SEOUL NATIONAL UNIVERSITY SCHOOL OF MECHANICAL AND AEROSPACE ENGINEERING

SYSTEM CONTROL	Fall 2014
HW #5	Assigned: November 4 (Tu)
	Due: November 11 (Tu)

1. Consider the unity-feedback system with the open-loop transfer function :

$$G(s) = \frac{10}{s+1}$$

Obtain the steady-state output of the system when it is subjected to each to the following inputs:

(a) $r(t) = \sin(t + 30^{\circ})$

Sol) $C_{ss}(t) = 0.905 \sin(t + 24.8^{\circ})$

(b) $r(t) = 2\cos(2t - 45^\circ)$

Sol) $C_{ss}(t) = 1.79\cos(2t - 55.3^{\circ})$

(c) $r(t) = \sin(t + 30^\circ) - 2\cos(2t - 45^\circ)$

Sol) $C_{ss}(t) = 0.905 \sin(t + 24.8^\circ) - 1.79 \cos(2t - 55.3^\circ)$

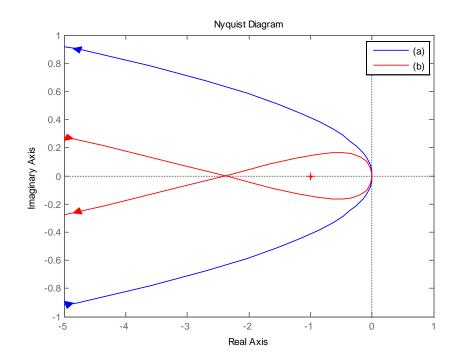
2. Sketch the polar plots of the open-loop transfer function

$$G(s)H(s) = \frac{K(T_a s + 1)(T_b s + 1)}{s^2(Ts + 1)}$$

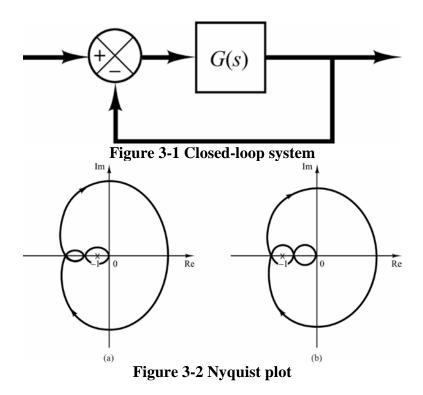
for the following two cases:

(a) $T_a > T > 0$, $T_b > T > 0$ (b) $T > T_a > 0$, $T > T_b > 0$

> Sol) Typical Nyquist curves for the cases (a) and (b) are shown below. (a) $T_a = 1.2$, $T_b = 0.5$, T = 0.4, K = 0.1(b) $T_a = 0.6$, $T_b = 1$, T = 2, K = 1



3. Consider the closed-loop system shown in Figure 3-1. G(s) has no poles in the right-half s plane. If the Nyquist plot of G(s) is as shown in Figure 3-2(a), is this system stable? If the Nyquist plot is as shown in Figure 3-2(b), is this system stable?

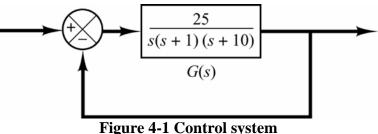


Sol) Since G(s) has no poles in the right-half s plane, the stability of the system can be studied by checking the enclosure of the -1 + j0 point by the Nyquist locus for $0 < \omega < \infty$.

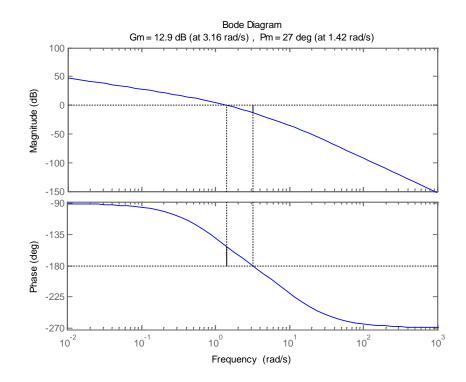
If the Nyquist plot of G(s) is as shown in Figure 3-2(a), then there is no enclosure of the -1+j0 point. Hence, the system is stable.

For the case of the Nyquist plot shown in Figure 3-2(b), the -1 + i0 point is enclosed by the Nyquist plot of G(jw) for $0 < \omega < \infty$. Hence, the system is unstable.

4. Consider the system shown in Figure 4-1. Draw a Bode diagram of the open-loop transfer function G(s). Determine the phase margin and gain margin.



Sol) A bode diagram of the system is shown below.



From this Bode diagram, we find the phase margin and gain margin to be 27° and 13 dB, respectively.

5. Consider a unity-feedback control system with the open-loop transfer function

$$G(s) = \frac{K}{s(s^2 + s + 4)}$$

Determine the value of the gain K such that the phase margin is 50° . What is the gain margin with this gain K?

Sol)
$$G(s) = \frac{K}{s(s^2 + s + 4)} = \frac{0.25K}{s(0.25s^2 + 0.25s + 1)}$$

The quadratic term in the denominator has the undamped natural frequency of 2 rad/sec and the damping ratio of 0.25. Define the frequency corresponding to the angle of -130° to be $\omega_{\rm l}$.

$$\angle G(j\omega_1) = -\angle j\omega_1 - \angle (1 - 0.25\omega_1^2 + j0.25\omega_1)$$
$$= -90^\circ - \tan^{-1}\frac{0.25\omega_1}{1 - 0.25\omega_1^2} = -130^\circ$$

Solving this last equation for ω_1 , we find $\omega_1 = 1.491 rad / \sec$. Thus, the phase angle becomes equal to -130° at $\omega = 1.491 rad / \sec$. At this frequency, the magnitude must be unity, or $|G(j\omega_1)| = 1$. The required gain K can be determined from

$$\left|G(j1.491)\right| = \left|\frac{0.25K}{(j1.491)(-0.555 + j0.3725 + 1)}\right| = 0.2890K$$

Setting |G(j1.491)| = 0.2890K = 1, we find

K = 3.46

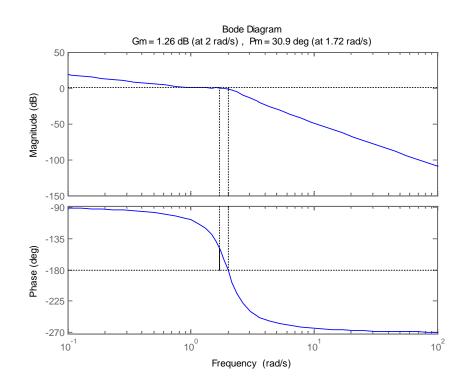
Note that the phase crossover frequency is a $\omega = 2rad / \sec$, since

$$\angle G(j2) = -\angle (j2) - \angle (-0.25 \times 2^2 + 0.25 \times j2 + 1) = -90^\circ - 90^\circ = -180^\circ$$

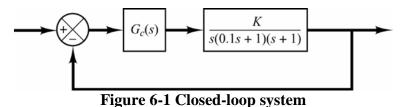
The magnitude |G(j2)| with K = 3.46 becomes

$$|G(j2)| = \left|\frac{0.865}{(j2)(-1+0.5j+1)}\right| = 0.865 = -1.26dB$$

Thus, the gain margin is 1.26dB. The Bode diagram of G(jw) with K=3.46 is shown below.



6. Referring to the closed-loop system shown in Figure 6-1, design a lead compensator $G_c(s)$ such that the phase margin is 45°, gain margin is not less than 8dB, and the static velocity error constant K_v is 4.0sec⁻¹. Plot unit-step and unit-ramp response curves of the compensated system with MATLAB.



Sol) Let us use the following lead compensator:

$$G_c(s) = K_c \alpha \frac{Ts+1}{\alpha Ts+1} = K_c \frac{s+\frac{1}{T}}{s+\frac{1}{\alpha T}}$$

Since K_v is specified as 4.0sec⁻¹, we have

$$K_{v} = \lim_{s \to 0} sK_{c} \alpha \frac{Ts+1}{\alpha Ts+1} \frac{K}{s(0.1s+1)(s+1)} = K_{c} \alpha K = 4$$

Let us set K=1 and define $K_c \alpha = \hat{K}$. Then,

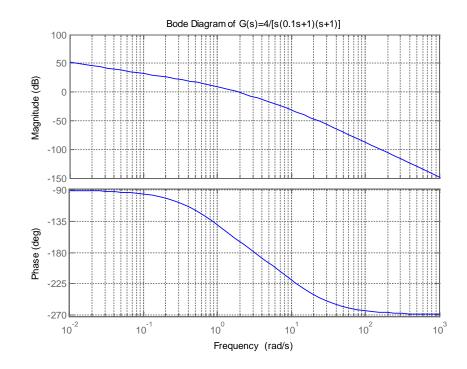
 $\hat{K} = 4$

Next, plot a Bode diagram of

$$\frac{4}{s(0.1s+1)(s+1)} = \frac{4}{0.1s^3 + 1.1s^2 + s}$$

The following MATLAB program produces the Bode diagram shown below.

num = [4]; den = [0.1 1.1 1 0]; bode(num,den) grid on title('Bode Diagram of G(s)=4/[s(0.1s+1)(s+1)]')



From this plot, the phase and gain margins are 17° and 8.7dB, respectively. Since the specifications call for a phase margin 45° , let us choose

$$\phi_{m} = 45^{\circ} - 17^{\circ} + 12^{\circ} = 40^{\circ}$$

(This means that 12° has been added to compensate for the shift in the gain crossover frequency.) The maximum phase lead is 40° . Since

$$\sin\phi_m = \frac{1-\alpha}{1+\alpha} \quad (\phi_m = 40^\circ)$$

 α is determined as 0.2174. Let us choose, instead of 0.2174, α to be 0.21.

Next step is to determine the corner frequencies $\omega = 1/T$ and $\omega = 1/(\alpha T)$ of the lead compensator. Note that the maximum phase-lead angle ϕ_m occurs at the geometric mean of the two corner frequencies, or $\omega = 1/(\sqrt{\alpha}T)$. The amount of the modification in the magnitude curve at $\omega = 1/(\sqrt{\alpha}T)$ due to the inclusion of the term $(Ts+1)/(\alpha Ts+1)$ is

$$\left|\frac{1+j\omega T}{1+j\omega\alpha T}\right|_{\omega=\frac{1}{\sqrt{\alpha T}}} = \frac{1}{\sqrt{\alpha}}$$

Note that

$$\frac{1}{\sqrt{\alpha}} = \frac{1}{\sqrt{0.21}} = 2.1822 = 6.7778 dB$$

We need to find the frequency point where, when the lead compensator is added, the total magnitude becomes 0 dB. The magnitude G(jw) is -6.7778 dB corresponds to $\omega = 2.81 rad / sec$. We select this frequency to be the new gain crossover frequency ω_c . Then we obtain

$$\frac{1}{T} = \sqrt{\alpha}\omega_c = \sqrt{0.21} \times 2.81 = 1.2877$$
$$\frac{1}{\alpha T} = \frac{\omega_c}{\sqrt{\alpha}} = \frac{2.81}{\sqrt{0.21}} = 6.1319$$

Hence

$$G_c(s) = \frac{4}{0.21} \frac{s + 1.2877}{s + 6.1319} = 4 \frac{0.7766s + 1}{0.16308s + 1}$$

The open-loop transfer function becomes as

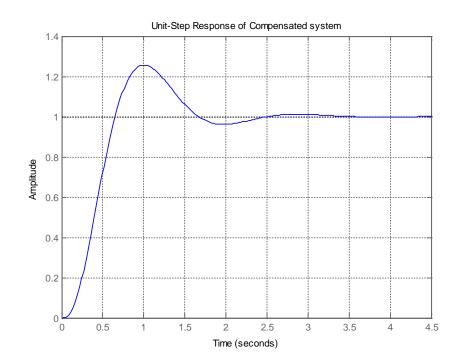
$$G_{c}(s)G(s) = 4 \frac{0.7766s + 1}{0.16308s + 1} \frac{1}{s(0.1s + 1)(s + 1)}$$
$$= \frac{3.1064s + 4}{0.01631s^{4} + 0.2744s^{3} + 1.2631s^{2} + s}$$

This open-loop transfer function yields phase margin of 45° and gain margin of 13 dB. So, the requirements on the phase margin and gain margin are satisfied. The closed-loop transfer function of the designed system is

$$\frac{C(s)}{R(s)} = \frac{3.1064s + 4}{0.01631s^4 + 0.2794s^3 + 1.2631s^2 + 4.1064s + 4}$$

The following MATLAB program produces the unit-step response curve as shown below.

```
numc = [3.1064 4];
denc = [0.01631 0.2794 1.2631 4.1064 4];
step(numc,denc)
grid
title('Unit-Step Response of Compensated system')
```



Similarly, the following MATLAB program produces the unit-ramp response curve as shown below.

```
numc = [3.1064 4];
denc = [0.01631 0.2794 1.2631 4.1064 4 0];
t = 0:0.01:5;
c = step(numc,denc,t);
plot(t,t,t,c)
grid
legend('Input', 'Output', 'location', 'northwest')
title('Unit-Ramp Response of Compensated system')
xlabel('Time (sec)')
ylabel('Unit-Ramp Input and System Output')
```

