

$$\tau = \mu \frac{d\langle u \rangle}{dy} - f \langle uv \rangle \quad \text{total shear stress } \tau_w$$

$$\tau_{(xy)} = \tau_w \left( 1 - \frac{y}{\delta_f} \right)$$

$$u_\tau = \sqrt{\frac{\tau_w}{\rho}} : \text{wall-shear velocity}$$

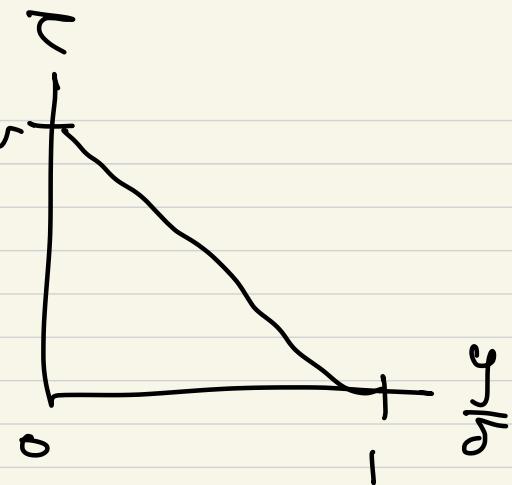
friction

$$\delta_v = \frac{\nu}{u_\tau} : \text{viscous length scale}$$

$$Re_\tau = \frac{u_\tau \delta}{\nu} = \frac{\delta}{\delta_v} : \text{friction Reynolds number}$$

$$y^+ = \frac{y u_\tau}{\nu} \quad + : \text{wall units}$$

$$u^+ = \frac{\langle u \rangle}{u_\tau}$$



viscous wall region :  $y^+ < 50$  direct effect of molecular viscosity on the shear stress  
 outer layer :  $y^+ > 50$  direct effect of viscosity is negligible.  
 viscous sublayer :  $y^+ < 5$  Reynolds shear stress is negligible

$$\text{As } Re \uparrow, \frac{\delta_r}{\delta} = \frac{D/u_c}{\delta} = \frac{2}{u_c \delta} = \frac{1}{Re_c} \downarrow$$

fraction of viscous wall region decreases

- Mean velocity profiles

channel flow is completely specified by

$$\rho, \nu, \delta \text{ and } -\frac{dp_w}{dx}$$

or " " "  $u_c = \sqrt{\tau_w/\rho} = \left(-\frac{\delta}{\rho} \frac{dp_w}{dx}\right)^{\frac{1}{2}}$

$$\Rightarrow \frac{\langle u \rangle}{u_c} = F_0 \left( \frac{y}{\delta}, Re_c \right) \text{ from dimensional analysis}$$

universal non-dimensional ft.

$$\frac{d\langle u \rangle}{dy} / \frac{u_\tau}{y} = \bar{\Phi} \left( \frac{y}{\delta}, \frac{y}{\delta} \right)$$

$$Re_\tau = \frac{y}{\delta} \frac{u_\tau}{\nu}$$

$$\bar{\Phi} \left( \frac{y}{\delta}, Re_\tau \right) \rightarrow \bar{\Phi} \left( \frac{y}{\delta}, \frac{y}{\delta} \right)$$

### Law of the wall

Prandtl (1925): at high  $Re \#$ , close to the wall ( $y/\delta \ll 1$ ), there is an inner layer in which  $\langle u \rangle$  is determined by the viscous scales.

$$\rightarrow \frac{d\langle u \rangle}{dy} / \frac{u_\tau}{y} = \bar{\Phi}_1 \left( \frac{y}{\delta} \right) \text{ for } \frac{y}{\delta} \ll 1$$

( $u^+ = \langle u \rangle / u_\tau$ ,  $y^+ = y u_\tau / \nu$ )

universal ft.

$$\rightarrow \frac{du^+}{dy^+} = \frac{1}{y^+} \bar{\Phi}_1(y^+) \xrightarrow{\text{integration}} u^+ = f_w(y^+) \quad \text{for } \frac{y}{\delta} \ll 1$$

inner layer

\* viscous sublayer  $u^+ = f_\omega(y^+)$

$$\text{@ wall, } \langle u \rangle = 0 \rightarrow 0 = f_\omega(0) \rightarrow \underline{f_\omega(0) = 0}$$

(y=0)

$$u^+ = f_\omega(y^+) \rightarrow \langle u \rangle = u_c f_\omega(y^+)_0$$

$$\frac{\partial \langle u \rangle}{\partial y} \Big|_{y=0} = u_c f'_\omega(y^+)_0 \cdot \frac{u_c}{\nu}$$

$$\underbrace{\rho v \frac{\partial \langle u \rangle}{\partial y}}_{\approx \nu} \Big|_{y=0} = \underbrace{\frac{\rho u_c^2}{\tau_{\text{visc}}} f'_\omega(0)}_{\approx \tau_{\text{visc}}} \Rightarrow \underline{f'_\omega(0) = 1}$$

$$f_\omega(y^+) = f_\omega(0) + y^+ \frac{f'_\omega(0)}{\nu} + \frac{1}{2} y^{+2} \frac{f''_\omega(0)}{\nu} + \dots$$

$\overset{\text{u}^+}{\underset{\text{u}^+}{\text{u}^+}}$

$$\Rightarrow \boxed{u^+ = y^+ + O(y^{+2})}$$

for  $y^+ < 5$

viscous sublayer

\* log law

At high Re, outer part of inner layer  $\rightarrow$  large  $y^+$

$$y^+ = \frac{y u_c}{\nu} = 0.1 \frac{u_c \delta}{\nu} \quad \text{i.e. } y^+ \simeq 0.1 \frac{\delta}{\nu} = 0.1 R_{\text{ec}} \gg 1$$

$$\downarrow \frac{y}{\delta} = 0.1$$

then, the viscosity has little effect.

$\rightarrow \Phi_1(\frac{y}{\delta})$  does not depend on  $y$

$\rightarrow \Phi_1$  is const.  $\Phi_1 = \frac{1}{K}$  for  $\frac{y}{\delta} \ll 1$  and  $y^+ \gg 1$

$$\therefore \frac{du^+}{dy^+} = \frac{1}{Ky^+} \rightarrow u^+ = \frac{1}{K} \ln y^+ + B$$

$k$ : von Karman constant

$$K = 0.41, B = 5.2$$

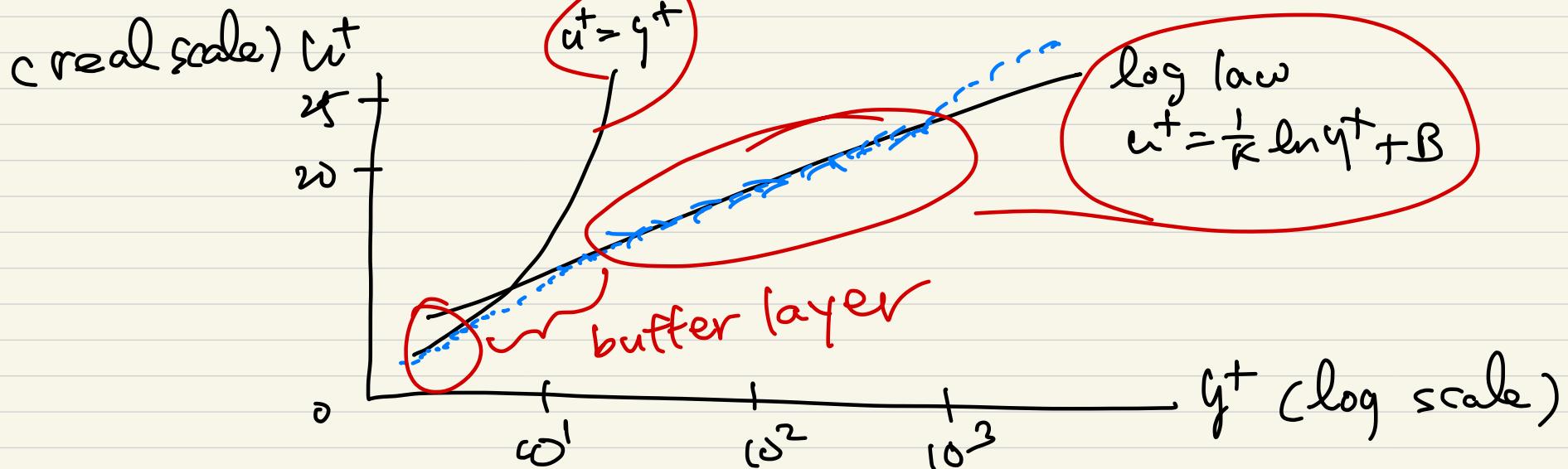
log law

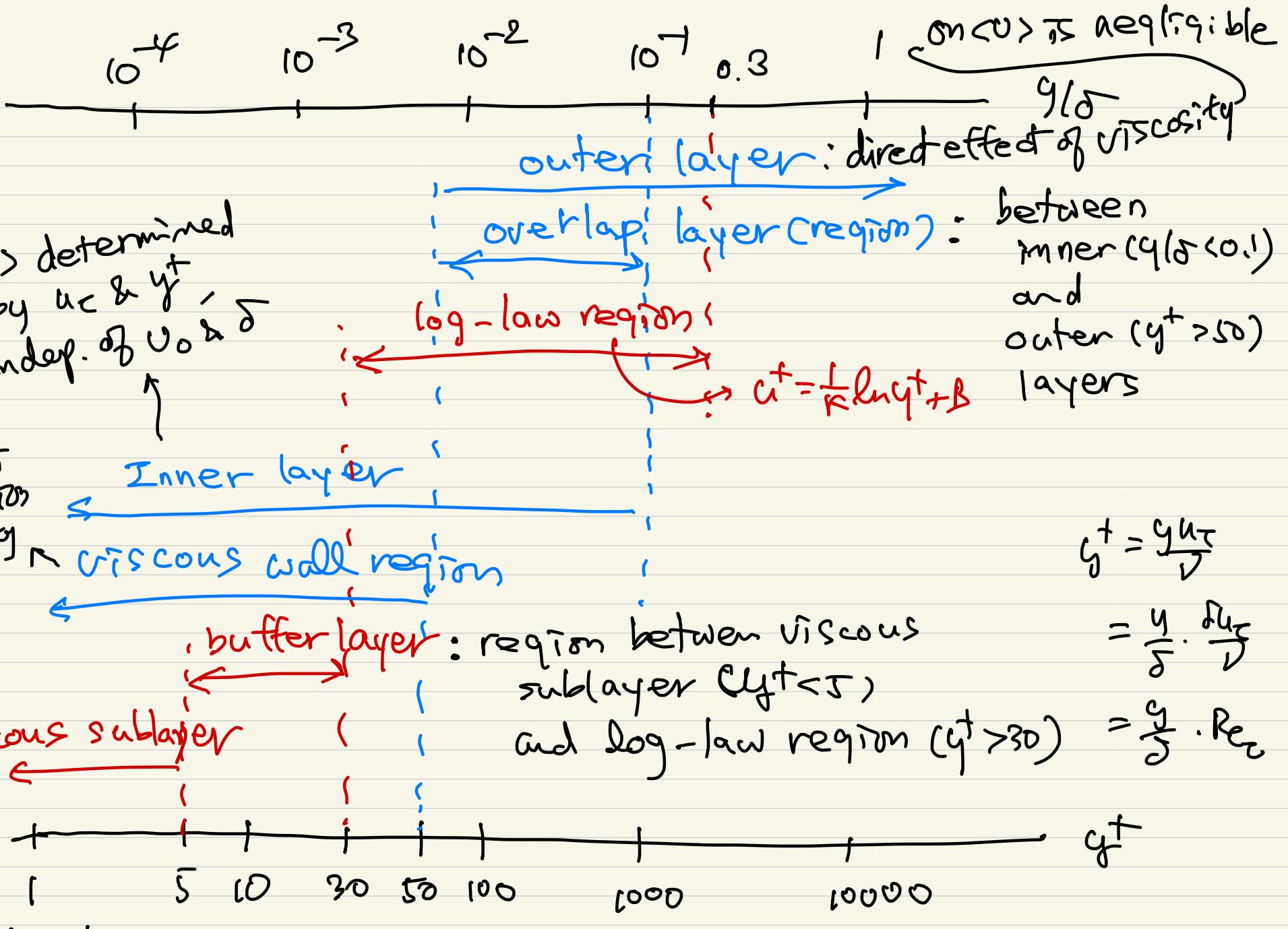
logarithmic law  
of the wall

von Karman (1930)

$$K = 0.41 \& B = 5.0$$

$$K = 0.4 \& B = 5.5$$



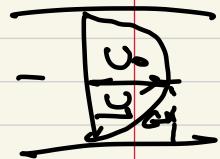


$u^+ = y^+$   
 Reynolds shear stress is negligible

\* Velocity - defect law

in outer layer ( $y^+ > 50$ ),  $\Phi(\frac{y}{\delta}, \frac{y}{\delta})$  is indep. of  $y$ .

$$\rightarrow \frac{d\langle u \rangle}{dy} / \frac{u_\tau}{y} = \Phi_0\left(\frac{y}{\delta}\right)$$



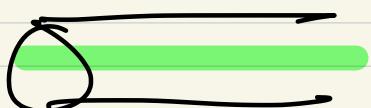
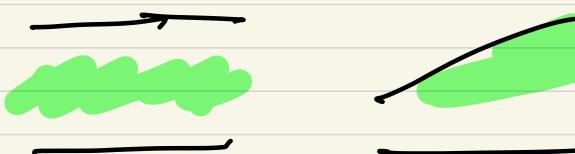
integrate  
from  $y$  to  $\delta$

$$\boxed{\frac{U_\infty - \langle u \rangle}{u_\tau} = F_D\left(\frac{y}{\delta}\right)}$$

velocity-defect law  
 $y^+ > 50$

not universal ft.

is different for diff. flows



At sufficiently high  $Re \#$ , there is an overlap region between inner layer ( $y/\delta < 0.1$ ) and outer layer ( $y^+ > 50$ ).

In this region,  $\frac{d\langle u \rangle}{dy} / \frac{u_\tau}{y} = \Phi_i\left(\frac{y}{\delta}\right)$  inner layer  
 $= \Phi_o\left(\frac{y}{\delta}\right)$  outer layer

= const

$$\rightarrow \frac{d\langle u \rangle}{dy} / \frac{u_\tau}{g} = \frac{1}{K} \quad \text{for } \delta_x \ll y \ll \delta$$

Millikan (1938)

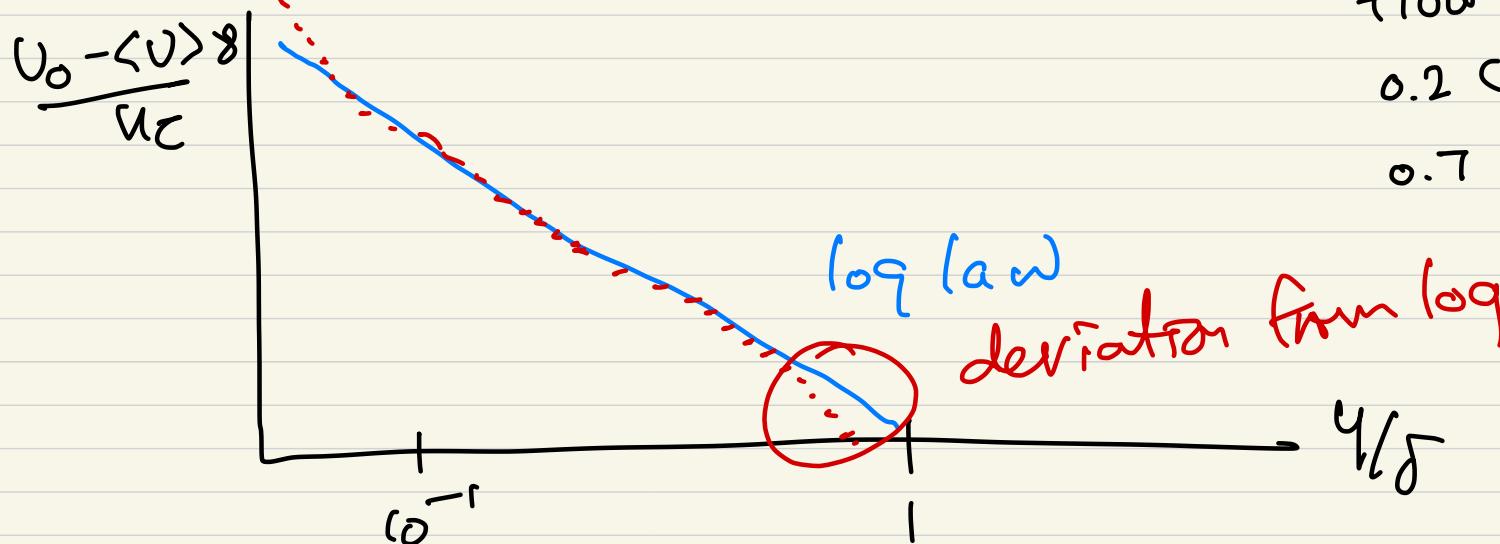
$$\hookrightarrow \text{log-law again} \rightarrow u^+ = \frac{1}{K} \ln \frac{y^+}{\delta} + B$$

Or,

$$\frac{U_0 - \langle u \rangle}{u_\tau} = F_D \left( \frac{y}{\delta} \right) = -\frac{1}{K} \ln \frac{y}{\delta} + B_1 \quad \text{for } \frac{y}{\delta} \ll 1$$

flow-dependent const.

0.2 (DNS) } small  
0.7 (exps.) }



log law  
deviation from log law exists.

## ① Friction law & Reynolds number

Relationships among  $U_0$ ,  $\bar{U}$  and  $u_\tau$

- bulk velocity  $\bar{U}$  from log law

$$\frac{U_0 - \langle U \rangle}{u_\tau} = -\frac{1}{k} \ln \frac{y}{\delta} + B_1$$

$$@ q = \delta, \langle U \rangle = U_0 \rightarrow B_1 = 0$$

$$\frac{U_0 - \bar{U}}{u_\tau} = \frac{U_0 - \frac{1}{\delta} \int_0^\delta \langle U \rangle dy}{u_\tau} = \frac{1}{\delta} \int_0^\delta \frac{U_0 - \langle U \rangle}{u_\tau} dy$$

$$\approx \frac{1}{\delta} \int_0^\delta \left( -\frac{1}{k} \ln \frac{y}{\delta} \right) dy = \frac{1}{k} \approx 2.4 \quad \text{agrees well with exp. data.}$$

$$\therefore \text{inner layer} \quad \frac{\langle U \rangle}{u_\tau} = \frac{1}{k} \ln \frac{y}{\delta} + B$$

$$\text{outer layer} \quad + \quad \frac{U_0 - \langle U \rangle}{u_\tau} = -\frac{1}{k} \ln \frac{y}{\delta} + B_1$$

$$\frac{U_0}{u_\tau} = \frac{1}{k} \ln \frac{\delta}{\delta_\tau} + B + B_1$$

$$\frac{\delta}{\delta_\tau} = \frac{\delta}{V/u_\tau} = \frac{\delta u_\tau}{V}$$

$$= \frac{\delta u_\tau}{V} \cdot \frac{u_\tau}{U_0}$$

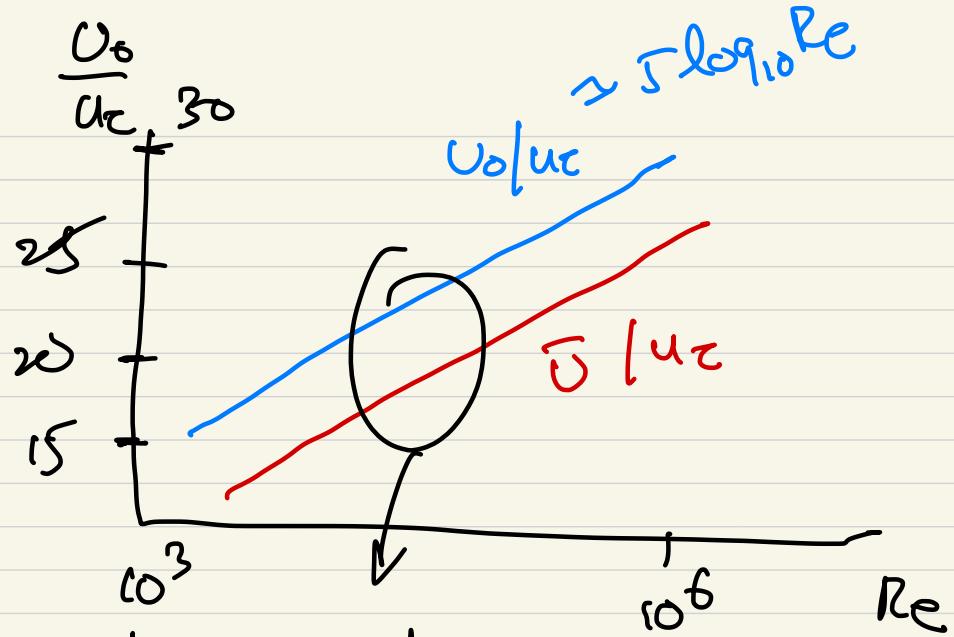
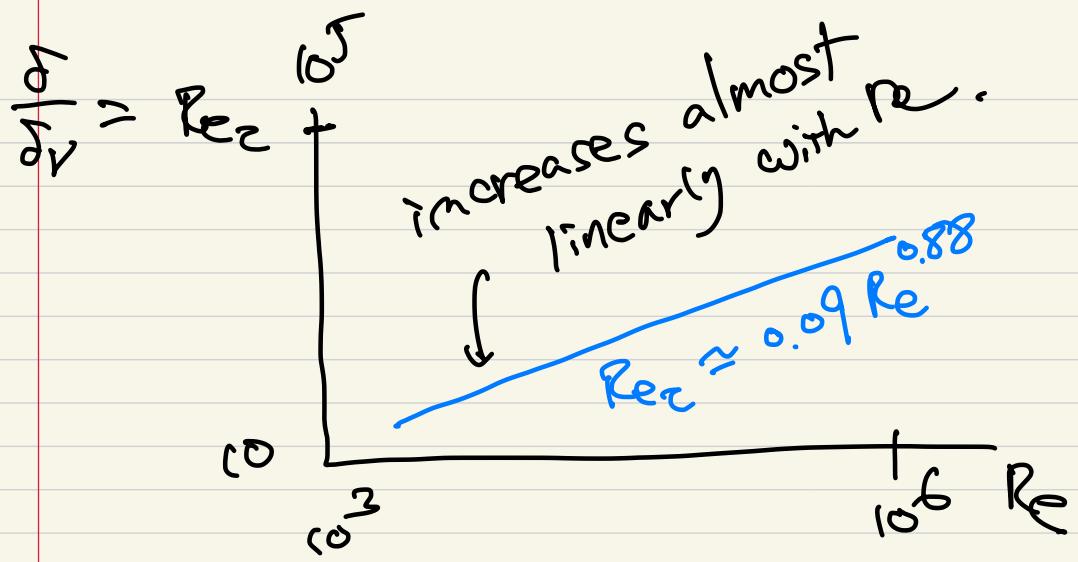
$$\rightarrow \frac{U_0}{U_\infty} = \frac{1}{K} \ln \left[ Re_0 \left( \frac{U_0}{U_\infty} \right)^{1/2} \right] + B + B_J$$

$$C_f = \frac{\tau_{cw}}{\frac{1}{2} \rho U_\infty^2} = 2 \left( \frac{U_\infty}{U_0} \right)^2 \rightarrow \frac{U_0}{U_\infty} = \sqrt{\frac{C_f}{2}}$$

$$\rightarrow \sqrt{\frac{2}{C_f}} = \frac{1}{K} \ln \left( Re_0 \sqrt{\frac{C_f}{2}} \right) + B + B_J$$

Given  $Re_0$ ,  $C_f$  (or  $\frac{U_0}{U_\infty}$ ) can be obtained,





velocity ratios  
increase very slowly  
with  $Re$  !!.

- At high  $Re$ , viscous length scale is very small.

e.g.,  $\delta = 2 \text{ cm}$ ,  $Re = 10^5$

$$Re_2 = 0.09 Re^{0.88} \approx 2260 = \frac{\delta}{\delta_V} \rightarrow \delta_V = 0.89 \times 10^{-5} \text{ m}$$

$$= 10^{-5} \text{ m}$$

$$y^+ = 100 \rightarrow y = 1 \text{ mm}!$$

significant fraction of the increase in mean velocity between wall & centerline occurs in the viscous wall region.

$\delta = 2\text{cm}$

$$\text{e.g., } \delta = 2\text{cm}, Re = 10^5 \rightarrow y^+ = 10 \rightarrow y = 0.1\text{ mm}$$

$$\langle U \rangle_{y^+=10} \approx 0.3 U_\infty$$

$$y^+ = \frac{y U_\infty}{\nu} = y_0^+$$

$$y^+ = \frac{20 y}{\delta U_\infty} = y_0^+ / Re_c = y_0^+ / (0.09 Re^{0.8})$$

$$\ln\left(\frac{y^+}{\delta}\right) = \ln(y_0^+/0.09) - 0.89 \ln Re$$

