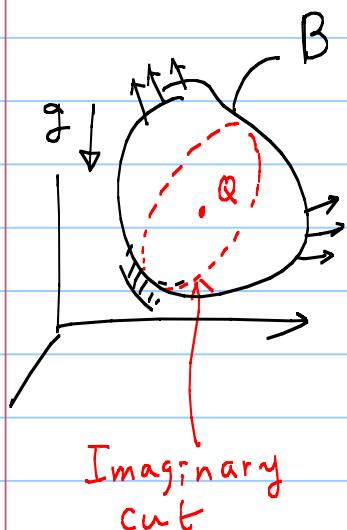


Lecture 1-4 Stress and Equilibrium Equations

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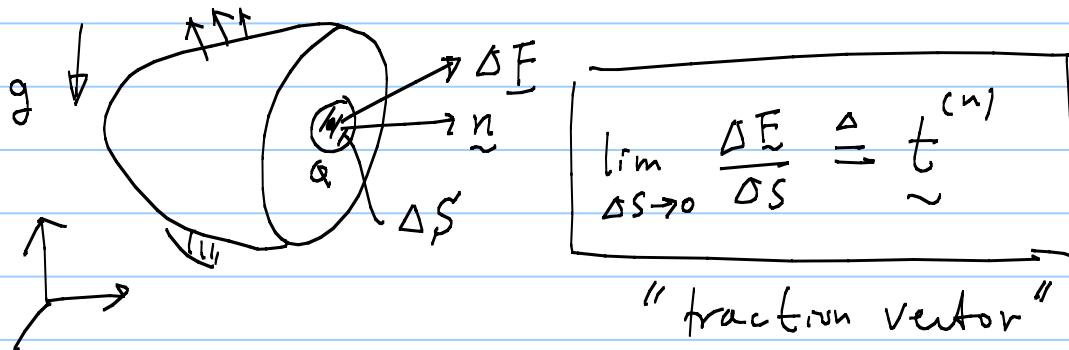
노트 제목



under force

{ Body force: gravity
Magnetic force
(no physical contact)
Surface or Contact force
(physical contact)

Now consider a imaginary cut around Q



At same Q, depending on n , $t^{(n)}$ will differ!

2.

Cauchy says:

$\underline{t}^{(n)}$ at $\underline{\alpha}$ for any n can be determined if $\underline{t}^{(1)}, \underline{t}^{(2)}, \underline{t}^{(3)}$ are known,

$$\left(\underline{t}^{(i)} = \text{fraction for } n = \underline{e}_i \right)$$

* $\underline{t}^{(i)}$ is a vector \rightarrow Let's write $\underline{t}^{(i)}$ as

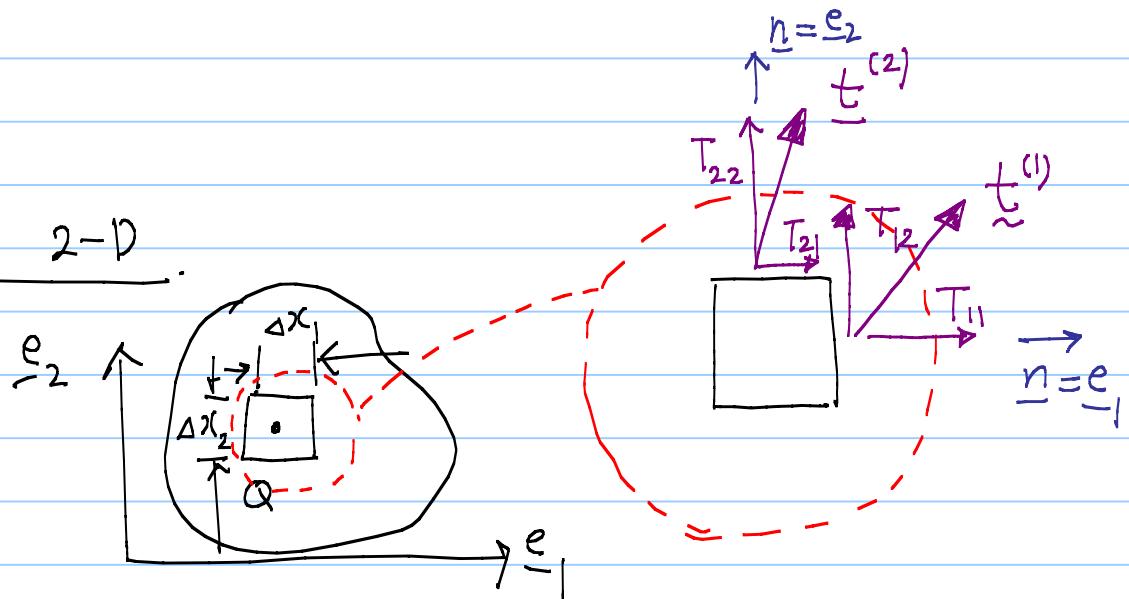
$$\begin{cases} \underline{t}^{(1)} \triangleq T_{11} \underline{e}_1 + T_{12} \underline{e}_2 + T_{13} \underline{e}_3 \\ \underline{t}^{(2)} \triangleq T_{21} \underline{e}_1 + T_{22} \underline{e}_2 + T_{23} \underline{e}_3 \\ \underline{t}^{(3)} \triangleq T_{31} \underline{e}_1 + T_{32} \underline{e}_2 + T_{33} \underline{e}_3 \end{cases}$$

or

$$\underline{t}^{(i)} \triangleq T_{ii} \underline{e}_{ik}$$

Meaning of $\underline{\underline{T}}$ (and $\underline{\underline{t}}^{(c)}$)

in 2-D.



* Cauchy shows

$$\underline{\underline{t}}^{(n)} = \underline{\underline{n}} \cdot \underline{\underline{T}} \cdot \underline{\underline{n}} \quad (\text{for any } \underline{\underline{n}})$$

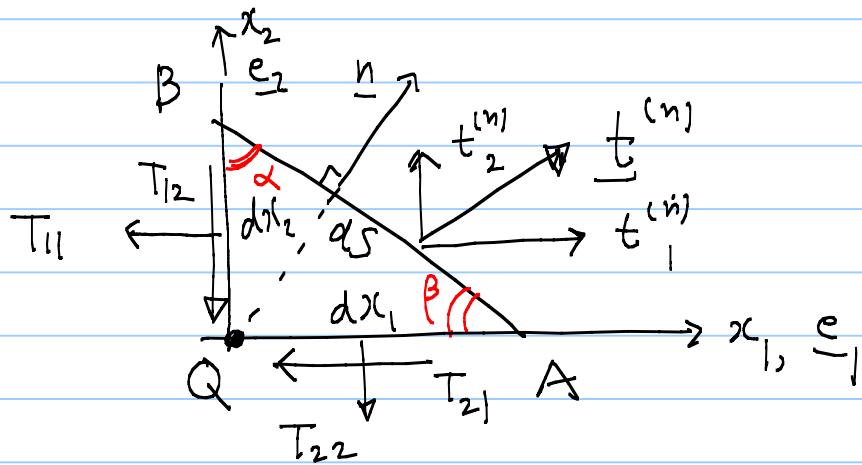
$$\underline{\underline{T}} = T_{ij} \underline{e}_i \otimes \underline{e}_j$$

"Cauchy Stress Tensor"

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\Rightarrow It says, "if you know 9 T_{ij} 's (actually 6 because $T_{ij} = T_{ji}$ as shall be shown), you know traction $\underline{t}^{(n)}$ for any \underline{n} "

2-D version of Cauchy's proof.



(In the limit of Δx_i)

(1) Geometric Relation

$$\underline{n} = (n_1, n_2)^T, n_1 = \cos\alpha, n_2 = \cos\beta$$

$$\begin{cases} dx_1 = ds \cos\beta = n_2 ds \\ dx_2 = ds \cos\alpha = n_1 ds \end{cases}$$

(2) Equilibrium (as $\delta x_i \rightarrow 0$)

$$\sum F_i = 0 :$$

$$\begin{aligned} t_1^{(n)} ds &= T_{11} dx_2 + T_{21} dx_1 \\ &= T_{11} n_1 ds + T_{21} n_2 ds \\ &= n_1 T_{11} ds + n_2 T_{21} ds \end{aligned}$$

$$\hookrightarrow t_1^{(n)} = n_1 T_{11} + n_2 T_{21}$$

$$\sum F_2 = 0 :$$

$$\hookrightarrow t_2^{(n)} = n_1 T_{12} + n_2 T_{22}$$

$$\Rightarrow t_i^{(n)} = n_j T_{ji}$$

$$t_i^{(n)} e_i = n_j T_{ji} e_i$$

$$= n_k \underbrace{\delta_{kj}}_{e_k \cdot e_j} T_{ji} e_i$$

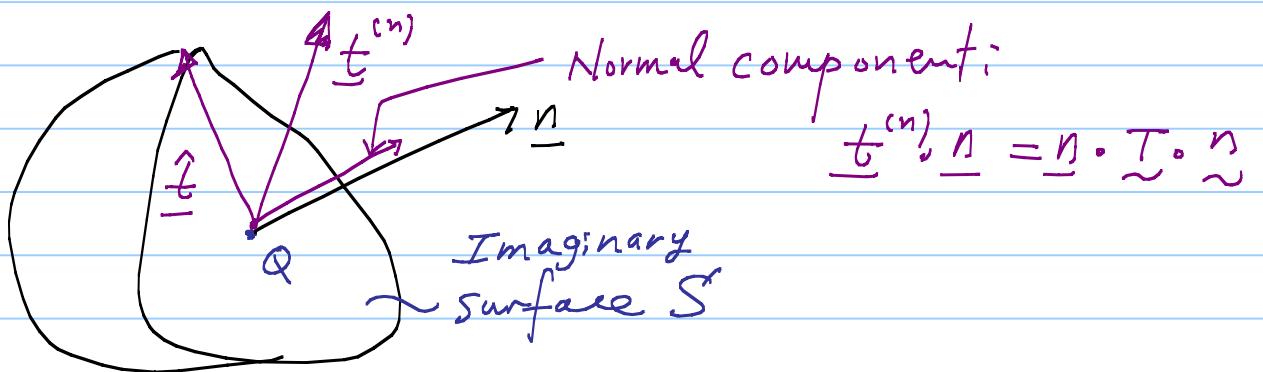
$$= (n_k e_k) \circ (T_{ji} e_j \otimes e_i)$$

$$\underline{i \cdot e} \quad \underline{t^{(n)}} = \underline{n} \circ \underline{T}$$



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Normal and Tangential Components



(In 3-D; \underline{t} be any unit vector
(lying on S)

Remark: $\underline{t}^{(n)} \cdot \underline{m}$ = Component of
traction $\underline{t}^{(n)}$ in the direction
of \underline{m}

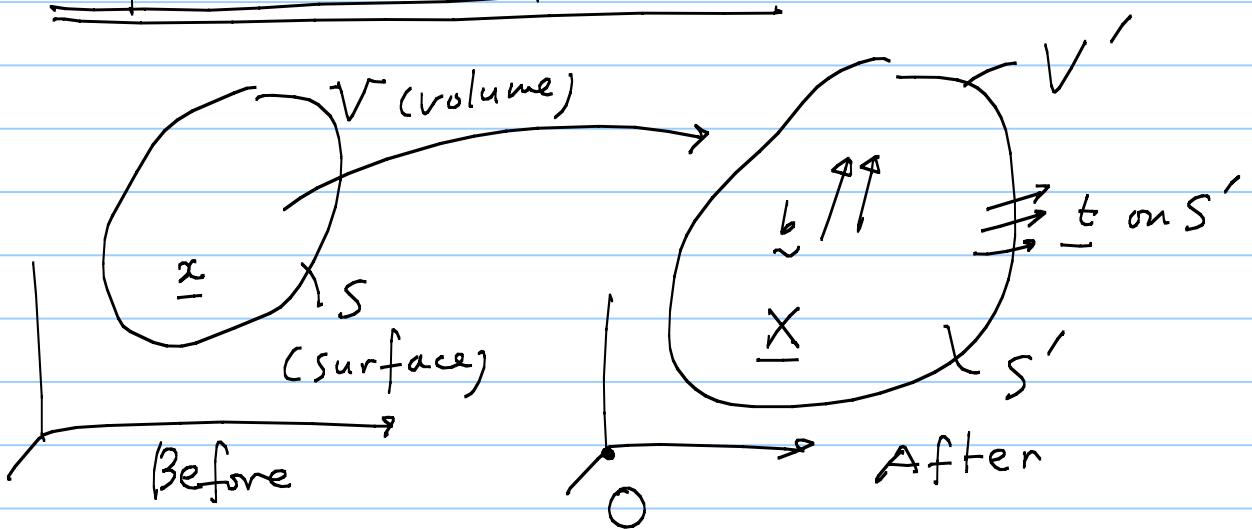
$$\underline{t}^{(n)} \cdot \underline{m} = \underline{n} \cdot \underline{T} \cdot \underline{m} = T_{nm}$$

for any $\underline{n}, \underline{m}$ $||\underline{n}|| = ||\underline{m}|| = 1$.

(Simplest case: $T_{12} = \underline{t}^{(1)} \cdot \underline{e}_2$)

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Equilibrium Equation



∅ Force equilibrium

$$\int_{S'} \underline{t} d\underline{S} + \int_{V'} \underline{b} dV = 0$$

$$\int_{S'} \underline{n} \cdot \underline{\tau} d\underline{S} = \int_{V'} \underline{\nabla}_x \cdot \underline{\tau} dV$$

Div.Th
(1.1.49 b) Based on Deformed State

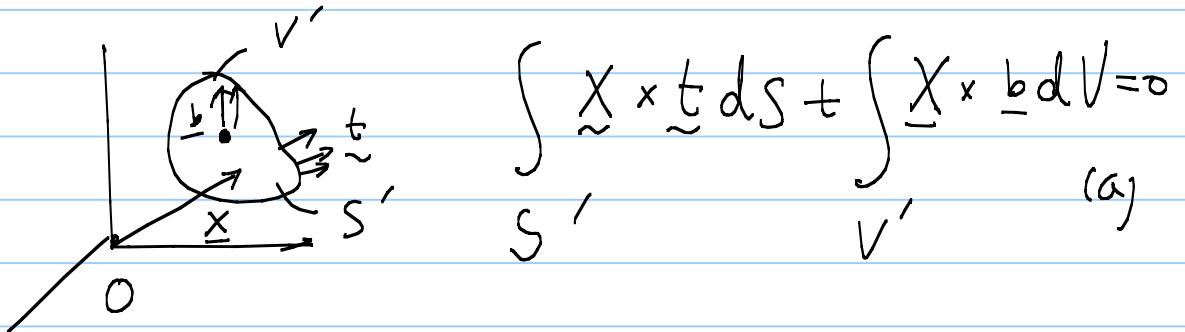
$$\therefore \int_{V'} (\underline{\nabla}_x \cdot \underline{\tau} + \underline{b}) dV = 0$$

δ

$$\Rightarrow \nabla_x \cdot \underline{\tau} + \underline{b} = 0$$

$$\left(\begin{array}{l} T_{ij,i} + b_j = 0 \\ \text{wrt } \underline{x} \end{array} \right)$$

② Moment Equilibrium



$$\int_{S'} \underline{x} \times \underline{t} d\underline{S} + \int_{V'} \underline{x} \times \underline{b} dV = 0 \quad (a)$$

Note

$$\int_{S'} \underline{x} \times \underline{t} d\underline{S} = \int_{S'} \underline{x} \times (\underbrace{\underline{n} \cdot \underline{\tau}}_{\text{vector}}) d\underline{S}$$

$$= - \int_{V'} (\underline{n} \cdot \underline{\tau}) \times \underline{x} dV \quad \underline{n} \cdot (\underline{\tau} \times \underline{x})$$

$$\text{Div} = - \int_{V'} \underline{\nabla}_X \cdot (\underline{T} \times \underline{X}) dV$$

$$= \int_{V'} \left[\underline{X} \times (\underline{\nabla}_X \cdot \underline{T}) + \underline{\epsilon} : \underline{T} \right] dV \quad (b)$$

To show the last equation, let's use the Cartesian tensor;

$$\underline{\nabla}_X \cdot (\underline{T} \times \underline{X})$$

$$= \underline{\epsilon}_{ij} \cdot \frac{\partial}{\partial X_j} (\underline{T} \times \underline{X})$$

$$= \underline{\epsilon}_{ij} \cdot \underbrace{\frac{\partial T}{\partial X_j}}_{\underline{\nabla}_X \cdot \underline{T}} \times \underline{X} + \underline{\epsilon}_{ij} \cdot \underline{T} \times \underbrace{\frac{\partial X}{\partial X_j}}_{\parallel}$$

$$\underline{\epsilon}_{ij} \cdot T_{jk} \underline{\epsilon}_j \otimes \underline{\epsilon}_k$$

$$= T_{ik} \underline{\epsilon}_k$$

Using $\underline{\epsilon}_k \times \underline{\epsilon}_i = \epsilon_{kilm} \underline{\epsilon}_m$

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$$\begin{aligned} &= (\nabla_x \cdot \underline{\tau}) \times \underline{x} + \underbrace{T_{ik} e_{kjm} e_m}_{-T_{ik} e_{ikm} e_m} \\ &\quad - \underbrace{e_m e_{mik} T_{ik}}_{-e : \underline{\tau}} \\ &\quad - \underline{e : \underline{\tau}} \end{aligned}$$

$$= -\underline{x} \times (\nabla_x \cdot \underline{\tau}) - \underline{e : \underline{\tau}}$$

(b) \rightarrow (a) :

$$\int_V' \underline{x} \times \left(\underbrace{\nabla_x \cdot \underline{\tau} + \underline{b}}_{!!} \right) dV + \int_V' \underline{e : \underline{\tau}} dV =$$

by ⁰ force eqm

$$\therefore \underline{e : \underline{\tau}} = 0$$

//

or

$$\epsilon_{mik} T_{ik} = 0$$

$$\Rightarrow \frac{1}{2} \underbrace{(\epsilon_{mik} - \epsilon_{mki})}_{\epsilon_{mik}} T_{ik} = 0$$

$$\therefore \epsilon_{mik} T_{ik} - \underbrace{\epsilon_{mki} T_{ik}}_{\substack{= \\ i \leftrightarrow k}} = 0$$

$$\epsilon_{mik} T_{ki}$$

$$\Rightarrow \epsilon_{mik} (T_{ik} - T_{ki}) = 0$$

$$\text{Thus } T_{ik} = T_{ki} \Leftrightarrow \underline{\underline{\Gamma}} = \underline{\underline{\Gamma}}^T$$

Summary : $\nabla_x \cdot \underline{\underline{\Gamma}} + \underline{b} = 0$
 $\underline{\underline{\Gamma}} = \underline{\underline{\Gamma}}^T$

- * applies to nonlinear deformation
- * $\underline{\underline{\Gamma}}$ defined in deformed state,

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Remark: if Stress Tensor is defined
in the undeformed Coord,
it is Piolar-Kirchhoff
Stress Tensor $\underline{\Sigma}$
(1st & 2nd)

(\bar{E}_m , \bar{E}_f : more complex in PK).

	Strain	Stress
Undeformed	E (Lagrangian)	T Cauchy
Deformed	E^* (Eulerian)	S P-K

Const. Rel.

Const. Rel.

For small deformation ($\frac{\partial \vec{x}_i}{\partial \vec{x}_0} \approx \frac{\partial \vec{x}_i}{\partial \vec{x}_0}$)

$$\underline{\tilde{E}} \approx \underline{\tilde{E}}^* \Rightarrow \underline{\tilde{\epsilon}}$$

$$\underline{\tilde{T}} \approx \underline{\tilde{S}} \Rightarrow \underline{\tilde{\Sigma}}$$

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Thus for linear Elasticity, the following equilibrium equation is used

$$\left\{ \begin{array}{l} \nabla \cdot \underline{\sigma} + \underline{b} = 0 \\ \frac{\partial \sigma_i}{\partial x_j} + b_j = 0 \quad (j=1, 2-3) \\ \text{more convenient} \\ \oplus \quad \underline{\sigma} = \underline{\sigma}^T \end{array} \right.$$

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for proof (or plasticity problems)

Sometimes, useful to use

$\left\{ \begin{array}{l} \text{deviatoric Stress tensor } \tilde{\sigma}' \text{ ("traceless")} \\ \text{and strain " } \tilde{\varepsilon}' \text{ ("traceless")} \end{array} \right.$

Such that

$$(*) \quad \tilde{\sigma}' = \tilde{\sigma} - \frac{1}{3} \operatorname{tr} \tilde{\sigma} \mathbb{1}$$

$$(\operatorname{tr} \tilde{\sigma}' = \operatorname{tr} \tilde{\sigma} - \frac{1}{3} \operatorname{tr} \tilde{\sigma} \operatorname{tr} \mathbb{1} = 0)$$

Then

$$\boxed{\tilde{\sigma}' = \tilde{\sigma} + p \mathbb{1} \quad \text{or} \quad \tilde{\sigma} = \tilde{\sigma}' - p \mathbb{1}} \quad (\alpha)$$

\downarrow

no net mean pressure

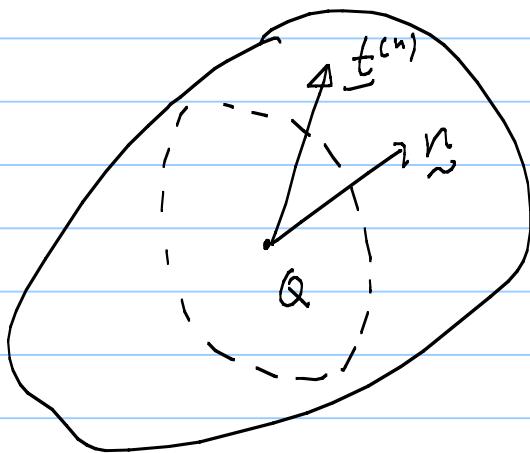
$$\boxed{\tilde{\varepsilon}' = \tilde{\varepsilon} - \frac{1}{3} (\operatorname{tr} \tilde{\varepsilon}) \mathbb{1} \quad \text{or} \quad \tilde{\varepsilon} = \tilde{\varepsilon}' + \frac{1}{3} e \mathbb{1}} \quad (\beta)$$

\downarrow

no net volume change

$e \Rightarrow$ volume change rate
(See problem 1.2-4)

Principal Stress and Direction



for some \underline{n} at Q

$$\underline{t}^{(n)} \parallel \underline{n}$$

$\Rightarrow \underline{n}$: principal direction

Thus, we have

$$\left\{ \begin{array}{l} \underline{t}^{(n)} = \lambda \underline{n} \rightarrow \underline{n} \cdot \underline{\sigma} = \lambda \underline{n} \\ \underline{\sigma}^T \underline{n} = \lambda \underline{n} \rightarrow \underline{\sigma} \cdot \underline{n} = \lambda \underline{n}. \end{array} \right.$$

"Eigenvalue problem"

For nontrivial sol;

$$[\sigma_{ij} - \lambda \delta_{ij}] = 0$$

$$\lambda^3 - I_1 \lambda^2 + I_2 \lambda - I_3 = 0$$

$$\left\{ \begin{array}{l} I_1 = \text{tr } \underline{\sigma} \\ I_2 = \frac{1}{2} [I_1^2 - \underline{\sigma} : \underline{\sigma}] \\ \quad \quad \quad \text{tr } \underline{\sigma}^T \underline{\sigma} \\ I_3 = \det \underline{\sigma} \end{array} \right.$$

If { principal stress: $\sigma^{(1)}, \sigma^{(2)}, \sigma^{(3)}$
" direction: $\underline{n}^{(1)}, \underline{n}^{(2)}, \underline{n}^{(3)}$

then

$$\underline{\sigma} = \sum_{i=1}^3 \sigma^{(i)} \underline{n}^{(i)} \otimes \underline{n}^{(i)}$$

$$I_1 = \sigma^{(1)} + \sigma^{(2)} + \sigma^{(3)}$$

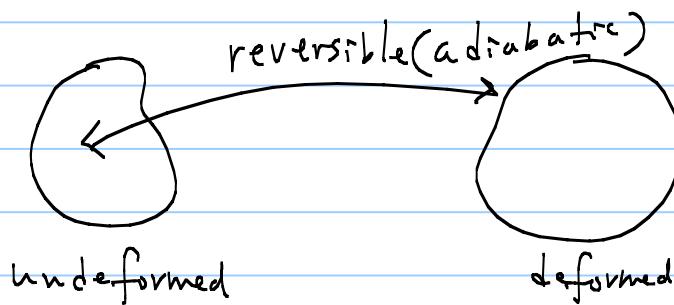
$$I_2 = \sigma^{(1)} \sigma^{(2)} + \sigma^{(2)} \sigma^{(3)} + \sigma^{(3)} \sigma^{(1)}$$

$$I_3 = \sigma^{(1)} \sigma^{(2)} \sigma^{(3)}$$



Constitutive Relation

Relation between $\underline{\epsilon}$ and $\underline{\sigma}$
 This is material property.



$\overbrace{\text{External Work Input } (W_{\text{external}})}$
 $\{ \text{Internal Energy Stored } (U_{\text{system}})$

Thus we will show

$$dW_{\text{ext}} = dU_{\text{sys}} \xrightarrow{\substack{\text{"Internal strain} \\ \text{energy}}} \quad$$

Where

$$\begin{aligned}
 dW_{\text{ext}} &= \int_V (\underline{b} \cdot d\underline{V}) \cdot d\underline{\alpha} \\
 &\quad + \int_S (\underline{t} \cdot d\underline{S}) \cdot d\underline{\alpha}
 \end{aligned}$$

(eq)

$$\begin{aligned}
 \text{Note: } & \int_S \underline{t} \cdot d\underline{u} dS \\
 &= \int_S \underline{n} \cdot (\underline{\sigma} \cdot d\underline{u}) dS \\
 \text{Div. } & \int_V \nabla \cdot (\underline{\sigma} \cdot d\underline{u}) dV \\
 &= \int_V (\nabla \cdot \underline{\sigma} \cdot d\underline{u} + \underline{\sigma} : \nabla d\underline{u}) dV \quad (b)
 \end{aligned}$$

(Let's check the last result:

$$\underline{e}_i \frac{\partial}{\partial x_i} \cdot (\underline{\sigma}_{jk} \underline{e}_j \otimes \underline{e}_k \cdot d\underline{u}_m \underline{e}_m) \xrightarrow{\text{Simplification}} S_{mk}$$

$$\underbrace{\underline{\sigma}_{jk} d\underline{u}_k \underline{e}_j}_{\underline{\sigma}_{jk} d\underline{u}_k \underline{e}_j}$$

$$= (\underline{\sigma}_{jk,i} d\underline{u}_k + \underline{\sigma}_{jk} d\underline{u}_{k,i}) S_{ij}$$

$$= \underline{\sigma}_{jk,j} d\underline{u}_k + \underline{\sigma}_{jk} d\underline{u}_{k,j}$$

$$= (\underline{\nabla} \cdot \underline{\sigma}) \cdot d\underline{u} + \underline{\sigma} : \underline{\nabla} d\underline{u}$$

Sym Tensor

Using $\underline{\sigma} : \nabla d\underline{u} \xrightarrow{\text{sym}} = \underline{\sigma} : d\underline{\nabla} \underline{u}$

$$= \underline{\sigma} : d\underline{\Xi}$$

Ξ only sym part

[Using $\underline{T}^{\text{sym}} : \underline{S}$

$$= \underline{T}^{\text{sym}} : (\underline{S}^{\text{sym}} + \underline{S}^{\text{anti}})$$

$$= \underline{T}^{\text{sym}} : \underline{S}^{\text{sym}}$$

$$\left(\because T^{\text{sym}} : S^{\text{antisym}} = T_{ij}^{\text{sym}} S_{ij}^{\text{anti}} \right.$$

$$\stackrel{(i \leftrightarrow j)}{=} T_{ji}^{\text{sym}} S_{ji}^{\text{anti}} = -T_{ji}^{\text{sym}} S_{ij}^{\text{anti}}$$

$$\left. 2 T_{ij}^{\text{sym}} S_{ij}^{\text{anti}} = 0 \right]$$

Using (a, b, c)

$$\begin{aligned} dW_{ext} &= \int_V \underline{b} \cdot d\underline{u} dV + \int_S \underline{t} \cdot d\underline{u} dS \\ &= \int_V (\underline{b} + \nabla \cdot \underline{u}) \cdot d\underline{u} dV + \int_V \underline{\sigma} : d\underline{\epsilon} dV \\ &\quad \text{by eqm} \end{aligned}$$

Let $dU_{sym} = \int_V \underline{\sigma} : d\underline{\epsilon} dV$

Then $\underline{dW_{ext}} = \underline{dU_{sym}}$

For reversible deforming process;

U_{sym} = path Independent
 (loading-process Independent)
 = only depends on final deformed state!

$$\begin{aligned} dU_{\text{sym}} &= \int \underline{\sigma} : d\underline{\varepsilon} dV \\ &\equiv \int_V dU(\underline{\varepsilon}) dV \end{aligned}$$

↑ only depends on $\underline{\varepsilon}$

U : Strain energy density
(defined on unit volume)

In this case

$$dU(\underline{\varepsilon}) = \underline{\sigma} : d\underline{\varepsilon}$$

$$\Leftrightarrow \underline{\sigma} = \frac{\partial U}{\partial \underline{\varepsilon}} \quad (\underline{\sigma}_{ij} = \frac{\partial U}{\partial \varepsilon_{ij}})$$

Because U is ftn of ε_{ij} , use the Taylor expansion for small ε_{ij}

$$\begin{aligned} U(\varepsilon_{ij}) &= U_0 + \beta_{ij} \varepsilon_{ij} + \frac{1}{2} C_{ijkl} \varepsilon_{ij} \varepsilon_{kl} \\ &\text{choose } = 0 + O(\varepsilon_{ij}^3) \end{aligned}$$

$$\underline{\sigma} \approx \underline{\beta} : \underline{\epsilon} + \frac{1}{2} \underline{\underline{\epsilon}} : \underline{\underline{\epsilon}} \quad (A)$$

$$\underline{\sigma}_{ij} = \underline{\beta}_{ij} + \underline{\underline{C}}_{ijk\ell} \underline{\epsilon}_{k\ell}$$

↑ pre-stress ↑ Elasticity Tensor
 (Let $\underline{\beta}_{ij} = 0$ denoting elastic coefficient
 for simplicity) (4th-order Tensor)

$$\underline{\underline{\epsilon}} = \underline{\underline{C}}_{ijk\ell} \underline{\epsilon}_i \otimes \underline{\epsilon}_j \otimes \underline{\epsilon}_k \otimes \underline{\epsilon}_\ell$$

$$\Rightarrow \left\{ \begin{array}{l} \underline{\sigma} = \underline{\underline{C}} : \underline{\epsilon} \\ \underline{\sigma}_{ij\cdot} = (\underline{\underline{C}}_{ijk\ell}) \underline{\epsilon}_{k\ell} \end{array} \right.$$

Constitutive Relation

$3 \times 3 \times 3 \times 3 = 81$ coefficients
but only 21 independent.

Basic Symmetry in C_{ijkl}

$$\textcircled{1} \quad C_{ijlk} = C_{klij}$$

$$C_{ijlk} = \frac{\partial^2 U}{\partial \varepsilon_{ij} \partial \varepsilon_{kl}} = \frac{\partial^2 U}{\partial \varepsilon_{kl} \partial \varepsilon_{ij}} = C_{klij}$$

(A)

$$\textcircled{2} \quad C_{ijlk} = C_{jilk}$$

$$\sigma_{ij} = C_{ijlk} \varepsilon_{lkl}$$

$$\sigma_{ji} = C_{jilk} \varepsilon_{lkl}$$

$$\text{Since } \sigma_{ij} = \sigma_{ji} \rightarrow C_{ijlk} = C_{jilk}$$

$$\textcircled{3} \quad C_{ijlk} = C_{ijlk}$$

$$\sigma_{ij} = C_{ijlk} \varepsilon_{lkl} \quad \text{--- (a)}$$

$$= C_{ijlk} \varepsilon_{ekl}$$

$$= C_{ijkl} \varepsilon_{ekl} \quad \text{--- (b)}$$

$$\therefore C_{ijlk} = C_{ijkl}$$

Thus, it is convenient to express the constitutive relation ($\sigma_{ij} = C_{ijk\ell} \epsilon_{k\ell}$) as

$$\left\{ \begin{array}{l} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{12} \\ \sigma_{13} \\ \sigma_{23} \end{array} \right\} = \left[\begin{array}{cccccc} C_{1111} & C_{1122} & C_{1133} & C_{1112} & C_{1113} & C_{1123} \\ C_{2211} & C_{2222} & C_{2233} & C_{2212} & C_{2213} & C_{2223} \\ C_{3311} & C_{3322} & C_{3333} & C_{3312} & C_{3313} & C_{3323} \\ C_{1211} & C_{1222} & C_{1233} & C_{1212} & C_{1213} & C_{1223} \\ C_{1311} & C_{1322} & C_{1333} & C_{1312} & C_{1313} & C_{1323} \\ C_{2311} & C_{2322} & C_{2333} & C_{2312} & C_{2313} & C_{2323} \end{array} \right] \left\{ \begin{array}{l} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{33} \\ \epsilon_{12} \\ \epsilon_{13} \\ \epsilon_{23} \end{array} \right\}$$

\uparrow
stiffness matrix.

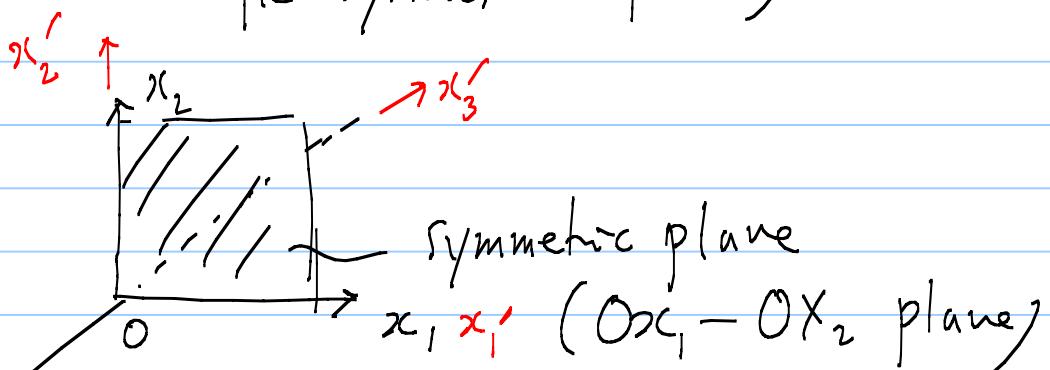
Symmetric matrix due to 1
Indep coeff:
 $1+2+\dots+6 = \frac{6 \times 7}{2} = 21$

Most Engineering Materials have
more Symmetries in material responses

- i) Monoclinic (single Symmetric plane)
- ii) Orthotropic (Two " ")
 \Leftrightarrow same as Three " "
- iii) Isotropic (2 indep.: E, ν)

i) Monoclinic Material

(Single Symmetric Plane)



Mirror-Image Material response

$$\Rightarrow C_{ij'k'l'} = C_{ijk'l} \quad \text{for the Coord. Sys shown in Fig.}$$

$$\text{OLD} \quad (\underline{e}_1, \underline{e}_2, \underline{e}_3) \quad \Rightarrow \quad \text{New} \quad (\underline{e}'_1 = \underline{e}_1, \underline{e}'_2 = \underline{e}_2, \underline{e}'_3 = -\underline{e}_3)$$

$$\text{Thus } \beta_{11'} = \beta_{1'1} = 1, \quad \beta_{22'} = \beta_{2'2} = 1 \\ \beta_{33'} = \beta_{3'3} = -1, \quad \text{other } \beta_{ij} = 0$$

$$(\text{Recall } \beta_{ij} = \beta_{j'i'} = \underline{e}_i \cdot \underline{e}_j)$$

Before using (*),
let's transform $C_{ijk\ell}$ to $C_{ij'k'\ell'}$ by
coordinate transformation rule:

$$C_{ij'k'\ell'} = \beta'_{1i} \beta'_{j'} \beta'_{k'} \beta'_{\ell'} C_{ijk\ell}$$

$$\begin{aligned} C_{1111} &= \beta'_{1i} \beta'_{1j} \beta'_{1k} \beta'_{1\ell} C_{ijk\ell} \\ &= C_{1111} \end{aligned}$$

$$\begin{aligned} C_{1133} &= \beta'_{1i} \beta'_{1j} \beta'_{3k} \beta'_{3\ell} C_{ijk\ell} \\ &= \underbrace{\beta'_{1i}}_{(1)} \underbrace{\beta'_{1j}}_{(1)} \underbrace{\frac{\beta'_{3k} \beta'_{3\ell}}{(-1)^2}}_{C_{1133}} C_{1133} \end{aligned}$$

Single appearance of (3')

$$= C_{1133}$$

$$\begin{aligned} C_{1123} &= \beta'_{1i} \beta'_{1j} \beta'_{2k} \beta'_{3\ell} C_{ijk\ell} \\ &= \underbrace{\beta'_{1i}}_{-1} \underbrace{\beta'_{1j}}_{-1} \underbrace{\beta'_{2k}}_{-1} \underbrace{\beta'_{3\ell}}_{-1} C_{1123} \\ &= -C_{1123} \quad -(*) \end{aligned}$$

Because

$$C_{i'j'k'l'} \equiv C_{ijk'l}$$

under the given Coord. Transf.

$$C_{1'1'2'3'} \equiv C_{1123} \quad (**)$$

We must have

$$C_{1123} = 0$$

\nwarrow single appearance of 3

Thus we have

$$\left\{ \begin{array}{l} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{12} \\ \sigma_{13} \\ \sigma_{23} \end{array} \right\} = \left[\begin{array}{cccccc} C_{1111} & C_{1122} & C_{1133} & C_{1112} & C_{1113} & C_{1123} \\ C_{2211} & C_{2222} & C_{2233} & C_{2212} & C_{2213} & C_{2223} \\ C_{3311} & C_{3322} & C_{3333} & C_{3312} & C_{3313} & C_{3323} \\ C_{1211} & C_{1222} & C_{1233} & C_{1212} & C_{1213} & C_{1223} \\ C_{1311} & C_{1322} & C_{1333} & C_{1312} & C_{1313} & C_{1323} \\ C_{2311} & C_{2322} & C_{2333} & C_{2312} & C_{2313} & C_{2323} \end{array} \right] \left\{ \begin{array}{l} \epsilon_{112} \\ \epsilon_{33} \\ \epsilon_{33} \\ 2\epsilon_{12} \\ 2\epsilon_{13} \\ 2\epsilon_{23} \end{array} \right\}$$

\Rightarrow Only 13 Independent Coeff.

ii) Orthotropic case

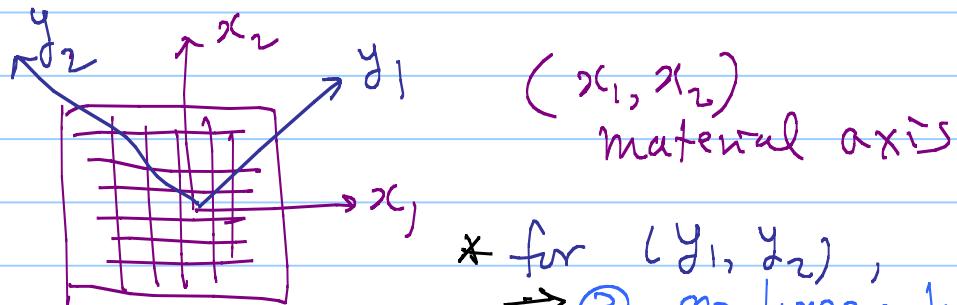
\Rightarrow two planes of symmetry

Add one more symmetry plane
to the Monoclinic material case,

$$\left\{ \begin{array}{l} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{12} \\ \sigma_{13} \\ \sigma_{23} \end{array} \right\} = \left[\begin{array}{cccccc} C_{1111} & C_{1122} & C_{1133} & 0 & 0 & 0 \\ C_{1122} & C_{2222} & C_{2233} & 0 & 0 & 0 \\ C_{1133} & C_{2233} & C_{3333} & 0 & 0 & 0 \\ 0 & 0 & 0 & C_{1212} & 0 & 0 \\ 0 & 0 & 0 & 0 & C_{1313} & 0 \\ 0 & 0 & 0 & 0 & 0 & C_{2323} \end{array} \right] \left\{ \begin{array}{l} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{33} \\ 2\epsilon_{12} \\ 2\epsilon_{13} \\ 2\epsilon_{23} \end{array} \right\}$$

* Observation:

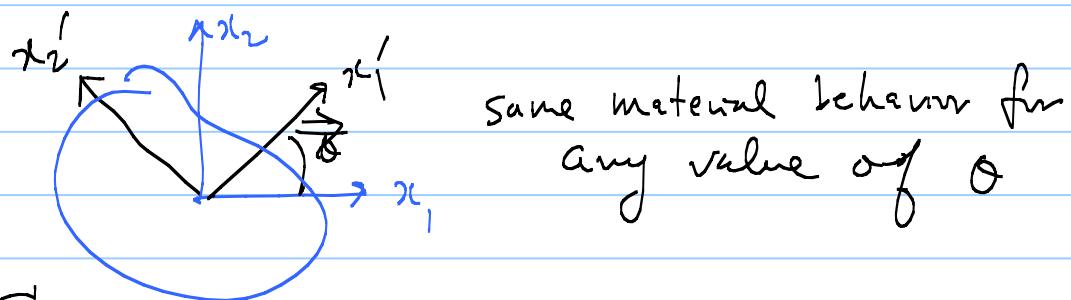
- ① 9 Indep coefficient
- ③ has three symmetry planes
- * ③ No coupling between Stretch and Shear (along the material axis)



* for (y_1, y_2) ,
 \Rightarrow ③ no longer holds.

iii) Isotropic

"Any" orthogonal axes of any orientation
 material axes



Show

$$\left\{ \begin{matrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{12} \\ \sigma_{13} \\ \sigma_{23} \end{matrix} \right\} = \left[\begin{matrix} \lambda + 2\mu & \lambda & \lambda & 0 & 0 & 0 \\ \lambda & \lambda + 2\mu & \lambda & 0 & 0 & 0 \\ \lambda & \lambda & \lambda + 2\mu & 0 & 0 & 0 \\ 0 & 0 & 0 & \mu & 0 & 0 \\ 0 & 0 & 0 & 0 & \mu & 0 \\ 0 & 0 & 0 & 0 & 0 & \mu \end{matrix} \right] \left\{ \begin{matrix} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{33} \\ 2\epsilon_{12} \\ 2\epsilon_{13} \\ 2\epsilon_{23} \end{matrix} \right\}$$

\Rightarrow 2 Indep coeff

$$C_{ijkl} = \lambda \delta_{ij} \delta_{kl} + \mu (\delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk})$$

λ, μ : Lamé constant
($\mu = G$; shear modulus)

Thus for Isotropic case, we have

$$\begin{aligned}\sigma_{ij} &= C_{ijkl} \varepsilon_{kl} \\ &= (\lambda \delta_{ij} \delta_{kl} + \mu \delta_{ik} \delta_{jl} + \delta_{il} \delta_{jk}) \varepsilon_{kl} \\ &= \lambda \varepsilon_{kk} \delta_{ij} + \mu (\underbrace{\delta_{ik} \varepsilon_{jk}}_{\downarrow \text{tr } \varepsilon} + \underbrace{\varepsilon_{ik} \delta_{jk}}_{\varepsilon_{ji}})\end{aligned}$$

$$\boxed{\begin{aligned}\sigma_{ij} &= \lambda \text{tr } \varepsilon \delta_{ij} + 2\mu \varepsilon_{ij} \\ \text{or} \\ \underline{\sigma} &= \lambda \text{tr } \varepsilon \underline{\underline{1}} + 2\mu \underline{\underline{\varepsilon}}\end{aligned}}$$

or

$$\tilde{\varepsilon} = -\frac{\lambda}{2\mu(3\lambda+2\mu)} \tilde{u} + \frac{1}{2\mu} \tilde{\sigma}$$

- * More common to use E (Young's modulus) and λ (Poisson's ratio) :

$$\tilde{\sigma}_{ij} = \frac{E}{1+\nu} \tilde{\varepsilon}_{ij} + \frac{\nu E}{(1+\nu)(1-2\nu)} \delta_{ij} \tilde{\varepsilon}_{kk}$$

$$\tilde{\varepsilon}_{ij} = \frac{1+\nu}{E} \tilde{\sigma}_{ij} - \frac{\nu}{E} \delta_{ij} \delta_{kk}$$

Then

$$\left\{ \begin{array}{l} \mu = \frac{E}{2(1+\nu)} \\ \lambda = \frac{\nu E}{(1+\nu)(1-2\nu)} \end{array} \right.$$

④ Restriction on values of E and ν

\star Claim: $E > 0$ and $-1 < \nu < \frac{1}{2}$

⑤ Proof:

Use: "Positive work must be given to deform an elastic body"

$$\Rightarrow U = \frac{1}{2} C_{ijkl} \varepsilon_{ij} \varepsilon_{kl} = \frac{1}{2} \sigma_{ij} \varepsilon_{ij} \geq 0$$

for any nonzero ε
and $U = 0$ only when $\varepsilon = 0$

Let's write

$$U = \frac{1}{2} \sigma_{ij} \varepsilon_{ij}$$

see (x, β) of page 14

$$= \frac{1}{2} (\sigma_{ij}' - \beta \delta_{ij}) (\varepsilon_{ij}' + \frac{1}{3} \epsilon \delta_{ij})$$

$$\begin{aligned}
 &= \frac{1}{2} (\sigma'_{ij} \varepsilon'_{ij} + \frac{1}{3} e \sigma'_{ii} - p \varepsilon_{ii} - \frac{1}{3} p e \delta_{ii}) \\
 &= \frac{1}{2} (\sigma'_{ij} \varepsilon'_{ij} - p e)
 \end{aligned}
 \tag{33}$$

Now check the relation between

- ① $(-p)$ and e
- ② σ'_{ij} and ε'_{ij}

① $(-p)$ and e relation $(-p = +\frac{1}{3} \operatorname{tr} \underline{\sigma})$

$$\cdot \operatorname{tr} \left\{ \sigma'_{ij} = \frac{E}{1+\nu} \varepsilon'_{ij} + \frac{\nu E}{(1+\nu)(1-2\nu)} \delta_{ij} \varepsilon_{kk} \right\}$$

$$\begin{aligned}
 \cdot \operatorname{tr} \underline{\sigma} &= \frac{E}{1+\nu} \operatorname{tr} \underline{\varepsilon} + \frac{3\nu E}{(1+\nu)(1-2\nu)} \operatorname{tr} \underline{\varepsilon} \\
 &= \frac{E \{ 1-2\nu + 3\nu \}}{(1+\nu)(1-2\nu)} \operatorname{tr} \underline{\varepsilon} \\
 &= \frac{E}{1-2\nu} \operatorname{tr} \underline{\varepsilon}
 \end{aligned}$$

Thus

$$\boxed{\sigma_{ij}} = +\frac{1}{3} \operatorname{tr} \tilde{\sigma} = \frac{E}{3(1-2\nu)} (\operatorname{tr} \tilde{\epsilon})^e$$

$$= \boxed{\text{Bulk modulus}} = \frac{E}{3(1-2\nu)} \quad (*)$$

③ $\underline{\sigma}'$ and $\underline{\epsilon}'$ relation

$$\boxed{\sigma_{ij}} = \tilde{\sigma}_{ij} + p \delta_{ij}$$

$$= \frac{E}{1+\nu} \epsilon_{ij} + \frac{\nu E}{(1+\nu)(1-2\nu)} \delta_{ij} e - k e \delta_{ij}$$

$$= \frac{E}{1+\nu} \left(\epsilon'_{ij} + \frac{1}{3} e \delta_{ij} \right) + \frac{\nu E}{(1+\nu)(1-2\nu)} e \delta_{ij}$$

$$- \frac{E}{3(1-2\nu)} e \delta_{ij}$$

$$= \frac{E}{1+\nu} \epsilon'_{ij} + \frac{E e}{3(1+\nu)(1-2\nu)} \left[\begin{matrix} (1-2\nu) + 3\nu \\ -1-\nu \end{matrix} \right]$$

$$= \frac{E}{1+\nu} \epsilon'_{ij} = \boxed{\dots} \quad (*)$$

Remark: can show

$$E = \frac{qK\mu}{3K+\mu} ; \nu = \frac{3K-2\mu}{6K+2\mu} \quad @$$

Thus,

$$\begin{aligned} U &= \frac{1}{2} (\sigma_{ij}' \varepsilon_{ij}' - b e) \\ &= \frac{1}{2} (2\mu \varepsilon_{ij}' \varepsilon_{ij}' + K e^2) \\ &= \frac{1}{2} \left[2\mu (\varepsilon_{11}'^2 + \varepsilon_{22}'^2 + \varepsilon_{33}'^2 + 2\varepsilon_{12}'^2 + 2\varepsilon_{23}'^2 + 2\varepsilon_{13}'^2) + K e^2 \right] \geq 0. \end{aligned}$$

Since all $\varepsilon_{ij}'^2, e^2 \geq 0$

$\mu > 0$ and $K > 0$

- ①, ②

$$\therefore E = @ \frac{qK\mu}{3K+\mu} > 0$$

Then

$$\left\{ \begin{array}{l} \mu = \frac{E}{2(1+\nu)} > 0 \rightarrow \nu > -1 \\ K = \frac{E}{3(1-2\nu)} \rightarrow \nu < +\frac{1}{2} \end{array} \right.$$

Finally,

$$E > 0, \quad -1 < \nu < \frac{1}{2}$$

$$\nu \Rightarrow -1$$

$$\nu \Rightarrow \frac{1}{2}$$

infinite
shear modulus
limit

Incompressibility
limit