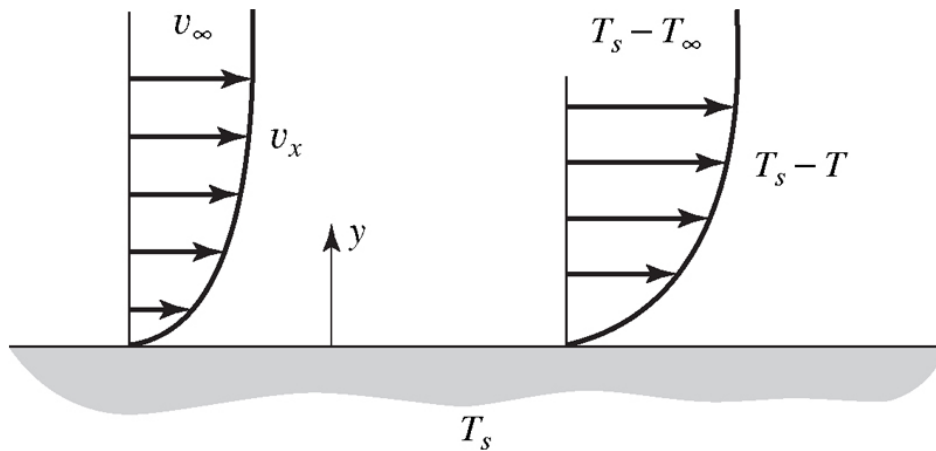


convective heat transfer



$$\rho \frac{D\mathbf{v}}{Dt} = -\nabla P + \mu \nabla^2 \mathbf{v} + \rho \mathbf{g}$$

$$\rho c_p \frac{DT}{Dt} = -k \nabla^2 T + \dot{q} + \boldsymbol{\tau} : \nabla \mathbf{v}$$

momentum diffusivity : $\nu \equiv \frac{\mu}{\rho}$

thermal diffusivity : $\alpha \equiv \frac{k}{\rho c_p}$

$$\text{Pr} \equiv \frac{\nu}{\alpha} = \frac{\mu c_p}{k}$$

$$h(T_s - T_\infty) = -k \left. \frac{\partial (T - T_s)}{\partial y} \right|_{y=0}$$

$$\frac{hL}{k} = \frac{\partial (T_s - T) / \partial y|_{y=0}}{(T_s - T_\infty) / L}$$

$$\text{Nu} = \frac{hL}{k} \equiv \frac{\text{conductive thermal resistance}}{\text{convective thermal resistance}}$$

dimensional analysis

forced convection

in a closed conduit

Variable	Symbol	Dimensions
Tube diameter	D	L
Fluid density	ρ	M/L^3
Fluid viscosity	μ	M/Lt
Fluid heat capacity	c_p	Q/MT
Fluid thermal conductivity	k	Q/tLT
Velocity	v	L/t
Heat-transfer coefficient	h	Q/tL^2T

of variables; 7

of dimensions; 5

of ranks; 4

of independent dimensionless groups; 3

$$\pi_1 = D^a k^b \mu^c v^d \rho$$

$$\pi_2 = D^e k^f \mu^g v^h c_p$$

$$\pi_3 = D^i k^j \mu^k v^l h$$

$$\pi_1 = \frac{Dv\rho}{\mu} = \text{Re}$$

$$\pi_2 = \frac{\mu c_p}{k} = \text{Pr}$$

$$\pi_3 = \frac{hD}{k} = \text{Nu}$$

$$\text{Nu} = f_1(\text{Re}, \text{Pr})$$

natural convection

$$\rho = \rho_0(1 - \beta\Delta T)$$

$$F_{\text{buoyant}} = (\rho_0 - \rho)g$$

$$F_{\text{buoyant}} = \beta g \rho_0 \Delta T$$

Variable	Symbol	Dimensions
Significant length	L	L
Fluid density	ρ	M/L^3
Fluid viscosity	μ	M/Lt
Fluid heat capacity	c_p	Q/MT
Fluid thermal conductivity	k	Q/LtT
Fluid coefficient of thermal expansion	β	$1/T$
Gravitational acceleration	g	L/t^2
Temperature difference	ΔT	T
Heat-transfer coefficient	h	Q/L^2tT

of variables; 9

of dimensions; 5

of ranks; 5

of independent dimensionless groups; 4

$$\pi_1 = L^a \mu^b k^c \beta^d c_p$$

$$\pi_2 = L^f \mu^g k^h g^j \rho$$

$$\pi_3 = L^k \mu^l k^m \beta^n g^o \Delta T$$

$$\pi_4 = L^p \mu^q k^r \beta^s g^t h$$

$$\pi_1 = \frac{\mu c_p}{k} = \text{Pr}$$

$$\pi_2 = \frac{L^3 g \rho^2}{\mu^2} \quad \pi_3 = \beta \Delta T$$

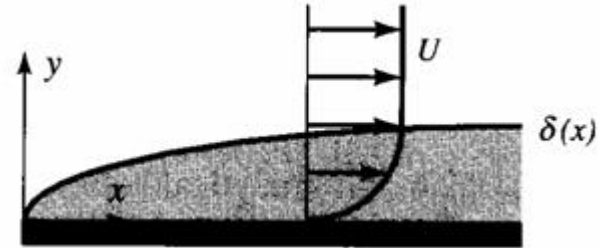
$$\pi_4 = \frac{hL}{k} = \text{Nu}$$

$$\text{Gr} \equiv \frac{\beta g \rho^2 L^3 \Delta T}{\mu^2}$$

$$\text{Nu} = f_3(\text{Gr}, \text{Pr})$$

Laminar boundary layer

No matter how turbulent the flow is far from the surface



Negligible viscous effect outside the boundary layer

$$\frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0$$

$$\rho \left(u_x \frac{\partial u_y}{\partial x} + u_y \frac{\partial u_y}{\partial y} \right) = -\frac{\partial p}{\partial y} + \mu \left(\frac{\partial^2 u_y}{\partial x^2} + \frac{\partial^2 u_y}{\partial y^2} \right)$$

$$u_y = 0; \quad \frac{\partial p}{\partial y} = 0 \quad \text{within the boundary layer}$$

$$\rho \left(u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} \right) = -\cancel{\frac{\partial p}{\partial x}} + \mu \left(\cancel{\frac{\partial^2 u_x}{\partial x^2}} + \frac{\partial^2 u_x}{\partial y^2} \right)$$

$$\frac{\partial u_x}{\partial x} + \frac{\partial u_y}{\partial y} = 0 \quad \rho \left(u_x \frac{\partial u_x}{\partial x} + u_y \frac{\partial u_x}{\partial y} \right) = \mu \frac{\partial^2 u_x}{\partial y^2}$$

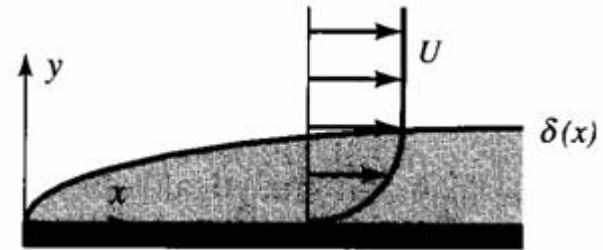
$$\frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2} = \nu \frac{\partial^3 \psi}{\partial y^3}$$

Boundary conditions

$$u_x = u_y = 0 \quad \text{on} \quad y = 0, \text{ for } x > 0$$

$$u_x = U \quad \text{for } x < 0$$

$$u_x \rightarrow U \quad \text{for } y \rightarrow \infty$$



$$\frac{\partial \psi}{\partial y} = -\frac{\partial \psi}{\partial x} = 0 \quad \text{on} \quad y = 0, \text{ for } x > 0$$

$$\frac{\partial \psi}{\partial y} \rightarrow U \quad \text{for } x < 0, \text{ and } y \rightarrow \infty$$

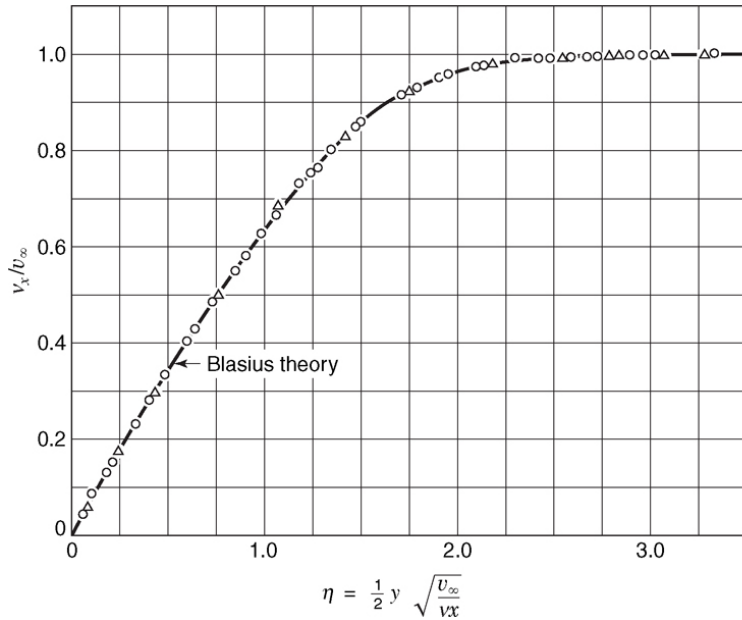
Blasius' similarity transformation

$$f(\eta) \equiv \frac{\psi}{(U\nu x)^{1/2}} \quad \eta \equiv \frac{y}{2} \left(\frac{U}{\nu x} \right)^{1/2}$$

$$f''' + ff'' = 0$$

$$f = f' = 0 \quad \text{at} \quad \eta = 0$$

$$f' \rightarrow 2 \quad \text{as} \quad \eta \rightarrow \infty$$



U_x becomes nearly U
along the boundary layer $y = \delta(x)$

$$\delta \left(\frac{U}{\nu x} \right)^{1/2} = 5$$

Boundary layer thickness

Friction coeff.

$$C_f \equiv \frac{-\mu(\partial u_x / \partial y)_{y=0}}{\frac{1}{2} \rho U^2} \equiv 0.664 \left(\frac{\nu}{xU} \right)^{1/2} = \frac{0.664}{(\text{Re}_x)^{1/2}}$$

Shear force

$$F_s = W \int_0^L -\mu \left(\frac{\partial u_x}{\partial y} \right)_{y=0} dx \quad \frac{F_s / LW}{\frac{1}{2} \rho U^2} \equiv \bar{C}_f = \frac{1.328}{(UL/\nu)^{1/2}} = \frac{1.328}{(\text{Re}_L)^{1/2}}$$

Local Reynolds number

$$\text{Re}_x \equiv \frac{xU\rho}{\mu}$$



$$\left. \frac{\partial v_x}{\partial y} \right|_{y=0} = v_\infty \left[\frac{0.332}{x} \text{Re}_x^{1/2} \right]$$

laminar boundary layer; Blasius

$$\frac{\partial T}{\partial t} + v_x \frac{\partial T}{\partial x} + v_y \frac{\partial T}{\partial y} = \alpha \left(\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} \right)$$

$$v_x \frac{\partial v_x}{\partial x} + v_y \frac{\partial v_x}{\partial y} = \nu \frac{\partial^2 v_x}{\partial y^2}$$

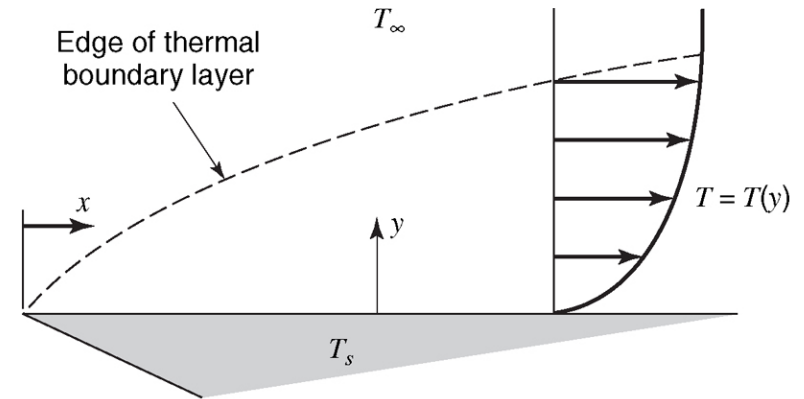
$$\frac{v_x}{v_\infty} = \frac{v_y}{v_\infty} = 0 \quad \text{at } y = 0$$

$$\frac{v_x}{v_\infty} = 1 \quad \text{at } y = \infty$$

$$v_x \frac{\partial T}{\partial x} + v_y \frac{\partial T}{\partial y} = \alpha \frac{\partial^2 T}{\partial y^2} \quad \text{for Pr}=1$$

$$\frac{T - T_s}{T_\infty - T_s} = 0 \quad \text{at } y = 0$$

$$\frac{T - T_s}{T_\infty - T_s} = 1 \quad \text{at } y = \infty$$



$$\left. \frac{\partial v_x}{\partial y} \right|_{y=0} = v_\infty \left[\frac{0.332}{x} \text{Re}_x^{1/2} \right]$$

$$\left. \frac{\partial T}{\partial y} \right|_{y=0} = (T_\infty - T_s) \left[\frac{0.332}{x} \text{Re}_x^{1/2} \right]$$

$$\frac{q_y}{A} = h_x (T_s - T_\infty) = -k \left. \frac{\partial T}{\partial y} \right|_{y=0}$$

$$\frac{h_x x}{k} = \text{Nu}_x = 0.332 \text{Re}_x^{1/2}$$

effect of Pr; Pohlhausen

$$\frac{\delta}{\delta_t} = \text{Pr}^{1/3}$$

$$\left. \frac{\partial T}{\partial y} \right|_{y=0} = (T_\infty - T_s) \left[\frac{0.332}{x} \text{Re}_x^{1/2} \right]$$

$$\left. \frac{\partial T}{\partial y} \right|_{y=0} = (T_\infty - T_s) \left[\frac{0.332}{x} \text{Re}_x^{1/2} \text{Pr}^{1/3} \right]$$

$$\frac{h_x x}{k} = \text{Nu}_x = 0.332 \text{Re}_x^{1/2}$$

$$\frac{h_x x}{k} = \text{Nu}_x = 0.332 \text{Re}_x^{1/2} \text{Pr}^{1/3}$$

$$\frac{hL}{k} = \text{Nu}_L = 0.664 \text{Re}_L^{1/2} \text{Pr}^{1/3}$$

fluid properties are evaluated at
the film temperature

$$T_f = \frac{T_s + T_\infty}{2}$$

; laminar boundary layer over a flat plate

(von Karman) $\text{Nu}_x = 0.36 \text{Re}_x^{1/2} \text{Pr}^{1/3}$

energy and momentum transfer analogy

Reynolds analogy ; for $Pr=1$

$$h(T_s - T_\infty) = -k \left. \frac{\partial}{\partial y} (T - T_s) \right|_{y=0}$$

$$\left. \frac{d v_x}{dy} \right|_{y=0} = \left. \frac{d}{dy} \left(\frac{T - T_s}{T_\infty - T_s} \right) \right|_{y=0} \quad h = \left. \frac{\mu c_p}{v_\infty} \frac{dv_x}{dy} \right|_{y=0}$$

coefficient of skin friction

$$C_f \cong \frac{\tau_0}{\rho v_\infty^2 / 2} = \frac{2\mu}{\rho v_\infty^2} \left. \frac{dv_x}{dy} \right|_{y=0} \quad h = \frac{C_f}{2} (\rho v_\infty c_p)$$

$$\boxed{\frac{h}{\rho v_\infty c_p} \equiv St = \frac{C_f}{2}}$$

Colburn analogy ; $0.5 < Pr < 50$, no form drag

$$\boxed{St Pr^{2/3} = \frac{C_f}{2}}$$

$$j_H = \frac{C_f}{2} \quad j_H = St Pr^{2/3}$$

Colburn j-factor

turbulent flow

$$\bar{v}_x \Big|_{y \pm L} - \bar{v}_x \Big|_y = \pm L \frac{d\bar{v}_x}{dy} \quad v'_x = \pm L \frac{d\bar{v}_x}{dy} \quad \tau_{yx}^{turb} = -\overline{\rho v'_x v'_y}$$

$$T \Big|_{y \pm L} - T \Big|_y = \pm L \frac{dT}{dy} \Big|_y \quad T' = \pm L \frac{d\bar{T}}{dy} \quad \frac{q_y}{A} \Big|_{turb} = \rho c_p \overline{(v'_y T')}$$

Prandtl analogy

$$St = \frac{C_f / 2}{1 + 5\sqrt{C_f / 2}(\text{Pr} - 1)}$$

von Karman analogy

$$St = \frac{C_f / 2}{1 + 5\sqrt{C_f / 2} \left\{ \text{Pr} - 1 + \ln \left[1 + \frac{5}{6}(\text{Pr} - 1) \right] \right\}}$$

convective heat transfer correlations

natural convection;

vertical plates, vertical cylinders,
horizontal plates, horizontal cylinders,
spheres, rectangular enclosures

forced convection for internal flow;

laminar flow, turbulent flow

forced convection for external flow;

flow parallel to plane surfaces, cylinders in crossflow,
single spheres, tube banks in crossflow,

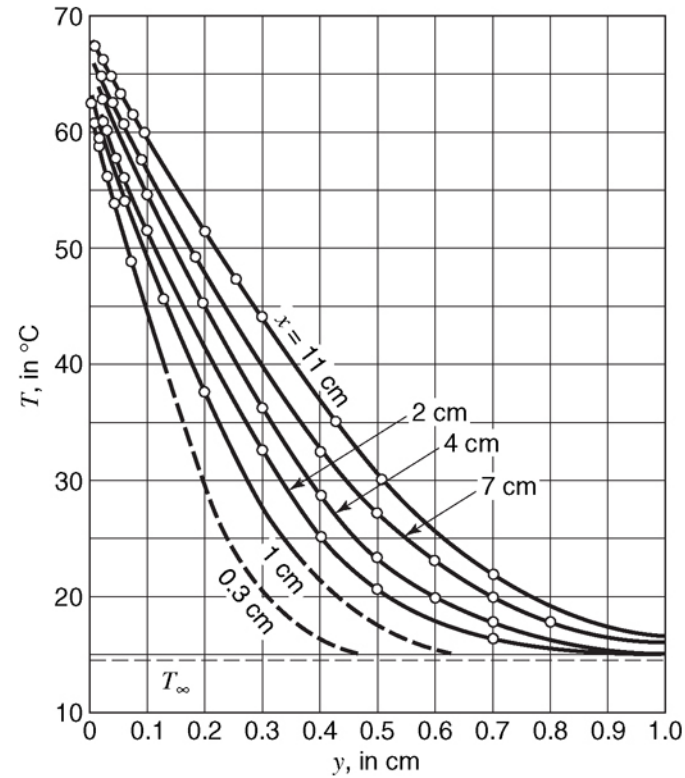
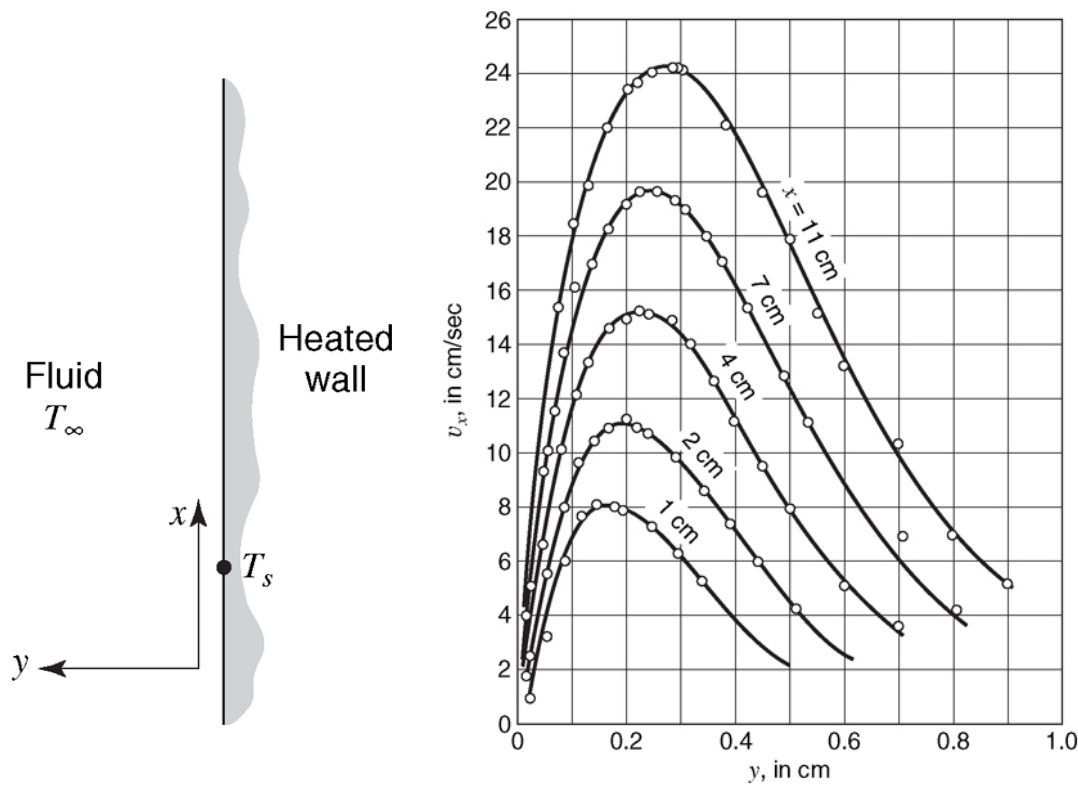
special considerations;

whether to evaluate fluid properties at bulk or film temperature

what significant length is used

what is the allowable Pr, Re range for a given set of data

natural convection; vertical plates



12.5cm high
 $T_s=65\text{C}$
 $T_{inf}=15\text{C}$

$$Nu = \left\{ 0.825 + \frac{0.387 Ra_L^{1/6}}{\left[1 + (0.492 / Pr)^{9/16} \right]^{8/27}} \right\}^2$$

$$Ra = Pr \cdot Gr$$

$$Pr = \frac{\mu c_p}{k}$$

$$Gr \equiv \frac{\beta g \rho^2 L^3 \Delta T}{\mu^2}$$

forced convection for internal flow

laminar flow; Graetz solution

$$v_x = 2v_{\text{avg}} \left[1 - \left(\frac{r}{R} \right)^2 \right] \quad v_x \frac{\partial T}{\partial x} = \alpha \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial T}{\partial r} \right) \right]$$

$$2v_{\text{avg}} \left[1 - \left(\frac{r}{R} \right)^2 \right] \frac{\partial T}{\partial x} = \alpha \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial T}{\partial r} \right) \right]$$

$$T = T_e \quad \text{at } x = 0 \quad \text{for } 0 \leq r \leq R$$

$$T = T_s \quad \text{at } x > 0, \quad r = R$$

$$\frac{\partial T}{\partial r} = 0 \quad \text{at } x > 0, \quad r = 0$$

$$\frac{T - T_e}{T_s - T_e} = \sum_{n=0}^{\infty} c_n f \left(\frac{r}{R} \right) \exp \left[-\beta_n^2 \frac{\alpha}{R v_{\text{avg}}} \frac{x}{R} \right]$$

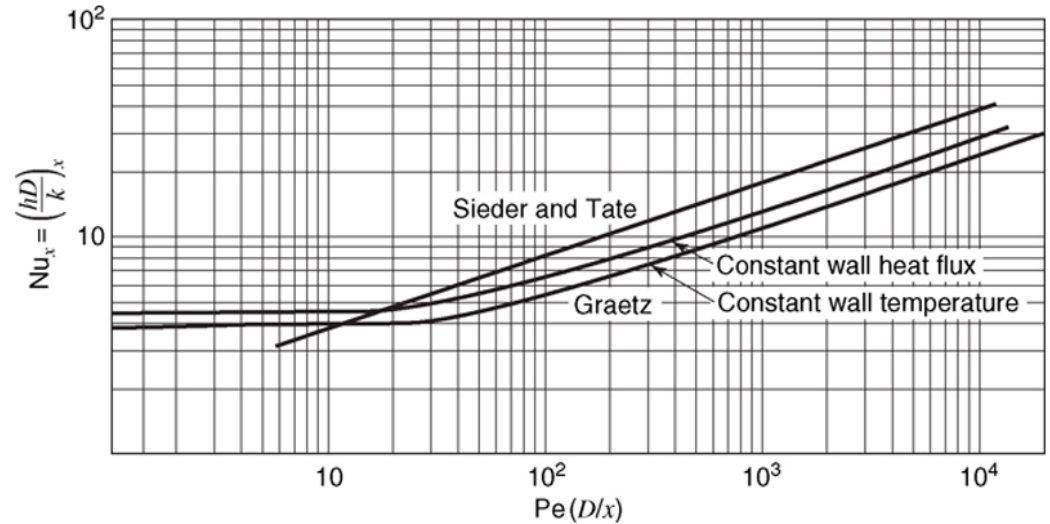
$$\frac{4}{\text{RePr} D / x} = \frac{4x / D}{\text{Pe}}$$

$$\text{Gz} \equiv \frac{\pi D}{4 x} \text{Pe}$$

$$\frac{T - T_e}{T_s - T_e} = \sum_{n=0}^{\infty} c_n f\left(\frac{r}{R}\right) \exp\left[-\beta_n^2 \frac{\alpha}{Rv_{\text{avg}}} \frac{x}{R}\right]$$

$$\text{Nu}_x = 4.364 \quad \text{for } q/A_{\text{wall}} = \text{constant}$$

$$\text{Nu}_x = 3.658 \quad \text{for } T_{\text{wall}} = \text{constant}$$



Sieder and Tate

$$\text{Nu}_D = 1.86 \left(\text{Pe} \frac{D}{L} \right)^{1/3} \left(\frac{\mu_b}{\mu_w} \right)^{0.14}$$

turbulent flow; Dittus and Boelter

$$\text{Nu}_D = 0.023 \text{Re}_D^{0.8} \text{Pr}^n$$

1. $n=0.4$ if the fluid is being heated, $n=0.3$ if the fluid is being cooled
2. all fluid properties are evaluated at the arithmetic-mean bulk temperature
3. $\text{Re} > 10^4$
4. $0.7 < \text{Pr} < 100$
5. $L/D > 60$

Colburn; $\text{St} = 0.023 \text{Re}_D^{-0.2} \text{Pr}^{-2/3}$

Sieder and Tate ; $\text{St} = 0.023 \text{Re}_D^{-0.2} \text{Pr}^{-2/3} \left(\frac{\mu_b}{\mu_w} \right)^{0.14}$

forced convection for external flow

flow parallel to plane surfaces

laminar;
$$\text{Nu}_x = 0.332 \text{Re}_x^{1/2} \text{Pr}^{1/3}$$

$$\text{Nu}_L = 0.664 \text{Re}_L^{1/2} \text{Pr}^{1/3}$$

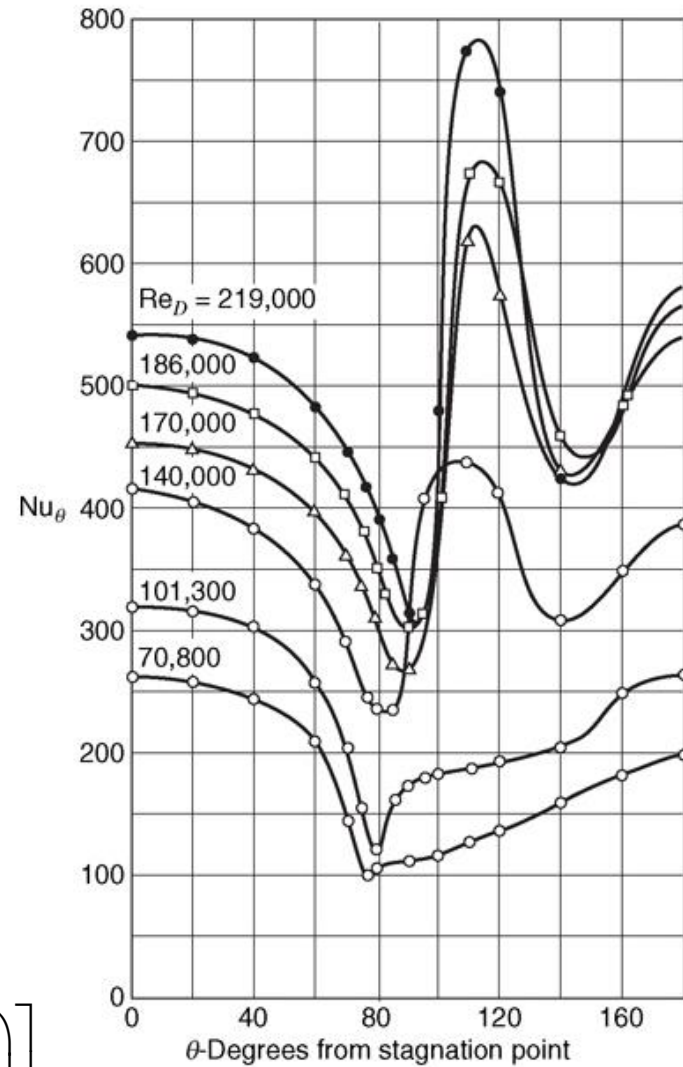
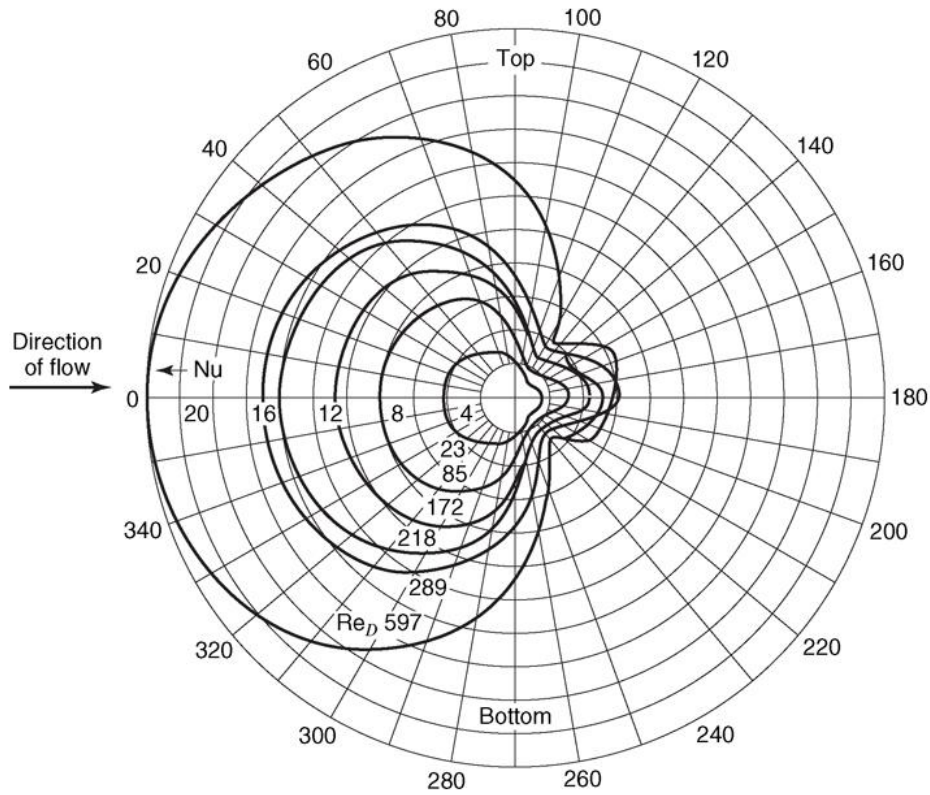
turbulent;
$$\text{St}_x \text{Pr}^{2/3} = \frac{C_{fx}}{2} \quad (\text{Colburn analogy})$$

$$C_{fx} = \frac{0.0576}{\text{Re}_x^{1/5}} \quad (\text{von Karman})$$

$$\text{Nu}_x = 0.0288 \text{Re}_x^{4/5} \text{Pr}^{1/3}$$

$$\text{Nu}_L = 0.036 \text{Re}_L^{4/5} \text{Pr}^{1/3}$$

cylinders in crossflow



$$Nu_D = 0.3 + \frac{0.62 Re_D^{1/2} Pr^{1/3}}{\left[1 + (0.4 / Pr)^{2/3}\right]^{1/4}} \left[1 + \left(\frac{Re_D}{282,000}\right)\right]$$