Chapter 8. Wave Forces

Potential flow approach



Figure 8-1 Potential flow around a circular cylinder.

$$\nabla^2 \phi = \frac{\partial^2 \phi}{\partial r^2} + \frac{1}{r} \frac{\partial \phi}{\partial r} + \frac{\partial^2 \phi}{r^2 \partial \theta^2} + \frac{\partial^2 \phi}{\partial z^2} = 0 \quad \text{with} \quad u_r = -\frac{\partial \phi}{\partial r}, \quad u_\theta = -\frac{1}{r} \frac{\partial \phi}{\partial \theta}, \quad u_z = -\frac{\partial \phi}{\partial z}$$
$$\phi(r, \theta, t) = U(t) r \left(1 + \frac{a^2}{r^2} \right) \cos \theta$$

$$u_r(a,\theta,t) = -\frac{\partial\phi}{\partial r}\Big|_{r=a} = -U(t)\left(1 - \frac{a^2}{r^2}\right)\cos\theta\Big|_{r=a} = 0$$

$$u_{\theta}(a,\theta,t) = -\frac{1}{r} \frac{\partial \phi}{\partial \theta} \bigg|_{r=a} = U(t) \left(1 + \frac{a^2}{r^2}\right) \sin \theta \bigg|_{r=a} = 2U(t) \sin \theta$$

Bernoulli equation at cylinder wall and far upstream:

$$\left[\frac{p(r,\theta)}{\rho} + gz + \frac{u_r^2 + u_\theta^2}{2} - \frac{\partial\phi}{\partial t}\right]_{r=a} = \left[\frac{p(r,\theta)}{\rho} + gz + \frac{u_r^2 + u_\theta^2}{2} - \frac{\partial\phi}{\partial t}\right]_{\substack{r=l>>a\\\theta=0}}$$

$$p(a,\theta) - p(l,0) = \rho \left[\frac{U^2(t)}{2} (1 - 4\sin^2 \theta) + 2a \frac{dU}{dt} \cos \theta - l \frac{dU}{dt} \right] + O(a^2 / l^2)$$

Steady flow term

Inertial term

Steady flow term:



Figure 8.2 Pressure distribution around cylinder for case of ideal flow. Note the low pressure at the sides, $\theta = 90^{\circ}$, and the symmetry with respect to $\theta = 0^{\circ}$ and $\theta = 90^{\circ}$.

pressure at front face = pressure at rear face \rightarrow net pressure force = 0



Figure 8.3 Calculation of elemental force in x direction. ΔF_x is positive in the downstream (-x) direction.

Drag force (=net force in *x*-direction):

$$dF_D = \int_0^{2\pi} p(a,\theta)a\cos\theta d\theta = \int_0^{2\pi} \left[\frac{\rho U^2(t)}{2}(1-4\sin^2\theta) + p(l,0)\right]a\cos\theta d\theta = 0$$

No force on cylinder in ideal steady flow (D'Alembert's paradox) ↑

· Potential flow assumption precludes the formation of boundary layers and a wake

Pressure distribution for real flow:



Figure 8.4 Measured pressure distributions around cylinders. (From Goldstein, 1938.)

$$dF_{D} = 2\int_{0}^{\theta_{s}} \frac{\rho U^{2}(t)}{2} (1 - 4\sin^{2}\theta) a \cos\theta d\theta + 2\int_{\theta_{s}}^{\pi} p_{\text{wake}} a \cos\theta d\theta$$
$$= \rho U^{2}(t) a \left[\int_{0}^{\theta_{s}} (1 - 4\sin^{2}\theta) \cos\theta d\theta + \int_{\theta_{s}}^{\pi} \frac{p_{\text{wake}}}{\rho U^{2}(t)/2} \cos\theta d\theta \right]$$
$$= C_{D}(\mathbf{R}) \frac{\rho D U^{2}(t)}{2}$$
$$= C_{D}(\mathbf{R}) \frac{\rho A U^{2}(t)}{2}$$

where D = 2a = diameter of cylinder, A = 2a = projected area/unit elevation, $\mathbf{R} = UD / v$ = Reynolds number, C_D = drag coefficient



Figure 8.5 Variation of drag coefficient, C_D with Reynolds number \mathbb{R} for a smooth circular cylinder. (From H. Schlichting, *Boundary Layer Theory*. Copyright © 1968 by McGraw-Hill Book Company. Used with the permission of McGraw-Hill Book Company.)

Unsteady flow:

$$p(a,\theta) - p(l,0) = \rho \left[\frac{U^2(t)}{2} (1 - 4\sin^2 \theta) + 2a \frac{dU}{dt} \cos \theta - l \frac{dU}{dt} \right]$$
$$dF_I = \int_0^{2\pi} \rho \frac{dU(t)}{dt} 2a^2 \cos^2 \theta d\theta - \int_0^{2\pi} \rho \frac{dU(t)}{dt} la \cos \theta d\theta$$
$$= 2\rho \pi a^2 \frac{dU}{dt}$$
$$= C_M \rho V \frac{dU}{dt}$$
where $V = \pi a^2$ = volume of cylinder per unit length

where $V = \pi a^{-1}$ = volume of cylinder per unit length $C_M = 1 + k_m$ = inertia coefficient, k_m = added mass coefficient \uparrow 1: pressure gradient to accelerate the fluid in the absence of cylinder k_m : additional local pressure gradient to accelerate the neighboring fluid around the cylinder (shape dependent, e.g. $k_m = 1$ for circular cylinder)

In the case where a cylinder is accelerating in quiescent water, $C_M = k_m$ because there is no pressure gradient in quiescent water.

Read textbook for more detailed explanation for the added mass coefficient, which is briefly summarized as follows:

Vertical buoyancy force in hydrostatic (quiescent) fluid is

 $F_{\rm B} = \rho g V$ (i.e. Archimedes principle)

Vertical pressure gradient in hydrostatic fluid is

$$\frac{\partial p}{\partial z} = -\rho g$$

Therefore

$$F_{\scriptscriptstyle B} = -\frac{\partial p}{\partial z} V$$

The buoyancy force in hydrostatic fluid is caused by the pressure difference in vertical direction (i.e. larger pressure on the lower side of a body than on the upper side). There is no horizontal buoyancy force in hydrostatic fluid because $\partial p / \partial x = 0$. However, $\partial p / \partial x \neq 0$ in an accelerating fluid. In the form similar to the vertical buoyancy force in hydrostatic fluid, the horizontal buoyancy-like force in accelerating fluid is

$$F_{\rm BH} = -\frac{\partial p}{\partial x}V = \rho \frac{du}{dt}V \qquad (\because \frac{du}{dt} = -\frac{1}{\rho}\frac{\partial p}{\partial x}, \text{ Euler equation})$$

Comparing with the inertia force,

$$dF_{I} = C_{M} \rho V \frac{du}{dt} = (1 + k_{m}) \rho V \frac{du}{dt}$$

$$\uparrow$$
'1' indicates the horizontal buoyancy-like force in accelerating fluid
$$\uparrow$$

This force exists even if there is no structure. If there is a structure, additional acceleration will occur around the structure, which is represented by k_m (added mass coefficient)

$$(1+k_m)\rho V = \rho V + k_m \rho V$$

$$\downarrow$$
mass of the structure added mass

If a structure is accelerating through a quiescent ideal fluid, since there is no pressure gradient in the fluid, the force would only be due to the added mass coefficient, i.e.

$$dF_I = C_M \rho V \frac{du}{dt} = k_m \rho V \frac{du}{dt}$$

Sarpkaya and Isaacson (1981), Mechanics of Wave Forces on Offshore Structures, gives

APPENDIX A

	SHAPE	ADDED MA	SS LENGTH MOTION	= p∀(Ca) e.g.	
(•c)	CIRCLE	ρπς2	4	$C_a = 1$	K	
b + a	ELLIPSE	ρπb ²				
	ELLIPSE	ρπa ²				• •
	PLATE	ρ π w ²				
$\frac{1}{2a}$	RECTANGL	<u>a/b</u> E - 1 - 10 1 5 1 2 1	.00 ρπa ² .14 " .21 " .36 "	<u>a/b</u> 1 1 0.5 1 0.2 1 0.1 2	.51 ρπa ² .70 " .98 " .23 " (Wen	del 1950)
	DIAMOND	2 0 1 0 0.5 0 0.2 0).85 ").76 ").67 ").61 "	(Wendel 1	950)	
	I-BEAM a/c = 2.0 b/c = 3.0	5 5 2	- .11 ρπa ²	(Patton 1	965)	
n SIDED	REGULAR POLYGON OR	n = 3 0 4 0 5 0 6 0 - 1	9.654 ρπa ² 9.787 " 9.823 " 9.867 " 9.000 "	(Wendel 1	950)	

 Table 2.3 Added Masses of Various Bodies

Forces due to real fluids

Morison equation:

$$dF = dF_D + dF_I = \frac{1}{2}C_D\rho Au \left| u \right| + C_M \rho V \frac{du}{dt}$$

Assuming constant C_D and C_M over the depth and using linear wave theory,

Total force =
$$F = \int_{-h}^{\eta} dF = \text{Eq.} (8.36)$$

Moment about seabed = $M = \int_{-h}^{h} (h+z) dF = \text{Eq.}$ (8.38)

To determine C_D and C_M for real fluid, minimize

$$\varepsilon^2 = \frac{1}{I} \sum_{i=1}^{I} (F_{mi} - F_{pi})^2$$

with respect to C_D and C_M for the data groups of approximately same Reynolds number.

 $C_D = f(\mathbf{R})$ as shown in Figure 8.9

 $C_M = 1.33$ as shown in Figure 8.10 (k_m reduces to 0.33 from 1.0 for potential flow)



Figure 8.9 Drag coefficient variation with Reynolds number as determined by Dean and Aagaard (1970). Copyright 1970 SPE-AIME.

Relative importance of inertia and drag forces:

$$\frac{(dF_I)_{\max}}{(dF_D)_{\max}} = \frac{C_M \pi kh}{C_D} \frac{D}{H}$$

Therefore, large $D/H \rightarrow$ inertia-dominant

small
$$D/H \rightarrow$$
 drag-dominant



.

Figure 8.13 H/D versus h/L_0 for condition of equal maximum drag and inertia force components.

$$\frac{(dF_I)_{\max}}{(dF_D)_{\max}} = \pi^2 \frac{C_M}{C_D} \frac{1}{u_m T / D} = \frac{\pi}{2} \frac{C_M}{C_D} \frac{1}{S / D}$$

where $u_m T / D$ = Keulegan-Carpenter number, and S / D = displacement parameter



$$u = u_m \cos \sigma t$$
; $S = \int_0^{T/4} u_m \cos \sigma t dt = \frac{u_m T}{2\pi}$

large $u_m T / D$ or $S / D \rightarrow$ drag-dominant

small $u_m T / D$ or $S / D \rightarrow$ inertia-dominant

Maximum total force:



Figure 8.18 Illustration of force component combination for the case of $|F_I| = 2 |F_D|$.

 $(F_T)_m$ occurs somewhere between $\sigma t = -90^\circ$ and $\sigma t = 0^\circ$.

At
$$(F_T)_m$$
, $dF_T/dt = 0 = -2F_D\sigma\cos(\sigma t)_m\sin(\sigma t)_m - F_I\sigma\cos(\sigma t)_m$ (8.67)

Two possible roots:

- 1) Dividing Eq. (8.67) by $\sigma \cos(\sigma t)_m$, $\sin(\sigma t)_m = -\frac{F_I}{2F_D} \rightarrow (F_T)_m = F_D + \frac{F_I^2}{4F_D}$
- 2) $\cos(\sigma t)_m = 0 \rightarrow (F_T)_m = F_I \leftarrow$ This solution must be taken if $F_I / 2F_D > 1$

Inertia force dominant case

Large structure \rightarrow Drag force is negligible \rightarrow Potential flow theory

MacCamy and Fuchs solution for circular cylinder:

Incident wave propagating in +x direction is

$$\phi_{I} = -\frac{gH}{2\sigma} \frac{\cosh k(h+z)}{\cosh kh} e^{i(kx-\sigma t)}$$
$$= -\frac{gH}{2\sigma} \frac{\cosh k(h+z)}{\cosh kh} \left[J_{0}(kr) + \sum_{m=1}^{\infty} 2i^{m} \cos m\theta J_{m}(kr) \right] e^{-i\sigma t}$$

which satisfies the Laplace equation, BBC, LKFSBC and LDFSBC.

Reflected wave from the cylinder is

$$\phi_{R} = \sum_{m=0}^{\infty} A_{m} \cos m\theta [J_{m}(kr) + iY_{m}(kr)]e^{-i\sigma t} \frac{\cosh k(h+z)}{\cosh kh}$$

which satisfies the Laplace equation, BBC, and radiation boundary condition for large kr.

No-flow condition at the cylinder, $\partial(\phi_I + \phi_R) / \partial r = 0$ at r = a, gives

$$\phi_{I+R} = \operatorname{Re}\left\{\frac{gH\cosh k(h+z)}{2\sigma\cosh kh}e^{-i\sigma t}\left\{\left[J_{0}(kr) - \frac{J_{1}(ka)}{J_{0}(ka) - iY_{0}(ka)}(J_{0}(kr) + iY_{0}(kr))\right] + 2\sum_{m=1}^{\infty}i^{m}\left[J_{m}(kr) - \frac{J_{m}(ka)}{J_{m}(ka) - iY_{m}(ka)}(J_{m}(kr) + iY_{m}(kr))\right]\right\}\cos m\theta\right\}$$

Using the unsteady form of Bernoulli equation to obtain the pressure (see Eq. 8.5), force per unit length of the cylinder is obtained:

$$dF_{I} = \frac{2\rho gH}{k} \frac{\cosh k(h+z)}{\cosh kh} G\left(\frac{D}{L}\right) \cos(\sigma t - \alpha)$$

where

$$\tan \alpha = \frac{J_1'(ka)}{Y_1'(ka)}; \quad G\left(\frac{D}{L}\right) = \frac{1}{\sqrt{J_1'(ka)^2 + Y_1'(ka)^2}}$$

Comparing with the general formula for inertial force,

$$dF_I = C_M \rho V \frac{\partial u}{\partial t}$$

we find $C_{M} = 4G(D/L) / \pi^{3} (D/L)^{2}$



Figure 8.19 Variation of inertia coefficient C_M and phase angle α of maximum force with parameter D/L.

 C_{M} and α reduce to 2.0 and 0, respectively, for small values of D/L, as predicted from potential flow theory.

Wave force on large rectangular objects:



Figure 8.20 Dynamic wave pressures on rectangular object.

 l_1 , l_2 , and l_3 = length in x, y, and z directions

z = -S at the bottom of the object

Calculate the horizontal inertia force F_x due to wave propagating in the x direction In the absence of the structure,

$$p(x, z, t) = \rho g \eta K_p(z) = \frac{\rho g H}{2} \frac{\cosh k(h+z)}{\cosh kh} \cos(kx - \sigma t)$$

$$P_{1}(x_{1},t) = \int_{-S}^{-S+l_{3}} l_{2} p(x_{1},z,t) dz = \frac{l_{2} l_{3} \rho g H \cos(kx_{1} - \sigma t)}{2 \cosh k h} \cosh k \left(h - S + \frac{l_{3}}{2}\right) \frac{\sinh\left(\frac{1}{2} k l_{3}\right)}{\frac{1}{2} k l_{3}}$$

$$P_{2}(x_{1}+l_{1},t) = \int_{-S}^{-S+l_{3}} l_{2} p(x_{1}+l_{1},z,t) dz$$
$$= \frac{l_{2}l_{3}\rho gH}{2\cosh kh} \cos[k(x_{1}+l_{1})-\sigma t)] \cosh k \left(h-S+\frac{l_{3}}{2}\right) \frac{\sinh\left(\frac{1}{2}kl_{3}\right)}{\frac{1}{2}kl_{3}}$$

$$F_{x} = P_{1} - P_{2} = \frac{l_{1}l_{2}l_{3}\rho gHk\cosh k(h-S+l_{3}/2)}{2\cosh kh} \frac{\sinh(kl_{3}/2)}{kl_{3}/2} \frac{\sin(kl_{1}/2)}{kl_{1}/2} \sin\left(k\left(x_{1}+\frac{l_{1}}{2}\right)-\sigma t\right)$$

This can be written in a more familiar form:

$$F_x = \rho V \frac{\sinh(kl_3/2)}{kl_3/2} \frac{\sin(kl_1/2)}{kl_1/2} \frac{\partial u}{\partial t}$$

where $\partial u / \partial t$ is evaluated at the center of the rectangular object.

For an object occupying from bottom to water surface, $S = l_3 = h$ gives

$$F_{x} = \frac{l_{1}l_{2}h\rho gHk\cosh(kh/2)}{2\cosh kh} \frac{\sinh(kh/2)}{kh/2} \frac{\sin(kl_{1}/2)}{kl_{1}/2} \sin\left(k\left(x_{1} + \frac{l_{1}}{2}\right) - \sigma t\right)$$

Accounting for the interaction of the structure with the waves,

$$F_x' = (1+k_m)F_x$$

Added mass coefficient, k_m , should be determined by experiments or Table 2.3 of

Sarpkaya and Isaacson (1981)

The vertical force can be calculated in a similar manner:

$$F_z = \rho V \frac{\sinh(kl_3/2)}{kl_3/2} \frac{\sin(kl_1/2)}{kl_1/2} \frac{\partial w}{\partial t}$$

where $\partial w / \partial t$ is again evaluated at the center of the object.

If the tank is situated on the bottom, such that the wave-induced pressure is not transmitted to the bottom of the tank,

$$F_{z} = -\rho V \frac{\coth(kl_{3})}{kl_{3}} \frac{\sin(kl_{1}/2)}{kl_{1}/2} \frac{\partial w}{\partial t}$$

where $\partial w / \partial t$ is now calculated at the center of the top of the object.

Spectral approach for irregular waves

For inertia-dominant case,

Total force
$$(F_I)_{\text{max}} = G(\sigma)H; \quad G(\sigma) = C_M \frac{\rho \pi D^2}{8k} \sigma^2$$

For the case in which both drag and inertia forces are important, the force on an element length ds at a distance s above the seabed is

$$dF = \left(\frac{C_D \rho D}{2} u_m^2 \cos \sigma t \mid \cos \sigma t \mid -\frac{C_M \rho \pi D^2}{4} u_m \sigma \sin \sigma t\right) ds$$

Linearizing,

$$dF = \left(\frac{C_{D_L}\rho D}{2}u_m \cos \sigma t - \frac{C_M \rho \pi D^2}{4}u_m \sigma \sin \sigma t\right) ds$$

where

$$C_{D_L} = C_D \sqrt{\frac{8}{\pi}} U_{rms}$$

Force spectrum is related to surface wave spectrum by

$$S_{dF}(\sigma) = |X_{dF}(\sigma)|^2 S_{\eta}(\sigma)$$

where

$$|X_{dF}(\sigma)|^{2} = \left[\frac{C_{D}\rho D}{2}\sqrt{\frac{8}{\pi}}U_{rms}X_{u}(\sigma,s)\right]^{2} + \left[\frac{C_{M}\rho\pi D^{2}}{4}X_{u}(\sigma,s)\sigma\right]^{2}$$
$$X_{u}(\sigma,s) = \frac{u_{m}}{|\eta|} = \sigma\frac{\cosh ks}{\sinh kh}$$
$$U_{rms} = \sqrt{\int_{0}^{\infty}|X_{u}(\sigma,s)|^{2}S_{\eta}(\sigma)d\sigma}$$

Total wave force over the entire water depth is

$$S_F(\sigma) = \left\{ \left(\frac{C_D \rho D}{2}\right)^2 \sigma \frac{8}{\pi} G_1(\sigma) + \left(\frac{C_M \rho \pi D^2}{4}\right)^2 G_2(\sigma) \right\} S_\eta(\sigma)$$

where

$$G_1(\sigma) = \frac{\int_0^h U_{rms}(s) \cosh ks ds}{\sinh kh}$$

$$G_2(\sigma) = \frac{\sigma^2}{k}$$