

Boundary layer approximation

$$\left(\frac{\partial}{\partial x} \ll \frac{\partial}{\partial y}, v \ll u \right)$$

$$\begin{cases} \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \\ u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = U \frac{dU}{dx} + \nu \frac{\partial^2 u}{\partial y^2} \end{cases} \quad \text{flat-plate}$$

Boundary layer eq.

b.c. $u=v=0$ @ $y=0$
 $u=U$ @ $y=\delta$

① Flat-plate bdry layer : $U = \text{const} \rightarrow \frac{dP}{dx} = 0, \frac{dU}{dx} = 0$

Blasius : $\frac{u}{U} = f(\eta), \quad \eta = y \sqrt{\frac{U}{\nu x}}$

$$\frac{\partial v}{\partial x} = \frac{\partial u}{\partial y} \frac{\partial y}{\partial x} = \nu f'' \cdot y \sqrt{\frac{U}{\nu}} \left(-\frac{1}{x}\right) x^{-\frac{3}{2}}$$

$$\frac{\partial u}{\partial y} = \frac{\partial u}{\partial \eta} \frac{\partial \eta}{\partial y} = \nu f'' \cdot \sqrt{\frac{U}{\nu x}} \ll$$

$$\frac{\partial^2 u}{\partial y^2} = \frac{\partial}{\partial y} \left(\nu f'' \sqrt{\frac{U}{\nu x}} \right) = \frac{\partial}{\partial \eta} \left(\nu f'' \sqrt{\frac{U}{\nu x}} \right) \frac{\partial \eta}{\partial y} = f''' \nu \sqrt{\frac{U}{\nu x}} \cdot \sqrt{\frac{U}{\nu x}}$$

$$\frac{\partial v}{\partial y} = -\frac{\partial u}{\partial x} \rightarrow v = -\frac{\partial}{\partial x} \int u dy = -\frac{\partial}{\partial x} \int f' U \sqrt{\frac{\nu x}{U}} dz \quad z = y \sqrt{\frac{U}{\nu x}}$$

$$= \dots = -\frac{1}{2} \sqrt{\frac{U \nu}{x}} f + \frac{1}{2} U x^{-1} f' y \quad \frac{u}{U} = f(z)$$

$$\Rightarrow \boxed{f''' + \frac{1}{2} f f'' = 0} \quad \text{Blasius eq. (exact eq.)}$$

$$\textcircled{1} y=0, u=v=0 \rightarrow \textcircled{1} z=0, f'(0)=0, f(0)=0$$

$$\textcircled{2} y \rightarrow \infty, u=U \rightarrow \textcircled{2} z \rightarrow \infty, f'(\infty)=1$$

Numerical method \rightarrow obtain f, f', f'', \dots

$$z = y \sqrt{\frac{U}{\nu x}} \rightarrow \delta = \frac{5.0}{\sqrt{U/\nu x}}$$

$$\delta = 5.0 \rightarrow \delta = \frac{5.0}{\sqrt{U/\nu x}} \rightarrow \delta = \frac{5.0}{\sqrt{Re_x}}$$

$z = y \sqrt{\frac{U}{\nu x}}$

$y[U/(\nu x)]^{1/2}$	u/U	$y[U/(\nu x)]^{1/2}$	u/U
0.0	0.0	2.8	0.81152
0.2	0.06641	3.0	0.84605
0.4	0.13277	3.2	0.87609
0.6	0.19894	3.4	0.90177
0.8	0.26471	3.6	0.92333
1.0	0.32979	3.8	0.94112
1.2	0.39378	4.0	0.95552
1.4	0.45627	4.2	0.96696
1.6	0.51676	4.4	0.97587
1.8	0.57477	4.6	0.98269
2.0	0.62977	4.8	0.98779
2.2	0.68132	5.0	0.99155
2.4	0.72899	∞	1.00000
2.6	0.77246		

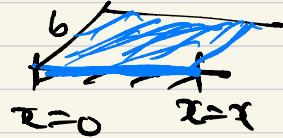
$$\text{Drag } D(x) = \int_0^x \tau_w dx \cdot b = 0.664 \rho^{1/2} \mu^{1/2} U^{3/2} x^{1/2}$$

$$\tau_w = \mu \frac{\partial u}{\partial y} \Big|_{y=0}$$

$$\frac{\partial u}{\partial y} \Big|_{y=0} = f''(\eta) U \sqrt{\frac{U}{\nu x}}$$

numerical sol.

$$\tau_w = \frac{0.332 \sqrt{\rho \mu} U^2}{\sqrt{x}}$$



Drag coefficient

$$C_D = \frac{D(L)}{\frac{1}{2} \rho U^2 b \cdot L} = \frac{1.328}{\sqrt{Re_L}}$$

$$C_f = \frac{\tau_w}{\frac{1}{2} \rho U^2} = \frac{0.664}{\sqrt{Re_x}}$$

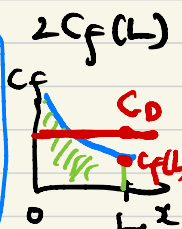
$$C_f \sim x^{-1/2}$$

$$\delta^* = \int_0^\delta \left(1 - \frac{u}{U}\right) dy$$

$$\int_0^\delta u dy = \int_0^{\eta(\delta)} f' U \sqrt{\frac{\nu x}{U}} d\eta$$

$$\frac{\delta^*}{x} = \frac{\delta}{x} - \frac{f(\xi(0)) \sqrt{\frac{\nu x}{U}}}{x}$$

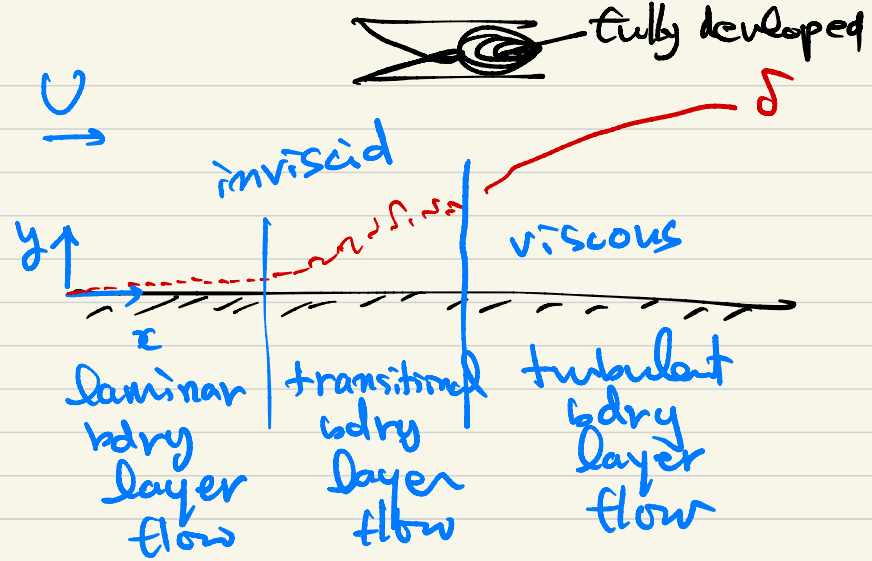
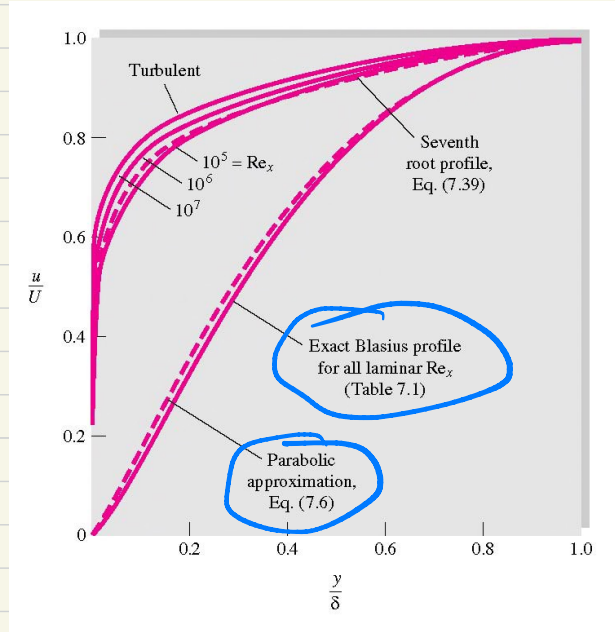
$$= \left[\underbrace{f(\eta(\delta))}_{\xi(0)} - f(0) \right] U \sqrt{\frac{\nu x}{U}}$$



$$= \frac{f(0)}{\sqrt{Re_x}} - \frac{f(0)}{\sqrt{Re_x}} = \frac{1.721}{\sqrt{Re_x}}$$

$$\theta = \int_0^\delta \frac{u}{U} \left(1 - \frac{u}{U}\right) dy \rightarrow \frac{\theta}{x} = \frac{0.664}{\sqrt{Re_x}}$$

$$H = \frac{\delta^*}{\theta} = \frac{1.721}{0.664} = 2.59$$

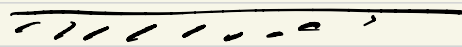


$$Re_x = \frac{Ux}{\nu}$$

$Re_{x,trans} \doteq 3 \times 10^6$ for smooth plate
 5×10^5 for rough plate

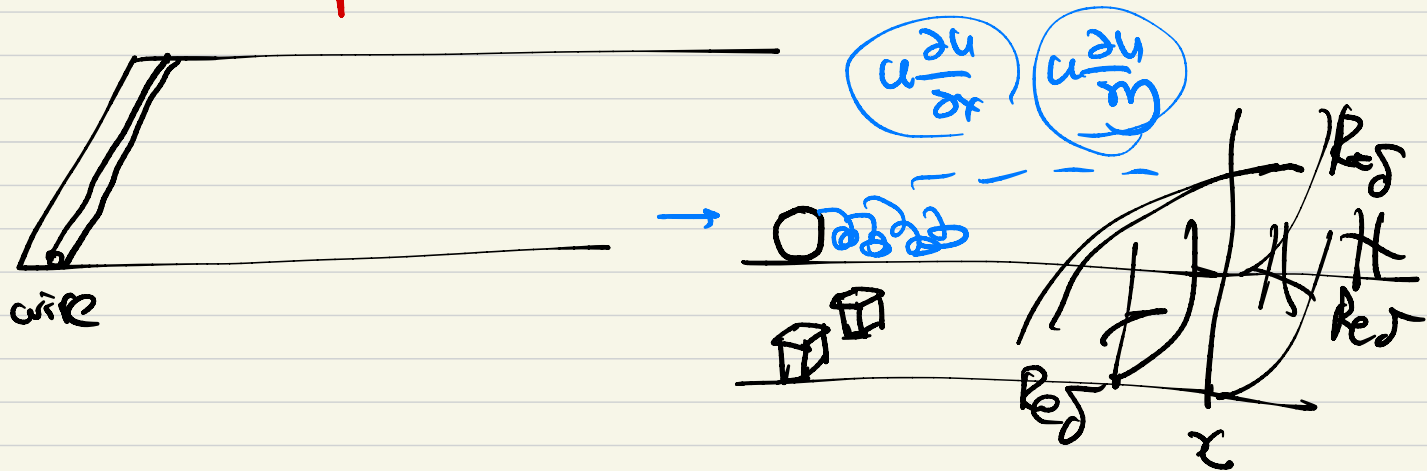
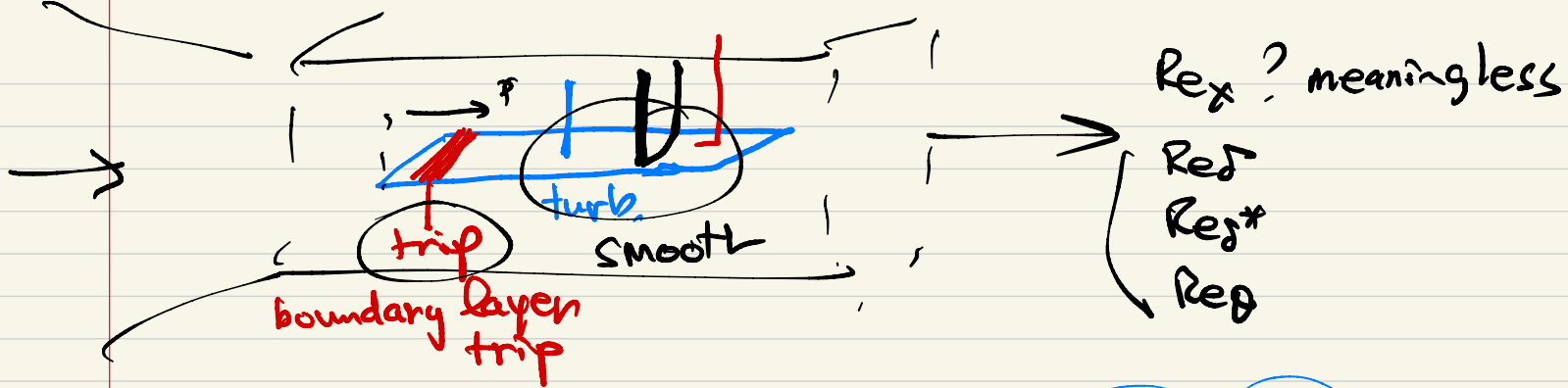
10 m/s
 \rightarrow

air



$$\frac{Ux_{trans}}{\nu} = 2 \times 10^6 \rightarrow x_{trans} = 4.5 \text{ m}$$

$$5 \times 10^5 \rightarrow x_{trans} = 0.75 \text{ m}$$



• Reynolds time averaging
velocity decomposition

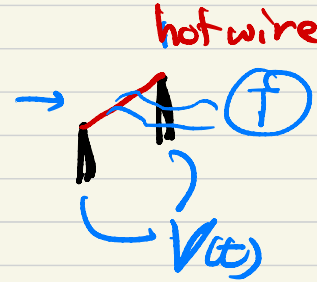
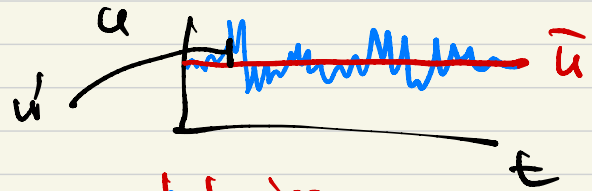
$$u(t) = \bar{u} + u'(t)$$

instantaneous velocity
순간속도

mean velocity
평균속도

fluctuating velocity
속도상동

turbulent flow



$$\overline{u} = \overline{\bar{u} + u'} \Rightarrow \boxed{\overline{u'} = 0}$$

Navier-Stokes eq.

$$\rho \frac{\partial u}{\partial t} + \rho \frac{\partial}{\partial x}(uu) + \rho \frac{\partial}{\partial y}(uv) + \rho \frac{\partial}{\partial z}(uw) = -\frac{\partial p}{\partial x} + \mu \nabla^2 u$$

$$\begin{aligned} \circ \frac{\partial}{\partial x}(uu) &= \frac{\partial}{\partial x}(\overline{uu}) = \frac{\partial}{\partial x}(\overline{(u+u')(u+u')}) = \frac{\partial}{\partial x}(\overline{u\bar{u} + u\bar{u}' + u'u + u'u'}) \\ &= \frac{\partial}{\partial x}(\bar{u}\bar{u} + \overline{u'u'}) \end{aligned}$$

$\circ \left(\begin{matrix} \bar{u}' = \bar{u} \\ \bar{u} = \bar{u}' \\ \bar{u} = \bar{u}' \end{matrix} \right)$

$$\overline{\frac{\partial}{\partial y}(uv)} = \frac{\partial}{\partial y}(\overline{uv} + \overline{u'v'})$$

$$\overline{\frac{\partial}{\partial z}(uw)} = \frac{\partial}{\partial z}(\overline{uw} + \overline{u'w'})$$

Reynolds-averaged
Navier-Stokes eq.

$$\Rightarrow \rho \frac{\partial}{\partial x}(\overline{u\bar{u}}) + \rho \frac{\partial}{\partial y}(\overline{u\bar{v}}) + \rho \frac{\partial}{\partial z}(\overline{u\bar{w}}) = -\frac{\partial \bar{p}}{\partial x}$$

$$+ \frac{\partial}{\partial x} \left(\mu \frac{\partial \bar{u}}{\partial x} - \rho \overline{u'u'} \right) + \frac{\partial}{\partial y} \left(\mu \frac{\partial \bar{u}}{\partial y} - \rho \overline{u'v'} \right) + \frac{\partial}{\partial z} \left(\mu \frac{\partial \bar{u}}{\partial z} - \rho \overline{u'w'} \right)$$

viscous stress

turbulent stress
or Reynolds stress

$f(\bar{u}, \bar{v}, \bar{w})$

$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} + \frac{\partial \bar{w}}{\partial z} = 0$$

turbulence modeling

closure problem

$$\overline{u'v'} \xrightarrow{\text{g.e.}} \frac{\partial}{\partial t}(\overline{u'v'}) + \dots = \frac{\partial}{\partial x}(\overline{u'u'u'}) + \dots$$