and it can be derived by dividing P_{cr} by A, the cross sectional area of the column.

Euler's formula

The corresponding critical stress : $\sigma_{cr} = \frac{P_{cr}}{A}$

 $=\frac{\pi^2 EI}{Al^2}$

 $=\pi^2 E\bigg(\frac{k}{l}\bigg)^2$

E = modulus of elasticity I = 2nd moment of area EI = flexural rigidity P = axial load

y = deflection of column A = area of column l = length of column

, where $k\left(k^2=I\,/\,A\right)$ is the radius of gyration of the section of the column.

The ratio (l/k), often called the slenderness ratio, is the main factor which governs the critical stress

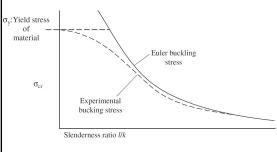
For large value of l/k the critical stress tends toward zero, and at small values of l/k it tends to infinity. In Euler's formula, the buckling stress may become infinite for a small value of l/k, however, buckling stress never goes up above the yield stress of the material in actual conditions, because the material would fail if the stress exceeded the yield stress.

1) The radius of gyration describes a circular ring whose area is the same as the area of interest.

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(1) Column Buckling

- Curve of Buckling Stress



by theoretical consideration, a horizontal line of yield stress connected to Euler buckling stress is specified as an upper limit of Euler's buckling curve.

$$\sigma_{cr} = a - b \left(\frac{l}{k} \right)$$
 Tetmayer's formula

$$\sigma_{cr} = a - b \left(\frac{\iota}{k} \right)$$
 Johnson's formula
$$\sigma_{cr} = \frac{a}{1 + b(1/k)^2}$$
 Rankine's formula

For example, one of the Classification Societies, ABS (American Bureau of Shipping) specifies the permissible load of a pillar or strut of mild steel material in the following equation:

$$\sigma_{cr} = 1.232 - 0.00452 \left(\frac{l}{k}\right) [ton \cdot f / cm^2]$$

From the above equation, we can see that the ABS formula is theoretically based on Tetmayer's experimental result.

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(1) Column Buckling

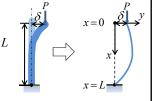
- Buckling of Thin Vertical Column Embedded at Its Base and Free at Its Top (1/2)

Suppose that a tin vertical homogeneous column is embedded at its base (x=0) and free at its top (x=L) and that a constant axial load P is applied to its free end.

The load either causes a small deflection δ , or does not cause such a deflection. In either case the differential equation for the deflection y(x) is

$$EI\frac{d^2y}{dx^2} = P(\delta - y) \quad \Box \rangle \quad EI\frac{d^2y}{dx^2} + Py = P\delta \quad \cdots (1)$$

(1) What is the predicted deflection when $\delta = 0$?



- The general solution of the differential equation (1) is $\frac{1}{D}$

$$y = c_1 \cos \sqrt{\frac{P}{EI}} x + c_2 \sin \sqrt{\frac{P}{EI}} x + \delta$$

- The boundary conditions of the differential equation (1) are $y_1(0) = y_2(0) = 0$
- If $\delta = 0$, this implies that $c_1 = c_2 = 0$ and y(x) = 0. That is, there is no deflection.

* Zill, D.G., Advanced Engineering Mathematics, 3rd edition, pp.166-174, 2006

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(1) Column Buckling

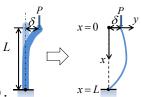
- Buckling of Thin Vertical Column Embedded at Its Base and Free at Its Top (2/2)

Suppose that a tin vertical homogeneous column is embedded at its base (x=0) and free at its top (x=L) and that a constant axial load P is applied to its free end.

The load either causes a small deflection δ , or does not cause such a deflection. In either case the differential equation for the deflection y(x) is

$$EI\frac{d^2y}{dx^2} = P(\delta - y) \qquad \Box \Rightarrow EI\frac{d^2y}{dx^2} + Py = P\delta \cdots (1)$$

(2) When $\delta \neq 0$, show that the Euler load for this column is one-fourth of the Euler load for the hinged



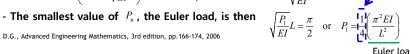
- If $\delta \neq 0$, the boundary conditions give, in turn, $c_1 = -\delta, \ c_2 = 0$.

$$y = \delta \left(1 - \cos \sqrt{\frac{P}{EI}} x \right)$$

- In order to satisfy the boundary condition $y(L) = \delta$, we must have

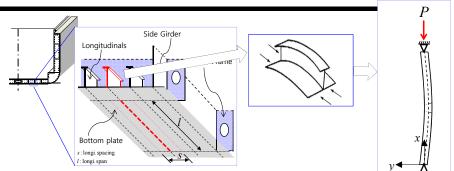
$$\delta = \delta \left(1 - \cos \sqrt{\frac{P}{EI}} L \right) \longrightarrow \cos \sqrt{\frac{P}{EI}} L = 0 \longrightarrow \sqrt{\frac{P}{EI}} L = n\pi/2$$

Zill, D.G., Advanced Engineering Mathematics, 3rd edition, pp.166-174, 2006



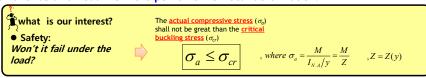
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(2) Buckling Strength of Stiffener

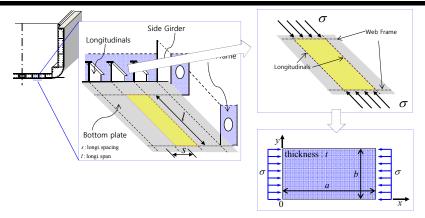


It is assumed that the stiffener is a fixed-end column supported by the web frames.

Hull girder bending moment is acting on the cross section of the ship as moment from the point view of global deformation. And it is acting on the each stiffener as axial load from the point view of local deformation.



(3) Buckling Strength of Plate (1/7)



A ship hull is a stiffened-plate structure, the plating supported by a system of transverse or longitudinal stiffeners.

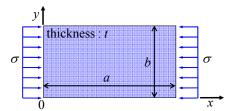
For practical design purpose, it is often assumed that the plate is simply supported at the all edges, since it gives the least critical stress and is on the safe side.

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(3) Buckling Strength of Plate (2/7)

Let us consider the rectangular plate with only supported edges as shown in this figure.



- σ : the uni-axial compressive stress
- v : Poisson's ratio
- ${\it E}$: Modulus of elasticity
- b: plate width
- t: thickness of the plate
- The equation of elastic buckling stress of the plate under uni-axial compressive stress:

$$\boxed{\frac{Et^3}{12(1-v^2)} \left(\frac{\partial^4 w}{\partial x^4} + 2 \frac{\partial^4 w}{\partial x^2 \partial y^2} + \frac{\partial^4 w}{\partial y^4} \right) + \sigma t \frac{\partial^2 w}{\partial x^2} = 0} \cdots (1)$$

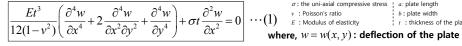
where, w = w(x, y) : deflection of the plate

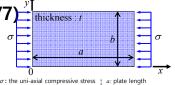
* Okumoto, Y., Design of Ship Hull Structures, pp.57-60, 2009

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(3) Buckling Strength of Plate (3/7)

The equation of elastic buckling stress of the plate under uni-axial compressive stress:





b: plate width
t: thickness of the plat

Because all four edges are simply supported, the boundary condition can be expressed in the form:

$$w(0,y) = w(a,y) = 0$$
 deformation at the edges are zero

Let us assume the following formula for the solution of the equation (1), so that the solution satisfies the boundary conditions.

$$w = f \sin\left(\frac{m\pi x}{a}\right) \cdot \sin\left(\frac{n\pi y}{b}\right) \cdots (2)$$

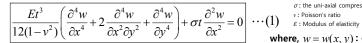
Okumoto, Y., Design of Ship Hull Structures, pp.57-60, 2009

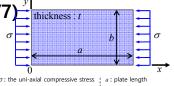
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(3) Buckling Strength of Plate (4/7)

The equation of elastic buckling stress of the plate under uni-axial compressive stress:





where, w = w(x, y): deflection of the plate

Substituting the formula (2) into the equation (1),

$$w = f \sin\left(\frac{m\pi x}{a}\right) \cdot \sin\left(\frac{n\pi y}{b}\right) \cdots (2)$$

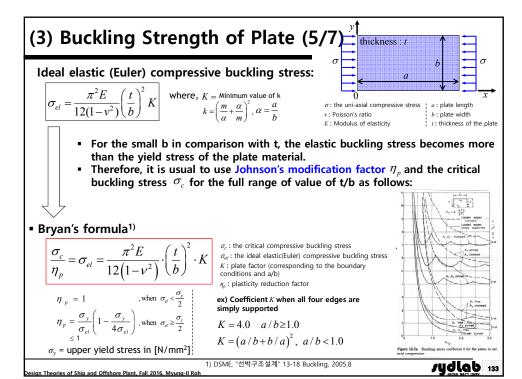
$$\sigma = \frac{Et^3}{12(1-v^2)} \frac{\pi^2}{b^2 t} \left(\frac{m}{\alpha} + n^2 \frac{\alpha}{m}\right)^2 \cdots (3) \quad \text{where, } \alpha = \frac{a}{b}$$

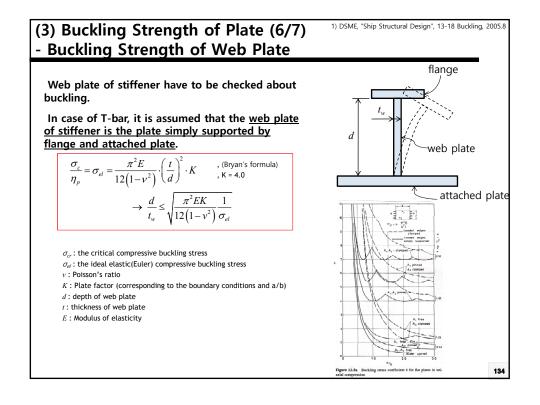
Elastic buckling stress is a minimum critical stress, therefore, we put n=1 in the equation (3),

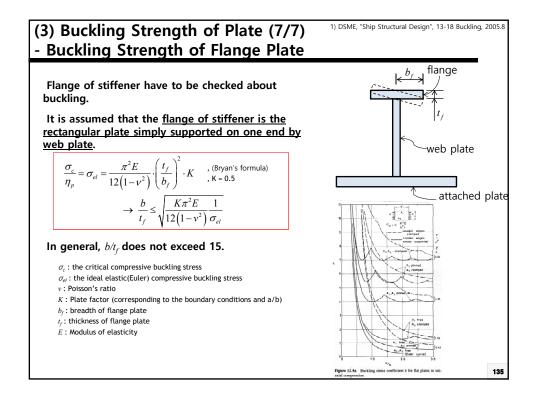
Ideal elastic (Euler) compressive buckling stress:

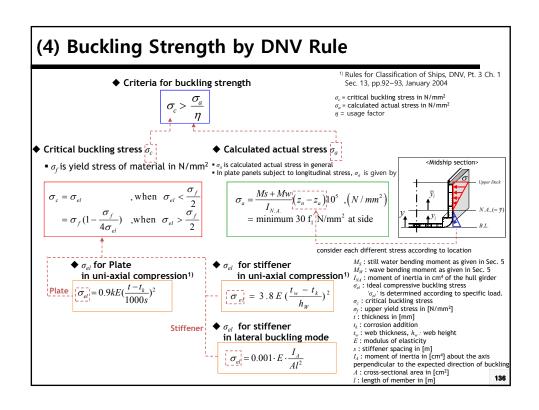
$$\sigma_{el} = \frac{\pi^2 E}{12(1-v^2)} \left(\frac{t}{b}\right)^2 K \quad \text{where, } K = \text{Minimum value of k, } k = \left(\frac{m}{\alpha} + \frac{\alpha}{m}\right)^2$$

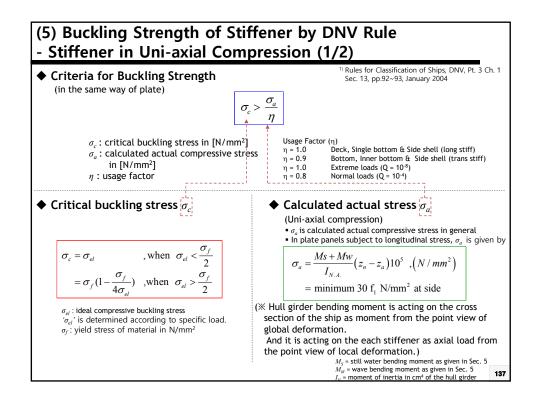
Okumoto, Y., Design of Ship Hull Structures, pp.57-60, 2009

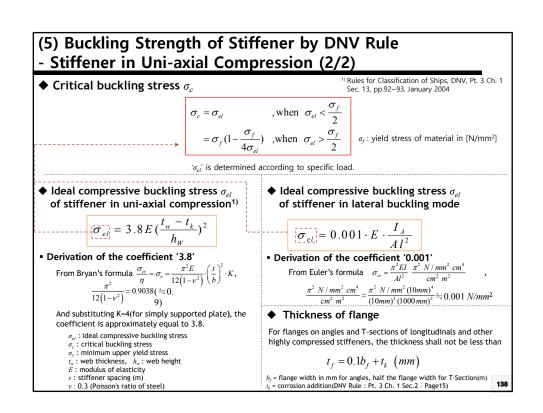


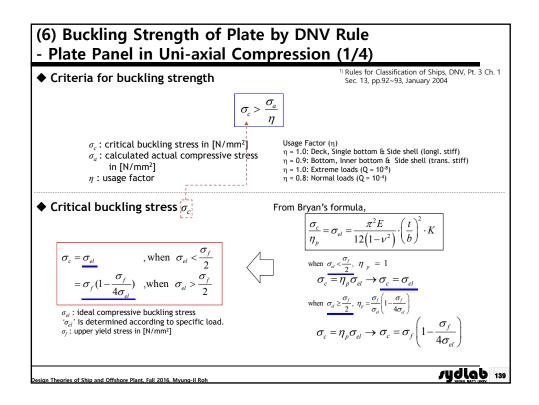


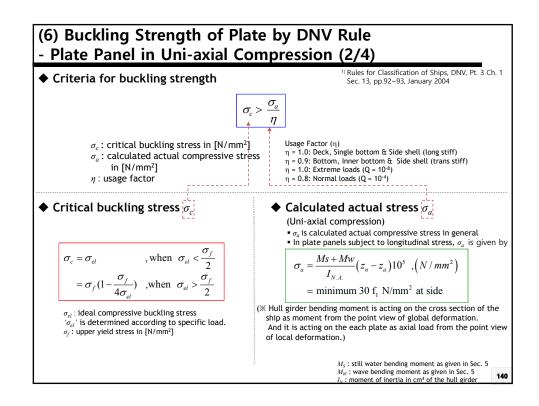










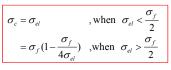


(6) Buckling Strength of Plate by DNV Rule

Plate Panel in Uni-axial Compression (3/4)

lacktriangle Critical buckling stress σ_c

Rules for Classification of Ships, DNV, Pt. 3 Ch. 1 Sec. 13, pp.92~93, January 2004



 $\sigma_{\!f}$: minimum upper yield stress of material in [N/mm²]

 σ_{el} is determined according to specific load.

• Ideal compressive buckling stress σ_{el} in uni-axial compression1)

$$|\vec{\sigma}_{el}| = 0.9 \vec{k} E \left(\frac{t - t_k}{1000 s}\right)^2$$

Derivation of the coefficient '0.9'

From Bryan's formula
$$\frac{\sigma_{cr}}{\eta} = \sigma_c = \frac{\pi^2 E}{12(1-\nu^2)} \cdot \left(\frac{t}{b}\right)^2 \cdot K$$
,
$$\frac{\pi^2}{12(1-\nu^2)} = \frac{3.141593^2}{12(1-0.3^2)} = 0.9038 \; (\rightleftharpoons 0.9)$$
 σ_{cl} : ideal compressive buckling stress

- σ_{el} : ideal compressive buckling stress σ_e : critical buckling stress σ_f : upper yield stress in N/mm² t: thickness (mm) t_e : corrosion addition E: modulus of elasticity

- s: stiffener spacing (m) v: 0.3 (Poisson's ratio of steel)

- lack factor k
- · For plating with longitudinal stiffeners (in direction of compression stress):

$$k = k_l = \frac{8.4}{\psi + 1.1}$$

· For plating with transverse stiffeners (perpendicular to compression stress):

$$k = k_s = c \left[1 + \left(\frac{s}{l} \right)^2 \right]^2 \frac{2.1}{\psi + 1.1}$$



 ψ = ratio between the smaller and the larger $\left(0 \leq \psi \leq 1\right)$ compressive stress (positive value) c =1.21 when stiffeners are angles or T sections

- =1.10 when stiffeners are bulb flats =1.05 when stiffeners are flat bars
- =1.30 when plating is supported by deep girders

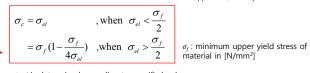
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(6) Buckling Strength of Plate by DNV Rule

- Plate Panel in Uni-axial Compression (4/4)

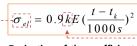
lacktriangle Critical buckling stress σ_c

Rules for Classification of Ships, DNV, Pt. 3 Ch. 1 Sec. 13, pp.92~93, January 2004



' σ_{el} ' is determined according to specific load.

 Ideal compressive buckling stress σ_{el} in uni-axial compression¹



Derivation of the coefficient '0.9'

From Bryan's formula
$$\frac{\sigma_{cr}}{\eta} = \sigma_c = \frac{\pi^2 E}{12(1-\nu^2)} \cdot \left(\frac{t}{b}\right)^2 \cdot K$$
,
$$\frac{\pi^2}{12(1-\nu^2)} = \frac{3.141593^2}{12(1-0.3^2)} = 0.9038 \ (\rightleftharpoons 0.903)$$

- $\sigma_{\it el}$: ideal compressive buckling stress
- σ_{el} : I deat compressive bucking σ_c : critical buckling stress σ_f : upper yield stress in N/mm² t: thickness (mm) t_k : corrosion addition E: modulus of elasticity

- s: stiffener spacing (m) v: 0.3 (Poisson's ratio of steel)

- - For plating with longitudinal stiffeners (in direction of compression stress): $k = k_l = \frac{8.4}{w + 1.1}$
 - For plating with transverse stiffeners (perpendicular to compression stress): $k=k_s=c\Bigg[1+\bigg(\frac{s}{l}\bigg)^2\Bigg]^2\frac{2.1}{\psi+1.1}$

Example) If $\psi = 1.0, c = 1.05, s/l = 1/10$

$$k = k_l = \frac{8.4}{1.0 + 1.1} = 4$$

Example) if
$$\psi = 1.0$$
, $c = 1.05$, $s / t = 1/10$
$$k = k_t = \frac{8.4}{1.0 + 1.1} = \frac{4}{1.0 + 1.1}$$

$$k = k_s = c \left[1 + \left(\frac{s}{t} \right)^2 \right]^2 \frac{2.1}{\psi + 1.1} = 1.05 \left[1 + \left(\frac{1}{10} \right)^2 \right]^2 \frac{2.1}{1.0 + 1.1} = \underline{1.071}$$
 Thus, the plate with longitudinal stiffeners can endure much

Thus, the plate with longitudinal stiffeners can endure much

6.5 Structural Design of Midship Section of a 3,700 TEU Container Ship

- (1) Data for Structural Design
- (2) Longitudinal Strength
- (3) Local Strength
- (4) Buckling Strength

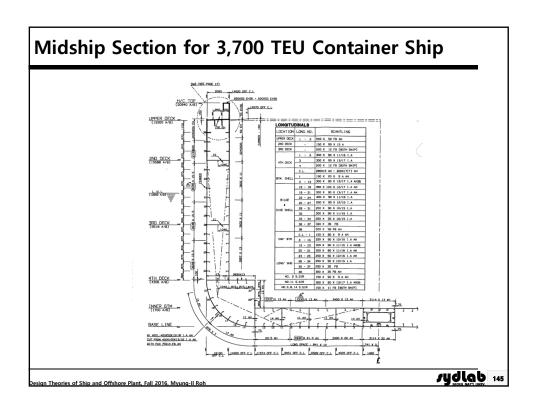
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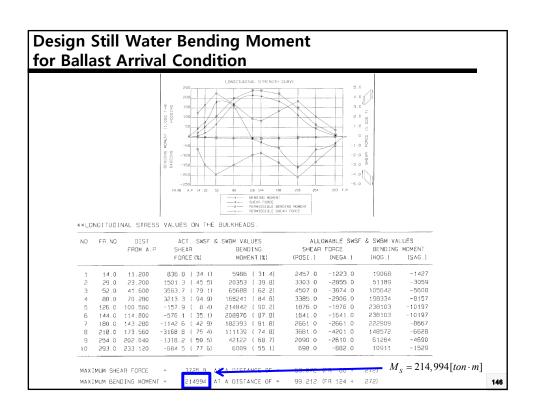
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(1) Data for Structural Design

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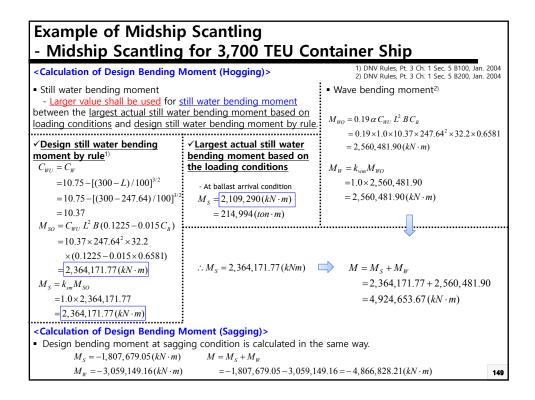


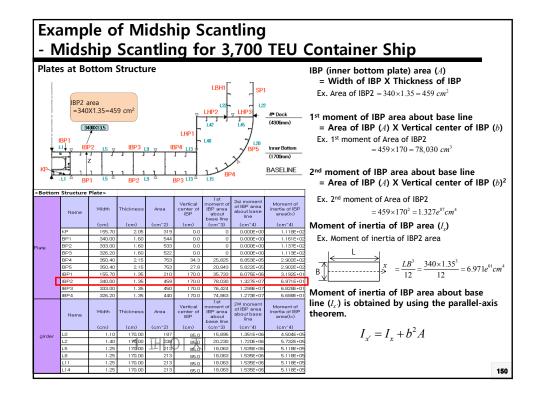


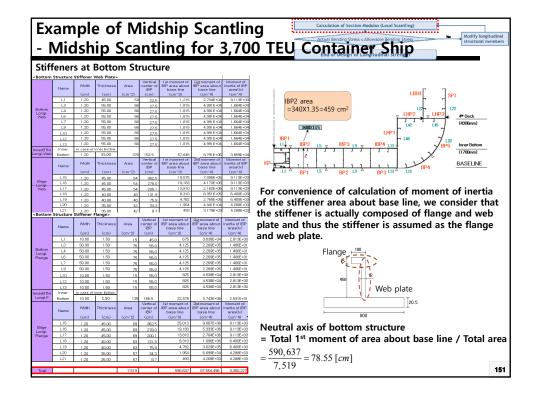
Design Still Water Bending Moment From ballast arrival **NOTES** condition, 1. DESIGN STILL WATER BENDING MOMENT IN SEAGOING CONDITION. $M_S = 214,994[ton \cdot m]$ HOGGING CONDITION : 238,000 TON-M (2,335,000 kN-M) 2. MIN. LEG LENGTH OF FILLET WELDING 4.5 EXCEPT AS SHOWN. 3. BOTH SIDES ARE SYMMETRICAL UNLESS OTHERWISE SHOWN. 4. SECTIONS ARE SHOWN IN LOOKING FORWARD AND ELEVATIONS ARE SHOWN TO PORT. 5. THE DETAILS NOT SHOWN IN THIS DRAWING ARE REFERRED TO "STRUCTURAL DETAILS FOR HULL" (DWG. NO. SF091.20) By calculating the section modulus and stress factor of the basis ship, we can assume the stress factor for the design ship. ydlab 147

(2) Longitudinal Strength ydlab 148

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Example of Midship Scantling

- Midship Scantling for 3,700 TEU Container Ship

Calculation of moment of inertia of sectional area from neutral axis

Area, neutral axis, 1st moment & 2nd moment about baseline, and moment of inertia of side structure, bulkhead structure, deck structure are calculated in the same way and the results are as follows:

Structure	Area	Neutral axis	1st moment of area about baseline	2nd moment of area about baseline	Moment of inertia of area	
Bottom	7,519	79	5.906E+05	8.755E+08	2.627E+07	
Side	3,135	1,158	3.630E+06	4.203E+09	1.261E+08	
Bulkhead	5,273	1,250	6.592E+06	8.242E+09	2.472E+08	
Deck	2,200	2,130	5.015E+06	1.208E+10	3.624E+08	
Total	18.127		1.583E+07	2.540E+10	7.620E+08	

Vertical location of neutral axis of midship section from baseline (\bar{h}) is calculated by using the above table.

$$\overline{h}$$
 = Total 1st moment of area about baseline / Total area
$$= \frac{1.583e^{07}}{18,127} = 873.2[cm]$$

Moment of inertia of area about neutral axis of midship section:

$$I_{Base,Total} = I_{N.A.,Total} + \overline{h}^2 \sum A_i$$

$$(Parallel-axis theorem)$$

$$I_{N.A.,Total} = I_{Base,Total} - \overline{h}^2 \sum A_i$$

$$= \sum (I_{Local,i} + A_i h_i^2) - \overline{h}^2 \sum A_i$$

$$= \sum I_{Local,i} + \sum A_i h_i^2 - \overline{h}^2 \sum A_i$$

$$= \sum I_{Local,i} + \sum A_i h_i^2 - \overline{h}^2 \sum A_i$$

$$= \sum I_{Local,i} + \sum A_i h_i^2 - \overline{h}^2 \sum A_i$$

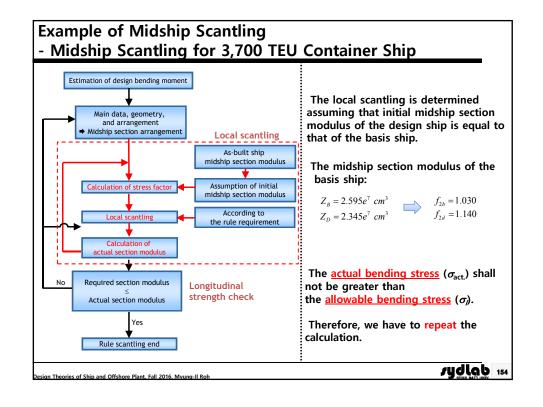
$$= (7.620e^{08} + 2.540e^{10}) - 873.2^2 \times 18,127 = 1.234e^{10} [cm^4]$$

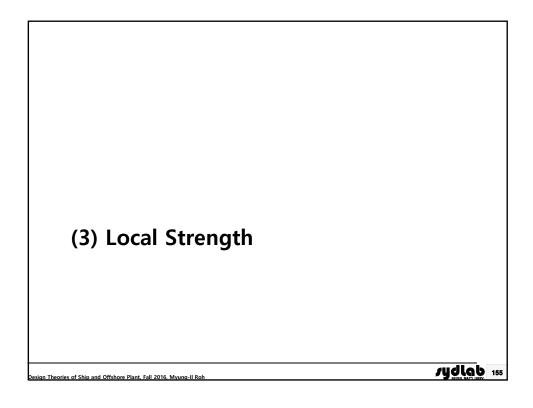
$$A_i \text{ area of structural member (cm)}$$

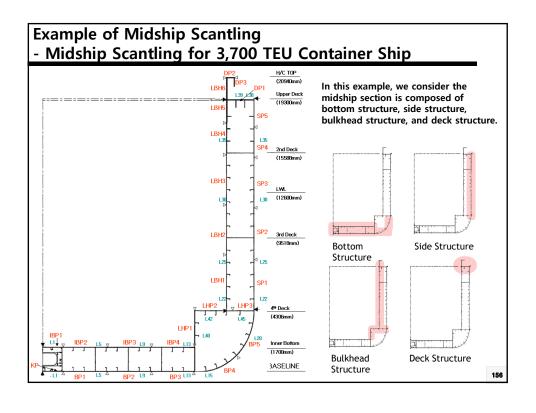
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Example of Midship Scantling - Midship Scantling for 3,700 TEU Container Ship $\frac{f_{2b,2d}}{Z_{b,d}} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$ 1) Assume section modulus • Bottom stress factor of the basis ship Deck stress factor of the basis ship $Z_B = 2.595e^7 \text{ cm}^3 \qquad , f_{2b} = 1.030$ $Z_D = 2.345e^7 \ cm^3 \ , f_{2d} = 1.140$ Deck section modulus ② Actual section modulus $Z_{\scriptscriptstyle D} = 2 \times I \, / \, y_{\scriptscriptstyle D} \,$ (port & starboard) Bottom section modulus $= 2 \times 1.234e^{10} / 1,226.8$ $Z_{\scriptscriptstyle B} = 2 \times I \, / \, y_{\scriptscriptstyle B}$ (port & starboard) $= 2.012e^{7} [cm^{3}]$ $=2\times1.234e^{10}/873.2$ $(y_D$: Vertical distance from N.A to deck=2094-873.2 = 1,226.8 cm) $=2.826e^{7}[cm^{3}]$ Because the section modulus at deck is smaller than $(y_B$: Vertical distance from N.A to bottom = 873.2cm) that of the basis ship, the stress factor will be increased. Because the section modulus at bottom is larger than that of the basis ship, the stress factor should be However, if HT-36 is used, then the stress factor can be ■ Bottom Stress Factor ■ Deck Stress Factor $f_{2b} = \frac{5.7(M_S + M_W)}{5.5}$ $f_{2d} = \frac{5.7(M_S + M_W)}{}$ $f_1 \times Z_B$ $f_1 \times Z_D$ $=\frac{5.7\times4,924,653.67}{5.7}=0.993$ $=\frac{5.7\times4,924,653.67}{3}=1.004$ $= \frac{1.0 \times 2.826e^7 - 0.993}{1.39 \times 2.012e^7} = 1.004$ 4 Because the allowable stress is increased, 3 Because the stress factor (f_{2b}) is decreased, the required section modulus is decreased. So, the allowable stress is increased. we can reduce the size of the structure member. $\sigma = 225 f_1 - 130 f_{2b} - 0.7 \sigma_{db}$ $Z = \frac{83l^2 spw_k}{s} [cm^3] \begin{vmatrix} e.g., \text{ Required section modulus} \\ \text{for longitudinals at inner bottom} \end{vmatrix}$ e.g., Allowable stress for longitudinals at inner bottom





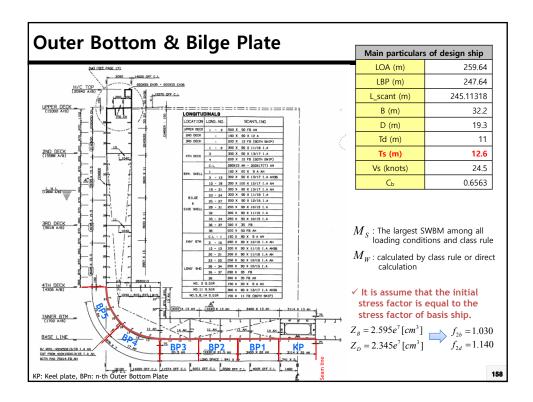


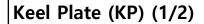
Example of Local Scantling

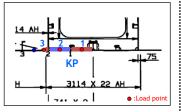
- **☑** Outer Bottom & Bilge plate
- **☑** Outer Bottom Longitudinals
- **☑** Inner Bottom Plate
- **☑** Inner Bottom Longitudinals
- **☑** Side Shell Plate
- **☑** Side Shell Longitudinals
- **☑** Deck Plate
- **☑** Deck Longitudinals
- **☑** Longitudinal Bulkhead Plate
- **☑** Longitudinal Bulkhead Longitudinals

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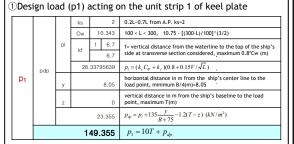
- ✓ Keel plate is composed of the three unit strips.
- ✓ Load point of the unit strip:
 - 1, 2: Midpoint
 - 3: Point nearest the midpoint
- ✓ Calculate the required thickness of each unit strip. And the thickest value shall be used for thickness of the plate.
- ✓ The material of keel plate of basis ship (NV-32) is used for that of design ship. $(f_1 = 1.28)$

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✓ Design Load

	DNV Rules, P	t. 3 Ch. 1 Sec. 6 Table B1, Jan. 2004
Structure	Load Type	$p(kN/m^2)$
Outer bottom	Sea pressure	$p_1 = 10T + p_{dp}$

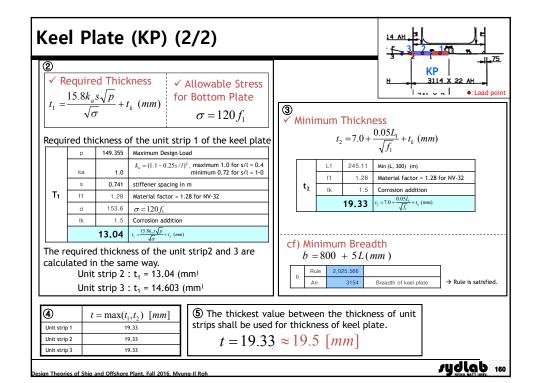
: Design load acting on the keel plate is only the sea pressure.

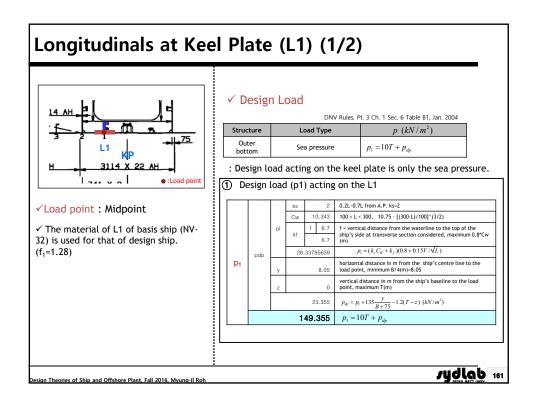


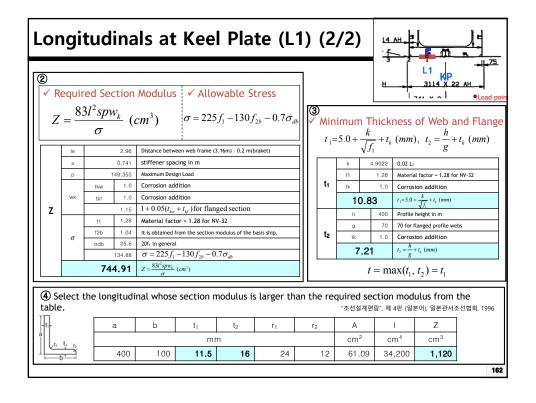
✓ The design loads of the unit strip 2 and 3 are calculated in the same way.

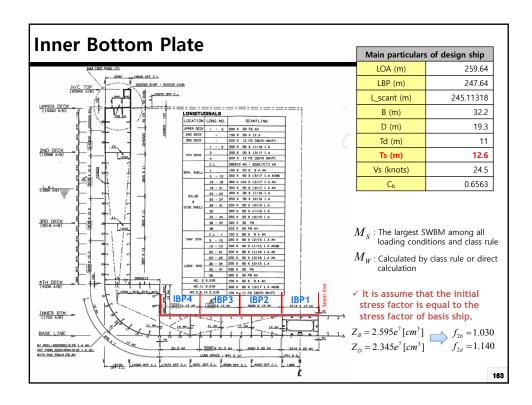
Unit strip 2: $p_1 = 149.355 (kN/m^2)$ Unit strip 3: $p_1 = 149.355 \text{ (kN/m}^2\text{)}$

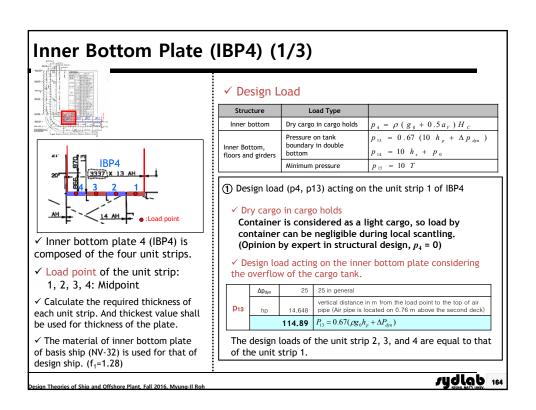
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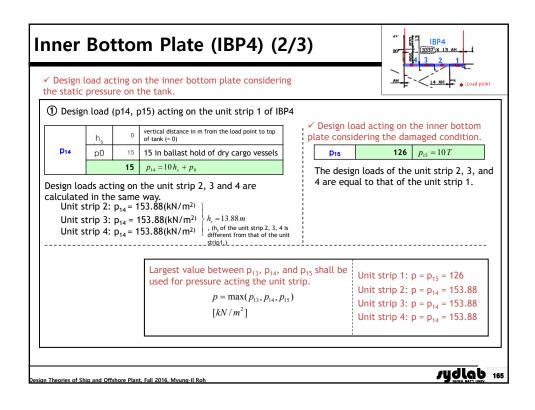


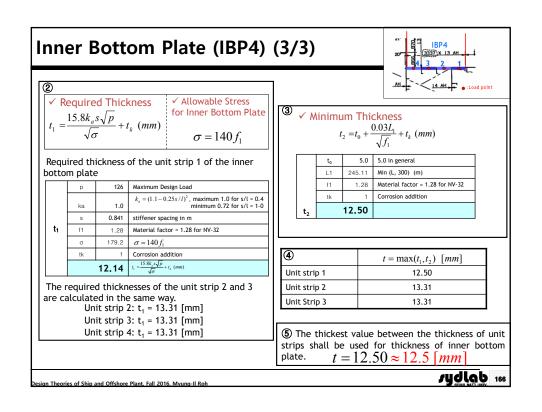


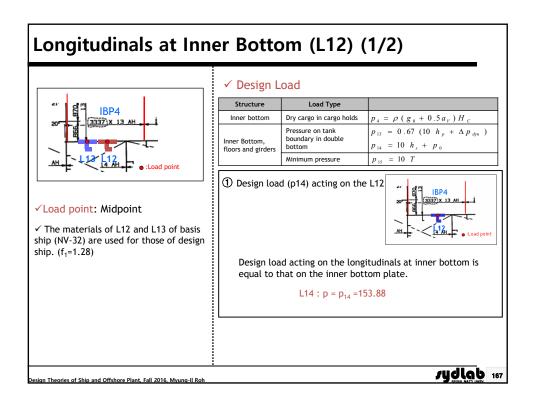


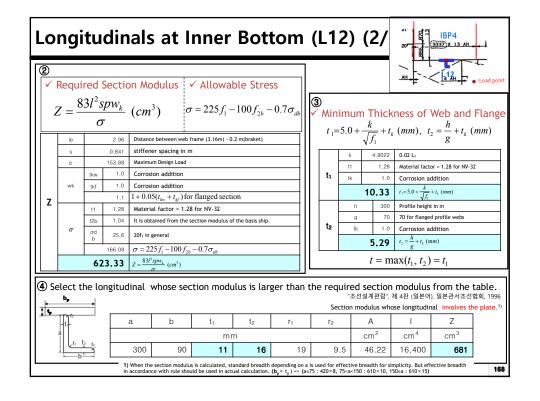


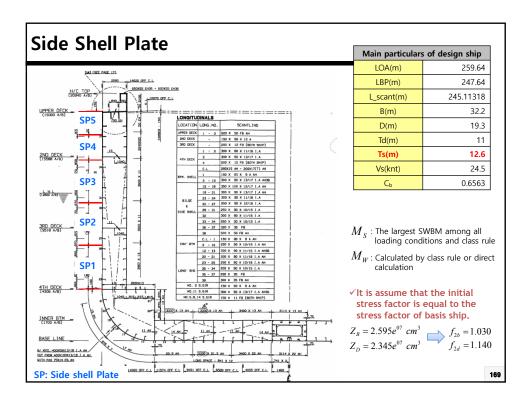


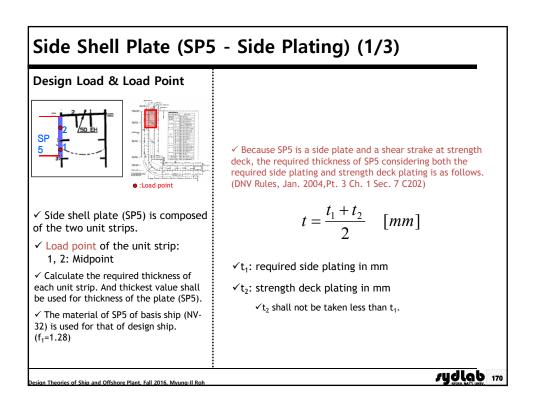


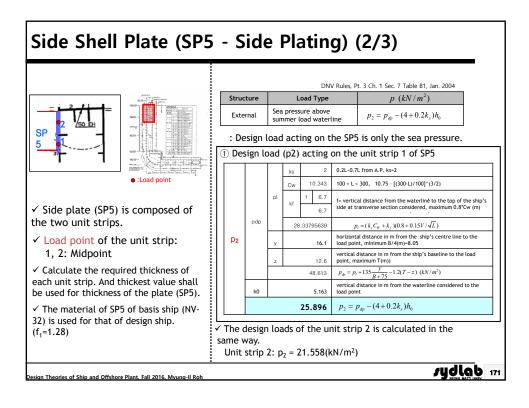


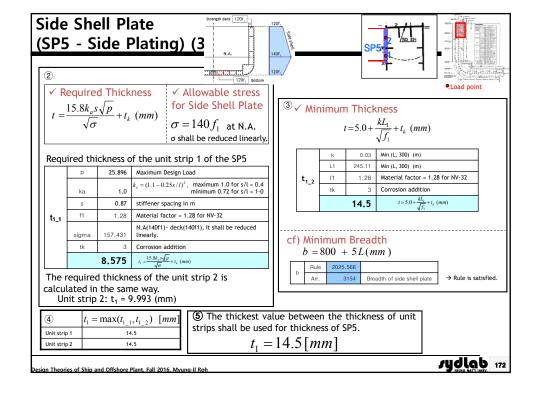


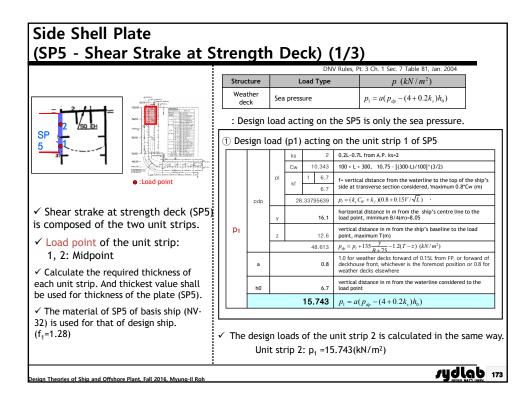


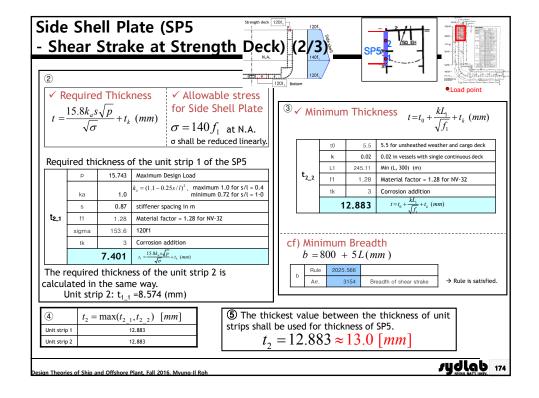






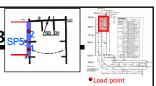






Side Shell Plate

(SP5 - Shear Strake at Strength Deck) (3/3



✓ Side shell plate(SP5)

$$t = \frac{t_1 + t_2}{2} \quad [mm]$$

 \checkmark t1 : required side plating in mm $t_1 = 14.5$

m < t2 : strength deck plating in mm $t_2 = 13.0$

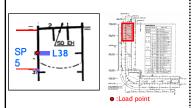
 \checkmark t2 shall not be taken less than t1. $\therefore t_2 = 14.5$

$$\therefore t = \frac{t_1 + t_2}{2} = \frac{14.5 + 14.5}{2} = 14.5 [mm]$$

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Longitudinals at Side Shell Plate (L38 - Deck Structure) (1/4)



✓ Load point: Midpoint

✓ The material of L38 of basis ship (NV-32) is used for that of design ship. $(f_1=1.28)$

✓ L38 to be considered is the longitudinals located between the side structure and deck structure.

: Design load acting on the L_{38} is only the sea pressure.

 \bigcirc Design load (p2) acting on the L₃₈ of the SP5

			_		
P ₂		lq	ks 2		0.2L-0.7L from A.P. ks=2
			Cw 10.343		3 100 < L < 300, 10.75 - [(300-L)/100]^(3/2)
			kf	f 6.	f= vertical distance from the waterline to the top of the ship's
			KI	6.7 side at transverse section considere	, side at transverse section considered, maximum 0.8*Cw (m)
	pdp		28.33795639		$p_i = (k_s C_W + k_f)(0.8 + 0.15V / \sqrt{L})$
		у	16.1		horizontal distance in m from the ship's centre line to the load point, minimum $B/4(m)=8.05$
		z	12.6		vertical distance in m from the ship's baseline to the load point, maximum T(m))
		48.613			$p_{\phi} = p_t + 135 \frac{y}{B + 75} - 1.2(T - z) (kN/m^2)$
	h0	5.598			vertical distance in m from the waterline considered to the load point
	23.982				$p_2 = p_{dp} - (4 + 0.2k_s)h_0$

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