

Lecture Note of Design Theories of Ship and Offshore Plant

Design Theories of Ship and Offshore Plant

Part I. Ship Design

Ch. 6 Structural Design

Fall 2017

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- ☑ **Ch. 6 Structural Design**
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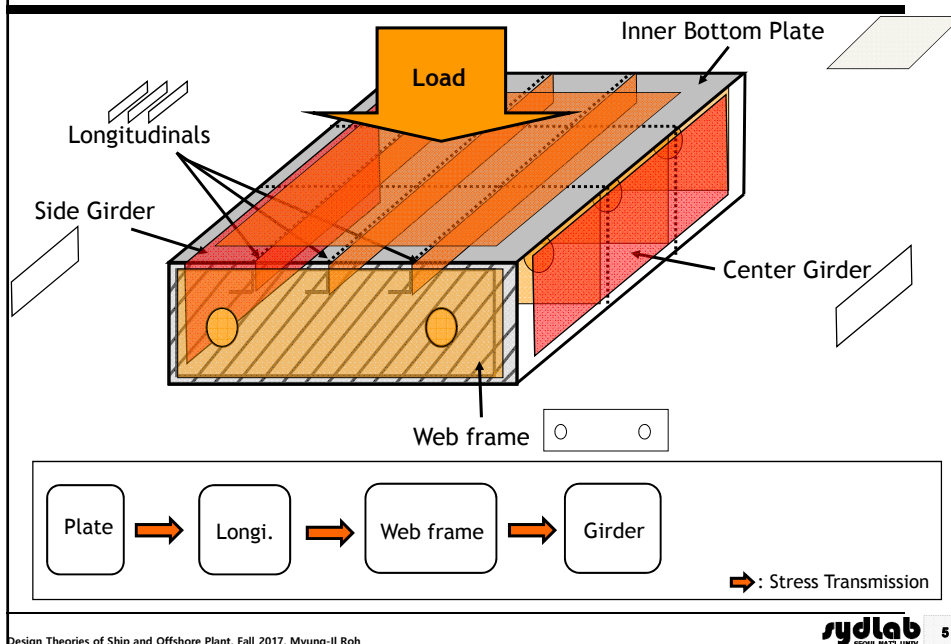
Ch. 6 Structural Design

- 6.1 Generals & Materials
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6.1 Generals & Materials

- (1) Stress Transmission
- (2) Principal Dimensions
- (3) Criteria for the Selection of Plate Thickness, Grouping of Longitudinal Stiffener
- (4) Material Factors

(1) Stress Transmission



(2) Principal Dimensions

DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec. 1
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The following principal dimensions are used in accordance with DNV rule.

1) Rule length (L or L_s)

: Length of a ship used for rule scantling procedure

$$0.96 \cdot L_{WL} < L < 0.97 \cdot L_{WL}$$

- Distance on [the summer load waterline](#) (L_{WL}) from the fore side of the stem to the axis of the rudder stock
- Not to be taken less than 96%, and need not be taken greater than 97%, of the extreme length on the summer load waterline (L_{WL})

Example of the calculation of rule length

L_{BP}	L_{WL}	$0.96 \cdot L_{WL}$	$0.97 \cdot L_{WL}$	L
250	261	250.56	253.17	250.56
250	258	247.68	250.26	250.00
250	255	244.80	247.35	247.35

2) Breadth

: Greatest moulded breadth in [m], measured at the summer load waterline

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(2) Principal Dimensions

DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec. 1
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3) Depth (D)

: Moulded depth defined as the vertical distance in [m] from baseline to moulded deck line at the uppermost continuous deck measured amidships

4) Draft (T)

: Mean moulded summer draft (**scantling draft**) in [m]

5) Block coefficient (C_B)

: To be calculated based on the rule length

$$C_B = \frac{\Delta}{1.025 \cdot L \cdot B \cdot T} \quad (\Delta : \text{Moulded displacement in sea water on draft } T)$$

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(3) Criteria for the Selection of Plate Thickness, Grouping of Longitudinal Stiffener

DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec. 1
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1) Criteria for the selection of plate thickness

➡ When selecting plate thickness, use the provided plate thickness.

- | |
|---------------------------|
| (1) 0.5 mm interval |
| (2) Above 0.25 mm: 0.5 mm |
| (3) Below 0.25 mm: 0.0 mm |

Ex) 15.75 mm ➡ 16.0 mm
15.74 mm ➡ 15.5 mm

2) Grouping of longitudinal stiffener

For the efficiency of productivity, each member is arranged by grouping longitudinal stiffeners. The grouping members should satisfy the following rule.

Average value but not to be taken less than 90% of the largest individual requirement (DNV).

Ex) The longitudinal stiffeners have design thickness of 100, 90, 80, 70, 60 mm. The average thickness is given by $80 \text{ mm} \times 5$. However, the average value is less than $100 \text{ mm} \times 90\% = 90 \text{ mm}$ of the largest individual requirement, 100 mm. Therefore, the average value should be taken $90 \text{ mm} \times 5$.

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(4) Material Factors

¹⁾ DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec.2

²⁾ James M. Gere, Mechanics of Materials 7th Edition, Thomson, Chap.1, pp.15~26

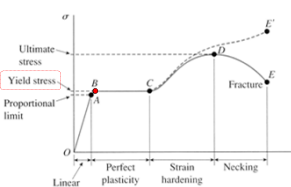
- The material factor f_t is included in the various formulae for scantlings and in expressions giving allowable stresses.¹⁾

Material Designation	Yield Stress (N/mm ²)	$\frac{\sigma}{\sigma_{NV-NS}}$	Material Factor (f_t)
NV-NS	235	$235/235 = 1.00$	1.00
NV-27	265	$265/235 = 1.13$	1.08
NV-32	315	$315/235 = 1.34$	1.28
NV-36	355	$355/235 = 1.51$	1.39
NV-40	390	$390/235 = 1.65$	1.47

* NV-NS: Normal Strength Steel (Mild Steel)

* NV-XX: High Tensile Steel

* High tensile steel: A type of alloy steel that provides better mechanical properties or greater resistance to corrosion than carbon steel. They have a carbon content between 0.05-0.25% to retain formability and weldability, including up to 2.0% manganese, and other elements are added for strengthening purposes.



* Yield Stress (σ_y) [N/mm²] or [MPa]: The magnitude of the load required to cause yielding in the beam.²⁾

* A: 'A' grade 'Normal Strength Steel'

* AH: 'A' grade 'High Tensile Steel'

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6.2 Global Hull Girder Strength (Longitudinal Strength)

- (1) Generals
- (2) Still Water Bending Moment (M_s)
- (3) Vertical Wave Bending Moment (M_w)
- (4) Section Modulus

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(1) Generals

Interest of "Ship Structural Design"

● Ship Structural Design



What is designer's **major** interest?

● Safety:

Won't it fail under the load?

a ship

} global

a stiffener
a plate

} local

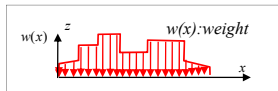


Let's consider the safety of the ship from the point of global strength first.

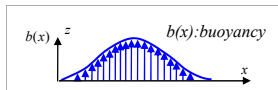
Dominant Forces Acting on a Ship



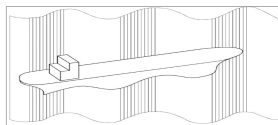
What are dominant forces acting on a ship in view of the longi. strength?



weight of light ship, weight of cargo, and consumables



hydrostatic force (buoyancy) on the submerged hull

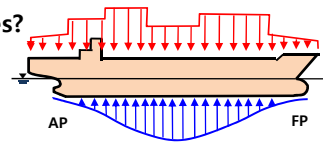


hydrodynamic force induced by the wave



What is the direction of the dominant forces?

The forces act in **vertical (lateral)** direction along the ship's length.



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Longitudinal Strength

: Overall strength of ship's hull which **resists** the bending moment, shear force, and torsional moment acting on a hull girder.

Longitudinal strength loads

: Load concerning the overall strength of the ship's hull, such as the bending moment, shear force, and torsional moment acting on a hull girder

Static longitudinal loads



Loads are caused by **differences between weight and buoyancy** in longitudinal direction in the still water condition

Hydrodynamic longitudinal loads



Loads are induced by **waves**

¹⁾ Okumoto, Y., Design of Ship Hull Structures, Springer, 2009, P.17

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Idealization of the Ship Hull Girder Structure



How can we idealize a ship as a structural member?

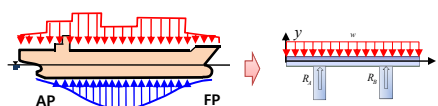
▪ **Structural member according to the types of loads**

- ① Axially loaded bar: structural member which supports **forces directed along the axis of the bar**
- ② Bar in torsion: structural member which supports torques (or couples) having **their moment about the longitudinal axis**
- ③ Beam: structural members subjected to **lateral loads**, that is, **forces or moments perpendicular to the axis of the bar**

Since a ship has a **slender shape** and **subject to lateral loads**, it will behave like a **beam** from the point view of structural member.



Ship is regarded as a **beam**.

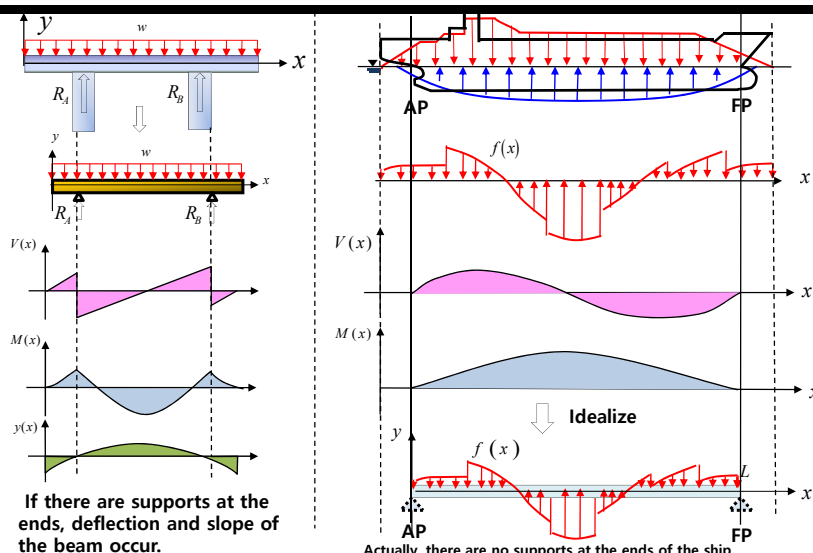


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Applying Beam Theory to a Ship



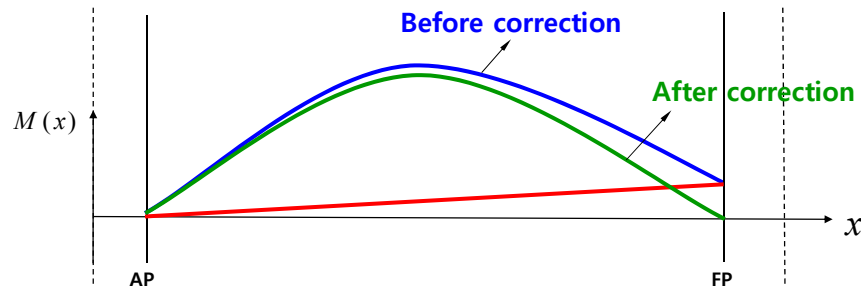
* James M. Gere, Mechanics of Materials, 6th Edition, Thomson, Ch. 4, p. 292

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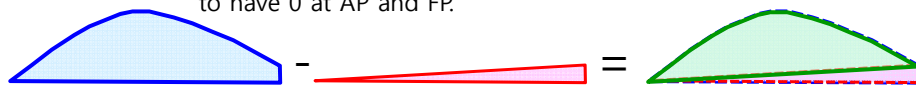
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Correction of a Bending Moment Curve



What if the bending moment is not zero at FP?
 ➔ The deflection and slope of the beam occur at FP.
 ➔ Thus, we correct the bending moment curve to have 0 at AP and FP.



* James M. Gere, Mechanics of Materials, 6th Edition, Thomson, Ch. 4, p. 292

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Actual Stress \leq Allowable Stress - Bending Stress and Allowable Bending Stress

1) DNV Rules, Pt. 3 Ch. 1 Sec. 5, Jan. 2004

The actual bending stress ($\sigma_{act.}$) shall not be greater than the allowable bending stress (σ_l).

M_S : Largest SWBM among all loading conditions and class rule

M_W : Calculated by class rule or direct calculation

$$\sigma_{act.} = \frac{|M_S + M_W|}{Z} 10^3 \text{ [kg / cm}^2\text{]}$$

$$\sigma_{act.} \leq \sigma_l$$

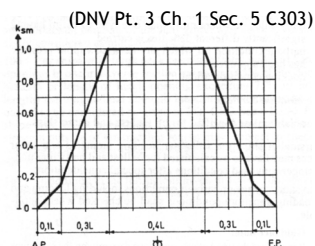


Fig. 2
Stillwater bending moment

$$\sigma_l = \sigma_{allow} = 175 f_1 \text{ [N / mm}^2\text{]} \text{ within } 0.4L \text{ amidship}$$

$$= 125 f_1 \text{ [N / mm}^2\text{]} \text{ within } 0.1L \text{ from A.P. or F.P.}$$

(f_1 : Material factor. Ex. Mild steel 1.0, HT-32 1.28, HT-36 1.39)

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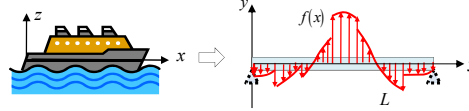
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Criteria of Structural Design (1/2)

• Ship Structural Design

a ship



The **actual bending stress** ($\sigma_{act.}$) shall not be greater than the **allowable bending stress** (σ_l).

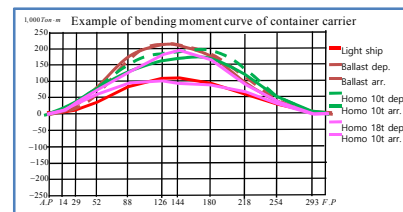
$$\sigma_{act.} \leq \sigma_l, \quad \sigma_{act.} = \frac{M}{Z} = \frac{|M_s + M_w|}{Z}$$

M_s : Largest SWBM among all loading conditions and class rule
 M_w : calculated by class rule or direct calculation

σ_l : allowable stress

For instance, allowable bending stresses by DNV rule are given as follows:

$$\begin{aligned} \sigma_l &= 175 f_1 \text{ [N/mm}^2\text{]} \quad \text{within 0.4L amidship} \\ &= 125 f_1 \text{ [N/mm}^2\text{]} \quad \text{within 0.1L from A.P. or F.P.} \end{aligned}$$



Actual bending moments at aft and forward area are smaller than that at the midship.



What is, then, the f_1 ?

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Criteria of Structural Design (2/2)

$$\sigma_{act.} \leq \sigma_l$$

$$\sigma_{act.} = \frac{M}{Z} = \frac{|M_s + M_w|}{Z}$$

- (1) Still Water Bending Moment (M_s)
- (2) Vertical Wave Bending Moment (M_w)
- (3) Section Modulus (Z)

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(2) Still Water Bending Moment (Ms)

Still Water Bending Moment (Ms)

$$\sigma_{act.} \leq \sigma_l$$

$$\sigma_{act.} = \frac{M}{Z} = \frac{M_s + M_w}{Z}$$

$\begin{cases} M_s: \text{Still water bending moment} \\ M_w: \text{Vertical wave bending moment} \end{cases}$

Hydrostatic loads along ship's length
caused by the weight & the buoyancy

$f_s(x)$: distributed loads in longitudinal
direction in still water



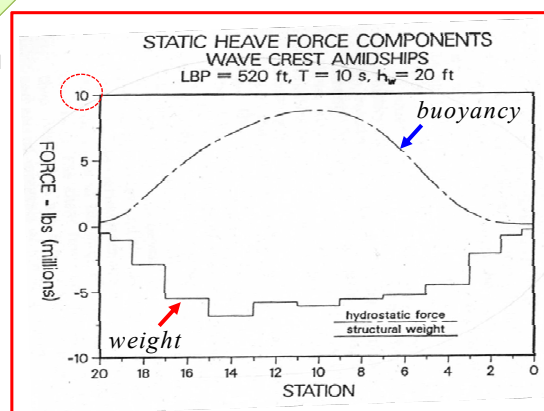
$V_s(x)$: still water shear force

$$V_s(x) = \int_0^x f_s(x) dx$$



$M_s(x)$: still water bending moment

$$M_s(x) = \int_0^x V_s(x) dx$$



Distributed Loads in Longitudinal Direction



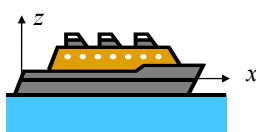
$$f(x) = f_s(x) + f_H(x)$$

$f(x)$: Distributed loads in longitudinal direction

$f_s(x)$: **Static longitudinal loads** in longitudinal direction

$f_H(x)$: **Hydrodynamic longitudinal loads** induced by wave

In still water



$$f_s(x) = b(x) + w(x)$$

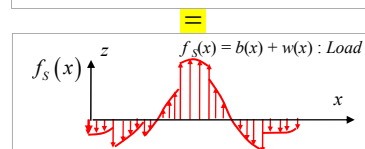
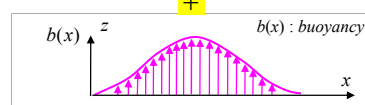
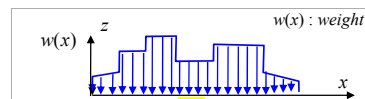
$b(x)$: Distributed buoyancy in longitudinal direction

$w(x) = LWT(x) + DWT(x)$

- $w(x)$: Weight distribution in longitudinal direction

- $LWT(x)$: Lightweight distribution

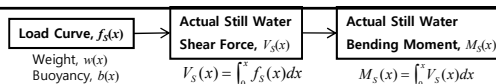
- $DWT(x)$: Deadweight distribution



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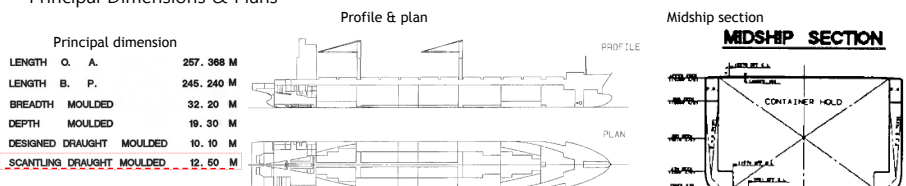
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Distributed Loads in Still Water



✓ Example of a 3,700 TEU Container Ship in Homogeneous 10 ton Scantling Condition

- Principal Dimensions & Plans



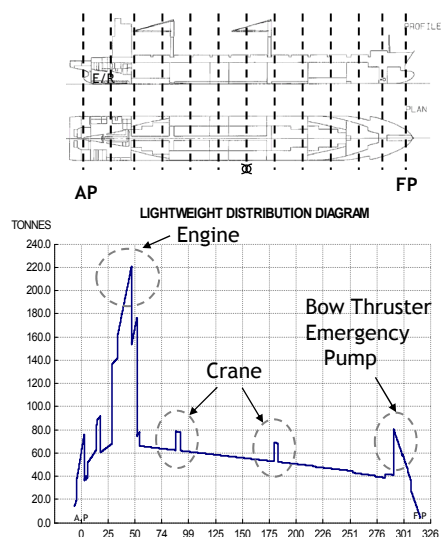
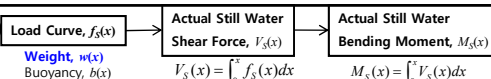
- Loading Condition: Homogeneous 10 ton Scantling Condition (Sailing state)

SAILING STATE				* Frame space: 800mm			
DRAUGHT F.P.	=	12.260 M	K.M.T.	=	14.889 M		
DRAUGHT MIDSHIP	=	12.457 M	KG (SOLID)	=	13.586 M		
DRAUGHT A.P.	=	12.654 M	GM (SOLID)	=	1.303 M		
TRIM BY STERN	=	.394 M	FREE SURF. CORR. (GG0)	=	.059 M		
PROPELLER T/D	=	160.3 %	G0M (FLUID)	=	1.244 M		
DISPLACEMENT	=	66813.6 T	KG0 ACTUAL (FLUID)	=	13.645 M		
DRAUGHT AT LCF	=	12.483 M	TRIM (DISA) / (MTCX100)	=	.394 M		
LCB FROM A.P.	=	115.677 M	FREE SURF. MOM.	=	3921 T-M		
LCG FROM A.P.	=	115.045 M	M.T.C.	=	1072.0 T-M		
TRIM LEVER : A	=	.632 M	LCF FROM A.P.	=	106.275 M		
DEGREE	=	.0 5.0 10.0 15.0 20.0 30.0 40.0 50.0 60.0 75.0					
KN	=	.000 1.296 2.591 3.882 5.168 7.614 9.592 10.930 11.697 11.959					
KG0XSIN0	=	.000 1.189 2.369 3.532 4.667 6.823 8.771 10.453 11.817 13.180					
GZ	=	.000 .107 .222 .350 .501 .791 .821 .477 -.120 -1.221					

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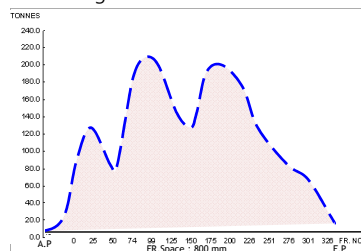
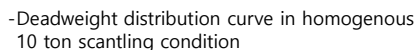
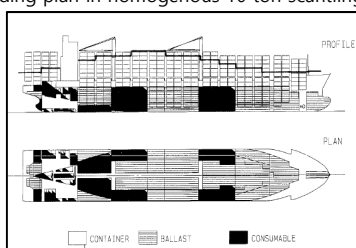
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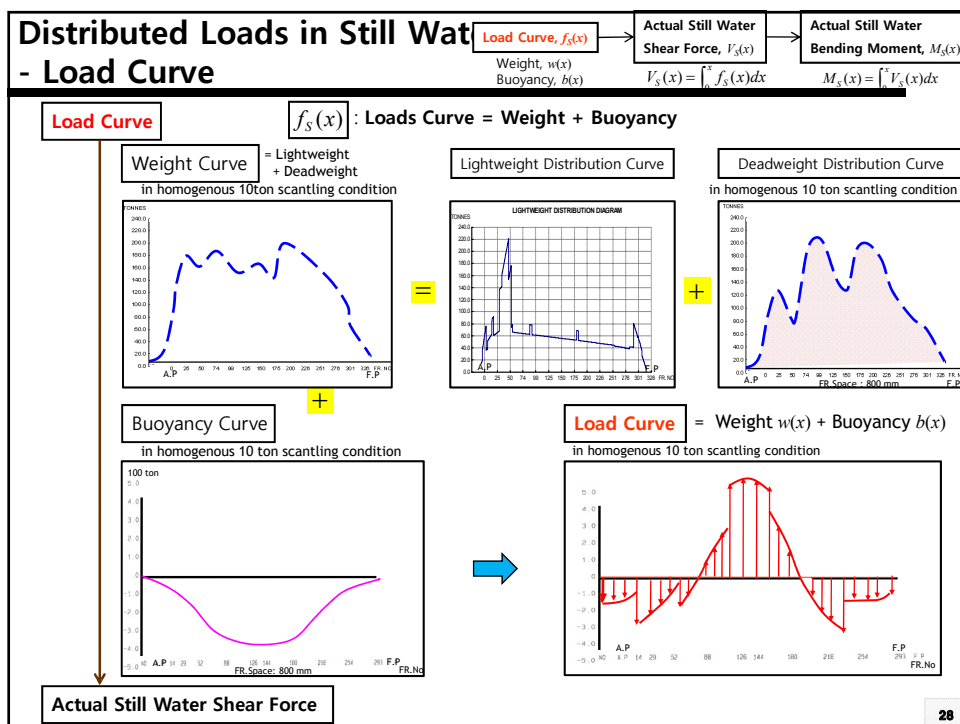
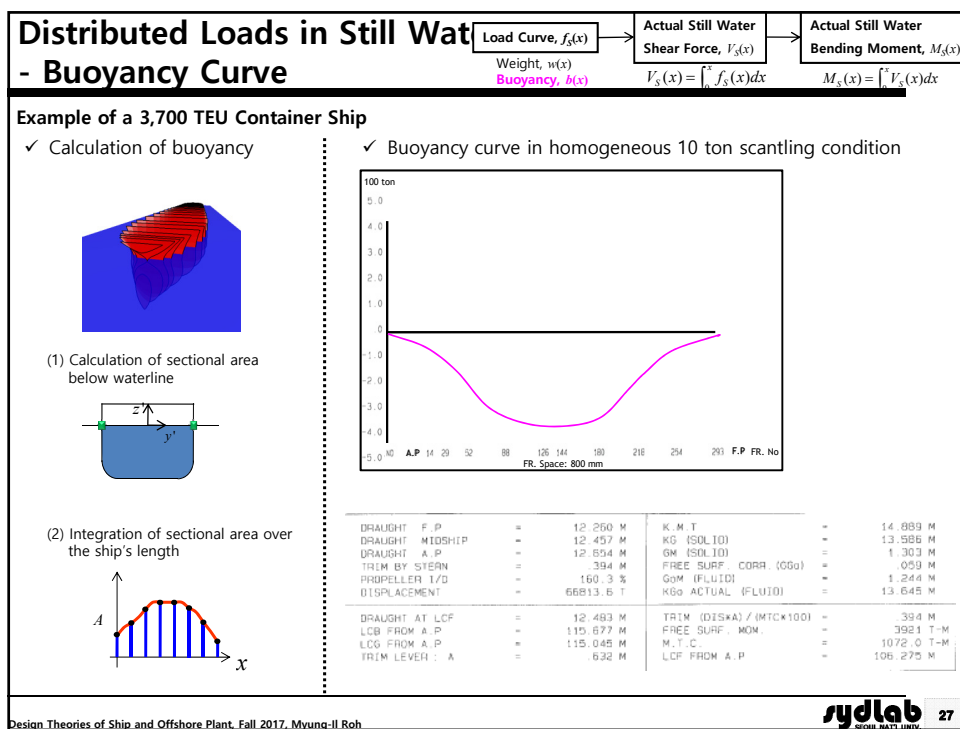
Example of a 3,700 TEU Container Ship

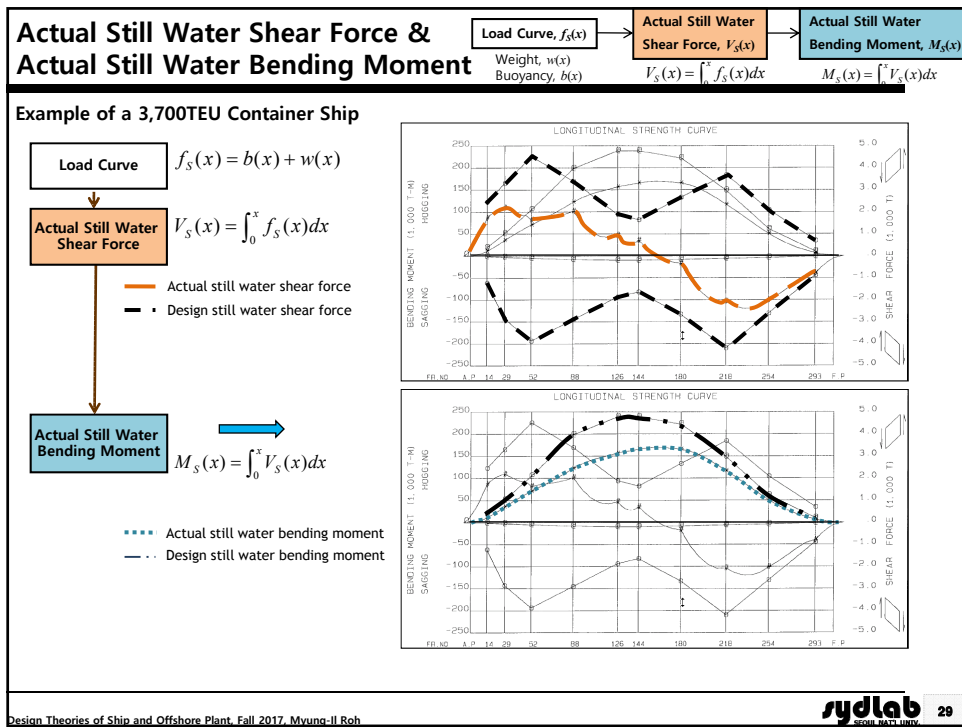


BILL No. 1 1329. 3,700 TEU CONTAINER VESSEL						
NO	AFT END	FORE END	WEIGHT	L.C.C.	MOMENT	
1	55.000	14.000	616.00	7.000	4312.0	
2	14.350	43.400	1387.10	31.400	4312.0	
3	43.400	232.320	7591.50	128.620	97648.7	
4	232.320	252.240	732.30	239.280	175224.7	
5	27.200	41.600	476.40	35.800	17055.1	
6	.000	245.240	300.00	122.620	1678.6	
7	43.400	232.320	340.00	134.000	4366.0	
8	-3.600	232.320	119.00	114.400	13613.6	
9	-3.400	2.400	151.90	.000	.0	
10	.000	252.240	224.00	120.000	26800.0	
11	202.240	232.320	137.90	217.000	99924.3	
12	43.400	202.240	1053.00	121.700	128150.1	
13	143.280	146.680	55.00	144.980	7973.9	
14	70.480	73.880	55.00	72.180	3969.9	
15	14.350	232.320	115.90	114.360	13254.3	
16	-3.400	232.320	128.00	114.360	14678.6	
17	232.320	245.240	118.30	238.600	28226.4	
18	36.000	170.000	3.00	81.000	243.0	
19	-5.000	4.000	50.00	-5.000	-25.0	
20	29.000	41.600	15.50	37.100	575.0	
21	-3.500	4.000	19.20	.000	.0	
22	4.000	11.200	34.30	7.600	260.7	
23	41.600	173.900	62.50	105.760	6610.0	
24	226.160	232.320	20.40	229.240	4376.5	
25	239.000	243.000	5.40	241.000	1061.0	
26	11.200	232.320	39.20	127.000	4770.6	
27	11.200	232.320	191.30	121.700	23281.2	
28	27.200	41.600	214.50	36.000	7722.0	
29	23.230	37.600	979.00	30.400	29674.0	
30	11.200	41.600	288.50	22.000	6336.0	
31	3.800	23.230	111.30	11.200	1246.6	
32	12.600	41.600	150.70	28.000	4219.6	
33	11.200	41.600	158.60	28.000	4445.0	
34	11.200	41.600	95.90	28.000	2685.2	
35	218.480	218.480	164.00	114.400	18066.0	
36	27.200	41.600	8.50	36.000	306.0	
37	11.200	41.600	43.00	30.000	1290.0	
38	27.200	41.600	4.30	36.000	154.8	
39	27.200	41.600	5.70	36.000	205.2	
LIGHT SHIP TOTAL =			15998.10	103.228	1651446.2	

Example of a 3,700 TEU Container Ship
-Loading plan in homogenous 10 ton scantling condition

Deadweight distribution in longitudinal direction
in homogenous 10 ton scantling condition[illegible]





Rule Still Water Bending Moment by the Classification Rule

Recently, **actual still water bending moment based on the loading conditions** is used for still water bending moment, because the rule still water bending moment is only for the tanker.

- The **design still water bending moments** to be taken less than (DNV Pt. 3 Ch. 1 Sec. 5 A105)

$$M_s = M_{SO} \text{ [kNm]} \quad (\text{rule still water bending moment})$$

$$M_{SO} = -0.065 C_{WU} L^2 B (C_B + 0.7) \quad \text{[kNm] in sagging}$$

$$= C_{WU} L^2 B (0.1225 - 0.015 C_B) \quad \text{[kNm] in hogging}$$

C_{WU} : Wave coefficient for unrestricted service

The design still water bending moment **shall not be less than the large of**: the **largest actual still water bending moment based on the loading conditions** and the **rule still water bending moment**.

Design SWBM = Max(Actual SWBM, Rule SWBM) + margin

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Rule Still Water Shear Force by the Classification Rule

- The design values of still water shear forces along the length of the ship are normally not to be taken less than

(Dnv Pt.3 Ch.1 Sec. 5 B107)

(rule still water shear force)

$$Q_S = k_{sq} Q_{SO} (kN)$$

$$Q_{SO} = 5 \frac{M_{SO}}{L} (kN)$$

$k_{sq} = 0$ at A.P. and F.P.

$= 1.0$ between 0.15L and 0.3L from A.P.

$= 0.8$ between 0.4L and 0.6L from A.P.

$= 1.0$ between 0.7L and 0.85L from A.P.

$M_{SO} = -0.065 C_{WU} L^2 B (C_B + 0.7)$ [kNm] in sagging

$= C_{WU} L^2 B (0.1225 - 0.015 C_B)$ [kNm] in hogging

C_{WU} : wave coefficient for unrestricted service

The design still water shear force shall not be less than the large of: the largest actual still water shear forces based on loading conditions and the rule still water shear force.

$$\text{Design SWSF} = \text{Max}(\text{Actual SWSF}, \text{Rule SWSF}) + \text{margin}$$

(3) Vertical Wave Bending Moment (Mw)

Vertical Wave Bending Moment (Mw)

$$\sigma_{act.} \leq \sigma_l, \quad \sigma_{act.} = \frac{M}{Z} = \frac{M_S + M_W}{Z}$$

$\begin{cases} M_S: \text{Still water bending moment} \\ M_W: \text{Vertical wave bending moment} \end{cases}$

Hydrodynamic loads induced by waves along ship's length

$f_W(x)$: distributed loads induced by waves
 = Froude-Krylov force + diffraction force
 + added mass force + damping force

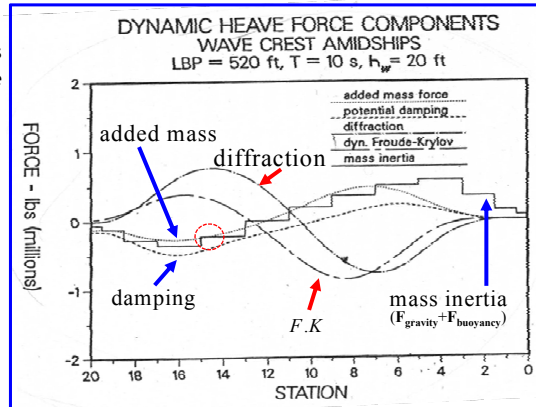
$$\begin{aligned} M\ddot{r} &= \sum F = (\text{Body Force}) + (\text{Surface Force}) \\ &= F_{gravity}(r) + F_{inertial}(r, \dot{r}, \ddot{r}) \\ &= F_{gravity} + F_{buoyancy}(r) + F_{F.K.}(r) + F_D(r) \\ &\quad + F_{R,Damping}(r, \dot{r}) + F_{R,Mass}(r, \ddot{r}) \end{aligned}$$

$V_W(x)$: vertical wave shear force

$$V_W(x) = \int_0^x f_W(x) dx$$

$M_W(x)$: vertical wave bending moment

$$M_W(x) = \int_0^x V_W(x) dx$$



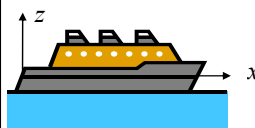
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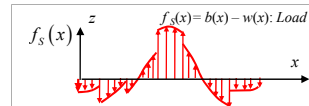
Dynamic Longitudinal Loads

In still water



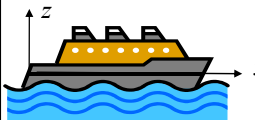
$$f_S(x) = b(x) + w(x)$$

$f_S(x)$: Distributed loads in longitudinal direction
 $f_b(x)$: Static longitudinal loads in longitudinal direction
 $f_w(x)$: Hydrodynamic longitudinal loads induced by wave

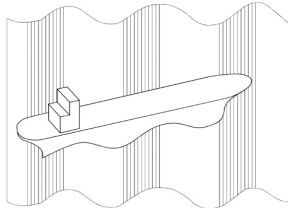


+

In wave

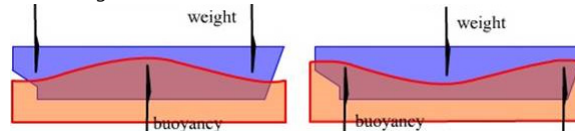


• Ship in oblique waves

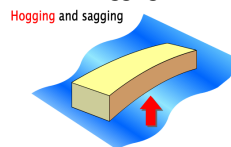


✓ Dynamic longitudinal loads
 : Loads are induced by waves

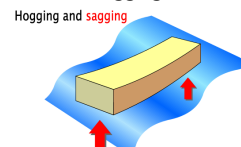
Vertical bending due to waves



Hogging



Sagging

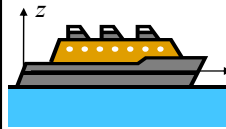


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Dynamic Longitudinal Loads

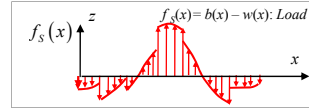
- Direct Calculation of Dynamic Longitudinal Loads (1/2)

In still water



$$f_s(x) = b(x) + w(x)$$

$f_s(x)$: Distributed loads in longitudinal direction
 $f_b(x)$: Static longitudinal loads in longitudinal direction
 $f_w(x)$: Hydrodynamic longitudinal loads induced by wave



+

In wave

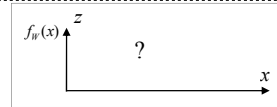


✓ Dynamic longitudinal loads
 : Loads are induced by waves

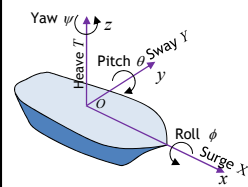
✓ Direct calculation of dynamic longitudinal loads

• from 6DOF motion of ship

$$\mathbf{x} = [X, Y, T, \phi, \theta, \psi]^T$$



✓ Ref. > 6 DOF motion of ship



$$f(x) = f_s(x)$$

$$+ f_w(x)$$

$$= b(x) + w(x) + f_D(x) + f_{F,K}(x) + f_R(x)$$

additional loads in wave

where,

$$f_R(x) = -a(x)\ddot{\mathbf{x}} - b(x)\dot{\mathbf{x}}$$

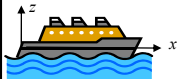
$f_D(x)$: Diffraction force at x
 $f_R(x)$: Radiation force at x by damping and added mass
 $f_{F,K}(x)$: Froude-Krylov force at x

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Dynamic Longitudinal Loads

- Direct Calculation of Dynamic Longitudinal Loads (2/2)

In wave



✓ Direct calculation of dynamic longitudinal loads

Load induced by Wave

$$f_w(x) = f_D(x) + f_{F,K}(x) + f_R(x)$$

$$\text{where, } f_R(x) = -a(x)\ddot{\mathbf{x}} - b(x)\dot{\mathbf{x}}$$

Actual Vertical Wave Shear Force

$$Q_w(x) = \int_0^x f_w(x) dx$$

Actual Vertical Wave Bending Moment

$$M_w(x) = \int_0^x Q_w(x) dx$$

Rule Values of Vertical Wave Bending Moments

- ✓ Direct calculation of dynamic longitudinal loads

- Loads are induced by waves

Actual Vertical Wave Shear Force

$$Q_w(x) = \int_0^x f_w(x) dx$$

Actual Vertical Wave Bending Moment

$$M_w(x) = \int_0^x Q_w(x) dx$$

Recently, rule values of vertical wave moments are used, because of the uncertainty of the direct calculation values of vertical wave bending moments.

Design VWBM = Min(Actual VWBM, Rule VWBM) + margin.....

The rule vertical wave bending moments amidships are given by:

$$M_w = M_{w0} \quad [kNm]$$

(DNV Pt.3 Ch.1 Sec.5 B201)

$$M_{w0} = -0.11\alpha C_w L^2 B(C_B + 0.7) \quad [kNm] \text{ in sagging}$$

$$= 0.19\alpha C_w L^2 B C_B \quad [kNm] \text{ in hogging}$$

$\alpha = 1.0$ for seagoing condition

$= 0.5$ for harbor and sheltered water conditions (enclosed fiords, lakes, rivers)

C_w : wave coefficient

C_B : block coefficient, not be taken less than 0.6

L	C _w
L ≤ 100	0.0792 · L
100 < L < 300	10.75 - [(300 - L)/100] ^{3/2}
300 ≤ L ≤ 350	10.75
L > 350	10.75 - [(L - 350)/150] ^{3/2}

Direct calculation values of vertical wave bending moments can be used for design vertical wave bending moment instead of the rule values of vertical wave moments, if the value of the direct calculation is smaller than that of the rule value.

Rule Values of Vertical Wave Shear Forces

- ✓ Direct calculation of dynamic longitudinal loads

- Loads are induced by waves

Load induced by Wave

$$f_w(x) = f_D(x) + f_{F,K}(x) + f_R(x)$$

$$\text{where, } f_R(x) = -a(x) \ddot{x} - b(x) \dot{x}$$

Actual Vertical Wave Shear Force

$$Q_w(x) = \int_0^x f_w(x) dx$$

Design VWSF = Min(Actual VWSF, Rule VWSF) + margin

The rule values of vertical wave shear forces along the length of the ship are given by:

(DNV Pt.3 Ch.1 Sec.5 B203)

$$\text{Positive shear force: } Q_{WP} = 0.3\beta k_{wqp} C_w LB(C_B + 0.7)$$

β : coefficient according to operating condition

$$\text{Negative shear force: } Q_{WN} = -0.3\beta k_{wqn} C_w LB(C_B + 0.7)$$

k_{wqp} , k_{wqn} : coefficients according to location in lengthwise
 C_w : wave coefficient

Direct calculation values of vertical wave shear forces can be used for vertical wave shear force instead of the rule values of vertical shear forces, if the value of the direct calculation is smaller than that of the rule value.

[Example] Rule Values of Still Water Bending Moments (Ms) and Vertical Wave Bending Moment (Mw)

Calculate L_s , $C_{B,SCANT}$, and vertical wave bending moment (M_w) at amidships ($0.5L$) of a ship in hogging condition for sea going condition.

Given: $L_{OA} = 332.0\text{ m}$, $L_{BP} = 317.2\text{ m}$, $L_{EXT} = 322.85\text{ m}$, $B = 43.2\text{ m}$, $T_s = 14.5\text{ m}$, $\Delta = 140,960\text{ ton}$

(Sol.) $L_s = 0.97 \times L_{EXT} = 0.97 \times 322.85 = 313.16$

$$C_{B,SCANT} = \Delta / (1.025 \times L_s \times B \times T_s) = \frac{140,906}{1.025 \times 313.16 \times 43.2 \times 14.5} = 0.701$$

$\alpha = 1.0$, for sea going condition,

$C_w = 10.75$, if $300 \leq L \leq 350$ (wave coefficient)

$k_{wm} = 1.0$ between $0.4L$ and $0.65L$ from A.P.(=0.0) and F.P

$$M_{wO} = 0.19 \times \alpha \times C_w \times L^2 \times B \times C_{B,SCANT} \quad (kNm)$$

$$= 0.19 \times 1.0 \times 10.75 \times 313.16^2 \times 43.2 \times 0.701 = 6,066,303 \quad (kNm)$$

at $0.5L$, $k_{wm} = 1.0$

$$M_w = 1.0 \times M_{wO}$$

$$\text{So, } M_w = 1.0 \times M_{wO} = 6,066,303 \quad (kNm)$$

$$M_s = M_{SO} \quad (kNm)$$

$$M_{SO} = -0.065 C_{wU} L^2 B (C_B + 0.7), \quad (\text{in sagging})$$

$$= C_{wU} L^2 B (0.1225 - 0.015 C_B), \quad (\text{in hogging})$$

$$M_w = M_{wO} \quad (kNm)$$

$$M_{wO} = -0.11 \alpha C_w L^2 B (C_B + 0.7), \quad (\text{in sagging})$$

$$= 0.19 \alpha C_w L^2 B C_B, \quad (\text{in hogging})$$

1) DSME, Ship Structural Design, 5-2 Load on Hull Structure, Example 4, 2005
Design Theories of Ship and Offshore Plant, Fall 2017, Myung-Il Roh

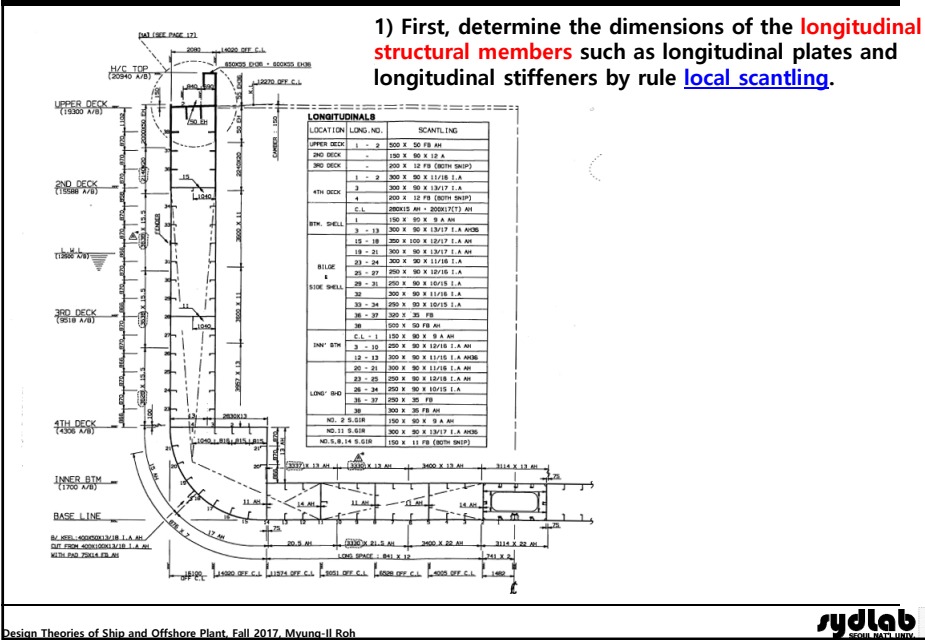
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(4) Section Modulus

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Example of Midship Section of a 3,700 TEU Container Ship



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Vertical Location of Neutral Axis about Baseline

2) Second, calculate the moment of sectional area about the base line.

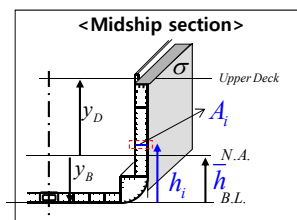
$$\sum h_i A_i$$

h_i : vertical center of structural member
 A_i : area of structural member

3) Vertical location of neutral axis from base line (\bar{h}) is, then, calculated by dividing the moment of area by the total sectional area.

$$\bar{h} = \frac{\sum h_i A_i}{A}$$

\bar{h} : vertical location of neutral axis
 A : total area



By definition, neutral axis pass through the centroid of the cross section.

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Midship Section Moment of Inertia about N.A

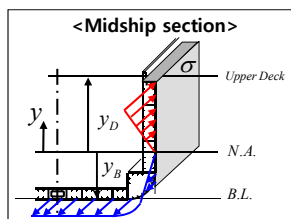
- The midship section moment of inertia about base line ($I_{B.L.}$)

$$I_{B.L.} = I_{N.A.} + A \bar{h}^2$$

- then calculate the midship section moment of inertia about neutral axis ($I_{N.A.}$) using $I_{B.L.}$.

$$I_{N.A.} = I_{B.L.} - A \bar{h}^2$$

Calculation of Section Modulus and Actual Stress at Deck and Bottom



σ : bending stress
 M_T : Total bending moment
 A : Total Area
 $I_{N.A.}$: Inertia moment of the midship section area about neutral axis (N.A.)
 B.L.: Base Line

Section modulus

$$Z_D = \frac{I_{N.A.}}{y_D}, \quad Z_B = \frac{I_{N.A.}}{y_B}$$

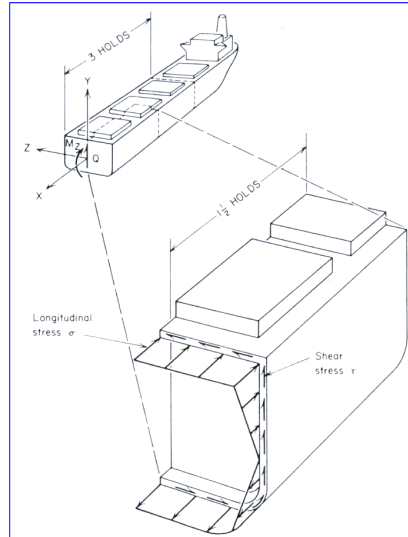
Calculation of Actual Stress at Deck and Bottom

$$\sigma_{Deck} = \frac{M}{Z_D} = \frac{M}{I_{N.A.} / y_D}$$

$$\sigma_{Bottom} = \frac{M}{Z_B} = \frac{M}{I_{N.A.} / y_B}$$

$$\sigma \leq \sigma_l, \quad \sigma = \frac{M}{Z} = \frac{M}{I_{N.A.} / y}$$

Global Hull Girder Strength (Longitudinal Strength) - Definition of the Longitudinal Strength Members



Application of hull girder load effects

※ Example of Requirement for Longitudinal Structural Member

DNV Rules for Classification of Ships
Part 3 Chapter 1 HULL STRUCTURE DESIGN SHIPS WITH
LENGTH 100 METERS AND ABOVE

Sec. 5 Longitudinal Strength

C 300 Section modulus

301 The requirements given in 302 and 303 will normally be satisfied when calculated for the midship section only, provided the following rules for tapering are complied with:

- Scantlings of all continuous longitudinal strength members shall be maintained within 0.4 L amidships.
- Scantlings outside 0.4 L amidships are gradually reduced to the local requirements at the ends, and the same material strength group is applied over the full length of the ship.

The section modulus at other positions along the length of the ship may have to be specially considered for ships with small block coefficient, high speed and large flare in the fore body or when considered necessary due to structural arrangement, see A106.

* Hughes, Ship Structural Design, John Wiley & Sons, 1983
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The Minimum Required Midship Section Modulus and Inertia Moment by DNV Rule

DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec. 5

The **midship section modulus** about the transverse neutral axis **shall not be less than:**
(Pt.3 Ch.1 Sec.5 C302)

$$Z_O = \frac{C_{WO}}{f_1} L^2 B (C_B + 0.7) \quad [\text{cm}^3]$$

C_{WO} : wave coefficient

L	C_{WO}
$L < 300$	$10.75 - [(300 - L)/100]^{1.2}$
$300 \leq L \leq 350$	10.75
$L > 350$	$10.75 - [(L - 350)/150]^{1.2}$

C_B is in this case not to be taken less than 0.60.

The **midship section moment of inertia** about the transverse neutral axis **shall not be less than:**
(Pt.3 Ch.1 Sec.5 C400)

$$I_{ship} = 3C_W L^3 B (C_B + 0.7) \quad [\text{cm}^4]$$

* DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec. 5
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Material Factors (f_1)

¹⁾ DNV Rules, Jan. 2004, Pt. 3 Ch. 1 Sec.2

²⁾ James M. Gere, Mechanics of Materials 7th Edition, Thomson, Chap.1, pp.15~26

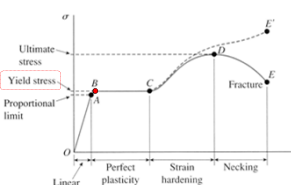
- The material factor f_1 is included in the various formulae for scantlings and in expressions giving allowable stresses.¹⁾

Material Designation	Yield Stress (N/mm ²)	$\frac{\sigma}{\sigma_{NV-NS}}$	Material Factor (f_1)
NV-NS	235	235/235 = 1.00	1.00
NV-27	265	265/235 = 1.13	1.08
NV-32	315	315/235 = 1.34	1.28
NV-36	355	355/235 = 1.51	1.39
NV-40	390	390/235 = 1.65	1.47

* NV-NS: Normal Strength Steel (Mild Steel)

* NV-XX: High Tensile Steel

* High tensile steel: A type of alloy steel that provides better mechanical properties or greater resistance to corrosion than carbon steel. They have a carbon content between 0.05-0.25% to retain formability and weldability, including up to 2.0% manganese, and other elements are added for strengthening purposes.



* Yield Stress (σ_y) [N/mm²] or [MPa]: The magnitude of the load required to cause yielding in the beam.²⁾

* A: 'A' grade 'Normal Strength Steel'

* AH: 'A' grade 'High Tensile Steel'

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Summary of Longitudinal Strength

$$\sigma \leq \sigma_l, \quad \sigma = \frac{M}{I_{N.A.}/y} = \frac{M}{Z}$$

Calculation of hull girder total shear force & bending moment

Still water shear forces Q_S
Still water bending moments M_S

(Q_S, M_S) based on the loading conditions

1. Weight curve $W(x)$
2. Buoyancy curve $B(x)$
3. Load curve $f_s(x) = W(x) + B(x)$
4. Shear force curve $Q_s = \int f_s dx$
5. Bending moment curve $M_s = \int Q_s dx$

(Q_S, M_S) Min. rule requirements

Larger value shall be used for the design still water bending moment between the largest actual still water bending moment based on loading conditions and still water bending moment by rule.

Wave shear force Q_W
Wave bending moment M_W

Direct calculation (Q_W, M_W)

1. Wave Load curve
 $f_w(x) = f_D(x) + f_{F.K.}(x) + f_R(x)$
2. Vertical Wave Shear force curve
 $Q_w = \int f_w dx$
3. Vertical Wave Bending moment curve
 $M_w = \int Q_w dx$

Class rule (Q_W, M_W)

Direct calculation values can be used for wave shear force and wave bending moment.

Calculation of section modulus (Local scantling)

Actual bending stress \leq Allowable bending stress

No

Modify longitudinal structural members

Yes

End of design of longitudinal strength

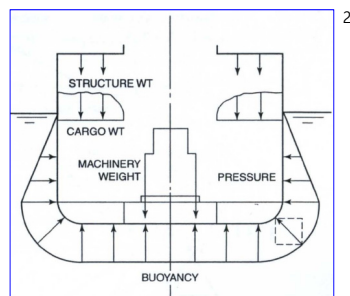
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6.3 Local Strength (Local Scantling)

- (1) Procedure of Local Scantling
- (2) Local Strength & Allowable Stress
- (3) Design Loads
- (4) Scantling of Plates
- (5) Scantling of Stiffeners
- (6) Sectional Properties of Steel Sections

Local Scantling

- Ship structure members are designed to endure the loads acting on the ship structure such as hydrostatic and hydrodynamic loads¹⁾.



- For instance, the structural member is subjected to:

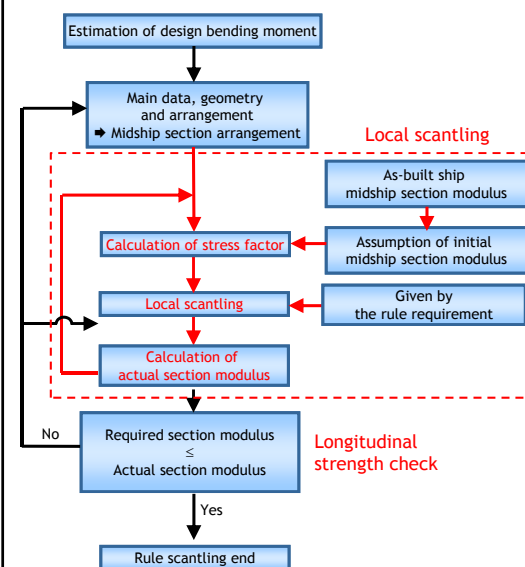
Hydrostatic pressure due to surrounding water
Internal loading due to self weight and cargo weight
Hydrodynamic load due to waves
Inertia force of cargo or ballast due to ship motion

¹⁾ Okumoto, Y., Takeda, Y., Mano, M., Design of Ship Hull Structures - a Practical Guide for Engineers, Springer, pp. 17-32, 2009

²⁾ Mansour, A., Liu, D., The Principles of Naval Architecture Series - Strength of Ships and Ocean Structures, The Society of Naval Architects and Marine Engineers, 2008

(1) Procedure of Local Scantling

Procedure of Local Scantling - Design Procedure of Structures



Ship structure design is carried out in accordance with the procedure shown in the figure.

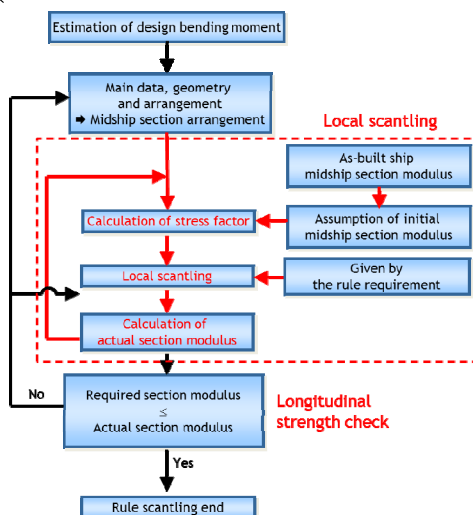
Each member is adjusted to have enough local strength given by the rule of Classification Societies based on the mechanics of materials.

This is called the "local scantling".

Design Procedure of Structures - Stress Factor



Why iteration is needed for the calculation of local scantling?



The actual midship section modulus at bottom or deck is needed.

However, the section modulus can be calculated after the scantlings of the members are determined.

➔ **Assumption!**

Therefore, the actual section modulus is calculated to be equal to the assumed section modulus by the iteration.

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Design Procedure of Structures - Stress Factor

1) DNV Rules, Pt. 3 Ch. 1 Sec. 6 C800, Jan. 2004



Why iteration is needed for the calculation of local scantling?

Example) Inner bottom longitudinals¹⁾

▪ Minimum plate thickness

$$t = \frac{15.8 k_a s \sqrt{p}}{\sqrt{\sigma}} \quad (mm)$$

▪ Minimum section modulus for longitudinal stiffener

$$Z = \frac{83 l^2 s p w_k}{\sigma} \quad (cm^3)$$

l : Stiffener span in m
 s : Stiffener spacing in m
 p : Design loads
 w_k : Section modulus corrosion factor in tanks, Sec.3 C1004

σ_{db} : Mean double bottom stress at plate flanges, normally not to be taken less than
= 20 t, for cargo holds in general cargo vessel
= 50 t, for holds for ballast
= 85 t, b/b for tanks for liquid cargo

Where, $\sigma = 225 f_1 - 100 f_{2b} - 0.7 \sigma_{db}$: Allowable stress of this structural part

f_1 : Material factor as defined in DNV Rules Pt. 3 Ch. 1 Sec.2

f_{2b} : stress factor

$$f_{2b,2d} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$$

M_S : Largest design SWBM²⁾ [kN-m]
 M_W : VWBM by class rule or direct calculation in [kN-m]

The actual midship section modulus at bottom or deck is needed. However, the section modulus can be calculated after the scantlings of the members are determined.

➔ **Assumption!**

Therefore, the actual section modulus is calculated to be equal to the assumed section modulus by the iteration.

2) Largest SWBM among all loading conditions and class rule

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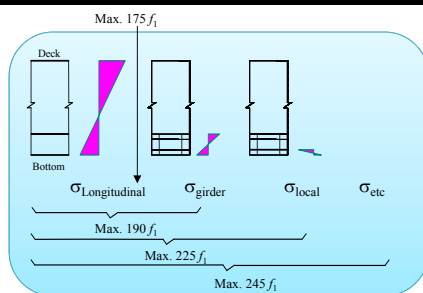
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(2) Local Strength & Allowable Stress

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Local Strength & Allowable Stresses - Allowable Stress for Local Strength



In the figure above, the meaning of the coefficients of the maximum allowable stresses is as follows:

- 245 f_i : Maximum Yield Stress
- 235 f_i : Proportional Limit
- 225 f_i : The maximum allowable stress for the local strength uses the value less than the maximum yield stress. In other words, 225 f_i is used for the yield stress, except for the other effects.

Relationship between load and stress

- 1) Longitudinal load induced by waves (Hogging or Sagging)

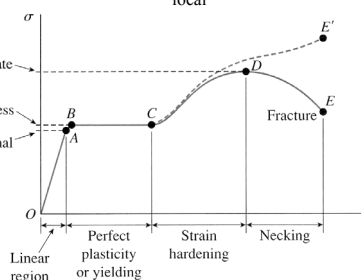
$\sigma_{\text{Longitudinal}}$

- 2) Cargo load

$\sigma_{\text{girder}} (\sigma_{db})$

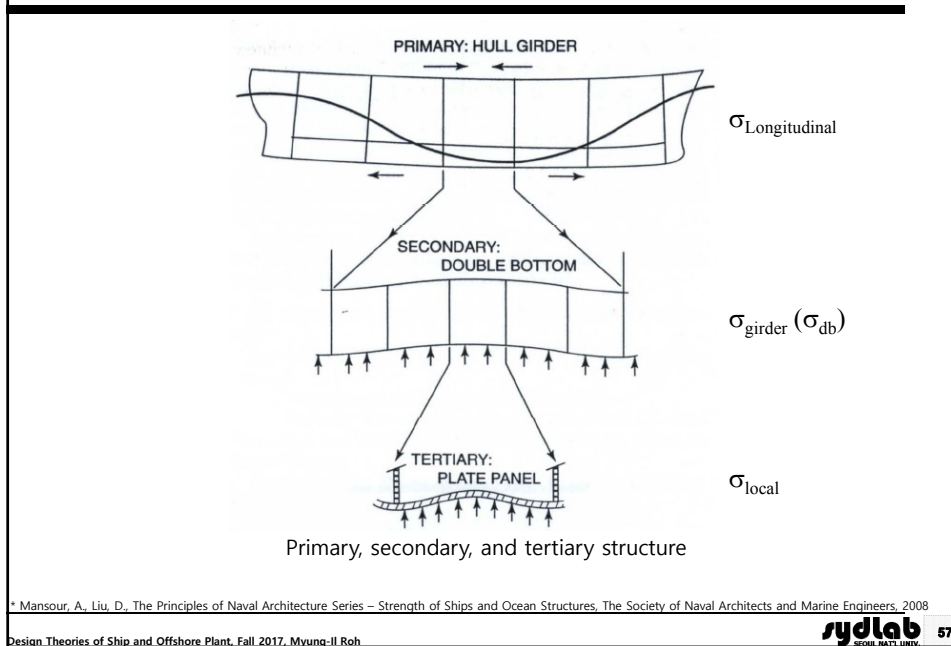
- 3) Ballasting load

σ_{local}



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Local Strength & Allowable Stresses



Allowable Stresses - Allowable Stress for Local Strength

Max. 175 f_1

Deck

Bottom

$\sigma_{\text{Longitudinal}}$

σ_{girder}

σ_{local}

σ_{etc}

Max. 190 f_1

Max. 225 f_1

Max. 245 f_1

L12

21990

21300

21300

Another interpretation of the figure

Example) Inner bottom longitudinals¹⁾

The section modulus requirement is given by:

$$Z = \frac{83l^2 spw_k}{\sigma} \text{ (cm}^3\text{)}$$

where, p is the local pressure on bottom structure.

The nominal allowable bending stress due to lateral pressure is used except for the longitudinal stress and the double bottom stress.

$$\sigma = 225 f_1 - 100 f_{2b} - 0.7 \sigma_{db}$$

The longitudinal stress is given by the stress factor.
And the double bottom stress is given by:

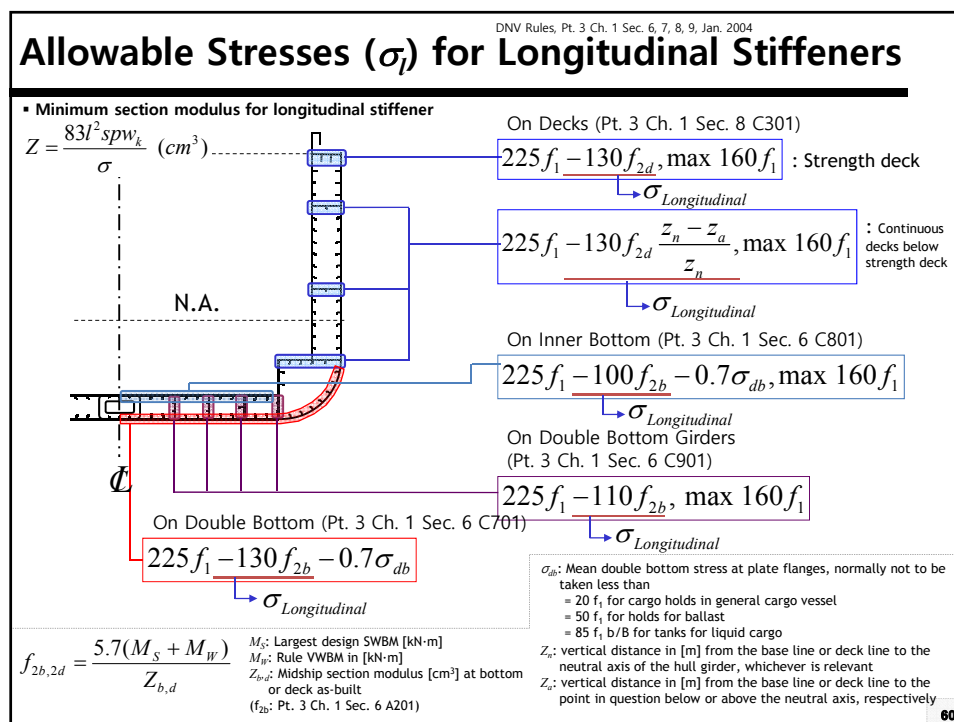
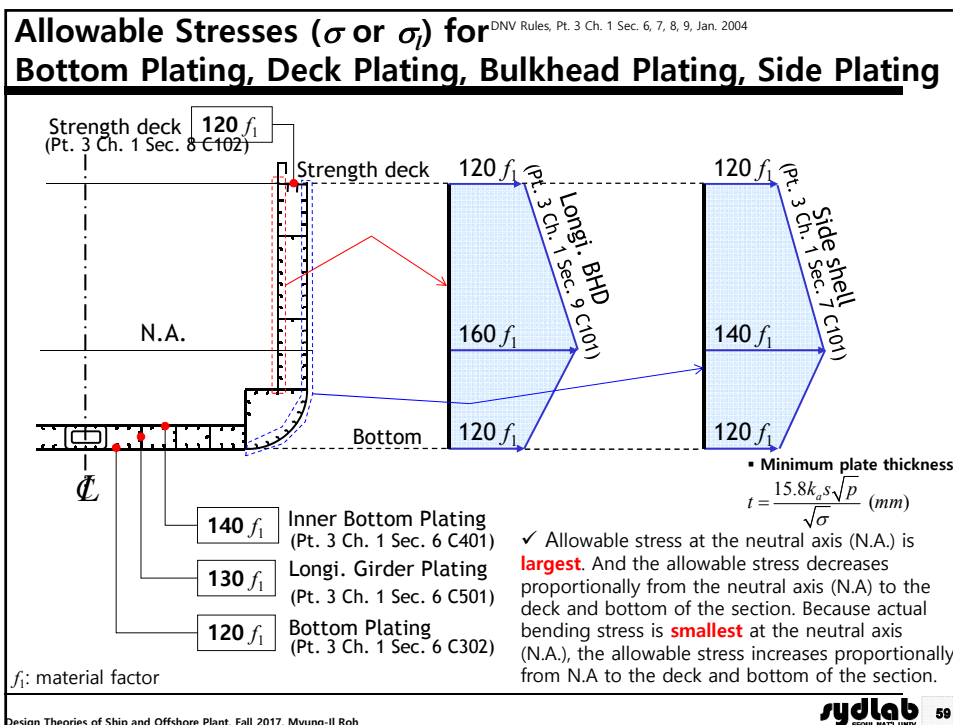
σ_{db} : Mean double bottom stress at plate flanges, normally not to be taken less than

- = 20 f_1 for cargo holds in general cargo vessel
- = 50 f_1 for holds for ballast
- = 85 f_1 b/B for tanks for liquid cargo

¹⁾ DNV Rules, Pt. 3 Ch. 1 Sec. 6 C800, Jan. 2004

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Allowable Stresses
- Longitudinal Stiffeners (1/6)

DNV Rules, Pt. 3 Ch. 1 Sec. 6, 7, 8, 9, Jan. 2004
DSME, DNV Rule Commentary Book, 1991.8

$$Z_{req.} = \frac{83I^2 \cdot s \cdot p \cdot w_k}{\sigma_l} \quad (cm^3)$$

✓ Calculation of $f_{2b,2d}$

$$f_{2b,2d} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$$

M_S : Largest design SWBM [kN·m]
 M_W : Rule VWBM in [kN·m]
 $Z_{b,d}$: Midship section modulus [cm³] at bottom or deck as-built

For example, 3,700 TEU Container Carrier: $I = 2.343e^{10} \text{ cm}^4$

Bottom: $y_B = 9.028e^2 \text{ cm}$
 $Z_B = 2.595e^7 \text{ cm}^3 \rightarrow f_{2b} = \frac{5.7(M_S + M_W)}{Z_b} = 1.030$

Deck: $y_D = 10.272e^2 \text{ cm}$
 $Z_D = 2.345e^7 \text{ cm}^3 \rightarrow f_{2d} = \frac{5.7(M_S + M_W)}{Z_d} = 1.140$

Section modulus of bottom is larger than that of deck, and thus the stress factor f_{2b} is smaller than f_{2d} .

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Allowable Stresses
- Longitudinal Stiffeners (2/6)

DNV Rules, Pt. 3 Ch. 1 Sec. 6, 7, 8, 9, Jan. 2004
DSME, DNV Rule Commentary Book, 1991.8

$$Z_{req.} = \frac{83I^2 \cdot s \cdot p \cdot w_k}{\sigma_l} \quad (cm^3)$$

✓ Calculation of Z_n, Z_a

Z_n : Vertical distance in [m] from the base line or deck line to the neutral axis of the hull girder, whichever is relevant

Z_a : Vertical distance in [m] from the base line or deck line to the point in question below or above the neutral axis

On Decks (Pt. 3 Ch. 1 Sec. 8 C301)

$$\sigma_l = 225f_1 - 130f_{2d} \frac{Z_n - Z_a}{Z_n}$$

$\sigma_{Longitudinal}$

$$f_{2b,2d} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$$

M_S : Largest design SWBM [kN·m]
 M_W : Rule VWBM in [kN·m]
 $Z_{b,d}$: Midship section modulus [cm³] at bottom or deck as-built

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Allowable Stresses

- Longitudinal Stiffeners (3/6)

DNV Rules, Pt. 3 Ch. 1 Sec. 6, 7, 8, 9, Jan. 2004
DSME, DNV Rule Commentary Book, 1991.8

$$Z_{req.} = \frac{83I^2 \cdot s \cdot p \cdot w_k}{\sigma_l} \quad (cm^3)$$

On Decks (Pt. 3 Ch. 1 Sec. 8 C301)

$$\sigma_l = 225f_1 - 130f_{2d} \frac{z_n - z_a}{z_n}$$

For example, 3,700 TEU Container Carrier:

$f_{2d} = 1.140$

Actual $f_1 = 1.28$ Assumption: $f_1 = 1.28$ $z_n = 10.272$, $z_a = 0.000 \rightarrow \frac{z_n - z_a}{z_n} = 1, \rightarrow \sigma_l = 139.800 \text{ [MPa]}$

Actual $f_1 = 1.0$ $z_n = 10.272$, $z_a = 3.712 \rightarrow \frac{z_n - z_a}{z_n} = 0.639, \rightarrow \sigma_l = 130.300 \text{ [MPa]}$

Actual $f_1 = 1.0$ $z_n = 10.272$, $z_a = 9.782 \rightarrow \frac{z_n - z_a}{z_n} = 0.048, \rightarrow \sigma_l = 217.886 \text{ [MPa]}$
 $= 160 \text{ [MPa]}$
 (Maximum: 160)

N.A.

\mathcal{L}

$f_{2b,2d} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$ M_S : Largest design SWBM [kN-m]
 M_W : Rule VWBM in [kN-m]
 $Z_{b,d}$: Midship section modulus [cm³] at bottom or deck as-built

Z_n : Vertical distance in m from the base line or deck line to the neutral axis of the hull girder, whichever is relevant
 Z_a : Vertical distance in m from the baseline or deck line to the point in question below or above the neutral axis

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Allowable Stresses

- Longitudinal Stiffeners (4/6)

DNV Rules, Pt. 3 Ch. 1 Sec. 6, 7, 8, 9, Jan. 2004
DSME, DNV Rule Commentary Book, 1991.8

$$Z_{req.} = \frac{83I^2 \cdot s \cdot p \cdot w_k}{\sigma_l} \quad (cm^3)$$

On Decks (Pt. 3 Ch. 1 Sec. 8 C301)

$$225f_1 - 130f_{2d} \frac{z_n - z_a}{z_n}$$

For example, 3,700 TEU Container Carrier:

$f_1 = 1.28$ $f_{2d} = 1.140$
 $z_n = 10.272$, $z_a = 0.000 \rightarrow \frac{z_n - z_a}{z_n} = 1, \rightarrow \sigma_l = 139.8 \text{ [MPa]}$

On Double Bottom (Pt. 3 Ch. 1 Sec. 6 C701)

$$225f_1 - 130f_{2b} - 0.7\sigma_{db}$$

For example, 3,700 TEU Container Carrier, Assumption: $\sigma_{db} = 0$

$f_1 = 1.28$ $f_{2b} = 1.030$
 $z_n = 9.208$, $z_a = 0.000 \rightarrow \frac{z_n - z_a}{z_n} = 1, \rightarrow \sigma_l = 154.1 \text{ [MPa]}$

N.A.

\mathcal{L}

$f_{2b,2d} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$ M_S : Largest design SWBM [kN-m]
 M_W : Rule VWBM in [kN-m]
 $Z_{b,d}$: Midship section modulus [cm³] at bottom or deck as-built

Z_n : vertical distance in m from the base line or deck line to the neutral axis of the hull girder, whichever is relevant
 Z_a : vertical distance in m from the base line or deck line to the point in question below or above the neutral axis
 σ_{db} : mean double bottom stress at plate flanges, normally not to be taken less than
 = $20 f_1$ for cargo holds in general cargo vessel
 = $50 f_1$ for holds for ballast
 = $85 f_1$ b/B for tanks for liquid cargo

Allowable stresses at deck are **smaller** than those at bottom, because the distance from N.A. to deck is **longer** than N.A. to bottom.

If the mean double bottom stress (σ_{db}) is considered as 20,
 $\sigma_l = 225f_1 - 130f_{2b} - 0.7\sigma_{db}$
 $= 225 \times 1.28 - 130 \times 1.030 - 0.7 \times 20 = 140.1 \text{ [MPa]}$

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Allowable Stresses - Longitudinal Stiffeners (5/6)

DNV Rules, Pt. 3 Ch. 1 Sec. 6, 7, 8, 9, Jan. 2004
DSME, DNV Rule Commentary Book, 1991.8

$$Z_{req.} = \frac{83I^2 \cdot s \cdot p \cdot w_k}{\sigma_i} \quad (cm^3)$$

On Side Shell
(Pt. 3 Ch. 1 Sec. 7 C301)

$$\frac{225f_1 - 130f_2 \frac{Z_n - Z_a}{Z_n}}{130}$$

$\sigma_{Longitudinal}$

130
which is lesser.

Max $130 f_1$

$225 f_1$

$$f_{2b,2d} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$$

M_S : Largest design SWBM [kN-m]
 M_W : Rule VWBM in [kN-m]
 $Z_{b,d}$: Midship section modulus [cm³] at bottom or deck as-built

Z_n : Vertical distance in m from the base line or deck line to the neutral axis of the hull girder, whichever is relevant
 Z_a : Vertical distance in m from the baseline or deck line to the point in question below or above the neutral axis

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Allowable Stresses - Longitudinal Stiffeners (6/6)

DNV Rules, Pt. 3 Ch. 1 Sec. 6, 7, 8, 9, Jan. 2004
DSME, DNV Rule Commentary Book, 1991.8

$$Z_{req.} = \frac{83I^2 \cdot s \cdot p \cdot w_k}{\sigma_i} \quad (cm^3)$$

On Longitudinal Bulkhead
(Pt. 3 Ch. 1 Sec. 9 C201)

$$\frac{225f_1 - 130f_2 \frac{Z_n - Z_a}{Z_n}}{160}$$

$\sigma_{Longitudinal}$

160
which is lesser.

Max $160 f_1$

$225 f_1$

$$f_{2b,2d} = \frac{5.7(M_S + M_W)}{Z_{b,d}}$$

M_S : Largest design SWBM [kN-m]
 M_W : Rule VWBM in [kN-m]
 $Z_{b,d}$: Midship section modulus [cm³] at bottom or deck as-built

Z_n : Vertical distance in m from the base line or deck line to the neutral axis of the hull girder, whichever is relevant
 Z_a : Vertical distance in m from the baseline or deck line to the point in question below or above the neutral axis

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- Minimum plate thickness

$$t = \frac{15.8 k_a s \sqrt{p}}{\sqrt{\sigma}} \text{ (mm)}$$

- Minimum section modulus for longitudinal stiffener

$$Z = \frac{83 l^2 s p w_k}{\sigma} \text{ (cm}^3\text{)}$$

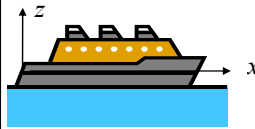
(3) Design Loads

Contents

- ☑ Ship Motion and Acceleration
- ☑ Combined Acceleration
- ☑ Design Probability Level
- ☑ Load Point
- ☑ Pressure & Force
 - Sea Pressure
 - Liquid Tank Pressure

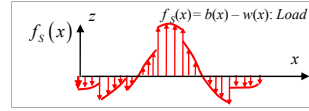
[Review] Loads in Wave

In still water



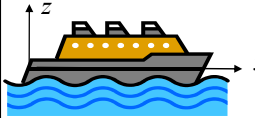
$$f_s(x) = b(x) + w(x)$$

$f_s(x)$: Distributed loads in longitudinal direction
 $f_b(x)$: Static longitudinal loads in longitudinal direction
 $f_w(x)$: Hydrodynamic longitudinal loads induced by wave

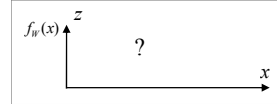


+

In wave



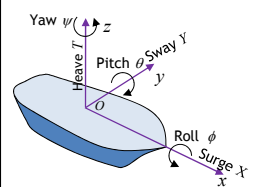
- ✓ Dynamic longitudinal loads
: Loads are induced by waves



- ✓ Direct calculation of dynamic longitudinal loads

- from 6DOF motion of ship
 $\mathbf{x} = [X, Y, T, \phi, \theta, \psi]^T$

✓ Ref. > 6 DOF motion of ship



$$f(x) = f_s(x) + f_w(x)$$

$$= b(x) + w(x) + f_D(x) + f_{F,K}(x) + f_R(x)$$

where,

$$f_R(x) = -a(x)\ddot{\mathbf{x}} - b(x)\dot{\mathbf{x}}$$

additional loads in wave

$f_D(x)$: Diffraction force at x
 $f_R(x)$: Radiation force at x by damping and added mass
 $f_{F,K}(x)$: Froude-Krylov force at x

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[Review] 6 DOF Equation of Motion of Ship



How to know $\ddot{\mathbf{x}}, \dot{\mathbf{x}}$?

By solving equations of motion, we can get the velocities and accelerations.

- ✓ Pressure acting on hull

$$\text{Linearized Bernoulli Eq. } P_{\text{Fluid}} = -\rho g z - \rho \frac{\partial \Phi}{\partial t} = -\rho g z - \rho \left(\frac{\partial \Phi_L}{\partial t} + \frac{\partial \Phi_D}{\partial t} + \frac{\partial \Phi_R}{\partial t} \right)$$

- ✓ Fluid force acting on hull

$$\mathbf{F}_{\text{Fluid}} = \iint_{S_b} P \mathbf{n} dS = - \iint_{S_b} \rho g z \mathbf{n} dS - \rho \iint_{S_b} \left(\frac{\partial \Phi_L}{\partial t} + \frac{\partial \Phi_D}{\partial t} + \frac{\partial \Phi_R}{\partial t} \right) dS$$

$$= \mathbf{F}_{\text{Static}} + \mathbf{F}_{F,K} + \mathbf{F}_D + \mathbf{F}_R$$

- ✓ 6 D.O.F equations of motion of a ship in waves

Newton's 2nd Law

$$\mathbf{M}\ddot{\mathbf{x}} = \sum \mathbf{F} = \mathbf{F}_{\text{Body}} + \mathbf{F}_{\text{Surface}} + \mathbf{F}_{\text{External}}$$

$$= \mathbf{F}_{\text{Gravity}} + \mathbf{F}_{\text{Fluid}} + \mathbf{F}_{\text{External}}$$

$$\mathbf{M}\ddot{\mathbf{x}} = \mathbf{F}_{\text{Gravity}} + \mathbf{F}_{\text{Static}} + \mathbf{F}_{F,K} + \mathbf{F}_D + \mathbf{F}_R + \mathbf{F}_{\text{External, dynamic}} + \mathbf{F}_{\text{External, static}}$$

$$\mathbf{M}\ddot{\mathbf{x}} = (\mathbf{F}_{\text{Gravity}} + \mathbf{F}_{\text{Static}}) + \mathbf{F}_{\text{Wave exciting}} - \mathbf{A}\ddot{\mathbf{x}} - \mathbf{B}\dot{\mathbf{x}} + \mathbf{F}_{\text{External, dynamic}} + \mathbf{F}_{\text{External, static}}$$

$\mathbf{F}_{\text{Restoring}} \quad \mathbf{F}_{\text{Wave exciting}} \quad \mathbf{F}_R = -\mathbf{A}\ddot{\mathbf{x}} - \mathbf{B}\dot{\mathbf{x}}$
 Added mass Damping Coefficient

$$\text{Linearization, } (\mathbf{F}_{\text{Restoring}} = (\mathbf{F}_{\text{Gravity}} + \mathbf{F}_{\text{Static}}) \approx -\mathbf{C}\mathbf{x})$$

$$(\mathbf{M} + \mathbf{A})\ddot{\mathbf{x}} + \mathbf{B}\dot{\mathbf{x}} + \mathbf{C}\mathbf{x} = \mathbf{F}_{\text{Wave exciting}} + \mathbf{F}_{\text{External, dynamic}} + \mathbf{F}_{\text{External, static}}$$

By solving equations of motion, we can get the velocities and accelerations of the ship!

$F_{F,K}$: Froude-Krylov force
 F_D : Diffraction force
 F_R : Radiation force

Φ_L : Incident wave velocity potential

Φ_D : Diffraction wave velocity potential

Φ_R : Radiation wave velocity potential

\mathbf{M}_A : 6×6 added mass matrix

\mathbf{B} : 6×6 damping coeff. matrix

\mathbf{C} : 6×6 restoring coeff. matrix

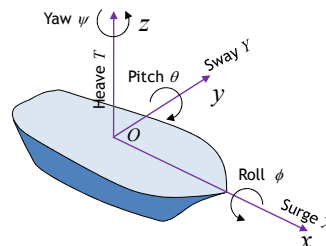
(1) Ship Motion and Acceleration - Empirical Formula of DNV Rule

DNV Rules, Pt. 3 Ch. 1 Sec. 4, Jan. 2004

✓ Ref. 6 DOF motion of ship

Common Acceleration Parameter	$a_0 = \frac{3C_W}{L} + C_V C_{V1}$
Surge Acceleration	$a_x = 0.2g_0 a_0 \sqrt{C_b}$
Combined Sway/Yaw Acceleration	$a_y = 0.3g_0 a_0$
Heave Acceleration	$a_z = 0.7g_0 \frac{a_0}{\sqrt{C_b}}$
Tangential Roll Acceleration	$a_r = \phi \left(\frac{2\pi}{T_r} \right)^2 R_r$
Tangential Pitch Acceleration	$a_p = \theta \left(\frac{2\pi}{T_p} \right)^2 R_p$

g_0 : standard acceleration of gravity
= 9.81 m/s²

Common Acceleration Parameter, a_0

$$a_0 = \frac{3C_W}{L} + C_V C_{V1}$$

$C_r = \frac{\sqrt{L}}{50}$, maximum 0.2
 $C_{r1} = \frac{V}{\sqrt{L}}$, minimum 0.8

 C_W = Wave coefficient

L	C_W
$L \leq 100$	$0.0792 \cdot L$
$100 < L < 300$	$10.75 - [(300 - L)/100]^{1/2}$
$300 \leq L \leq 350$	10.75
$L > 350$	$10.75 - [(L - 350)/150]^{1/2}$

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(1) Ship Motions and Accelerations - Roll Angle & Roll Period

DNV Rules, Pt. 3 Ch. 1 Sec. 4, Jan. 2004

✓ Roll angle

$$\phi = \frac{50c}{B + 75} \quad (\text{rad})$$

$c = (1.25 - 0.025 T_R) k$
 $k = 1.2$ for ships without bilge keel
 $= 1.0$ for ships with bilge keel
 $= 0.8$ for ships with active roll damping facilities
 $T_R =$ as defined in 402, not to be taken greater than 30.

✓ Roll period

$$T_R = \frac{2k_r}{\sqrt{GM}} \quad (\text{s})$$

$k_r = 0.39B$ for ships with even transverse distribution of mass
 $= 0.35B$ for tankers in ballast
 $= 0.25B$ for ships loaded with ore between longitudinal bulkheads
 $GM = 0.07B$ in general
 $= 0.12B$ for tankers and bulk carriers

✓ Pitch angle

$$\theta = 0.25 \frac{a_0}{C_B} \quad (\text{rad})$$

$$a_0 = \frac{3C_W}{L} + C_V C_{V1}$$

✓ Pitch period

$$T_P = 1.8 \sqrt{\frac{L}{g_0}} \quad (\text{s})$$

g_0 : standard acceleration of gravity
= 9.81 m/s²

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(2) Combined Acceleration

- Combined Vertical Acceleration (a_v)

✓ The acceleration along the ship's vertical axis considering combined effect of heave, pitch & roll motion¹⁾

$$a_v = \frac{k_v g_o a_o}{C_b}$$

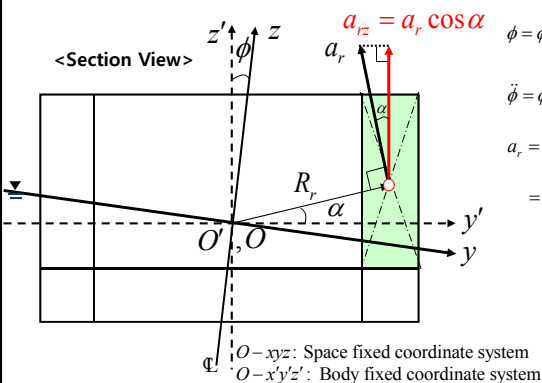
K_v = Acceleration distribution factor along the length of vessel
= 0.7 between 0.3L and 0.6L from A.P.
 a_o = Common Acceleration Parameter

$$a_v = \max \left\{ \sqrt{a_z^2 + a_{rz}^2}, \sqrt{a_z^2 + a_{pz}^2} \right\}$$

Heave $a_z = 0.7 g_o \frac{a_o}{C_b}$
Acceleration⁴⁾

Vertical component of
tangential roll acceleration

Vertical component of
tangential pitch
acceleration



a_r : tangential roll acceleration
 R_r : distance in m from the center of the mass
to the axis of rotation
 α : angle of center of mass about
the body fixed coordinate system
 ϕ : roll angle
 ϕ^A : roll angle amplitude²⁾
 T_R : period of roll³⁾
 g_o : standard acceleration of gravity
= 9.81 m/s²

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(2) Combined Acceleration

- Combined Vertical Acceleration (a_v)

✓ The acceleration along the ship's vertical axis considering combined effect of heave, pitch & roll motion¹⁾

$$a_v = \frac{k_v g_o a_o}{C_b}$$

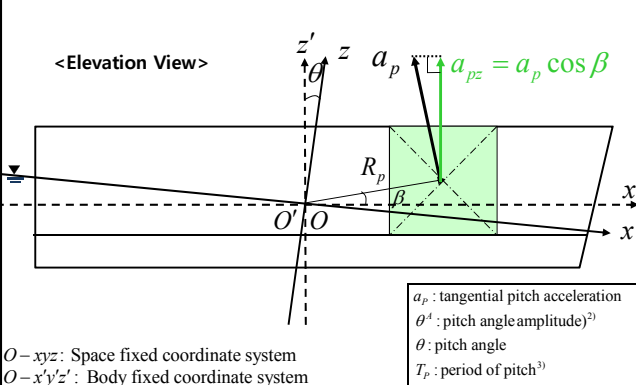
K_v = Acceleration distribution factor along the length of vessel
= 0.7 between 0.3L and 0.6L from A.P.
 a_o = Common Acceleration Parameter

$$a_v = \max \left\{ \sqrt{a_z^2 + a_{rz}^2}, \sqrt{a_z^2 + a_{pz}^2} \right\}$$

Heave $a_z = 0.7 g_o \frac{a_o}{C_b}$
acceleration⁴⁾

Vertical component of
tangential roll acceleration

Vertical component of
tangential pitch
acceleration



a_p : tangential pitch acceleration
 θ^A : pitch angle amplitude²⁾
 θ : pitch angle
 T_p : period of pitch³⁾

R_p : distance in m from the center of the mass
to the axis of rotation
 β : angle of center of mass about
the body fixed coordinate system

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1) DNV Rules, Pt. 3 Ch. 1 Sec. 4 B700, Jan. 2004

(2) Combined Acceleration

- Combined Transverse Acceleration (a_t)

✓ The acceleration along the ship's transverse axis considering combined effect of sway, yaw & roll motion¹⁾

$$a_t = \sqrt{a_y^2 + (g_0 \sin \phi + a_{ry})^2}$$

Combined sway & yaw acceleration
 $a_y = 0.3g_0a_0$
Transverse component of acceleration of gravity by roll angle
Transverse component of the tangential roll acceleration

<Section View>

$a_{ry} = a_r \sin \alpha$

$a_r = \phi^4 \left(\frac{2\pi}{T_r} \right)^2 R_r$

a_r : tangential roll acceleration
 R_r : distance in m from the center of the mass to the axis of rotation
 ϕ : roll angle
 ϕ^4 : roll angle amplitude
 T_r : period of roll³⁾
 g_0 : standard acceleration of gravity
 $= 9.81 \text{ m/s}^2$

$O-xyz$: Space fixed coordinate system
 $O-x'y'z'$: Body fixed coordinate system

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1) DNV Rules, Pt. 3 Ch. 1 Sec. 4 B800, Jan. 2004

(2) Combined Acceleration

- Combined Longitudinal Acceleration (a_l)

✓ The acceleration along the ship's longitudinal axis considering combined effect of surge & pitch motion¹⁾

$$a_l = \sqrt{a_x^2 + (g_0 \sin \theta + a_{px})^2}$$

Surge acceleration
 $a_x = 0.2g_0a_0\sqrt{C_b}$
Longitudinal component of gravitational acceleration by pitch angle
Longitudinal component of the pitch acceleration

<Elevation View>

$a_{px} = a_p \sin \alpha$

$a_p = \theta^4 \left(\frac{2\pi}{T_p} \right)^2 R_p$

a_p : tangential pitch acceleration
 R_p : distance in m from the center of the mass to the axis of rotation
 θ : pitch angle
 θ^4 : pitch angle amplitude²⁾
 T_p : period of pitch³⁾
 g_0 : standard acceleration of gravity
 $= 9.81 \text{ m/s}^2$

$O-xyz$: Space fixed coordinate system
 $O-x'y'z'$: Body fixed coordinate system

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(2) Combined Acceleration

- [Example] Vertical Acceleration

(Example) Calculate the vertical acceleration of a given ship at 0.5L (amidships) by DNV Rule.

[Dimension] $L_s=315.79$ m, $V=15.5$ knots, $C_B=0.832$

$$a_v = \frac{k_v g_0 a_0}{C_b}$$

K_v = Acceleration distribution factor along the length of vessel
 = 0.7 between 0.3L and 0.6L from A.P.
 a_0 = Common Acceleration Parameter
 g_0 = Standard acceleration of gravity ($=9.81\text{m/sec}^2$)

$$\begin{aligned} \text{(Sol.) } a_v &= (k_v g_0 a_0) / C_B = (0.7 \times 9.81 \times 0.277) / 0.832 \\ &= 2.286 \text{ (m/sec}^2\text{)} \end{aligned}$$

where, $k_v = 0.7$ at mid ship

$$a_0 = 3 C_W / L + C_v C_{v1} = 3 \times 10.75 / 315.79 + 0.2 \times 0.872 = 0.277$$

$$C_v = L^{0.5} / 50 = 315.79^{0.5} / 50 = 0.355 \text{ or Max. } 0.2$$

$$= 0.2$$

$$C_{v1} = V / L^{0.5} = 15.5 / 315.79^{0.5} = 0.872 \text{ or Min. } 0.8$$

$$= 0.872$$

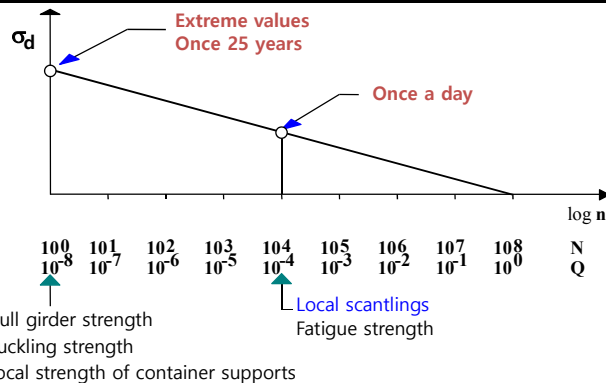
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(3) Design Probability Level

1) DNV, Fatigue Assessment of Ship Structures, p.18, 2003

Probability Level¹⁾



Design Probability Level²⁾

- ✓ Number of waves that the ship experiences during the ship's life (for 25 years): about 10^8
 - ➔ The ship is designed to endure the extreme wave (10^{-8} probability) which the ship encounters once for 25 years.
 - (Extreme condition: Ship motion and acceleration are given as extreme values.)
- ✓ In case of design pressure, use the reduced value of 10^{-4} (Reduction value = $0.5 \times$ Extreme value)

Ex) Liquid Tank Pressure: Pressure, P_1 , considering vertical acceleration

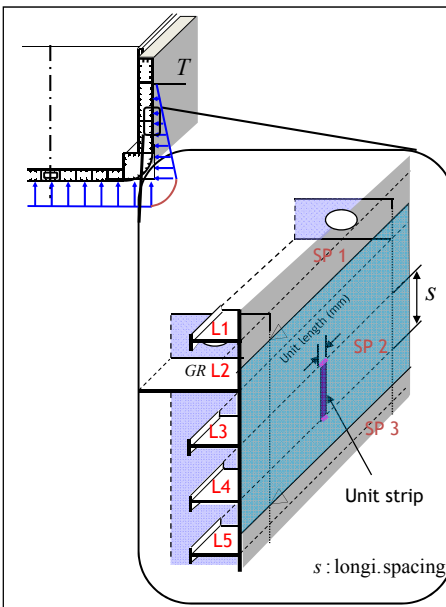
$$p_1 = \rho (g_0 + 0.5 a_v) h$$

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(4) Load Point
- Horizontally Stiffened Plate

DNV Rules, Pt. 3 Ch. 1 Sec. 4 A202, Jan. 2004

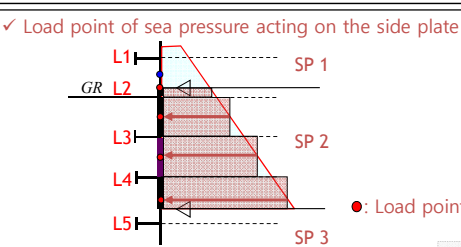
$$p_s = \rho g_o \left[0.67(h_t + \phi b) - 0.12\sqrt{H b_t \phi} \right]$$


✓ **The pressure at the load point** is considered as **uniform load** of unit strip

✓ **Definition of load point**

- General
: Midpoint of stiffened plate field
- Seam & butt (In case two plates are welded)
 - When considered plate includes the midpoint of stiffened plate field
: Midpoint of stiffened plate field
 - When considered plate does not include the midpoint of stiffened plate field
: Nearest seam or butt line from midpoint

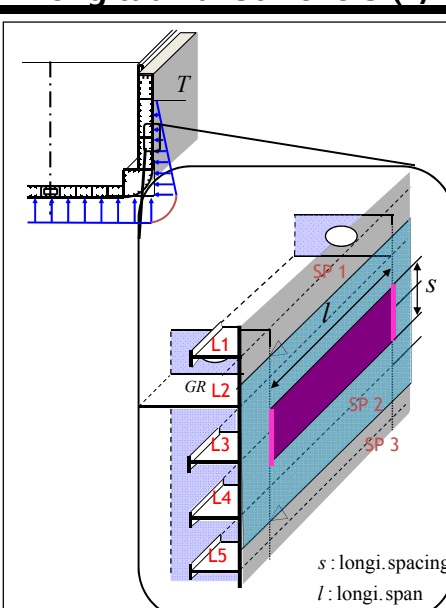
✓ **Load point of sea pressure acting on the side plate**



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(4) Load Point
- Longitudinal Stiffeners (1/2)

DNV Rules, Pt. 3 Ch. 1 Sec. 4 A202, Jan. 2004

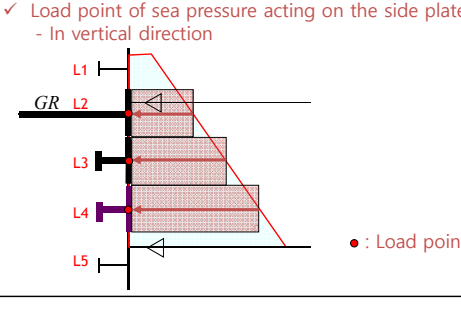


✓ **The pressure at the load point** is considered as **uniform load**

✓ **Definition of load point**

- In vertical direction
: The point of intersection between a plate and a stiffener
- In longitudinal direction
: Midpoint of span

✓ **Load point of sea pressure acting on the side plate - In vertical direction**

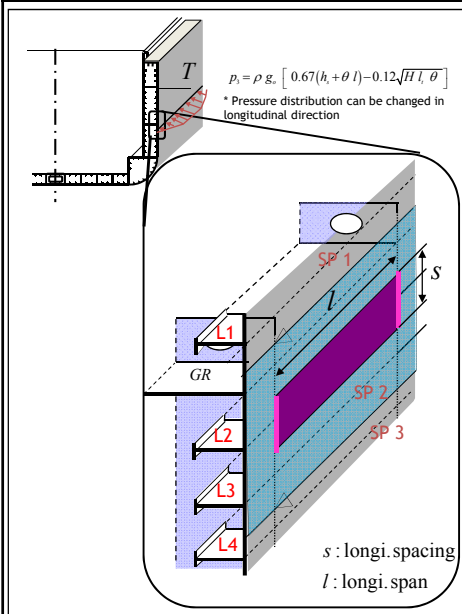


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(4) Load Point

- Longitudinal Stiffeners (2/2)

DNV Rules, Pt. 3 Ch. 1 Sec. 4 A202, Jan. 2004



✓ The pressure at the load point is considered as uniform load

✓ Definition of load point

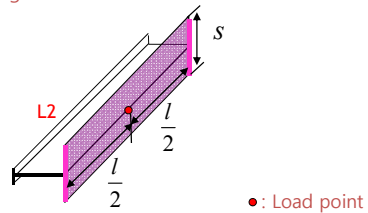
1. In vertical direction

: The point of intersection between a plate and a stiffener

2. In longitudinal direction

: Midpoint of span

✓ Load point of sea pressure acting on the side plate - In longitudinal direction



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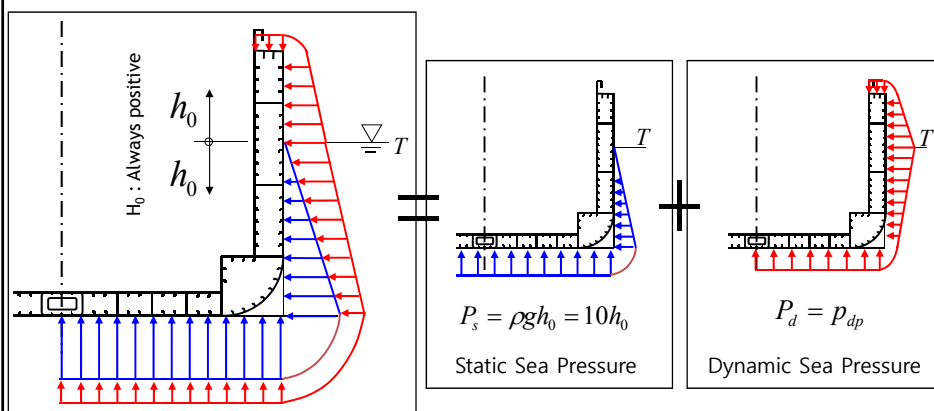
(5) Pressure and Force

- Sea Pressure

DNV Rules, Pt. 3 Ch. 1 Sec. 4 C201, Jan. 2004

✓ Sea pressures = Static sea pressure + Dynamic sea pressure

$$P = P_s + P_d$$



(5) Pressure and Force

- Liquid Tank Pressure (1/7)

DNV Rules, Pt. 3 Ch. 1 Sec. 4 C300, Jan. 2004

✓ The pressure in full tanks shall be taken as the greater of $p_1 \sim p_5$ ¹⁾

$$p_1 = \rho (g_0 + 0.5a_v) h_s$$

P_1 : Considering vertical acceleration

$$p_2 = \rho g_0 \left[0.67(h_s + \phi b) - 0.12\sqrt{H b_l \phi} \right]$$

P_2 : Considering rolling motion

$$p_3 = \rho g_0 \left[0.67(h_s + \theta l) - 0.12\sqrt{H l_l \theta} \right]$$

P_3 : Considering pitching motion

$$p_4 = 0.67(\rho g_0 h_p + \Delta P_{dyn})$$

P_4 : Considering overflow

$$p_5 = \rho g_0 h_s + p_o$$

P_5 : Considering tank test pressure

a_v : Vertical acceleration

ϕ : Roll angle

b : The largest athwart ship distance in [m] from the load point to the tank corner at top of tank

h_l & l_l : Breadth and length in [m] of top of tank

ρ : Density of liquid cargo

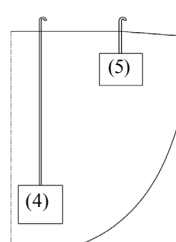
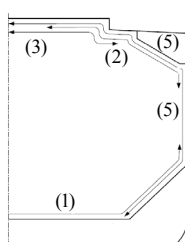
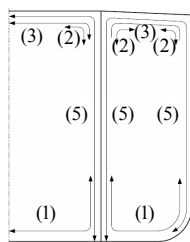
h_s : Vertical distance from the load point to tank top in tank

h_p : Vertical distance from the load point to the top of air pipe

p_o : 25 kN/m² general

Δp_{dyn} : Calculated pressure drop

✓ Maximum pressure is different depending on locations



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(5) Pressure and Force

- Liquid Tank Pressure (2/7)

$$p_1 = \rho (g_0 + 0.5a_v) h_s$$

P_1 : Considering vertical acceleration

$$p_2 = \rho g_0 \left[0.67(h_s + \phi b) - 0.12\sqrt{H b_l \phi} \right]$$

P_2 : Considering rolling motion

$$p_3 = \rho g_0 \left[0.67(h_s + \theta l) - 0.12\sqrt{H l_l \theta} \right]$$

P_3 : Considering pitching motion

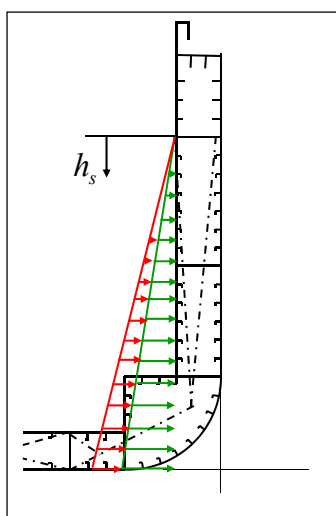
$$p_4 = 0.67(\rho g_0 h_p + \Delta P_{dyn})$$

P_4 : Considering overflow

$$p_5 = \rho g_0 h_s + p_o$$

P_5 : Considering tank test pressure

✓ Design pressure P_1 considering vertical acceleration (General)



$$P_1 = \underbrace{\rho g_0 h_s}_{\text{Static Pressure}} + \underbrace{0.5 \rho a_v h_s}_{\text{Dynamic Pressure}}$$

Static Pressure Dynamic Pressure

Reduced value of 10^{-4} by probability level is used.
(Reduction value = $0.5 \times$ Extreme value)

$$p = \rho (g_0 + 0.5a_v) h_s$$

a_v : Vertical acceleration

h_s : vertical distance in m from load point to top of tank

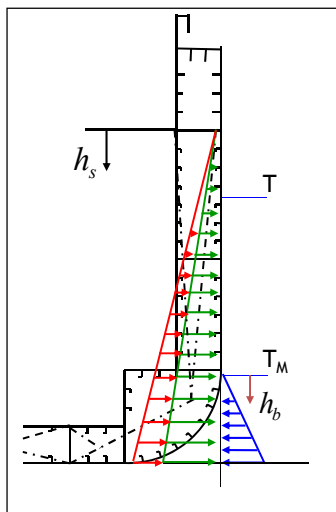
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(5) Pressure and Force - Liquid Tank Pressure (3/7)

$p_1 = \rho(g_0 + 0.5a_v)h_s$	P ₁ : Considering vertical acceleration
$p_2 = \rho g_s \left[0.67(h_s + \phi b) - 0.12\sqrt{H} h_s \phi \right]$	P ₂ : Considering rolling motion
$p_3 = \rho g_s \left[0.67(h_s + \theta l) - 0.12\sqrt{H} l \theta \right]$	P ₃ : Considering pitching motion
$p_4 = 0.67(\rho g h_s + \Delta P_s)$	P ₄ : Considering overflow
$p_5 = \rho g_s h_s + p_s$	P ₅ : Considering tank test pressure

✓ Design pressure P_1 considering vertical acceleration (In case of side shell)



In case of side shell, the effect of sea pressure is considered.

$$P = \underbrace{\rho g_0 h_s}_{\text{Static Pressure}} + \underbrace{0.5 \rho a_v h_s}_{\text{Dynamic Pressure}} - \underbrace{10 h_b}_{\text{Sea Pressure}}$$

When we consider the design pressure, the largest value shall be applied. The liquid cargo pressure acting on the side shell is the highest when the sea pressure is the lowest, i.e. in case of minimum draft.

$$p = \rho(g_0 + 0.5a_v)h_s - 10h_b$$

h_b : vertical distance in m from load point to minimum design draft
 $= 2 + 0.02L$ for Tanker
 $= 0.35 T$ for Dry Cargo
 $(T$: Rule Draft)

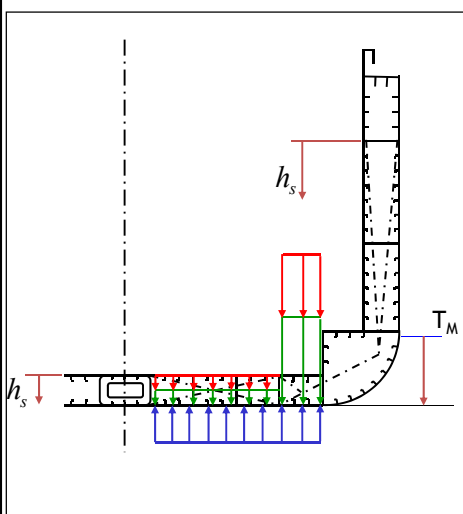
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(5) Pressure and Force - Liquid Tank Pressure (4/7)

$p_1 = \rho(g_0 + 0.5a_v)h_s$	P ₁ : Considering vertical acceleration
$p_2 = \rho g_s \left[0.67(h_s + \phi b) - 0.12\sqrt{H} h_s \phi \right]$	P ₂ : Considering rolling motion
$p_3 = \rho g_s \left[0.67(h_s + \theta l) - 0.12\sqrt{H} l \theta \right]$	P ₃ : Considering pitching motion
$p_4 = 0.67(\rho g h_s + \Delta P_s)$	P ₄ : Considering overflow
$p_5 = \rho g_s h_s + p_s$	P ₅ : Considering tank test pressure

✓ Design pressure P_1 considering vertical acceleration (In case of bottom shell)



In case of bottom shell, the effect of sea pressure is considered

$$P = \underbrace{\rho g_0 h_s}_{\text{Static Pressure}} + \underbrace{0.5 \rho a_v h_s}_{\text{Dynamic Pressure}} - \underbrace{10 T_M}_{\text{Sea Pressure}}$$

When we consider the design pressure, the largest value shall be applied. The liquid cargo pressure acting on the bottom shell is the highest when the sea pressure is the lowest, i.e. in case of minimum draft.

$$p = \rho(g_0 + 0.5a_v)h_s - 10T_M$$

T_M : vertical distance in m from load point to minimum design draft
 $= 2 + 0.02L$ for Tanker
 $= 0.35 T$ for Dry Cargo
 $(T$: Rule Draft)

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(5) Pressure and Force

DNV Rules, Pt. 3 Ch. 1 Sec. 4 B800, Jan. 2004

- Example) Calculation of P_1 Pressure

(Example) When the tank is filled up, calculate the P_1 pressure of inner bottom and deck by using vertical acceleration ($a_v=2.286 \text{ m/s}^2$) and dimensions of tank which is given below.

[Dimension] Inner bottom height: 3.0 m, Deck height: 31.2m, $\rho = 1.025 \text{ ton/m}^3$

$$P_1 = \rho(g_0 + 0.5a_v)h_s$$

ρ = density (ton/m^3)

a_v = Vertical acceleration

g_0 = Standard acceleration of gravity ($=9.81 \text{ m/sec}^2$)

h_s : virtual distance in m from load point to top of tank

(Sol.) $a_v = 2.286 \text{ m/s}^2$

① Inner Bottom

$$h_s = 31.2 - 3.0 = 28.8 \text{ m}$$

$$\begin{aligned} P_1 &= \rho(g_0 + 0.5a_v)h_s \\ &= 1.025(9.81 + 0.5 \times 2.286) \times 28.2 \\ &= 316.6 \text{ kN/m}^2 \end{aligned}$$

② Deck

$$h_s = 31.2 - 31.2 = 0 \text{ m}$$

$$\begin{aligned} P_1 &= \rho(g_0 + 0.5a_v)h_s \\ &= 1.025(9.81 + 0.5 \times 2.286) \times 0 \\ &= 0 \text{ kN/m}^2 \end{aligned}$$

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(5) Pressure and Force

- Liquid Tank Pressure (5/7)

$$p_1 = \rho(g_0 + 0.5a_v)h_s$$

P_1 : Considering vertical acceleration

$$p_2 = \rho g_0 [0.67(h_s + \phi b) - 0.12\sqrt{H\phi b_t}]$$

P_2 : Considering rolling motion

$$p_3 = \rho g_0 [0.67(h_s + \phi l) - 0.12\sqrt{H\phi l_t}]$$

P_3 : Considering pitching motion

$$p_4 = 0.67(\rho g_0 h_s + \Delta P_{ov})$$

P_4 : Considering overflow

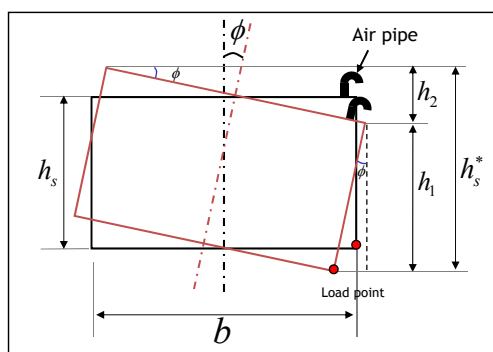
$$p_5 = \rho g_0 h_s + p_{ov}$$

P_5 : Considering tank test pressure

DSME 선박구조설계 5-3

DNV Rules, Pt. 3 Ch. 1 Sec. 4, Jan. 2004

✓ Design pressure P_2 considering the rolling motion



When the ship is rolling, the higher static pressure is applied.

Assumption: $\phi \ll 1$

$$h_1 = h_s \cos \phi \approx h_s$$

$$h_2 = b \sin \phi \approx b \phi$$

$$\begin{aligned} \therefore h_s^* &= h_1 + h_2 \\ &= (h_s + b \phi) \end{aligned}$$

$$p_2 = \rho g_0 [0.67(h_s + \phi b) - 0.12\sqrt{H\phi b_t}]$$

H : Height in m of the tank
 b : Breadth in m of top of tank

In case of rolling of a ship, two third (≈ 0.67) of actual pressure is applied considering pressure drop by overflow.

The filling ratio of the most tank is about 98%. That (about 2%) is considered.

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(5) Pressure and Force - Liquid Tank Pressure (6/7)

$$p_1 = \rho(g_z + 0.5a_z)h_i$$

$$p_2 = \rho g_z \left[0.67(h_i + \phi b) - 0.12\sqrt{H h_i \phi} \right]$$

$$p_3 = \rho g_z \left[0.67(h_i + \theta l) - 0.12\sqrt{H l \theta} \right]$$

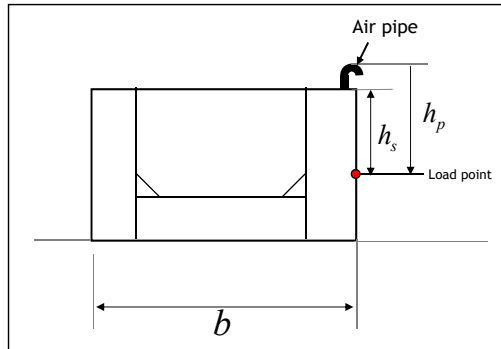
$$p_4 = 0.67(\rho g_0 h_p + \Delta P_{dyn})$$

$$p_5 = \rho g_0 h_s + p_o$$

P₁: Considering vertical acceleration
P₂: Considering rolling motion
P₃: Considering pitching motion
P₄: Considering overflow
P₅: Considering tank test pressure

DSME, 선박구조설계 5-3
DNV Rules, Pt. 3 Ch. 1 Sec. 4, Jan. 2004

✓ Design pressure P₄ considering the tank overflow



The liquid of tank is filled up to air pipe in case of tank overflow.
So, h_p is used for calculating static pressure.

h_p = vertical distance in m from the load point to the top of air pipe

$$p = 0.67(\rho g_0 h_p + \Delta P_{dyn})$$

Calculated pressure drop
Generally, 25kN/m²

In case of rolling of a ship, two third (=0.67) of actual pressure is applied considering pressure drop by overflow.

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(5) Pressure and Force - Liquid Tank Pressure (7/7)

$$p_1 = \rho(g_z + 0.5a_z)h_i$$

$$p_2 = \rho g_z \left[0.67(h_i + \phi b) - 0.12\sqrt{H h_i \phi} \right]$$

$$p_3 = \rho g_z \left[0.67(h_i + \theta l) - 0.12\sqrt{H l \theta} \right]$$

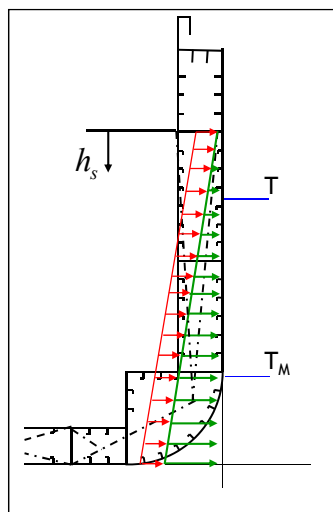
$$p_4 = 0.67(\rho g_0 h_p + \Delta P_{dyn})$$

$$p_5 = \rho g_0 h_s + p_o$$

P₁: Considering vertical acceleration
P₂: Considering rolling motion
P₃: Considering pitching motion
P₄: Considering overflow
P₅: Considering tank test pressure

DSME, 선박구조설계 5-3
DNV Rules, Pt. 3 Ch. 1 Sec. 4, Jan. 2004

✓ Design pressure P₅ considering the tank test pressure



Over-pressure is applied in order to have the water head of 'tank height + 2.5' [m] in case of tank test for leakage.
(Water head of over-pressure of tank test: 2.5m)

$$p = \rho g_0 h_s + p_o$$

$$p_o = \rho g_0 \times 2.5$$

$$= 10 \times 2.5$$

$$= 25 \text{ kN/m}^2$$

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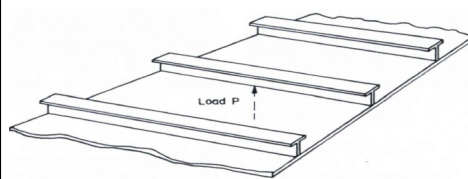
(4) Scantling of Plates

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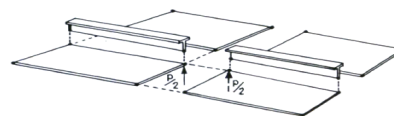
sydlab 91
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Scantling of Plates (1/3)

Use of eccentric beam element



(a) Beams attached to plating



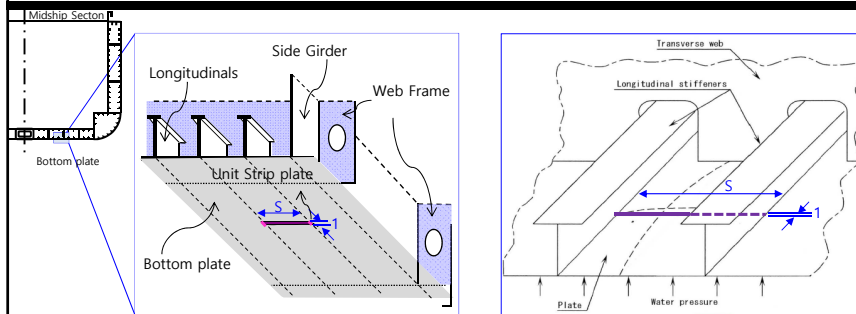
(b) Structural model using eccentric beam element

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Scantling of Plates (2/3)

p : "pressure" on the load point for the stiffener

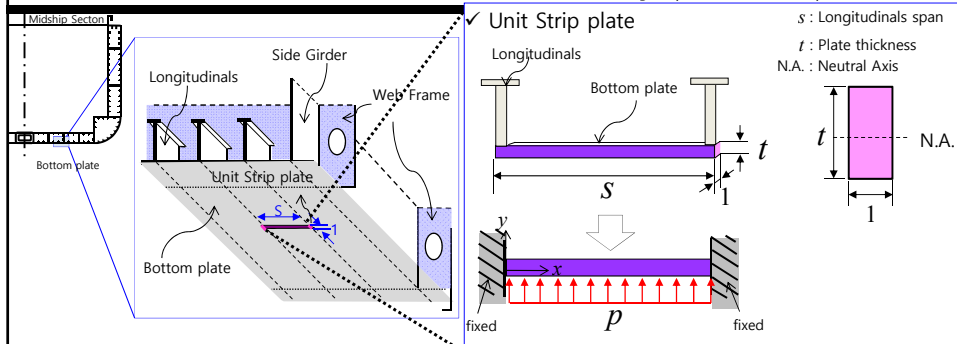


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Scantling of Plates (3/3)

p : "pressure" on the load point for the stiffener



Assumption 1. Cut off the **unit strip plate** supported by the **longitudinals** or **girder**. And consider the unit strip plate as a "**fixed-end beam**" which has a span ' s ', thickness ' t '.

Assumption 2. Consider the lateral load of the beam as a uniformly distributed load. (Assume the pressure on the load point as an intensity of uniformly distributed load.)

Assumption 3. The design of plates is based on the **plastic design**.

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Comparison between Stiffener and Plate

p : "pressure" on the load point for the stiffener

s : Stiffener spacing
 l : Stiffener span

s : Stiffener spacing

✓ Longitudinal stiffener attached to the plate

l : Stiffener span
 s : Stiffener spacing

$$M = \frac{1}{12} p \cdot s \cdot l^2$$

✓ Unit strip plate

s : Stiffener spacing
1 : Unit length of strip

$$M_p = \frac{1}{16} p \cdot 1 \cdot s^2$$

Comparison of the Elastic and Plastic Design of the Plate - Overview

Flexure formula

Plastic Design

Plastic moment (M_p)

$$M_p = \frac{p \cdot 1 \cdot s^2}{16}$$

Plastic section modulus (Z_p)

$$Z_p = \frac{1 \cdot t^2}{4} = \frac{t^2}{4}$$

Substituting formula:

$$\sigma = \frac{M_p}{Z_p} = \frac{ps^2}{4t^2}, t = \frac{s\sqrt{p}}{2\sqrt{\sigma}}$$

↓ assumption: $\sigma = \sigma_l$

$$t_{req.} = \frac{15.8 k_a s \sqrt{p}}{\sqrt{\sigma_l}} + t_k \text{ (mm)}$$

Elastic Design

Elastic moment (M)

$$M = \frac{p \cdot 1 \cdot s^2}{12}$$

Elastic section modulus (Z)

$$Z = \frac{1 \cdot t^2}{6} = \frac{t^2}{6}$$

Substituting formula:

$$\sigma = \frac{M}{Z} = \frac{ps^2}{2t^2}, t = \frac{s\sqrt{p}}{\sqrt{2}\sqrt{\sigma}}$$

↓ assumption: $\sigma = \sigma_l$

$$t_{req.} = \frac{22.4 k_a s \sqrt{p}}{\sqrt{\sigma_l}} + t_k \text{ (mm)}$$

k_a = correction factor for aspect ratio of plate field t_k = corrosion addition

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Comparison of the Elastic and Plastic Design

- [Example] Thickness Requirements

Plastic moment (M_p)

$$M_p = \frac{p \cdot l \cdot s^2}{16}$$

Plastic section modulus (Z_p)

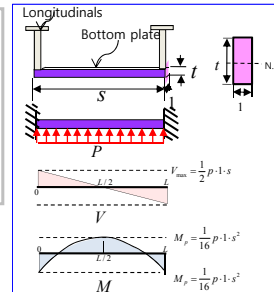
$$Z_p = \frac{1 \cdot t^2}{4} = \frac{t^2}{4}$$

Elastic moment (M)

$$M = \frac{p \cdot l \cdot s^2}{12}$$

Elastic section modulus (Z)

$$Z = \frac{1 \cdot t^2}{6} = \frac{t^2}{6}$$



① A mild steel plate carries the uniform pressure of 100 kN/m² on a span length of 800 mm.

Compare the **thickness requirement** depending on the plastic design and elastic design.

$$t_{req. plastic} = \frac{15.8 k_a s \sqrt{p}}{\sqrt{\sigma_l}} = \frac{15.8 \times 1 \times 0.8 \times \sqrt{100}}{\sqrt{235}} = 8.24 \text{ (mm)}$$

$$t_{req. elastic} = \frac{22.4 k_a s \sqrt{p}}{\sqrt{\sigma_l}} = \frac{22.4 \times 1 \times 0.8 \times \sqrt{100}}{\sqrt{235}} = 11.69 \text{ (mm)}$$

The **thickness requirement** of the plate **of plastic design** is **smaller than** that of the **elastic design** at the same pressure and on the same span.

k_a = correction factor for aspect ratio of plate field

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Comparison of the Elastic and Plastic Design

- [Example] Design Pressure

Plastic moment (M_p)

$$M_p = \frac{p \cdot l \cdot s^2}{16}$$

Plastic section modulus (Z_p)

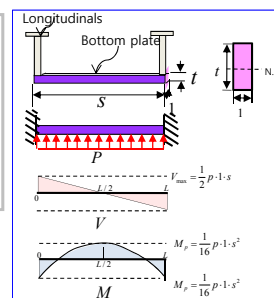
$$Z_p = \frac{1 \cdot t^2}{4} = \frac{t^2}{4}$$

Elastic moment (M)

$$M = \frac{p \cdot l \cdot s^2}{12}$$

Elastic section modulus (Z)

$$Z = \frac{1 \cdot t^2}{6} = \frac{t^2}{6}$$



② A mild steel plate has a thickness of 10 mm on a span length of 800 mm.

Compare the **design pressure** that the maximum stresses of the plate reaches the yield stress depending on the plastic design and elastic design.

$$p_{plastic} = \frac{t^2 \sigma_l}{15.8^2 s^2} = \frac{10^2 \times 235}{15.8^2 \cdot 0.8^2} = 147 \text{ [kN / m}^2\text{]}$$

$$p_{elastic} = \frac{t^2 \sigma_l}{22.4^2 s^2} = \frac{10^2 \times 235}{22.4^2 \cdot 0.8^2} = 73 \text{ [kN / m}^2\text{]}$$

The **design pressure** of **plastic design** that reaches the yield stress, is **higher** than that of the **elastic design** on the same span with the same thickness.

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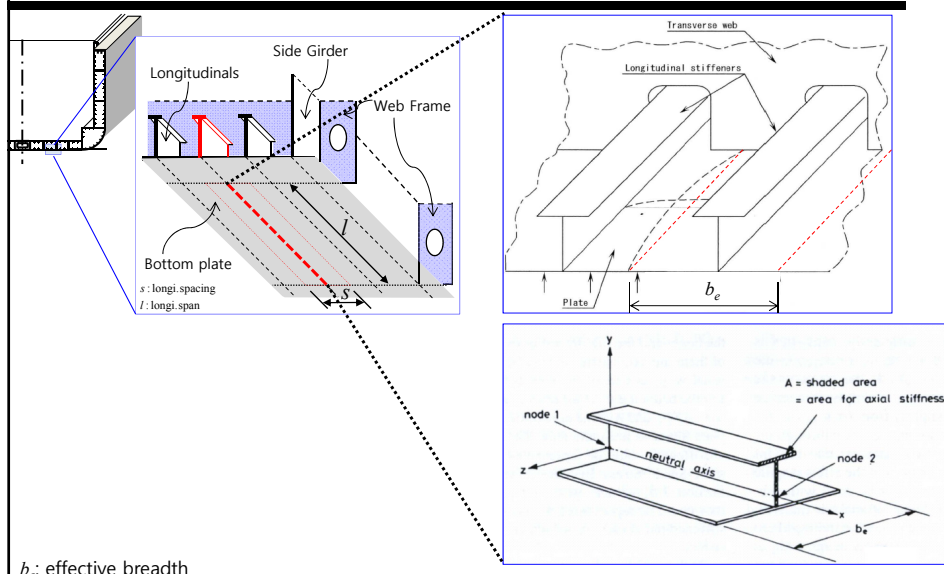
(5) Scantling of Stiffeners

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Scantling of Stiffeners (1/3)

p : "pressure" on the load point for the stiffener



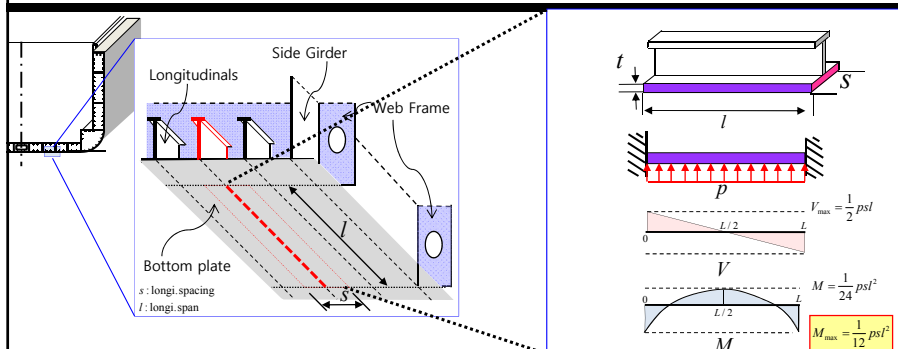
* Okumoto, Y., Takeda, Y., Mano, M., Design of Ship Hull Structures - a Practical Guide for Engineers, Springer, pp. 17-32, 2009

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Scantling of Stiffeners (2/3)

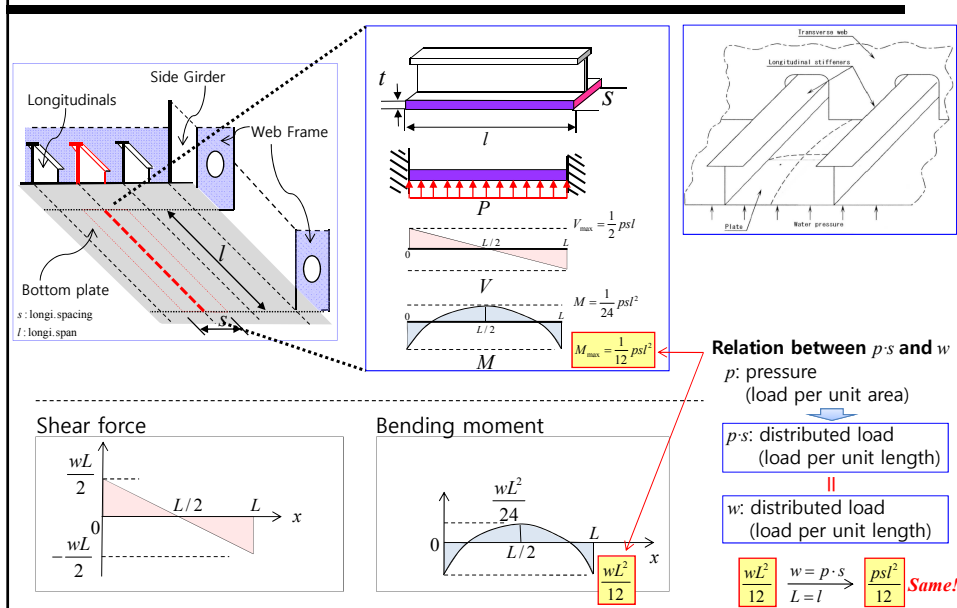
p : "pressure" on the load point for the stiffener



- Assumption 1. Cut off the **stiffener and attached plate with effective breadth**. Sectional properties of stiffener are calculated including attached plate.
- Assumption 2. Consider the stiffener and attached plate as a **"fixed-end beam"** supported by the **web frames**.
- Assumption 3. Consider the **lateral load** of the beam as a **uniformly distributed load**. (Assume the **"pressure"** on the load point as an intensity of uniformly distributed load.)
- Assumption 4. The design of stiffener is based on the **elastic design** (when the load is removed, the material returns to its original dimensions)

Scantling of Stiffeners (3/3)

p : "pressure" on the load point for the stiffener



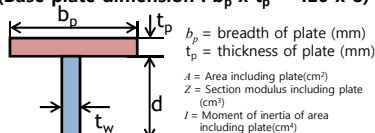
(6) Sectional Properties of Steel Sections

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sydlab 103

Sectional Properties of Steel Sections for Ship Building¹⁾ (1/12)

<Sectional properties of steel sections including attached plate>

¹⁾ "조선설계편람", 제 4판 (일본어), 일본관서조선협회, 1996(Base plate dimension : $b_p \times t_p = 420 \times 8$)

d	t_w	6	9	11	12.7	14
50	A	3.00	4.5	5.50	6.35	7.00
	Z	6.05	8.81	10.6	12.1	13.3
	I	31.2	44.5	53.0	59.7	75.2
65	A	3.90	5.85	7.15	8.26	9.10
	Z	9.55	14.0	16.8	19.3	21.1
	I	62.3	88.8	105	119	129
75	A	4.50	6.75	8.25	9.53	10.5
	Z	12.3	18.1	21.8	25.0	27.3
	I	91.4	130	154	174	189
90	A	5.40	8.10	9.90	11.4	12.6
	Z	17.2	25.3	30.5	34.8	38.0
	I	150	214	252	284	307
100	A	6.00	9.00	11.0	12.7	14.0
	Z	20.9	30.6	37.0	42.2	46.1
	I	200	284	335	376	407
125	A	7.50	11.3	13.8	15.9	17.5
	Z	31.7	46.4	55.8	63.6	69.5
	I	370	521	612	685	738
150	A	9.00	13.5	16.5	19.1	21.0
	Z	44.7	65.2	78.3	89.1	97.2
	I	614	856	1000	1120	1200

d	t_w	16	19	22	25.4	28	32	35	38
200	A	32.0	38.0	44.0	50.8	56.0	64.0	70.0	76.0
	Z	215	259	305	359	401	469	521	576
	I	3900	4730	5600	6640	7460	8790	9830	10900
250	A	40.0	47.5	55.0	63.5	70.0	80.0	87.5	95.0
	Z	325	390	458	536	597	694	769	845
	I	7120	8600	10100	11900	13400	15600	17400	19200
300	A	48.0	57.0	66.0	76.2	84.0	96.0	105.0	114.0
	Z	455	546	639	746	829	961	1060	1160
	I	11700	14000	16500	19300	21600	25100	27800	30700
350	A	56.0	66.5	77.0	88.9	98.0	112.0	122.5	133.0
	Z	606	726	847	988	1100	1270	1400	1530
	I	17700	21200	24800	29100	32400	37600	41600	45700
400	A	64.0	76.0	88.0	101.6	112.0	128.0	140.0	152.0
	Z	776	928	1080	1260	1400	1610	1780	1940
	I	25300	30300	35400	41400	46000	53300	58900	64600
450	A	72.0	85.5	99.0	114.3	126.0	144.02	157.5	171.0
	Z	965	1150	1340	1560	1730	2000	2200	2400
	I	34700	41500	48500	56500	62800	72600	80100	87700
500	A	80.0	95.0	110.0	127.0	140.0	160.0	175.0	190.0
	Z	1170	1400	1630	18907	2100	2420	2660	2900
	I	46000	55000	64200	74700	82900	95700	10500	11500

$$A = 42 \times 0.8 + 15 \times 1.4 = 21 \text{ [cm}^2\text{]}$$

$$Z_{\text{Top}} = 349.6 \text{ [cm}^3\text{]}$$

$$Z_{\text{Bottom}} = 97.2 \text{ [cm}^3\text{]}$$

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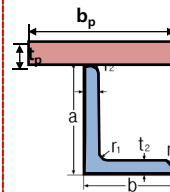
Sectional Properties of Steel Sections for Ship Building¹⁾ (2/12)

¹⁾ "조선설계편람", 제 4판 (일본어), 일본관서조선협회, 1996

<Sectional properties of steel sections including attached plate>

- Use the standard dimension of plate depending on "a" ($b_p \times t_p$) => ($a \leq 75 : 420 \times 8$, $75 < a < 150 : 610 \times 10$, $150 \leq a : 610 \times 15$)

Symbol Unit	Dimension mm						Area		
	a	b	t ₁	t ₂	r ₁	r ₂	A	I	Z
mm							cm ²	cm ⁴	cm ³
Equal angle	L						T		
	50		6		6.5	4.5	5.64		
	65		6		8.5	4	7.53		
	65		8		8.5	6	9.76		
	75		6		8.5	4	8.73		
	75		9		8.5	6	12.69		
	75		12		8.5	6	16.56	90.1	18.7
	90		10		10	7	17.00	191	31.9
	90		13		10	7	21.71	229	39.7
	100		10		10	7	19.00	284	42.5
	100		13		10	7	24.31	369	58.2
	130		9		12	6	11.74	433	71.6
	130		12		12	8.5	19.76	767	96.0
	130		15		12	8.5	36.75	905	117
	150		12		14	7	34.77	1030	119
	150		15		14	10	42.74	1220	147
	150		19		14	10	53.38		
	200		20		17	12	76.00		
	200		25		17	12	93.75		
	200		29		17	12	107.6		
Unequal angle	L						T		
	100	75	7		10	5	11.87	674	72.5
	100	75	10		10	7	16.50	860	96.2
	125	75	7		10	5	13.62	110	97.2
	125	75	10		10	7	19.00	1420	130
	150	90	9		12	6	20.94	2490	181
	150	90	12		12	8.5	27.36	3060	230
	150	90	12		12	8.5	27.36	3060	230



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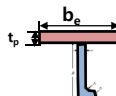
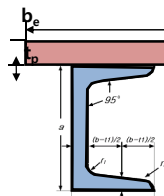
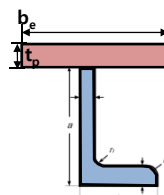
Sectional Properties of Steel Sections for Ship Building¹⁾ (3/12)

¹⁾ "조선설계편람", 제 4판 (일본어), 일본관서조선협회, 1996

<Sectional properties of steel sections including attached plate>

- Use the standard dimension of plate depending on "a" ($b_p \times t_p$) => ($a \leq 75 : 420 \times 8$, $75 < a < 150 : 610 \times 10$, $150 \leq a : 610 \times 15$)

Symbol Unit	Dimension mm						Area		
	a	b	t ₁	t ₂	r ₁	r ₂	A	I	Z
mm							cm ²	cm ⁴	cm ³
Unequal angle	L						T		
	200	90	9	14	14	7.0	29.66	5870	340
	250	90	10	15	17	8.5	37.47	10300	494
	250	90	12	16	17	8.5	42.95	11000	540
	300	90	11	16	19	9.5	46.22	16400	681
	300	90	13	17	19	9.5	52.67	17600	743
	400	100	11.5	16	24	12	61.09	34200	1120
	400	100	13	18	24	12	68.59	36700	1230
	450	125	11.5	18	24	12	73.11	51200	1570
	450	150	11.5	15	24	12	73.45	51700	1590
	500	150	11.5	18	24	12	83.6	70400	2020
	550	150	12	21	24	12	95.91	93300	2520
	600	150	12.5	23	24	12	107.6	118000	3000
Channels	C						T		
	150	75	6.5	10	10	5	23.71	2160	154
	200	90	8	13.5	14	7	38.65	5650	322
	250	90	9	13	14	7	44.07	9420	439
	250	90	11	14.5	17	8.5	51.17	10500	499
	300	90	9	13	14	7	48.57	14300	567
	300	90	10	15.5	19	9.5	55.74	16000	646
	300	90	12	16	19	9.5	61.90	16900	693
	380	100	10.5	16	18	9	69.39	29900	989
	380	100	13	20	24	12	85.71	34900	1190
Bulb flats	I						T		
	180	32.5	9.5	-	7	2	21.06	2860	172
	200	36.5	10	-	8	2	25.23	4160	231
	230	41	11	-	9	2	31.98	6610	330
	250	45	12	-	10	2	38.13	8960	424



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6.4 Buckling Strength

- (1) Column Buckling
- (2) Buckling Strength of Stiffener
- (3) Buckling Strength of Plate
- (4) Buckling Strength by DNV Rule
- (5) Buckling Strength of Stiffener by DNV Rule
- (6) Buckling Strength of Plate by DNV Rule

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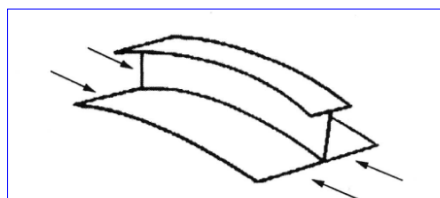
Buckling

James M. Gere, Mechanics of Materials 6th Edition, Thomson, pp. 748-762.
Rules for classification of ships, Det Norske Veritas, January 2004, Pt. 3 Ch. 1 Sec. 13

- **Definition:** The phenomenon where lateral deflection may arise in the athwart direction* against the axial working load

*선측(船側)에서 선측으로 선체를 가로지르는

- This section covers buckling control for plate and longitudinal stiffener.



Flexural buckling of stiffeners plus plating

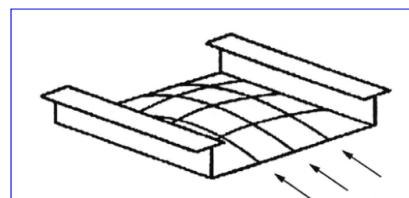


Plate alone buckles between stiffeners

* Mansour, A., Liu, D., The Principles of Naval Architecture Series - Strength of Ships and Ocean Structures, The Society of Naval Architects and Marine Engineers, 2008

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SEOUL NATA LINDY

James M. Gere, Mechanics of Materials 6th Edition, Thomson, pp. 748-762

(1) Column Buckling

- The Equation of the Deflection Curve

- Differential equation for column buckling: $EIy'' + Py = 0$

Using the notation $k^2 = \frac{P}{EI}$, $y'' + k^2 y = 0$

General solution of the equation: $y = C_1 \sin kx + C_2 \cos kx$

Boundary conditions:
 $y(0) = 0, y(L) = 0$
 $y(0) = C_2 = 0$
 $y(L) = C_1 \sin kL = 0$

1) If $C_1 = 0, y = 0$ (trivial solution).
 2) If $\sin kL = 0$, ($\sin kL = 0$: buckling equation)

① If $kL = 0, y = 0$ (trivial solution).
 ② If $kL = n\pi$ ($n=1, 2, 3$) or $P = \left(\frac{n\pi}{L}\right)^2 EI$, it is nontrivial solution.

$\therefore y = C_1 \sin kx = C_1 \sin \frac{n\pi x}{L}, n=1, 2, 3, \dots$

$\frac{P}{EI} = k^2, k = \frac{n\pi}{L}$

E = modulus of elasticity
 I = 2nd moment of the section area
 EI = flexural rigidity
 P = axial load
 y = deflection of column
 L = length of column

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[Example] Mode Shapes of a Cantilevered I-beam

Lateral bending (1st mode) Torsional bending (1st mode) Vertical bending (1st mode)

Lateral bending (2nd mode) Torsional bending (2nd mode) Vertical bending (2nd mode)

* Reference: <https://en.wikipedia.org/wiki/Bending>

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(1) Column Buckling
- Critical Stress

James M. Gere, Mechanics of Materials 6th Edition, Thomson, pp. 748-762

▪ Differential equation for column buckling : $EIy'' + Py = 0$

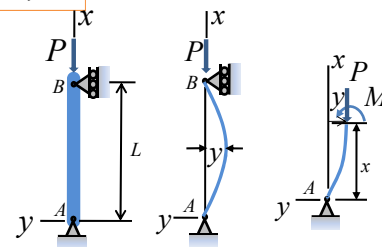
The equation of the deflection curve : $y = C_1 \sin \frac{n\pi x}{L}, n = 1, 2, 3, \dots$

The critical loads : $P = k^2 EI = \left(\frac{n\pi}{L} \right)^2 EI$

The lowest critical load (n=1) : $P_{cr} = \left(\frac{\pi}{L} \right)^2 EI = \frac{\pi^2 EI}{L^2}$

The corresponding critical stress : $\sigma_{cr} = \frac{P_{cr}}{A} = \frac{\pi^2 EI}{AL^2}$
Euler's formula

$\frac{P}{EI} = k^2, k = \frac{n\pi}{L}$



E = modulus of elasticity
 I = 2nd moment of area
 EI = flexural rigidity
 P = axial load
 y = deflection of column
 A = area of column
 L = length of column

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(1) Column Buckling
- Critical Load

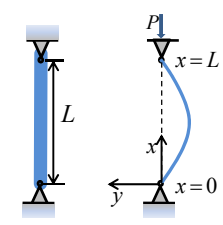
James M. Gere, Mechanics of Materials 6th Edition, Thomson, pp. 748-762

▪ Differential equation for column buckling : $y'' + \lambda y = 0, y(0) = 0, y(L) = 0$
, where $\lambda = P / EI$

The equation of the deflection curve : $y_n(x) = C_1 \sin(n\pi x / L)$

The critical loads : $P_n = n^2 \pi^2 EI / L^2, n = 1, 2, 3, \dots$

The lowest critical load (n=1) : $P_{cr} = P_1 = \pi^2 EI / L^2$



E = modulus of elasticity
 I = 2nd moment of area
 EI = flexural rigidity
 P = axial load
 y = deflection of column
 A = area of column
 L = length of column

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(1) Column Buckling - Critical Buckling Stress

A **critical buckling stress** is often used instead of a buckling load and it can be derived by dividing P_{cr} by A , the cross sectional area of the column.

Euler's formula

$$\text{The corresponding critical stress : } \sigma_{cr} = \frac{P_{cr}}{A} = \frac{\pi^2 EI}{Al^2} = \pi^2 E \left(\frac{k}{l} \right)^2$$

E = modulus of elasticity
 I = 2nd moment of area
 EI = flexural rigidity
 P = axial load
 y = deflection of column
 A = area of column
 l = length of column

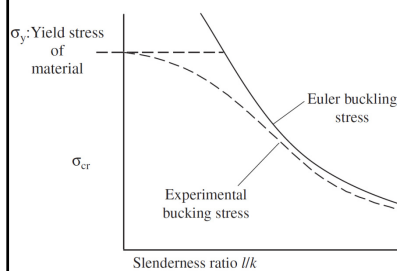
, where k ($k^2 = I / A$) is the **radius of gyration**¹⁾ of the section of the column.

The ratio (l/k) , often called the **slenderness ratio**, is the main factor which governs the critical stress

For large value of l/k the critical stress tends toward zero, and at small values of l/k it tends to **infinity**. In Euler's formula, the buckling stress may become infinite for a small value of l/k , however, buckling stress never goes up above the yield stress of the material in actual conditions, because the material would fail if the stress exceeded the yield stress.

1) The radius of gyration describes a circular ring whose area is the same as the area of interest.

(1) Column Buckling - Curve of Buckling Stress



by theoretical consideration, a horizontal line of yield stress connected to Euler buckling stress is specified as an upper limit of Euler's buckling curve.

$$\sigma_{cr} = a - b \left(\frac{l}{k} \right) \quad \text{Tetmayer's formula}$$

$$\sigma_{cr} = a - b \left(\frac{l}{k} \right)^2 \quad \text{Johnson's formula}$$

$$\sigma_{cr} = \frac{a}{1 + b \left(l/k \right)^2} \quad \text{Rankine's formula}$$

For example, one of the Classification Societies, ABS (American Bureau of Shipping) specifies the permissible load of a pillar or strut of mild steel material in the following equation:

$$\sigma_{cr} = 1.232 - 0.00452 \left(\frac{l}{k} \right) \quad [ton \cdot f / cm^2]$$

From the above equation, we can see that the ABS formula is theoretically based on Tetmayer's experimental result.

(1) Column Buckling**- Buckling of Thin Vertical Column Embedded at Its Base and Free at Its Top (1/2)**

Suppose that a thin vertical homogeneous column is embedded at its base ($x=0$) and free at its top ($x=L$) and that a constant axial load P is applied to its free end.

The load either causes a small deflection δ , or does not cause such a deflection. In either case the differential equation for the deflection $y(x)$ is

$$EI \frac{d^2 y}{dx^2} = P(\delta - y) \quad \Rightarrow \quad EI \frac{d^2 y}{dx^2} + Py = P\delta \quad \dots (1)$$

(1) What is the predicted deflection when $\delta = 0$?

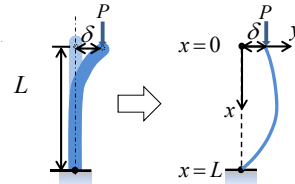
- The general solution of the differential equation (1) is

$$y = c_1 \cos \sqrt{\frac{P}{EI}} x + c_2 \sin \sqrt{\frac{P}{EI}} x + \delta$$

- The boundary conditions of the differential equation (1) are

$$y(0) = y'(0) = 0$$

- If $\delta = 0$, this implies that $c_1 = c_2 = 0$ and $y(x) = 0$. That is, there is no deflection.



* Zill, D.G., Advanced Engineering Mathematics, 3rd edition, pp.166-174, 2006

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(1) Column Buckling**- Buckling of Thin Vertical Column Embedded at Its Base and Free at Its Top (2/2)**

Suppose that a thin vertical homogeneous column is embedded at its base ($x=0$) and free at its top ($x=L$) and that a constant axial load P is applied to its free end.

The load either causes a small deflection δ , or does not cause such a deflection. In either case the differential equation for the deflection $y(x)$ is

$$EI \frac{d^2 y}{dx^2} = P(\delta - y) \quad \Rightarrow \quad EI \frac{d^2 y}{dx^2} + Py = P\delta \quad \dots (1)$$

(2) When $\delta \neq 0$, show that the Euler load for this column is one-fourth of the Euler load for the hinged column?

- If $\delta \neq 0$, the boundary conditions give, in turn, $c_1 = -\delta$, $c_2 = 0$.

Then

$$y = \delta \left(1 - \cos \sqrt{\frac{P}{EI}} x \right)$$

- In order to satisfy the boundary condition $y(L) = \delta$, we must have

$$\delta = \delta \left(1 - \cos \sqrt{\frac{P}{EI}} L \right) \quad \longrightarrow \quad \cos \sqrt{\frac{P}{EI}} L = 0 \quad \longrightarrow \quad \sqrt{\frac{P}{EI}} L = n\pi/2$$

- The smallest value of P_n , the Euler load, is then

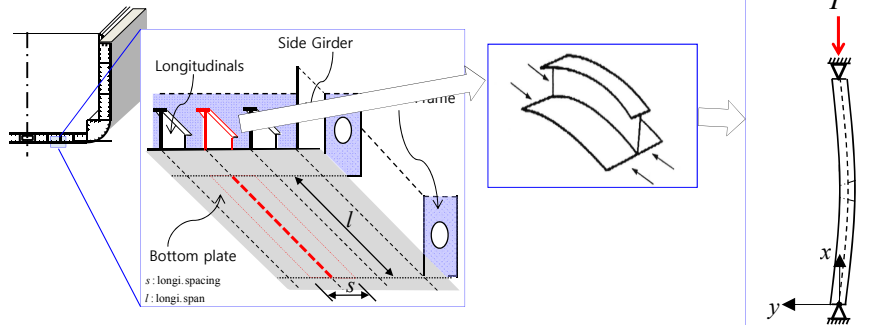
$$\sqrt{\frac{P_1}{EI}} L = \frac{\pi}{2} \quad \text{or} \quad P_1 = \frac{1}{4} \left(\frac{\pi^2 EI}{L^2} \right)$$

Euler load

* Zill, D.G., Advanced Engineering Mathematics, 3rd edition, pp.166-174, 2006

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(2) Buckling Strength of Stiffener



It is assumed that the stiffener is a fixed-end column supported by the web frames.

Hull girder bending moment is acting on the cross section of the ship as moment from the point view of global deformation. And it is acting on the each stiffener as axial load from the point view of local deformation.

what is our interest?

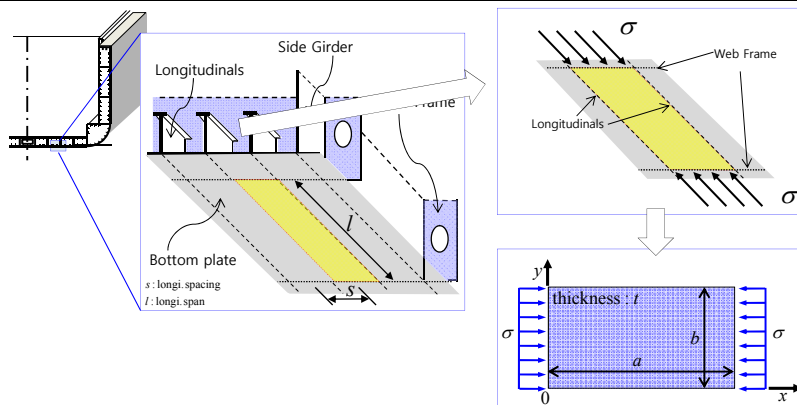
- Safety:
Won't it fail under the load?

The actual compressive stress (σ_a) shall not be great than the critical buckling stress (σ_{cr})

$$\sigma_a \leq \sigma_{cr}$$

, where $\sigma_a = \frac{M}{I_{N.A.}/y} = \frac{M}{Z}$, $Z = Z(y)$

(3) Buckling Strength of Plate (1/7)

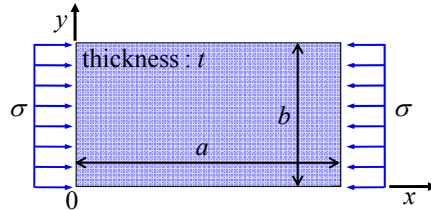


A ship hull is a stiffened-plate structure, the plating supported by a system of transverse or longitudinal stiffeners.

For practical design purpose, it is often assumed that the plate is simply supported at the all edges, since it gives the least critical stress and is on the safe side.

(3) Buckling Strength of Plate (2/7)

Let us consider the rectangular plate with only supported edges as shown in this figure.



σ : the uni-axial compressive stress
 ν : Poisson's ratio
 E : Modulus of elasticity
 a : plate length
 b : plate width
 t : thickness of the plate

- The equation of elastic buckling stress of the plate under uni-axial compressive stress:

$$\frac{Et^3}{12(1-\nu^2)} \left(\frac{\partial^4 w}{\partial x^4} + 2 \frac{\partial^4 w}{\partial x^2 \partial y^2} + \frac{\partial^4 w}{\partial y^4} \right) + \sigma t \frac{\partial^2 w}{\partial x^2} = 0 \quad \dots(1)$$

where, $w = w(x, y)$: deflection of the plate

* Okumoto, Y., Design of Ship Hull Structures, pp.57-60, 2009

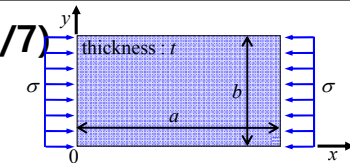
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(3) Buckling Strength of Plate (3/7)

- The equation of elastic buckling stress of the plate under uni-axial compressive stress:

$$\frac{Et^3}{12(1-\nu^2)} \left(\frac{\partial^4 w}{\partial x^4} + 2 \frac{\partial^4 w}{\partial x^2 \partial y^2} + \frac{\partial^4 w}{\partial y^4} \right) + \sigma t \frac{\partial^2 w}{\partial x^2} = 0 \quad \dots(1)$$



σ : the uni-axial compressive stress
 ν : Poisson's ratio
 E : Modulus of elasticity
 a : plate length
 b : plate width
 t : thickness of the plate

where, $w = w(x, y)$: deflection of the plate

- Because all four edges are simply supported, the boundary condition can be expressed in the form:

$$\begin{aligned} w(0, y) = w(a, y) = 0 \\ w(x, 0) = w(x, b) = 0 \end{aligned} \quad \leftarrow \text{deformation at the edges are zero}$$

- Let us assume the following formula for the solution of the equation (1), so that the solution satisfies the boundary conditions.

$$w = f \sin\left(\frac{m\pi x}{a}\right) \cdot \sin\left(\frac{n\pi y}{b}\right) \quad \dots(2)$$

where, m, n are integers presenting the number of half-wave of buckles.

* Okumoto, Y., Design of Ship Hull Structures, pp.57-60, 2009

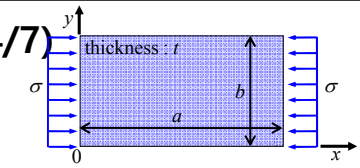
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(3) Buckling Strength of Plate (4/7)

- The equation of elastic buckling stress of the plate under uni-axial compressive stress:

$$\frac{Et^3}{12(1-\nu^2)} \left(\frac{\partial^4 w}{\partial x^4} + 2 \frac{\partial^4 w}{\partial x^2 \partial y^2} + \frac{\partial^4 w}{\partial y^4} \right) + \sigma t \frac{\partial^2 w}{\partial x^2} = 0 \quad \dots(1)$$



σ : the uni-axial compressive stress
 ν : Poisson's ratio
 E : Modulus of elasticity
 a : plate length
 b : plate width
 t : thickness of the plate

where, $w = w(x, y)$: deflection of the plate

- Substituting the formula (2) into the equation (1),

$$w = f \sin\left(\frac{m\pi x}{a}\right) \cdot \sin\left(\frac{n\pi y}{b}\right) \quad \dots(2)$$

$$\sigma = \frac{Et^3}{12(1-\nu^2)} \frac{\pi^2}{b^2 t} \left(\frac{m}{\alpha} + n^2 \frac{\alpha}{m} \right)^2 \quad \dots(3) \quad \text{where, } \alpha = \frac{a}{b}$$

- Elastic buckling stress is a minimum critical stress, therefore, we put $n=1$ in the equation (3),

Ideal elastic (Euler) compressive buckling stress:

$$\sigma_{el} = \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t}{b} \right)^2 K \quad \text{where, } K = \text{Minimum value of } k, k = \left(\frac{m}{\alpha} + \frac{\alpha}{m} \right)^2$$

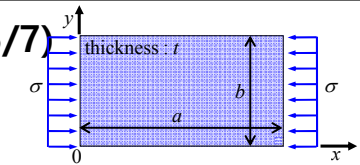
* Okumoto, Y., Design of Ship Hull Structures, pp.57-60, 2009
 Design Theories of Ship and Offshore Plant, Fall 2017, Myung-Il Roh

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(3) Buckling Strength of Plate (5/7)

Ideal elastic (Euler) compressive buckling stress:

$$\sigma_{el} = \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t}{b} \right)^2 K \quad \text{where, } K = \text{Minimum value of } k, k = \left(\frac{m}{\alpha} + \frac{\alpha}{m} \right)^2, \alpha = \frac{a}{b}$$



σ : the uni-axial compressive stress
 ν : Poisson's ratio
 E : Modulus of elasticity
 a : plate length
 b : plate width
 t : thickness of the plate

- For the small b in comparison with t , the elastic buckling stress becomes more than the yield stress of the plate material.
- Therefore, it is usual to use **Johnson's modification factor** η_p and the critical buckling stress σ_c for the full range of value of t/b as follows:

- Bryan's formula¹⁾**

$$\frac{\sigma_c}{\eta_p} = \sigma_{el} = \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t}{b} \right)^2 \cdot K$$

σ_c : the critical compressive buckling stress
 σ_{el} : the ideal elastic(Euler) compressive buckling stress
 K : plate factor (corresponding to the boundary conditions and a/b)
 η_p : plasticity reduction factor

$$\eta_p = 1, \text{ when } \sigma_{el} < \frac{\sigma_y}{2}$$

$$\eta_p = \frac{\sigma_y}{\sigma_{el}} \left(1 - \frac{\sigma_y}{4\sigma_{el}} \right), \text{ when } \sigma_{el} \geq \frac{\sigma_y}{2}$$

$$\sigma_y = \text{upper yield stress in [N/mm}^2\text{]}$$

ex) Coefficient K when all four edges are simply supported

$$K = 4.0 \quad a/b \geq 1.0$$

$$K = (a/b + b/a)^2, \quad a/b < 1.0$$

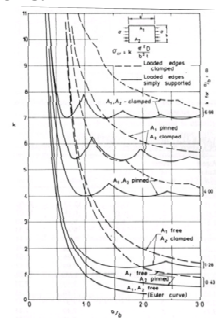


Figure 13.5a Buckling stress coefficient K for flat plates in uni-axial compression

1) DSME, "선박구조설계" 13-18 Buckling, 2005.8
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(3) Buckling Strength of Plate (6/7) - Buckling Strength of Web Plate

1) DSME, "Ship Structural Design", 13-18 Buckling, 2005.8

Web plate of stiffener have to be checked about buckling.

In case of T-bar, it is assumed that the web plate of stiffener is the plate simply supported by flange and attached plate.

$$\frac{\sigma_c}{\eta_p} = \sigma_{el} = \frac{\pi^2 E}{12(1-\nu^2)} \cdot \left(\frac{t}{d}\right)^2 \cdot K, \quad (\text{Bryan's formula}), \quad K = 4.0$$

$$\rightarrow \frac{d}{t_w} \leq \sqrt{\frac{\pi^2 E K}{12(1-\nu^2)} \frac{1}{\sigma_{el}}}$$

σ_c : the critical compressive buckling stress

σ_{el} : the ideal elastic(Euler) compressive buckling stress

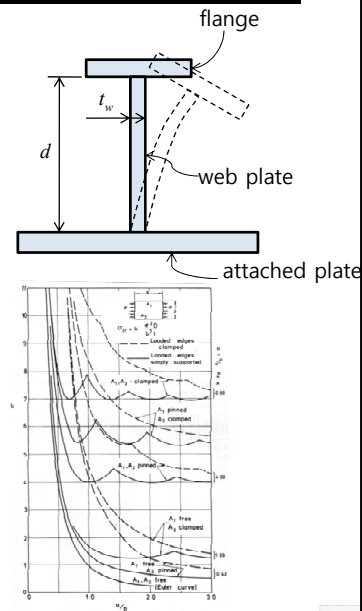
ν : Poisson's ratio

K : Plate factor (corresponding to the boundary conditions and a/b)

d : depth of web plate

t : thickness of web plate

E : Modulus of elasticity

Figure 12.5a Buckling stress coefficient k for the plates in uni-axial compression

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(3) Buckling Strength of Plate (7/7) - Buckling Strength of Flange Plate

1) DSME, "Ship Structural Design", 13-18 Buckling, 2005.8

Flange of stiffener have to be checked about buckling.

It is assumed that the flange of stiffener is the rectangular plate simply supported on one end by web plate.

$$\frac{\sigma_c}{\eta_p} = \sigma_{el} = \frac{\pi^2 E}{12(1-\nu^2)} \cdot \left(\frac{t_f}{b_f}\right)^2 \cdot K, \quad (\text{Bryan's formula}), \quad K = 0.5$$

$$\rightarrow \frac{b}{t_f} \leq \sqrt{\frac{K \pi^2 E}{12(1-\nu^2)} \frac{1}{\sigma_{el}}}$$

In general, b/t_f does not exceed 15.

σ_c : the critical compressive buckling stress

σ_{el} : the ideal elastic(Euler) compressive buckling stress

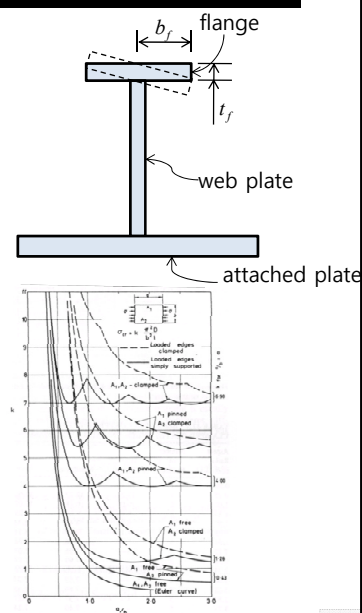
ν : Poisson's ratio

K : Plate factor (corresponding to the boundary conditions and a/b)

b_f : breadth of flange plate

t_f : thickness of flange plate

E : Modulus of elasticity

Figure 12.5a Buckling stress coefficient k for the plates in uni-axial compression

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(4) Buckling Strength by DNV Rule

◆ **Criteria for buckling strength**

$$\sigma_c > \frac{\sigma_a}{\eta}$$

1) Rules for Classification of Ships, DNV, Pt. 3 Ch. 1 Sec. 13, pp.92~93, January 2004

σ_c = critical buckling stress in N/mm²
 σ_a = calculated actual stress in N/mm²
 η = usage factor

◆ **Critical buckling stress σ_c**

- σ_f is yield stress of material in N/mm²
- σ_a is calculated actual stress in general
- In plate panels subject to longitudinal stress, σ_a is given by

◆ **Calculated actual stress σ_a**

$\sigma_a = \frac{Ms + Mw}{I_{N.A.}} (z_n - z_a) 10^5, (N/mm^2)$
 = minimum 30 f_i N/mm² at side

◆ **σ_{el} for Plate in uni-axial compression¹⁾**

Plate: $\sigma_{el} = 0.9kE \left(\frac{t - t_k}{1000s} \right)^2$

◆ **σ_{el} for stiffener in uni-axial compression¹⁾**

Stiffener: $\sigma_{el} = 3.8 E \left(\frac{t_w - t_k}{h_w} \right)^2$

◆ **σ_{el} for stiffener in lateral buckling mode**

Stiffener: $\sigma_{el} = 0.001 \cdot E \cdot \frac{I_A}{Al^2}$

consider each different stress according to location

M_s : still water bending moment as given in Sec. 5
 M_w : wave bending moment as given in Sec. 5
 $I_{N.A.}$: moment of inertia in cm⁴ of the hull girder
 σ_{el} : ideal compressive buckling stress
 σ_a : critical buckling stress
 σ_f : upper yield stress in [N/mm²]
 t : thickness in [mm]
 t_k : corrosion addition
 t_w : web thickness, h_w : web height
 E : modulus of elasticity
 s : stiffener spacing in [m]
 I_A : moment of inertia in [cm⁴] about the axis perpendicular to the expected direction of buckling
 A : cross-sectional area in [cm²]
 l : length of member in [m]

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(5) Buckling Strength of Stiffener by DNV Rule - Stiffener in Uni-axial Compression (1/2)

◆ **Criteria for Buckling Strength**
 (in the same way of plate)

$$\sigma_c > \frac{\sigma_a}{\eta}$$

1) Rules for Classification of Ships, DNV, Pt. 3 Ch. 1 Sec. 13, pp.92~93, January 2004

σ_c : critical buckling stress in [N/mm²]
 σ_a : calculated actual compressive stress in [N/mm²]
 η : usage factor

◆ **Critical buckling stress σ_c**

$\sigma_c = \sigma_{el}$, when $\sigma_{el} < \frac{\sigma_f}{2}$
 $= \sigma_f \left(1 - \frac{\sigma_f}{4\sigma_{el}} \right)$, when $\sigma_{el} > \frac{\sigma_f}{2}$

σ_{el} : ideal compressive buckling stress
 σ_a : calculated actual compressive stress in [N/mm²]
 σ_f : yield stress of material in N/mm²

◆ **Calculated actual stress σ_a**
 (Uni-axial compression)

- σ_a is calculated actual compressive stress in general
- In plate panels subject to longitudinal stress, σ_a is given by

$\sigma_a = \frac{Ms + Mw}{I_{N.A.}} (z_n - z_a) 10^5, (N/mm^2)$
 = minimum 30 f_i N/mm² at side

(※ Hull girder bending moment is acting on the cross section of the ship as moment from the point view of global deformation.
 And it is acting on the each stiffener as axial load from the point view of local deformation.)

M_s : still water bending moment as given in Sec. 5
 M_w : wave bending moment as given in Sec. 5
 $I_{N.A.}$: moment of inertia in cm⁴ of the hull girder

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(5) Buckling Strength of Stiffener by DNV Rule - Stiffener in Uni-axial Compression (2/2)

◆ Critical buckling stress σ_c

¹⁾ Rules for Classification of Ships, DNV, Pt. 3 Ch. 1
Sec. 13, pp.92~93, January 2004

$$\sigma_c = \sigma_{el} \quad , \text{when } \sigma_{el} < \frac{\sigma_f}{2}$$

$$= \sigma_f \left(1 - \frac{\sigma_{el}}{4\sigma_f}\right) \quad , \text{when } \sigma_{el} > \frac{\sigma_f}{2}$$

σ_f : yield stress of material in [N/mm²]

' σ_{el} ' is determined according to specific load.

◆ Ideal compressive buckling stress σ_{el} of stiffener in uni-axial compression¹⁾

$$\sigma_{el} = 3.8 E \left(\frac{t_w - t_k}{h_w} \right)^2$$

▪ Derivation of the coefficient '3.8'

From Bryan's formula $\frac{\sigma_{cr}}{\eta} = \sigma_c = \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t}{b} \right)^2 \cdot K$,

$$\frac{\pi^2}{12(1-\nu^2)} = 0.9038 (\approx 0.9)$$

And substituting K=4 (for simply supported plate), the coefficient is approximately equal to 3.8.

σ_{el} : ideal compressive buckling stress
 σ_c : critical buckling stress
 σ_f : minimum upper yield stress
 t_w : web thickness, t_k : web height
 E : modulus of elasticity
 s : stiffener spacing (m)
 ν : 0.3 (Poisson's ratio of steel)

◆ Ideal compressive buckling stress σ_{el} of stiffener in lateral buckling mode

$$\sigma_{el} = 0.001 \cdot E \cdot \frac{I_A}{A I^2}$$

▪ Derivation of the coefficient '0.001'

From Euler's formula $\sigma_{cr} = \frac{\pi^2 E I}{A l^2} = \frac{\pi^2 N / mm^2 \cdot cm^4}{cm^2 \cdot m^2}$,

$$\frac{\pi^2 N / mm^2 \cdot cm^4}{cm^2 \cdot m^2} = \frac{\pi^2 N / mm^2 (10mm)^4}{(10mm)^2 (1000mm)^2} \approx 0.001 N/mm^2$$

◆ Thickness of flange

For flanges on angles and T-sections of longitudinals and other highly compressed stiffeners, the thickness shall not be less than

$$t_f = 0.1 b_f + t_k \quad (mm)$$

b_f = flange width in mm for angles, half the flange width for T-section(m)
 t_k = corrosion addition (DNV Rule : Pt. 3 Ch. 1 Sec.2 - Page15)

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(6) Buckling Strength of Plate by DNV Rule - Plate Panel in Uni-axial Compression (1/4)

◆ Criteria for buckling strength

¹⁾ Rules for Classification of Ships, DNV, Pt. 3 Ch. 1
Sec. 13, pp.92~93, January 2004

$$\sigma_c > \frac{\sigma_a}{\eta}$$

σ_c : critical buckling stress in [N/mm²]
 σ_a : calculated actual compressive stress in [N/mm²]
 η : usage factor

Usage Factor (η)

$\eta = 1.0$: Deck, Single bottom & Side shell (longl. stiff)
 $\eta = 0.9$: Bottom, Inner bottom & Side shell (trans. stiff)
 $\eta = 1.0$: Extreme loads ($Q = 10^{-8}$)
 $\eta = 0.8$: Normal loads ($Q = 10^{-4}$)

◆ Critical buckling stress σ_c

$$\sigma_c = \sigma_{el} \quad , \text{when } \sigma_{el} < \frac{\sigma_f}{2}$$

$$= \sigma_f \left(1 - \frac{\sigma_{el}}{4\sigma_f}\right) \quad , \text{when } \sigma_{el} > \frac{\sigma_f}{2}$$

σ_{el} : ideal compressive buckling stress
' σ_{el} ' is determined according to specific load.
 σ_f : upper yield stress in [N/mm²]

From Bryan's formula,

$$\frac{\sigma_c}{\eta_p} = \sigma_{el} = \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t}{b} \right)^2 \cdot K$$

when $\sigma_{el} < \frac{\sigma_f}{2}$, $\eta_p = 1$

$$\sigma_c = \eta_p \sigma_{el} \rightarrow \sigma_c = \sigma_{el}$$

when $\sigma_{el} \geq \frac{\sigma_f}{2}$, $\eta_p = \frac{\sigma_f}{\sigma_{el}} \left(1 - \frac{\sigma_{el}}{4\sigma_f}\right)$

$$\sigma_c = \eta_p \sigma_{el} \rightarrow \sigma_c = \sigma_f \left(1 - \frac{\sigma_{el}}{4\sigma_f}\right)$$

(6) Buckling Strength of Plate by DNV Rule - Plate Panel in Uni-axial Compression (2/4)

◆ Criteria for buckling strength

¹⁾ Rules for Classification of Ships, DNV, Pt. 3 Ch. 1
Sec. 13, pp.92~93, January 2004

$$\sigma_c > \frac{\sigma_a}{\eta}$$

σ_c : critical buckling stress in [N/mm²]
 σ_a : calculated actual compressive stress
in [N/mm²]
 η : usage factor

Usage Factor (η)
 $\eta = 1.0$: Deck, Single bottom & Side shell (long stiff)
 $\eta = 0.9$: Bottom, Inner bottom & Side shell (trans stiff)
 $\eta = 1.0$: Extreme loads ($Q = 10^{-8}$)
 $\eta = 0.8$: Normal loads ($Q = 10^{-4}$)

◆ Critical buckling stress σ_c

$$\sigma_c = \sigma_{el} \quad , \text{ when } \sigma_{el} < \frac{\sigma_f}{2}$$

$$= \sigma_f \left(1 - \frac{\sigma_f}{4\sigma_{el}}\right) \quad , \text{ when } \sigma_{el} > \frac{\sigma_f}{2}$$

σ_{el} : ideal compressive buckling stress
' σ_{el} ' is determined according to specific load.
 σ_f : upper yield stress in [N/mm²]

◆ Calculated actual stress σ_a

(Uni-axial compression)

- σ_a is calculated actual compressive stress in general
- In plate panels subject to longitudinal stress, σ_a is given by

$$\sigma_a = \frac{Ms + Mw}{I_{N.A.}} (z_n - z_a) 10^5 \quad , (N/mm^2)$$

$$= \text{minimum } 30 f_t \text{ N/mm}^2 \text{ at side}$$

(※ Hull girder bending moment is acting on the cross section of the ship as moment from the point view of global deformation. And it is acting on the each plate as axial load from the point view of local deformation.)

M_s : still water bending moment as given in Sec. 5
 M_w : wave bending moment as given in Sec. 5
 $I_{N.A.}$: moment of inertia in cm⁴ of the hull girder

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(6) Buckling Strength of Plate by DNV Rule - Plate Panel in Uni-axial Compression (3/4)

◆ Critical buckling stress σ_c

¹⁾ Rules for Classification of Ships, DNV, Pt. 3 Ch. 1
Sec. 13, pp.92~93, January 2004

$$\sigma_c = \sigma_{el} \quad , \text{ when } \sigma_{el} < \frac{\sigma_f}{2}$$

$$= \sigma_f \left(1 - \frac{\sigma_f}{4\sigma_{el}}\right) \quad , \text{ when } \sigma_{el} > \frac{\sigma_f}{2}$$

σ_f : minimum upper yield stress of material in [N/mm²]

' σ_{el} ' is determined according to specific load.

◆ Ideal compressive buckling stress σ_{el} in uni-axial compression¹⁾

$$\sigma_{el} = 0.9 k E \left(\frac{t - t_k}{1000 s} \right)^2$$

▪ Derivation of the coefficient '0.9'

From Bryan's formula $\frac{\sigma_{cr}}{\eta} = \sigma_c = \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t}{b} \right)^2 \cdot K$,

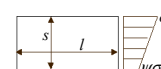
$$\frac{\pi^2}{12(1-\nu^2)} = \frac{3.141593^2}{12(1-0.3^2)} = 0.9038 \quad (\approx 0.9)$$

σ_{el} : ideal compressive buckling stress
 σ_c : critical buckling stress
 σ_f : upper yield stress in N/mm²
 t : thickness (mm)
 t_k : corrosion addition
 E : modulus of elasticity
 s : stiffener spacing (mm)
 ν : 0.3 (Poisson's ratio of steel)

◆ factor k

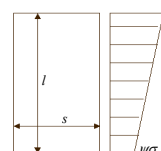
- For plating with longitudinal stiffeners
(in direction of compression stress):

$$k = k_l = \frac{8.4}{\psi + 1.1}$$



- For plating with transverse stiffeners
(perpendicular to compression stress):

$$k = k_s = c \left[1 + \left(\frac{s}{l} \right)^2 \right] \frac{2.1}{\psi + 1.1}$$



ψ = ratio between the smaller and the larger compressive stress (positive value)

$c = 1.21$ when stiffeners are angles or T sections
 $= 1.10$ when stiffeners are bulb flats
 $= 1.05$ when stiffeners are flat bars
 $= 1.30$ when plating is supported by deep girders

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(6) Buckling Strength of Plate by DNV Rule - Plate Panel in Uni-axial Compression (4/4)

◆ Critical buckling stress σ_c

¹⁾ Rules for Classification of Ships, DNV, Pt. 3 Ch. 1
Sec. 13, pp.92~93, January 2004

$$\sigma_c = \sigma_{el} \quad , \text{when } \sigma_{el} < \frac{\sigma_f}{2}$$

$$= \sigma_f \left(1 - \frac{\sigma_{el}}{4\sigma_f}\right) \quad , \text{when } \sigma_{el} > \frac{\sigma_f}{2}$$

σ_f : minimum upper yield stress of material in [N/mm²]

' σ_{el} ' is determined according to specific load.

◆ Ideal compressive buckling stress σ_{el} in uni-axial compression¹⁾

$$\sigma_{el} = 0.9 k E \left(\frac{t - t_k}{1000s} \right)^2$$

▪ Derivation of the coefficient '0.9'

From Bryan's formula $\frac{\sigma_{cr}}{\eta} = \sigma_c = \frac{\pi^2 E}{12(1-\nu^2)} \left(\frac{t}{b} \right)^2 \cdot K$,

$$\frac{\pi^2}{12(1-\nu^2)} = \frac{3.141593^2}{12(1-0.3^2)} = 0.9038 \quad (\approx 0.9)$$

σ_{el} : ideal compressive buckling stress
 σ_c : critical buckling stress
 σ_f : upper yield stress in N/mm²
 t : thickness (mm)
 t_k : corrosion addition
 E : modulus of elasticity
 s : stiffener spacing (m)
 ν : 0.3 (Poisson's ratio of steel)

◆ factor k

- For plating with longitudinal stiffeners (in direction of compression stress): $k = k_l = \frac{8.4}{\psi + 1.1}$
- For plating with transverse stiffeners (perpendicular to compression stress): $k = k_s = c \left[1 + \left(\frac{s}{l} \right)^2 \right]^2 \frac{2.1}{\psi + 1.1}$

Example) If $\psi = 1.0, c = 1.05, s/l = 1/10$

$$k = k_l = \frac{8.4}{1.0 + 1.1} = 4$$

$$k = k_s = c \left[1 + \left(\frac{s}{l} \right)^2 \right]^2 \frac{2.1}{\psi + 1.1} = 1.05 \left[1 + \left(\frac{1}{10} \right)^2 \right]^2 \frac{2.1}{1.0 + 1.1} = 1.071$$

Thus, the plate with longitudinal stiffeners can endure much stress than the plate with transverse stiffeners

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