

Chap-5. Drag, scale and the Reynolds number

* Dimensional Analysis

Step 2 : List the dimensions of $Q; Q_1, Q_2, \dots, Q_n$

Step 3 : Pick from the indep. variables Q_1, \dots, Q_n a "complete", "dimensionally independent" subset Q_1, \dots, Q_k ($k \leq n$). Then express the dimensions of the remaining independent variables Q_{k+1}, \dots, Q_n and the dependent variable Q in terms of products of powers of Q_1, \dots, Q_k .

* dimensionally independent : each of the members of a subset has a dimension that cannot be expressed as a product of powers of the dimensions of the remaining members

* complete : all the remaining quantities Q_{k+1}, \dots, Q_n have dimensions which can be expressed as products of powers of the dimensions of Q_1, \dots, Q_k

Step 4 : $Q^* = \frac{Q}{Q_1^a \dots Q_k^b}$: dimensionless (monodimensional)

$Q_{k+1}^* = \frac{Q_{k+1}}{Q_1^p \dots Q_k^q} \dots$

Define dimensionless forms of the dependent variable Q and the $n-k$ remaining independent variables Q_{k+1}, \dots, Q_n by dividing each one with the product of powers of Q_1, \dots, Q_k →

which has the same dimension as the quantity it divides.

Step 5 : Final result : $Q^* = f_n(Q_{k+1}^*, \dots, Q_n^*)$

Why) An alternative form of eq (1) :

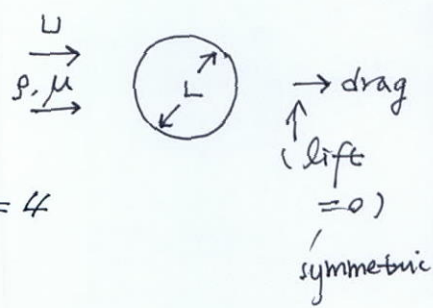
$$Q^* = f_n(\underbrace{Q_1, \dots, Q_k}_{\text{dimensionless}}; \underbrace{Q_{k+1}^*, Q_{k+2}^*, \dots, Q_n^*}_{\text{dimensionless}})$$

· principle of dimensional homogeneity →
 Q_1, \dots, Q_k : should be absent.

The Buckingham pi theorem

When a complete relationship between dimensional physical quantities is expressed in dimensionless form, the number of independent quantities which appear in it is reduced from the original n to $n-k$, where k is the maximum number of the original n quantities that are dimensionally independent.

e.g.
~~Ex 5.2~~ force on sphere/cylinder



Step 1 : $F = f_n(L, U, \rho, \mu)$ $n=4$

Step 2 :

F :	MLT^{-2}	$kg \cdot m/s^2$
L :	L	m
U :	LT^{-1}	m/s
ρ :	ML^{-3}	kg/m^3
μ :	$ML^{-1}T^{-1}$	$kg/m \cdot s$

17
8
Step 3

fundamental dimensions: L, M, T.

of fund. dim. = # dimensionally indep. var. = k

Fund. dim.	Quantity
L	L
T	U
M	ρ

k=3

Step 4

$$F: MLT^{-2} = (ML^{-3})(LT^{-1})^2 L^2 \\ \sim \rho U^2 L^2$$

$$F^* = \frac{F}{\rho U^2 L^2}$$

$$\mu: ML^{-1}T^{-1} = (ML^{-3})(LT^{-1})L \\ \sim \rho UL$$

$$\mu^* = \frac{\mu}{\rho UL}$$

$$n-k = 4-3 = 1.$$

Step 5

$$\frac{F}{\rho U^2 L^2} = f\left(\frac{\mu}{\rho UL}\right) \quad \text{OR} \quad f\left(\frac{\rho UL}{\mu}\right) = f(Re_L)$$

Ex 5.3 power input P to a centrifugal pump.

Step 1: $P = f_n(Q, D, \Omega, \rho, \mu)$ $n=5$

Step 2: $P: \frac{kg \cdot m^2}{s^2}$, $Q: \frac{m^3}{s}$, $D: m$

$\Omega: s^{-1}$, $\rho: kg/m^3$, $\mu: kg/m \cdot s$

step 3.

$$k=3$$

$$m \rightarrow D \text{ (m)}$$

$$s \rightarrow \Omega \text{ (s}^{-1}\text{)}$$

$$kg \rightarrow \rho \text{ (kg/m}^3\text{)}$$

step 4.

$$P: \frac{kg \cdot m^2}{s^3} = \left(\frac{kg}{m^3}\right) \left(\frac{1}{s^3}\right) m^5 \sim \rho \Omega^3 D^5$$

$$P^* = \frac{P}{\rho \Omega^3 D^5}$$

$$Q: \frac{m^3}{s} = m^3 \cdot \left(\frac{1}{s}\right) \sim D^3 \Omega$$

$$Q^* = \frac{Q}{D^3 \Omega}$$

$$\mu: \frac{kg}{m \cdot s} = \left(\frac{kg}{m^3}\right) \cdot \left(\frac{1}{s}\right) \cdot m^2 \sim \rho \Omega D^2$$

$$\mu^* = \frac{\mu}{\rho \Omega D^2}$$

$$\left\{ \begin{array}{l} n-k=2 \end{array} \right.$$

step 5.

$$\frac{P}{\rho \Omega^3 D^5} = f\left(\frac{Q}{\Omega D^3}, \frac{\mu}{\rho \Omega D^2}\right)$$

Ex. 5.4.



step 1

$$Q = f(R, \mu, \frac{dP}{dx}) \quad n=3$$

step 2.

$$Q: \frac{m^3}{s}$$

$$R: m$$

$$\mu: \frac{kg}{m \cdot s}$$

$$\frac{dP}{dx}: \frac{kg}{m^2 \cdot s^2}$$

step 3.

m	R
s	μ
kg	dP/dx

$k=3$

step 4. $Q: \frac{m^3}{s} = \left(\frac{kg}{m^2 \cdot s^2}\right) \left(\frac{m \cdot s}{kg}\right) \cdot m^4 \sim \frac{dp}{dx} \frac{1}{\mu} \cdot R^4$

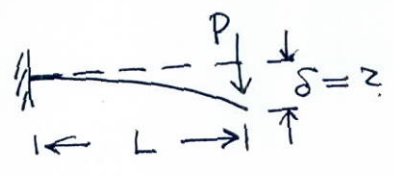
$$Q^* = \frac{Q}{\frac{1}{\mu} R^4 \left(\frac{dp}{dx}\right)}$$

step 5. When $m-k=0$: $Q^* = \text{const.}$

$$\frac{Q}{\frac{1}{\mu} R^4 \left(\frac{dp}{dx}\right)} = \text{const}$$

$$Q = \frac{c}{\mu} R^4 \left(\frac{dp}{dx}\right)$$

Ex. 5.5.



step 1: $\delta = f(P, L, E, I)$ $m=4$

- step 2:
- $\delta: m$
 - $P: kg \cdot m/s^2$
 - $L: m$
 - $E: \frac{kg}{m \cdot s^2}$
 - $I: m^4$

step 3

m	L
kg	E
s	P

$k=3$

step 4. $\delta: m = m \sim L$

$$\delta^* = \frac{\delta}{L}$$

$$I: m^4 \sim L^4$$

$$\sim \left(\frac{kg \cdot m}{s^2}\right)^2 \left(\frac{m \cdot s^2}{kg}\right)^2 \sim P^2 E^2$$

Return to step 3

$$m = \left(\frac{\text{kg} \cdot \text{m}}{\text{s}^2}\right)^{1/2} \left(\frac{\text{m} \cdot \text{s}^2}{\text{kg}}\right)^{1/2}$$

$$L \sim \left(\frac{P}{E}\right)^{1/2}$$

$\therefore L, P, E$: not independent dimensionally.

m	L	k=2
kg/s ²	E	

step 4 $\delta^* = \frac{\delta}{L}$

P: $\frac{\text{kg} \cdot \text{m}}{\text{s}^2} = \left(\frac{\text{kg}}{\text{s}^2}\right) \cdot \text{m}^2 = EI^2$ $P^* = \frac{P}{EI^2}$

I: $\text{m}^4 \sim L^4$ $I^* = \frac{I}{L^4}$

step 5. $\frac{\delta}{L} = f\left(\frac{P}{EI^2}, \frac{I}{L^4}\right)$

If we take $\delta = f(P, L, EI)$,

$\frac{\delta}{L} = f\left(\frac{PL^2}{EI}\right)$ // "physical argument" in linear regime: $\delta \propto P$ $\frac{\delta}{L} = c \frac{PL^2}{EI}$

5.4 Nondimensionalization of the basic equations

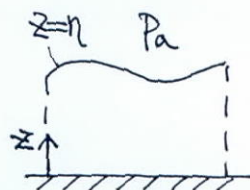
continuity : $\nabla \cdot \vec{V} = 0$

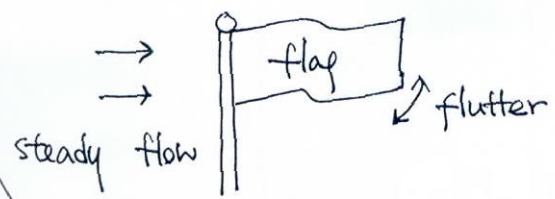
N.S. : $\rho \frac{d\vec{V}}{dt} = -\nabla p + \mu \nabla^2 \vec{V} + \rho \vec{g}$

B.C. fixed solid surface $z=0$: $\vec{V} = 0$

free surface $z=\eta$: $w = \frac{d\eta}{dt}$

$p = p_a - \sigma \left(\frac{1}{R_1} + \frac{1}{R_2}\right)$





• voice generation

§ Other dimensionless parameters

from energy eq.

- Prandtl number $Pr = \frac{\mu C_p}{k} = \frac{\nu}{\alpha}$
- Eckert number $Ec = \frac{U^2}{C_p T_0}$
- Grashof number $Gr = \frac{\beta \Delta T g L^3 \rho^2}{\mu^2}$
- Wall temp. ratio $\frac{T_w}{T_0}$

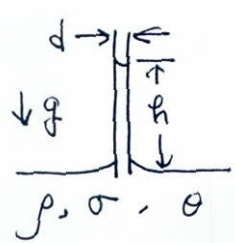
• roughness ratio $\frac{\epsilon}{L}$

• Drag coefficient $C_D = \frac{D}{\frac{1}{2} \rho U^2 A} = \frac{\text{drag force}}{\text{dynamic force}}$

• Lift coefficient $C_L = \frac{L}{\frac{1}{2} \rho U^2 A} = \frac{\text{lift force}}{\text{dynamic force}}$

⋮

~~Ex. prob.~~ e.g.



capillary rise problem

(a) $h = f(d, \rho g, \sigma, \theta)$

- h : m
- d : m
- ρg : $\frac{kg}{m^3} \cdot \frac{m}{s^2} = \frac{kg}{m^2 s^2}$
- σ : $\frac{kg}{s^2}$
- θ : dimensionless

fund dim	quant
m	d^{\checkmark}
s	σ^{\checkmark}
kg	ρg

 $k=2$ (not 3)

$$h^* = \frac{h}{d}$$

$$\rho g: \frac{\text{kg}}{\text{m}^2 \cdot \text{s}^2} = \frac{\text{kg}}{\text{s}^2} \cdot \frac{1}{\text{m}^2} \rightarrow \frac{\sigma}{d^2}$$

$$\rho g^* = \frac{\rho g d^2}{\sigma}$$

$$\therefore \frac{h}{d} = f\left(\frac{\rho g d^2}{\sigma}, \theta\right) \quad \text{: end of dimensional analysis}$$

+ physical argument. $h \sim \sigma$

$$\frac{h}{d} = F(\theta) \frac{\sigma}{\rho g d^2}$$

$$\text{or } \frac{h \rho g d}{\sigma} = F(\theta)$$

(b) $h = 3 \text{ cm}$. $\begin{cases} d' = \frac{1}{2}d \\ \sigma' = \frac{1}{2}\sigma \\ \rho' = 2\rho \end{cases}$ θ same $h' = ?$

$$\frac{h \rho g d}{\sigma} = \frac{h' \rho' g' d'}{\sigma'}$$

$$h' = h \left(\frac{\rho}{\rho'}\right) \left(\frac{d}{d'}\right) \left(\frac{\sigma'}{\sigma}\right) = h \left(\frac{1}{2}\right) (2) \left(\frac{1}{2}\right) = \frac{1}{2}h = 1.5 \text{ cm}$$

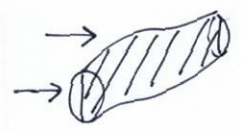
* Bond number. $Bo = \frac{\rho g d^2}{\sigma} = \frac{\text{gravity}}{\text{surface tension}}$.

§ Re and drag coefficient.

$$C_D = \frac{F_D}{\frac{1}{2} \rho S U^2}$$

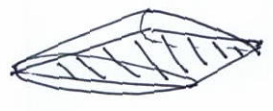
S (area)

① frontal area

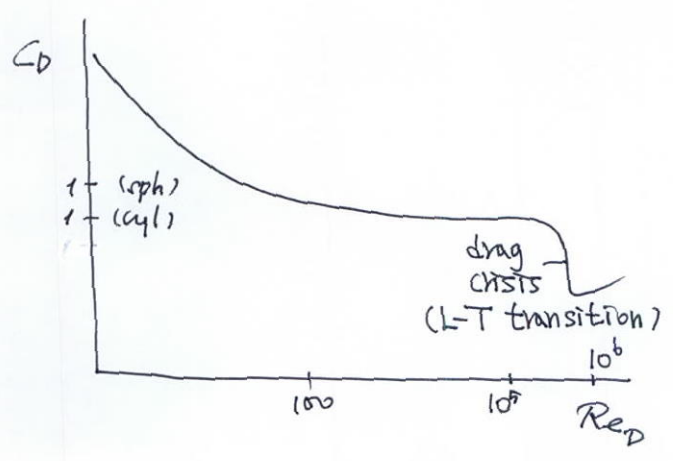


② wetted area : total surface exposed to flow

③ plan form area



④ $V^{2/3}$

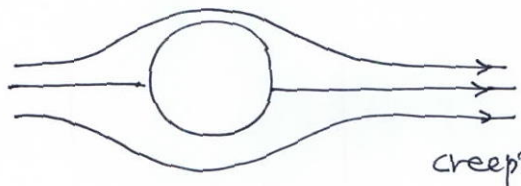


for cylinder
 $S = \pi D L$

for sphere
 $S = \frac{\pi}{4} D^2$

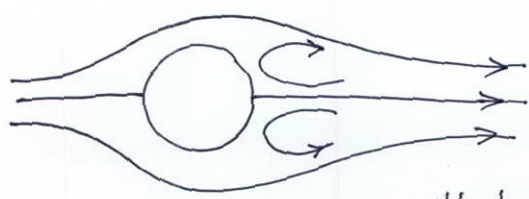
§ Flow around a cylinder

$Re < 10$



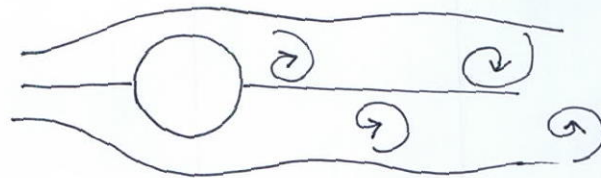
creeping flow

$10 < Re < 40$



attached vortices

$40 < Re < 200,000$



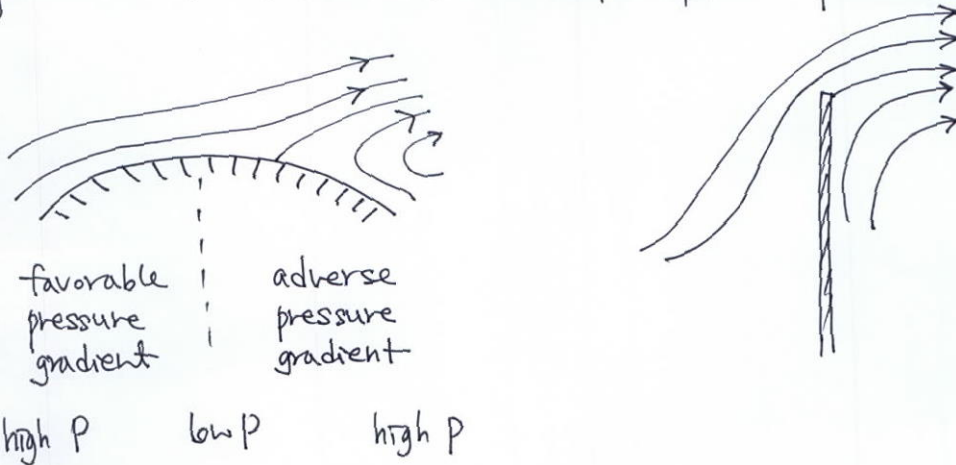
wide wake, $\theta_{sep} = 82^\circ$
von Karman vortex trail

$Re > 200,000$



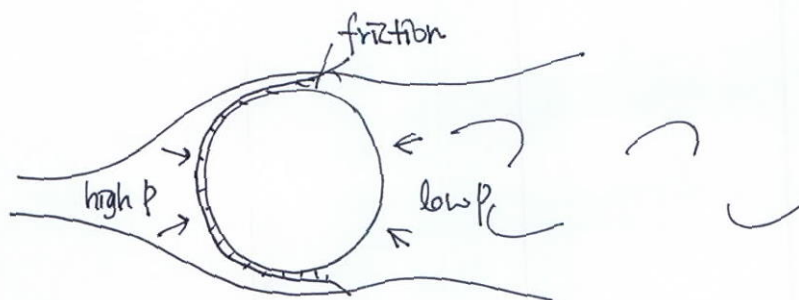
fully turbulent wake
 $\theta_{sep} = 120^\circ$

separation of flow (boundary-layer separation)



Shape and two kinds of drag

overall drag = skin friction + pressure drag



if pressure drag dominates, (high Re)

$$F_D \sim \rho U^2 S$$

$$C_D = \frac{F_D}{\frac{1}{2} \rho U^2 S} \sim k \text{ (const.)}$$

if skin friction dominates, (low Re)

$$F_D \sim \mu \frac{U}{L} \cdot L^2 \sim \mu U L$$

$$C_D = \frac{F_D}{\frac{1}{2} \rho U^2 L^2} \sim \frac{\mu U L}{\frac{1}{2} \rho U^2 L^2} \sim \frac{\mu}{\rho U L} \sim \frac{1}{Re}$$

· read discussion on Table 5.2 (p. 97 - 98)

* How to reduce drag (C_D)

· high Re - pressure drag dominates

→ "streamlining"

(forward-directed pressure from the rear nearly counterbalances the dynamic pressure on the front)

vs. blunt body (bluff body)

(energy lost in the wake
~ pressure decreases)

$$C_D, \text{ airplane} \sim 0.02 C_D, \text{ sphere} \quad (\text{same frontal area})$$

· main disadvantage of streamlining as a scheme for drag reduction

~ if assumes a particular orientation to the flow.

∴ minor change in wind direction

~ wipe out the benefits of a streamlined (stationary) body.

· low Re - streamlining increases surface area

: skin friction ↑

- may outweigh reduction in pressure drag.

- roughness

e.g. dimples on golf balls

only effective (for bluff bodies
within a range of Re (25,000 ~ 150,000)

promoting turbulence → narrowing wake region
~ decreasing pressure drag.

Fig. 5.8.