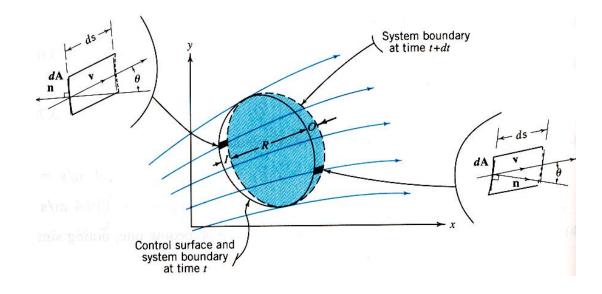


# Continuity, Energy, and Momentum Equations (3)







# Lecture 7 Continuity, Energy, and Momentum Equation<sup>356</sup> (3)

#### Contents

- 7.1 Linear Momentum Equation for Finite Control Volumes
- 7.2 The Moment of Momentum Equation for Finite Control Volumes

#### Objectives

- Derive the momentum equation by applying Newton's 2<sup>nd</sup> law of motion and Reynolds Transport Theorem
- Derive the moment of the momentum equation by applying Newton's 2<sup>nd</sup>
   law to <u>rotating fluid masses</u>





#### 7.1.1 Momentum Principle

The momentum equation can be derived from <u>Newton's 2nd law</u> of motion  $d\vec{a} = d(m\vec{a}) = d\vec{M}$ 

$$\vec{F} = m\vec{a} = m\frac{dq}{dt} = \frac{a(mq)}{dt} = \frac{dM}{dt}$$
(7.1)

$$\vec{M} = \underline{\text{linear momentum vector}} = m\vec{q}$$
  
 $\vec{F} = \underline{\text{external force}}$ 

 $\vec{F} = \begin{bmatrix} \text{boundary (surface) forces:} \\ \text{tangential to boundary - pressure,} \\ \vec{F}_{s} \end{bmatrix}$ 

body forces - force due to gravitational or magnetic fields,  $ec{F}_b$ 



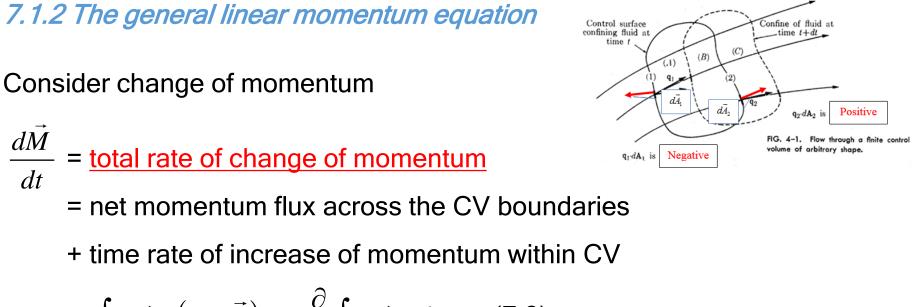


$$\vec{F}_p + \vec{F}_s + \vec{F}_b = \frac{d\vec{M}}{dt}$$

 $\vec{F}_{b} = \int_{CV} f_{b}(\rho dV)$ , where  $f_{b}$  = body force per unit mass





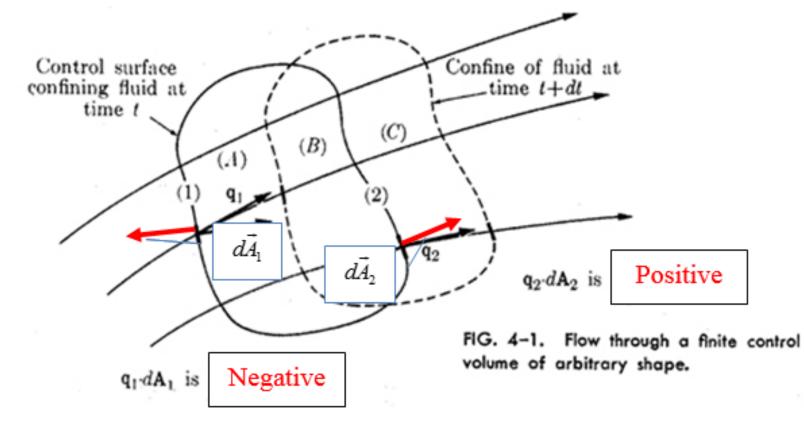


$$= \oint_{CS} \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} \vec{q} \rho \, dV \quad (7.3) \leftarrow \text{Reynolds Transport Theorem}$$

where  $\vec{q}\rho(\vec{q}\cdot d\vec{A})$  = momentum flux = velocity × mass per time  $d\vec{A}$  = vector unit area pointing <u>outward</u> over the control surface











#### Substitute (7.3) into (7.2)

$$\vec{F}_{p} + \vec{F}_{s} + \vec{F}_{b} = \oint_{CS} \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} \vec{q} \rho \, dV$$
(7.4)

For steady flow and negligible body forces

$$\vec{F}_{p} + \vec{F}_{s} = \oint_{CS} \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right)$$
(7.5)

- Eq. (7.4)
  - It is applicable to both ideal fluid systems and <u>viscous fluid systems</u> involving <u>friction and energy dissipation</u>.
  - It is applicable to both <u>compressible fluid and incompressible fluid</u>.





- Combined effects of friction, energy loss, and heat transfer appear implicitly in the magnitude of the external forces, with corresponding effects on the local flow velocities.
- Knowledge of the internal conditions is not necessary.
- We can consider only external conditions.





#### 7.1.3 Inertial control volume for a generalized apparatus

• Three components of the forces

$$x - dir.: \vec{F}_{p_{x}} + \vec{F}_{s_{x}} + \vec{F}_{b_{x}} = \oint_{CS} u\rho \left(\vec{q} \cdot d\vec{A}\right) + \frac{\partial}{\partial t} \int_{CV} u\rho \, dV$$
$$y - dir.: \vec{F}_{p_{y}} + \vec{F}_{s_{y}} + \vec{F}_{b_{y}} = \oint_{CS} v\rho \left(\vec{q} \cdot d\vec{A}\right) + \frac{\partial}{\partial t} \int_{CV} v\rho \, dV$$
$$z - dir.: \vec{F}_{p_{z}} + \vec{F}_{s_{z}} + \vec{F}_{b_{z}} = \oint_{CS} w\rho \left(\vec{q} \cdot d\vec{A}\right) + \frac{\partial}{\partial t} \int_{CV} w\rho \, dV$$
(7.6)

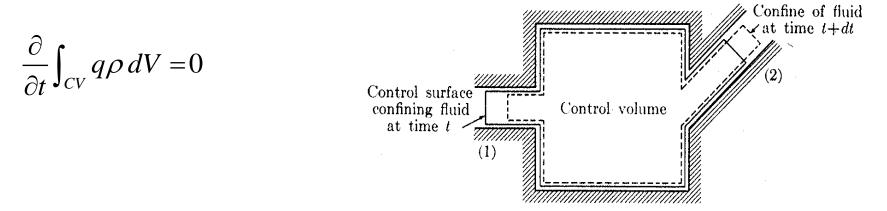




For flow through generalized apparatus

$$x - dir.: \vec{F}_{p_x} + \vec{F}_{s_x} + \vec{F}_{b_x} = \int_2 u\rho \, dQ - \int_1 u\rho \, dQ + \frac{\partial}{\partial t} \int_{CV} u\rho \, dV$$

• For 1D steady flow,







• Velocity and density are constant normal to the flow direction.

$$\begin{aligned} x - dir.: \quad \vec{F}_{p_{x}} + \vec{F}_{s_{x}} + \vec{F}_{b_{x}} &= \sum F_{x} = (V_{x}\rho Q)_{2} - (V_{x}\rho Q)_{1} \\ &= V_{x_{2}}\rho_{2}Q_{2} - V_{x_{1}}\rho_{1}Q_{1} = Q\rho(V_{x_{2}} - V_{x_{1}}) = Q\rho(V_{x_{out}} - V_{x_{in}}) \\ y - dir.: \quad \sum F_{y} = (V_{y}\rho Q)_{2} - (V_{y}\rho Q)_{1} \qquad \rho_{1}Q_{1} = \rho_{2}Q_{2} = Q\rho \\ z - dir.: \quad \sum F_{z} = (V_{z}\rho Q)_{2} - (V_{z}\rho Q)_{1} \end{aligned}$$

Control surface confining fluid

at time t

Control volume

where V = average velocity in flow direction

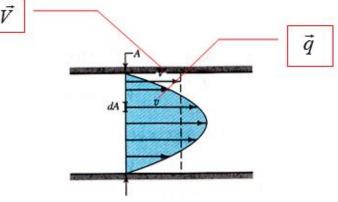




• Non-uniform velocity profile

If velocity varies over the cross section, then introduce momentum flux coefficient

$$\int \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right) = K_m \vec{V} \left( \rho V A \right) \qquad \vec{V}$$
$$\int \vec{q} \rho \ dQ = K_m \vec{V} \rho Q$$
$$K_m = \frac{\int \vec{q} \rho dQ}{\vec{V} \rho Q}$$







#### where

V = magnitude of average velocity over cross section = Q/A

 $\vec{V}$  = average velocity vector

 $K_m$  = momentum flux coefficient  $\geq 1$ 

=  $\int 1.33$  for laminar flow (pipe flow)

<u>1.03-1.04 for turbulent flow (smooth pipe)</u>

$$\sum F_{x} = (K_{m}V_{x}\rho Q)_{2} - (K_{m}V_{x}\rho Q)_{1}$$
$$\sum F_{y} = (K_{m}V_{y}\rho Q)_{2} - (K_{m}V_{y}\rho Q)_{1}$$
$$\sum F_{z} = (K_{m}V_{z}\rho Q)_{2} - (K_{m}V_{z}\rho Q)_{1}$$





[Cf] Energy correction coefficient

$$K_e = \frac{\int \frac{\rho}{2} q^2 dQ}{\frac{\rho}{2} \vec{V}^2 Q}$$

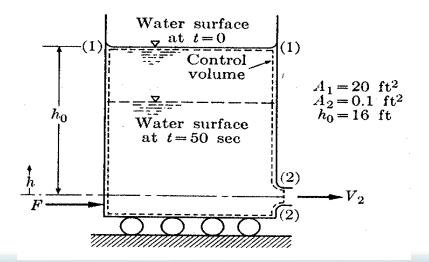




[Example 7-1] Continuity, energy, and linear momentum with <u>unsteady flow</u>

A large tank mounted on rollers is filled with water to a depth of 16 ft above a discharge port. At time t = 0, the fast-acting valve on the discharge nozzle is opened. Determine depth *h*, discharge rate Q, and force F necessary to keep the tank stationary at t = 50 sec.

Assumptions:  $V_1 \approx 0$   $\rho = \text{const.}$   $p_2 = p_{atm} = 0$  $h_2 = 0 \text{ (datum)}$ 





Apply continuity, energy, and linear momentum equations

(7.7) 
$$\int_{CV} \frac{\partial \rho}{\partial t} dV + \oint_{CS} \rho \vec{q} \cdot d\vec{A} = 0$$

(7.8) 
$$\frac{\delta Q}{dt} - \frac{\delta W_{shaft}}{dt} - \frac{\delta W_{shear}}{dt}$$
$$= \oint_{CS} \left( \frac{p}{\rho} + u + gh + \frac{q^2}{2} \right) \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} e\rho \, dV$$

(7.9) 
$$\vec{F}_p + \vec{F}_s + \vec{F}_b = \oint_{CS} \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} \vec{q} \rho \, dV$$





i) Use integral form of continuity equation, Eq. (7.7)

$$\frac{\partial}{\partial t} \int_{CV} \rho \, dV = \int \rho q_n dA_1 - \int \rho q_n dA_2$$
  

$$dV = A_1 dh , \quad \rho q_n dA_1 = 0 \quad \text{(because no inflow across the Section 1)}$$
  

$$\therefore \quad \rho A_1 \frac{\partial}{\partial t} \int_0^h dh = -\rho V_2 A_2$$
  

$$A_1 \frac{dh}{dt} = -V_2 A_2 \qquad (A)$$

ii) Energy equation, Eq. (7.8)

~ no shaft work

 $\sim$  heat transfer and temperature changes <u>due to friction are negligible</u>





 $\frac{\delta Q}{dt} - \frac{\delta W_{shaft}}{dt} - \frac{\delta W_{shear}}{dt}$ 

$$= \oint_{CS} \left( \frac{p}{\rho} + u + gh + \frac{q^2}{2} \right) \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} e\rho \, dV$$
II
$$e = \text{ energy per unit mass} = u + gh + \frac{q^2}{2}$$

$$I = \oint_{CS} \left( u + \frac{p}{\rho} + gh + \frac{q^2}{2} \right) \rho \left( \vec{q} \cdot d\vec{A} \right)$$

$$= \left( u + \frac{p}{\rho} + gh + \frac{q^2}{2} \right)_2 \rho V_2 A_2 - \left( u + \frac{p}{\rho} + gh + \frac{q^2}{2} \right)_1 \rho V_1 A_2$$





$$= \left(u + \frac{p}{\rho} + gh + \frac{q^2}{2}\right)_2 \rho V_2 A_2 \qquad (V_1 \approx 0)$$
  
$$II = \frac{\partial}{\partial t} \int_{CV} e\rho \, dV = \frac{\partial}{\partial t} \int_{CV} \left(u + gh + \frac{q^2}{2}\right) \rho dV \qquad A_1 \, dh$$

∴ nearly constant in the tank

except near the nozzle

$$= A_1 \rho \frac{\partial}{\partial t} \int_0^h (u + gh) dh$$

$$\therefore 0 = \left(u + \frac{p}{\rho} + gh + \frac{q^2}{2}\right)_2 \rho V_2 A_2 + A_1 \rho \frac{\partial}{\partial t} \int_0^h (u + gh) dh$$





Assume 
$$\rho = \text{const.}$$
,  $p_2 = p_{atm} = 0$ ,  $h_2 = 0$  (datum)

$$0 = uV_{2}A_{2} + \frac{V_{2}^{2}}{2}V_{2}A_{2} + uA_{1}\frac{dh}{dt} + A_{1}gh\frac{dh}{dt}$$
(B)  
e (A) into (B)
$$A_{1}\frac{dh}{dt} = -V_{2}A_{2}$$

Substitute

$$0 = uV_2A_2 + \frac{V_2^2}{2}V_2A_2 + u(-V_2A_2) + gh(-V_2A_2)$$

$$\therefore \quad \frac{V_2^2}{2}V_2A_2 = ghV_2A_2$$

$$V_2 = \sqrt{2gh} \tag{C}$$





#### Substitute (C) into (A)

$$A_2 \sqrt{2gh} = -A_1 \frac{dh}{dt}$$
$$\frac{dh}{\sqrt{h}} = -\frac{A_2}{A_1} \sqrt{2g} dt$$

#### Integrate

$$\int_{h_0}^{h} \frac{dh}{\sqrt{h}} = \int_{0}^{t} -\frac{A_2}{A_1} \sqrt{2g} dt$$
$$h = \left(h_0^{\frac{1}{2}} - \frac{A_2}{A_1} \frac{\sqrt{2g}}{2}t\right)^2$$

$$\left\{\int_{h_0}^{h} h^{-\frac{1}{2}} dh = \left[\frac{1}{2h^2}\right]_{h_0}^{h}\right\}$$





$$h = \left(\sqrt{16} - \frac{0.1}{20} \frac{\sqrt{2(32.2)}}{2}t\right)^2$$
$$= \left(4 - 0.0201t\right)^2$$

At 
$$t = 50 \sec$$
,  $h = (4 - 0.0201 \times 50)^2 = 8.98 ft$ 

$$V_2 = \sqrt{2gh} = \sqrt{2(32.2)(8.98)} = 24.05$$
 fps

$$Q_2 = (VA)_2 = 24.05(0.1) = 2.405$$
 cfs





ii) Momentum equation, Eq. (7.4)  

$$\vec{F}_{p} + \vec{F}_{s} + \vec{F}_{b} = \oint_{CS} \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} \vec{q} \rho \, dV$$

II = Time rate of <u>change of momentum inside CV is negligible</u> if <u>tank area</u>  $(A_1)$  is large compared to the nozzle area  $(A_2)$ .

$$I = \oint_{CS} \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right) = \int q_n \rho q_n dA_2 - \int q_n \rho q_n dA_1 = V_2 \rho V_2 A_2$$
  
$$\therefore F_{px} = V_2 \rho V_2 A_2 = V_2 \rho Q_2$$
  
$$F_{px} = (24.05)(1.94)(2.405) = 112 \text{ lb}$$





7.2.1 The Moment of momentum principle for inertial reference systems

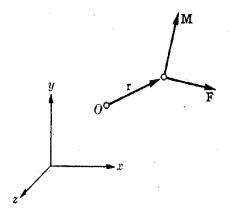
Apply Newton's 2nd law to rotating fluid masses

→ The vector sum of all the external moments acting on a fluid mass  $(\vec{r} \times \vec{F})$ equals the time rate of change of the moment of momentum (angular momentum) vector  $(\vec{r} \times \vec{M})$  of the fluid mass.

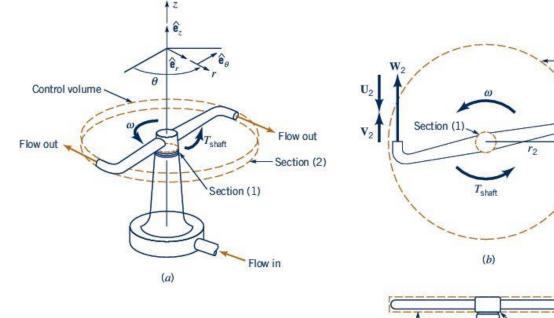
Example: rotary lawn sprinklers, ceiling

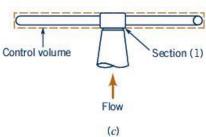
fans, wind turbines











Control volume

Section (2)

V<sub>02</sub>

W2

 $U_2 = r_2 \omega$ 





$$T = \vec{r} \times \vec{F} = \frac{d}{dt} (\vec{r} \times \vec{M})$$

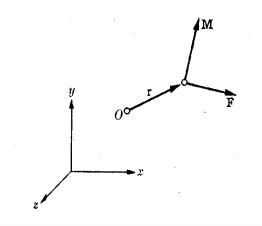
(7.10)

#### where

T = torque

 $\vec{r}$  = position vector of a mass in an arbitrary curvilinear motion

 $\vec{M}$  = linear momentum







[Re] Derivation of (7.10)

Eq. (7.1): 
$$\vec{F} = \frac{d\vec{M}}{dt}$$

Take the vector cross product of  $\vec{r}$ 

$$\vec{r} \times \vec{F} = \vec{r} \times \frac{d\vec{M}}{dt}$$

By the way,

$$\frac{d}{dt} \left( \vec{r} \times \vec{M} \right) = \frac{d\vec{r}}{\frac{dt}{dt}} \times \vec{M} + \vec{r} \times \frac{d\vec{M}}{dt}$$





$$I = \frac{d\vec{r}}{dt} \times \vec{M} = \vec{q} \times m \vec{q} = 0 \quad \left( \because \frac{d\vec{r}}{dt} = \vec{q} \right)$$
$$\left( \therefore \vec{q} \times \vec{q} = |\vec{q}| |\vec{q}| \sin 0^\circ = 0 \right)$$
$$\therefore \quad \left( \vec{r} \times \frac{d\vec{M}}{dt} \right) = \frac{d}{dt} (\vec{r} \times \vec{M})$$
$$\therefore \therefore \quad \vec{r} \times \vec{F} = \frac{d}{dt} (\vec{r} \times \vec{M})$$

where  $\vec{r} \times \vec{M}$  = angular momentum (moment of momentum)





[Re] Torque  $\vec{T} = \vec{r} \times \vec{F}$ 

• translational motion  $\rightarrow$ 

Force - linear acceleration

• rotational motion  $\rightarrow$ 

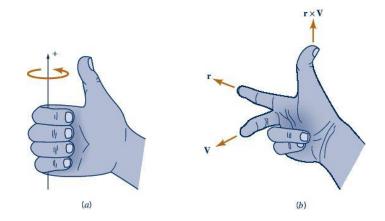
Torque - angular acceleration [Re] Vector Product

$$\vec{V} = \vec{a} \times \vec{b}$$

Magnitude =  $|\vec{v}| = |\vec{a}| \times |\vec{b}| \sin \gamma$  = area of parallelogram

direction = perpendicular to plane of  $\vec{a}$  and  $\vec{b} \rightarrow$  right-handed triple







$$\vec{b} \times \vec{a} = -(\vec{a} \times \vec{b})$$
$$(k\vec{a}) \times \vec{b} = k(\vec{a} \times \vec{b})$$
$$\vec{a} \times (\vec{b} + \vec{c}) = (\vec{a} \times \vec{b}) + (\vec{a} \times \vec{b})$$

• External moments arise from external forces

$$\begin{pmatrix} \vec{r} \times \vec{F}_p \end{pmatrix} + \begin{pmatrix} \vec{r} \times \vec{F}_s \end{pmatrix} + \begin{pmatrix} \vec{r} \times \vec{F}_b \end{pmatrix} = \frac{d}{dt} (\vec{r} \times \vec{M})$$

$$\vec{T}_p \qquad \vec{T}_s \qquad \vec{T}_b \qquad \vec{T$$

(7.11)

where  $\vec{T}_{p}$ ,  $\vec{T}_{s}$ ,  $\vec{T}_{b}$  = external torque



7.2.2 The general moment of momentum equation

(7.3): 
$$\frac{dM}{dt} = \oint_{CS} \vec{q} \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} \vec{q} \rho \, dV$$

$$\therefore \quad \frac{d}{dt} \left( \vec{r} \times \vec{M} \right) = \oint_{CS} \left( \vec{r} \times \vec{q} \right) \rho \left( \vec{q} \cdot d\vec{A} \right) + \frac{\partial}{\partial t} \int_{CV} \left( \vec{r} \times \vec{q} \right) \rho \, dV$$

$$\vec{T}_{p} + \vec{T}_{s} + \vec{T}_{b} = \oint_{CS} (\vec{r} \times \vec{q}) \rho (\vec{q} \cdot d\vec{A}) + \frac{\partial}{\partial t} \int_{CV} (\vec{r} \times \vec{q}) \rho \, dV \quad (7.12)$$

$$x - dir.: \quad \left| \left( \vec{r} \times \vec{q} \right)_{yz} \right| = r_{yz} q_{yz} \sin\left(\frac{\pi}{2} - \alpha_{yz}\right) = (rq \cos \alpha)$$
  
angle between  $q_{yz}$  and  $r_{yz}$ 





$$x - dir.: \quad \vec{T}_{px} + \vec{T}_{sx} + \vec{T}_{bx} = \oint_{CS} (rq \cos \alpha)_{yz} \rho (\vec{q} \cdot d\vec{A}) + \frac{\partial}{\partial t} \int_{CV} (rq \cos \alpha)_{yz} \rho dV$$

$$y - dir.: \quad \vec{T}_{py} + \vec{T}_{sy} + \vec{T}_{by} = \oint_{CS} (rq \cos \alpha)_{zx} \rho (\vec{q} \cdot d\vec{A}) + \frac{\partial}{\partial t} \int_{CV} (rq \cos \alpha)_{zx} \rho dV$$

$$z - dir.: \quad \vec{T}_{pz} + \vec{T}_{sz} + \vec{T}_{bz} = \oint_{CS} (rq \cos \alpha)_{xy} \rho(\vec{q} \cdot d\vec{A}) + \frac{\partial}{\partial t} \int_{CV} (rq \cos \alpha)_{xy} \rho dV$$

$$(7.13)$$



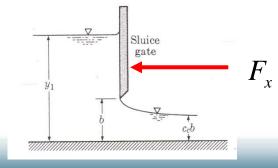


Homework Assignment # 3

Due: 1 week from today

7-1. Derive the <u>equation for the volume rate</u> of flow per unit width for the sluice gate shown in Fig. 4-20 in terms of the geometric variable *b*,  $y_{1}$ , and  $C_{C}$ . Assume the pressure in hydrostatic at  $y_1$  and  $c_c b$  and the velocity is constant over the depth at each of these sections.

7-2. Derive the expression for the total force per unit width exerted by the sluice gate on the fluid in terms of vertical distances shown in Fig. 4-20.

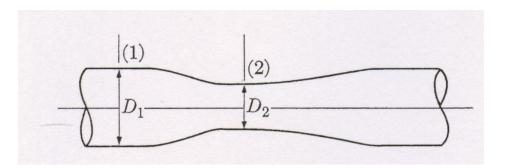






#### **Problems**

7-3. Consider the flow of an incompressible fluid through the <u>Venturi</u> <u>meter</u> shown in Fig. 4-22. Assuming uniform flow at sections (1) and (2) neglecting all losses, find the <u>pressure difference</u> between these sections as a function of the flow rate *Q*, the diameters of the sections, and the density of the fluid,  $\rho$ . Note that for a given configuration, *Q* is a function of only the pressure drop and fluid density.



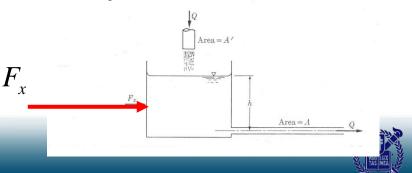




#### **Problems**

7-4. Water flows into a tank from a supply line and out of the tank through a horizontal pipe as shown in Fig. 4-23. The rates of inflow and outflow are the same, and the water surface in the tank remains a distance h above the discharge pipe centerline. All velocities in the tank are negligible compared to those in the pipe. The head loss between the tank and the pipe exit is  $H_{i}$  (a) Find the <u>discharge</u> Q in terms of h, A, and  $H_{L}$  (b) What is the <u>horizontal force</u>,  $F_{\chi}$  required to keep the tank from moving? (c) If the supply line has an area A', what is the vertical force exerted on the water in the tank by the vertical jet?





7-5. Derive the <u>one-dimensional continuity equation</u> for the <u>unsteady</u>, <u>non-uniform flow</u> of an <u>incompressible liquid in a horizontal open channel</u> as shown in Fig. 4-29. The channel has a rectangular cross section of a constant width, *b*. Both the depth,  $y_0$  and the mean velocity, *V* are functions of *x* and *t*.

