

# Lect 3.2: Torsion for Typical Sections

노트 제목

1

Approach

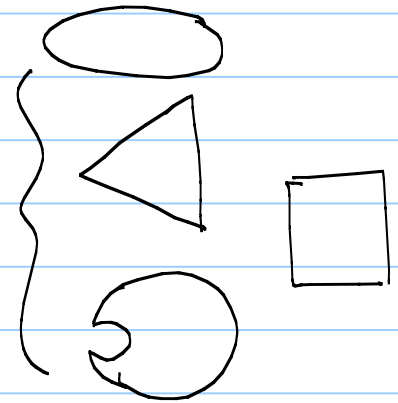
Analogy!

i) Prandtl Stress Function Approach

$$\begin{cases} \nabla^2 \psi = -2 & \text{in } R & (a) \\ \psi = 0 & \text{on } C & (b) \end{cases}$$

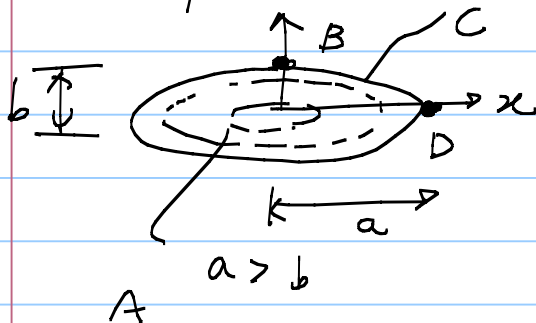
ii) Membrane problem

$$\begin{cases} \nabla^2 \eta = -p/T & \text{in } R \\ \eta = 0 & \text{on } C \end{cases}$$



$$\boxed{\begin{aligned} \psi &\sim \eta \\ \tau_{\text{stress}} &\sim -\frac{d\eta}{dx} \\ \text{shear stress} & \\ C_t &= 2\mu \int \psi dA \sim \int \eta dA \end{aligned}}$$

< Elliptic cross section >



$$C: \left(\frac{2x}{a}\right)^2 + \left(\frac{2y}{b}\right)^2 - 1 = 0$$

$$\text{Try: } \psi = m \left( \frac{x^2}{a^2} + \frac{y^2}{b^2} - 1 \right)$$

→ satisfies (b)

and satisfies (a)

$$\text{if } m = -\frac{a^2 b^2}{a^2 + b^2}$$

$\tau_B$ : max  
(membrane Analogy)

$$\left\{ \begin{array}{l} \sigma_{zx} = \mu \alpha \frac{\partial \psi}{\partial y} = -2\mu \alpha \frac{a^2 y}{a^2 + b^2} \\ \sigma_{zy} = -\mu \alpha \frac{\partial \psi}{\partial x} = \frac{2\mu \alpha a^2 x}{a^2 + b^2} \end{array} \right.$$

$$C_t = 2\mu \int_R \psi dA = \frac{\pi \mu a^3 b^3}{a^2 + b^2}$$

Q:  $C_t|_{\max} = ?$  for given  $\pi ab = A_0$  (Area)

$$C_t = \frac{\pi \mu \left(\frac{A_0}{\pi}\right)^3}{a^2 + \left(\frac{A_0}{\pi a}\right)^2}$$

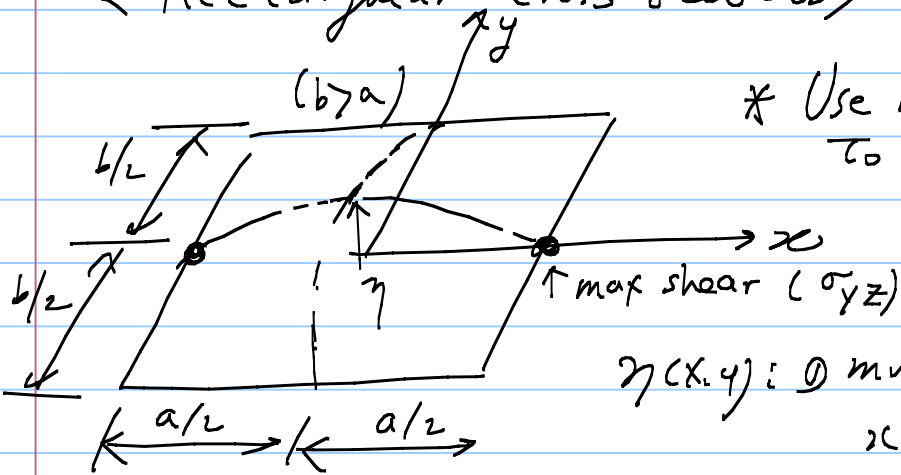
becomes max when

$$a^2 = \left(\frac{A_0}{\pi a}\right)^2$$

$$\pi a^2 = A_0$$

$\therefore$  Circular section has the largest torsional rigidity.

< Rectangular Cross sections >



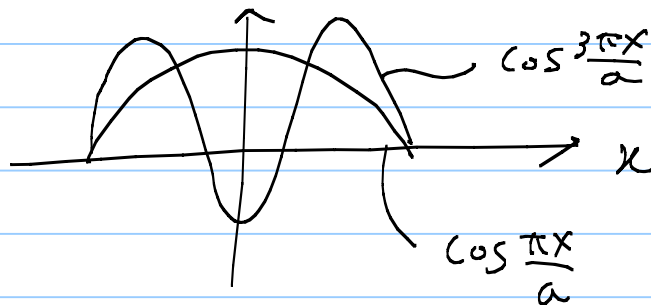
\* Use Membrane Analogy  
to find solution:

①  $\eta(x, y)$  must be sym wnt  
x and y axes

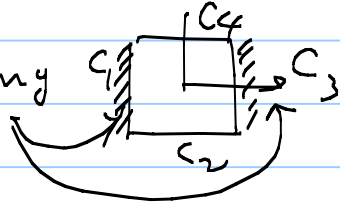
②  $\eta = 0$  along the  
boundary

$$\begin{aligned} \nabla^2 \eta &= -\frac{\phi}{T} \text{ in } R \text{ ①} \\ \eta &= 0 \text{ on } C \text{ ②} \end{aligned}$$

$\therefore$  Let 
$$\eta(x, y) = \sum_{n=0}^{\infty} \cos \frac{(2n+1)\pi x}{a} Y_n(y) \quad \text{③}$$



- i) ③ satisfies BC's along  $C_1, C_3$
- ii) Substitute ③ into ① and check BC's along other sides  $C_2, C_4$



$$-\frac{\rho}{T} = \frac{\partial^2 \eta}{\partial x^2} + \frac{\partial^2 \eta}{\partial y^2}$$

$$= \sum_{n=0}^{\infty} \left\{ - \left[ \left( \frac{2n+1}{a} \right) \pi \right]^2 Y_n(y) + Y_n''(y) \right\} \times \cos \frac{2n+1}{a} \pi x \quad \text{④}$$

Let  $\gamma_n = \frac{(2n+1)\pi}{a}$ ,

Use  $-\frac{\rho}{T} = \left( \frac{-\rho}{T} \right) \frac{4}{a} \sum_{n=0}^{\infty} \frac{(-1)^n}{\gamma_n} \cos \gamma_n x$

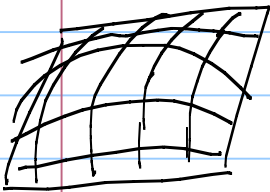
Then ④ becomes

$$\boxed{\frac{d^2 Y_n(y)}{dy^2} - \gamma_n^2 Y_n(y) = -\frac{4\rho}{aT} \frac{(-1)^n}{\gamma_n}} \quad \text{⑤}$$

for every  $n$

Solution to (5)

$$\begin{aligned} Y_n &= Y_n^c + Y_n^p \\ &= A_n \sinh \delta_n y + B_n \cosh \delta_n y \\ &\quad + \frac{4P}{aT} \frac{(-1)^n}{\delta_n^3} \end{aligned}$$



• Using the solution symmetry wrt  $y=0$ ,  
 $A_n = 0$

• Using the BC at  $y = \pm b/2$

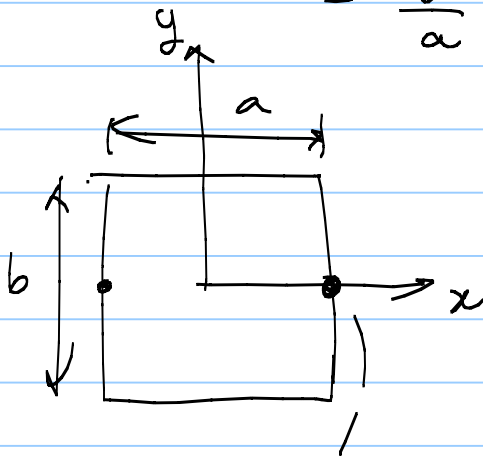
$$B_n \cosh \frac{b\delta_n}{2} + \frac{4P}{aT} \frac{(-1)^n}{\delta_n^3} = 0$$

$$\rightarrow B_n = -\frac{4P}{aT} \frac{(-1)^n}{\delta_n^3} \cdot \frac{1}{\cosh b\delta_n/2}$$

$$\therefore Y_n(y) = \frac{4P}{aT} \frac{(-1)^n}{\delta_n^3} \left[ 1 - \frac{\cosh \delta_n y}{\cosh b\delta_n/2} \right]$$

$$\therefore \psi(x, y) = \frac{2T}{\rho} \eta(x, y)$$

$$= \frac{8}{a} \sum_{n=0}^{\infty} \frac{(-1)^n}{\gamma_n^3} \left[ 1 - \frac{\cosh \gamma_n y}{\cosh b \gamma_n / 2} \right] \cos \gamma_n x$$



$b > a$

max shear

$$\underline{\underline{\sigma_{zy} \left( x = \pm \frac{a}{2}, y = 0 \right)}}$$

$$\sigma_{zy} = -\mu \alpha \frac{\partial \psi}{\partial x}$$

$$= \frac{8\mu\alpha}{a} \sum_{n=0}^{\infty} \frac{(-1)^n}{\gamma_n^2}$$

$$\times \left[ 1 - \frac{\cosh \gamma_n y}{\cosh b \gamma_n / 2} \right] \sin \gamma_n x$$

$$\therefore \tau_{\max} = \sigma_{zy} \Big|_{\left(\frac{a}{2}, 0\right)}$$

$$= \frac{8\mu\alpha}{a} \sum_{n=0}^{\infty} \frac{(-1)^n}{\gamma_n^2} \left[ 1 - \frac{1}{\cosh b \gamma_n / 2} \right] \underbrace{\sin \left( n \left( \frac{a}{2} + \frac{\pi}{2} \right) \right)}_{(-1)^n}$$

$$= \frac{8\mu\alpha}{a} \sum_{n=0}^{\infty} \left( \frac{1}{\gamma_n^2} - \frac{1}{\gamma_n^2 \cosh b\gamma_n/2} \right)$$

Note  $\sum_{n=0}^{\infty} \frac{1}{\gamma_n^2} = \frac{a^2}{\pi^2} \sum_{n=0}^{\infty} \frac{1}{(2n+1)^2}$

$$= \frac{a^2}{\pi^2} \times \frac{\pi^2}{8} \Rightarrow \frac{a^2}{8}$$

↑ sum of an infinite series

∴

$$\tau_{\max} = \mu\alpha a \left[ 1 - \frac{8}{a^2} \sum_{n=0}^{\infty} \frac{1}{\gamma_n^2 \cosh(\gamma_n b/2)} \right]$$

( $b \gg a$ )

$$C_t = \frac{M_z}{\alpha} = 2\mu \int \psi \, dA$$

$$= \frac{\mu a^3 b}{3} \left[ 1 - \frac{92}{\pi^2} \left( \frac{a}{b} \right) \sum_{n=0}^{\infty} \frac{1}{(2n+1)^5} \tanh \gamma_n b/2 \right]$$

Compare



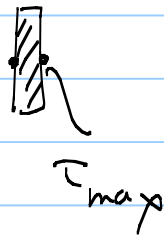
$$\tau_{\max} = \mu\alpha a$$



$$\tau_{\max} = 0.675 \mu\alpha a$$

## <Special Cases>

i)  $b/a \rightarrow \infty$   $\tau_{\max} = \mu \alpha a$



$$C_t = \frac{1}{3} \mu a^3 b \quad (\alpha = M_t / C_t)$$

$$= \left(\frac{a}{b}\right) \left[ \frac{1}{3} \mu (ab)^2 \right] \rightarrow 0$$

Area

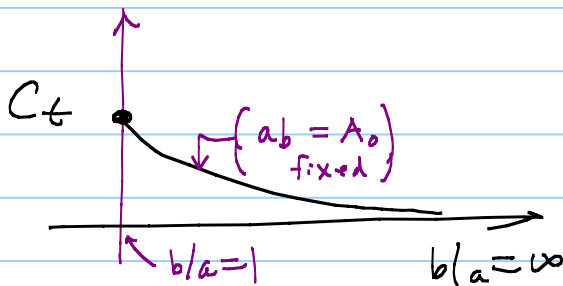
ii)  $b = a$   $\tau_{\max} = 0.675 \mu \alpha a$



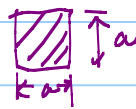
$$C_t = 0.422 \left( \frac{1}{3} \mu a^3 b \right)$$

$$= 0.422 \left( \frac{a}{b} \right) \left[ \frac{1}{3} \mu (ab)^2 \right]$$

" 1



Comparison:  $A_0 = \pi r_0^2 = a^2$  ( $a = \sqrt{\pi} r_0$ )



Circular cylinder

$$C_t = \frac{\mu A^2}{2\pi} \approx 0.159 \mu A_0^2$$

$$\tau_{\max} = 0.637 \frac{M_z}{r_0^3}$$

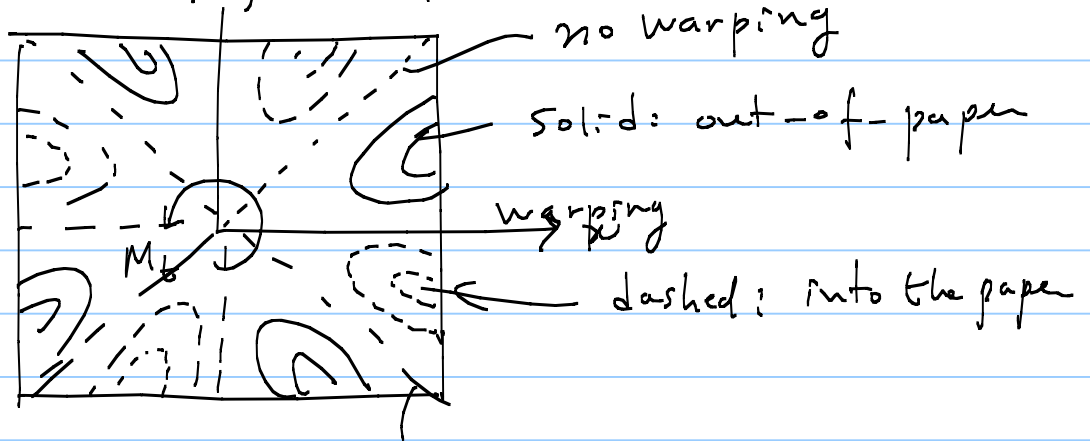
square cylinder

$$C_t = 0.141 \mu A_0^2 \quad (89\%)$$

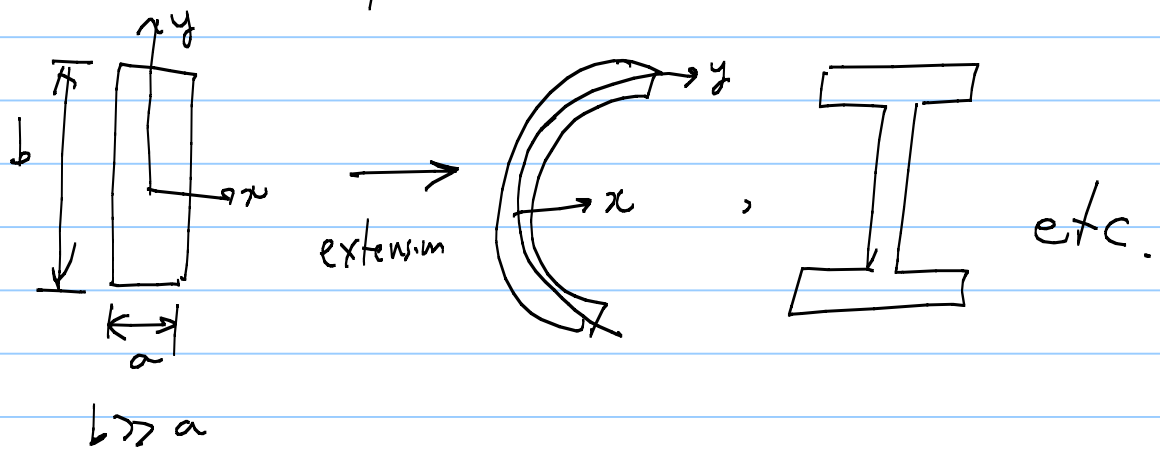
$$\tau_{\max} = 0.863 \frac{M_z}{r_0^3} \quad (135\%)$$



Check the warping ftn  $\phi$



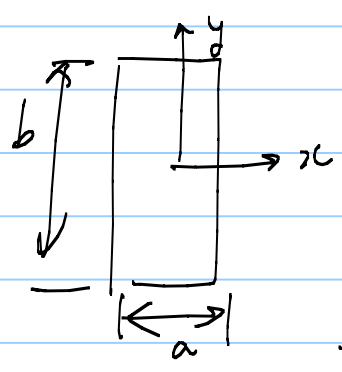
### <Narrow Rectangular Sections>



$$C_t = \frac{1}{3} \mu a^3 b$$

$$\tau_{max} = \mu \alpha a = M_z / \frac{1}{3} a^2 b$$

### Analysis for Narrow Rectangular Case



$$\frac{\partial^2 \eta}{\partial x^2} + \frac{\partial^2 \eta}{\partial y^2} = -\frac{\rho}{T}$$

( $\eta = 0$  along boundary)

Introduce Dimensionless Coord:

$$b \gg a$$

$$X = \frac{x}{a}, \quad Y = \frac{y}{b} \quad |X|, |Y| \leq \frac{1}{2}$$

//

Then,

$$\frac{\partial^2 \eta}{\partial X^2} + \frac{a^2}{b^2} \frac{\partial^2 \eta}{\partial Y^2} = -\frac{a^2 \rho}{T} \quad (*)$$

as  $b \rightarrow \infty$ , (\*) becomes

$$\frac{\partial^2 \eta}{\partial X^2} = -\frac{a^2 \rho}{T}$$

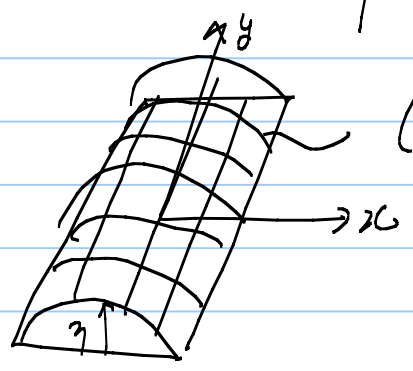
With  $X = ax$

$$\frac{\partial^2 \eta}{\partial x^2} = -\frac{\rho}{T}$$

Using  $\eta = 0$  at  $x = \pm a/2$

$$\eta = -\frac{\rho}{2T} \left( x^2 - \left(\frac{a}{2}\right)^2 \right)$$

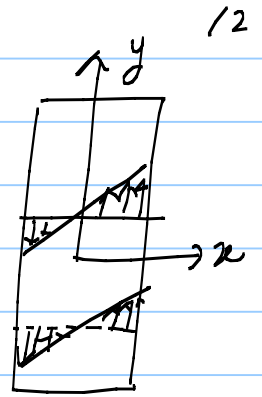
$$\therefore \psi = \frac{2T}{\rho} \eta = \left(\frac{a}{2}\right)^2 - x^2$$



$$\left(\frac{a}{2}\right)^2 - x^2$$

\* not valid near  $y = \pm b/2$

• Stress:  $\sigma_{yz} = -\mu\alpha \frac{\partial \psi}{\partial x} = 2\mu\alpha x$   
 $\sigma_{xz} = \mu\alpha \frac{\partial \psi}{\partial y} = 0$



$\therefore \tau_{max} = \sigma_{yz} \Big|_{x=\pm a/2} = \mu\alpha a$

$\left\{ \begin{array}{l} \star \sigma_{yz} \Big|_{x=0} = 0 \text{ (along the centerline)} \end{array} \right.$

• Torsional rigidity

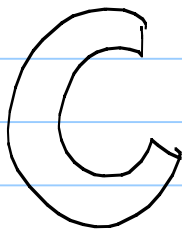
$$C_t = 2\mu \int \psi^2 dA$$

$$= 2\mu \int_{-a/2}^{a/2} \left[ \left(\frac{a}{2}\right)^2 - x^2 \right] b dx$$

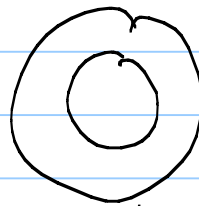
$$= \frac{1}{3} \mu a^3 b$$

# Extension to thin-walled open sections

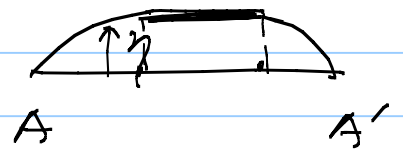
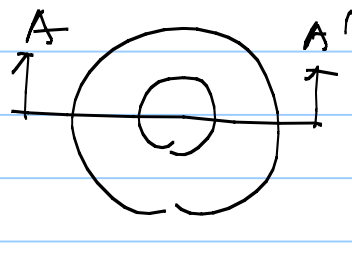
open



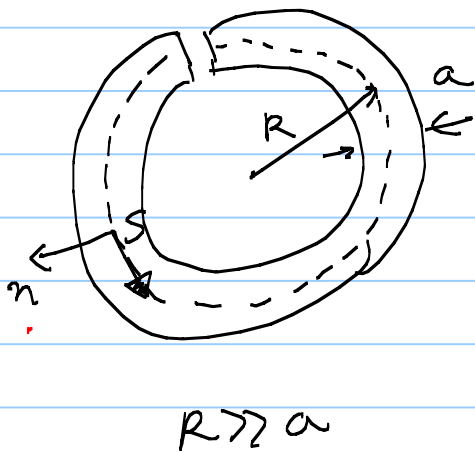
closed



Behaviour - completely different !!



< Example >



$$\begin{cases} \sigma_{zy} = 2\mu\alpha x \\ \text{but } x \leftrightarrow n, y \leftrightarrow S \end{cases}$$

Thus

$$\begin{cases} \sigma_{sz} = 2\mu\alpha n \\ \sigma_{sz}(n=0) = 0 \end{cases} \left( \begin{matrix} \sigma_{sz}^b \\ \sigma_{sz}^c = 0 \end{matrix} \right)$$

↑ centerline

$\Rightarrow \star \boxed{\begin{aligned} \epsilon_{sz}^0(s, z) &= \epsilon_{sz}(n=0, s, z) = 0 \\ &\text{along the middle line} \end{aligned}}$

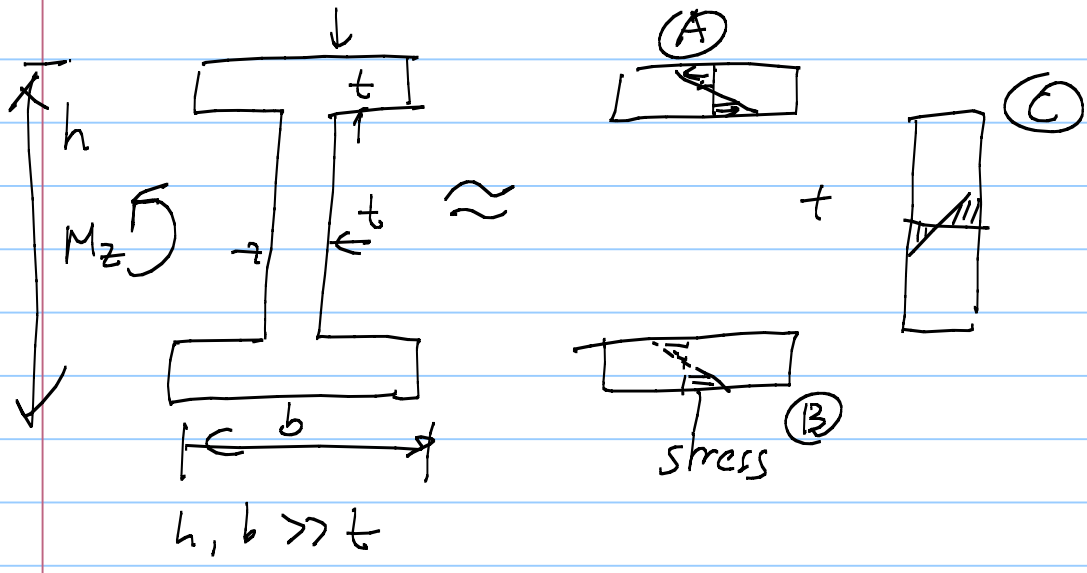
$$C_t = \frac{1}{3} \mu a^3 b$$

$$= \frac{1}{3} \mu a^3 (2\pi R)$$

$$b = 2\pi R$$

< Another Example >

↳ visualize the membrane deformation



$$\begin{aligned} C_t &= C_t^{(A)} + C_t^{(B)} + C_t^{(C)} \\ &= \frac{1}{3} \mu b t^3 + \frac{1}{3} \mu b t^3 + \frac{1}{3} \mu h t^3 \\ &= \frac{1}{3} \mu (2b + h) t^3 \end{aligned}$$

$$\begin{aligned} \tau_{\max} &= 2\mu \alpha n \\ &= 2\mu \alpha \left(\frac{t}{2}\right) = \mu \alpha t \\ &= \mu t \cdot \frac{M_z}{C_t} = \frac{3M_z}{(2b+h)t^2} \end{aligned}$$

(In this analysis, the detailed stress distribution near the corner is not considered).