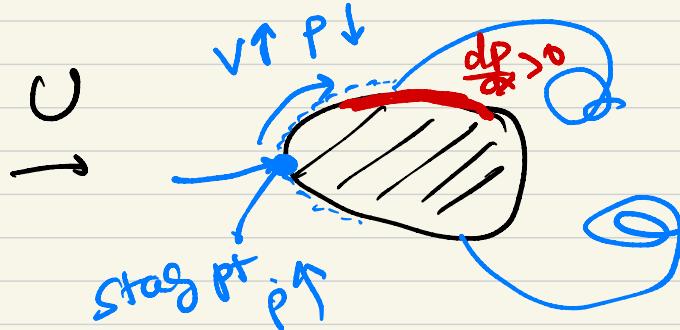


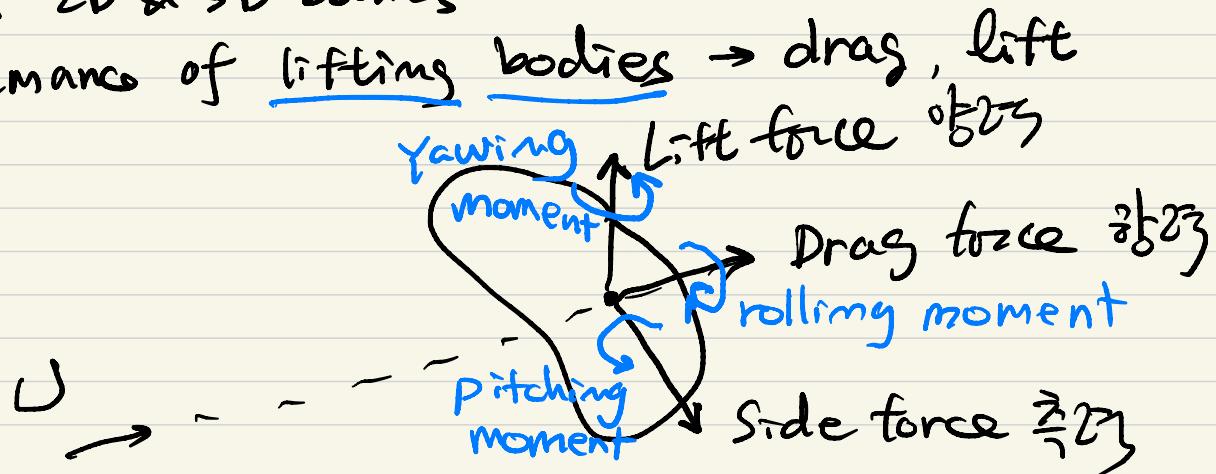
7.6 Experimental external flows



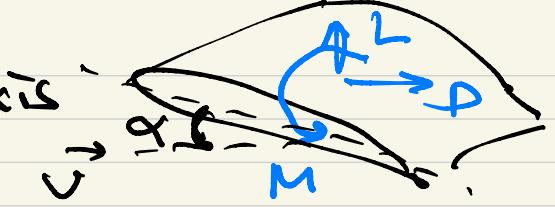
sep. occurs
→ bdry layer approx X
theory fails.
↓
No theory
↓
experiment or CFD

- Drag of 2D & 3D bodies

Performance of lifting bodies → drag, lift



Symmetry about the lift-drag axis
 → side force, yawing & rolling moment = 0 α : angle of attack
 drag, lift, pitching moment $\neq 0$



Two planes of symmetry

→ drag $\neq 0$ $L(t) \neq 0$, $\boxed{I = 0}$.



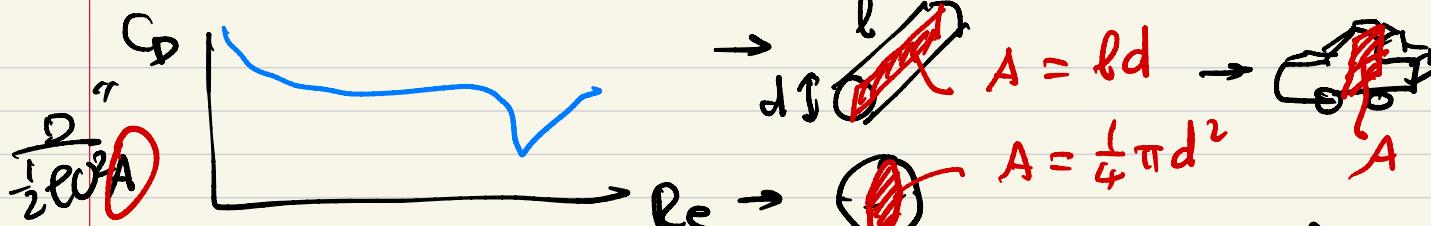
- Similarity (geometric, kinematic, dynamic)
 $[U]$ $[L, T]$ $[L, T, M]$

$$\stackrel{\triangle}{=} \frac{U}{d} \cdot \frac{f}{\rho} \cdot d \quad \text{Drag} = f(U, d, \rho, \mu)$$

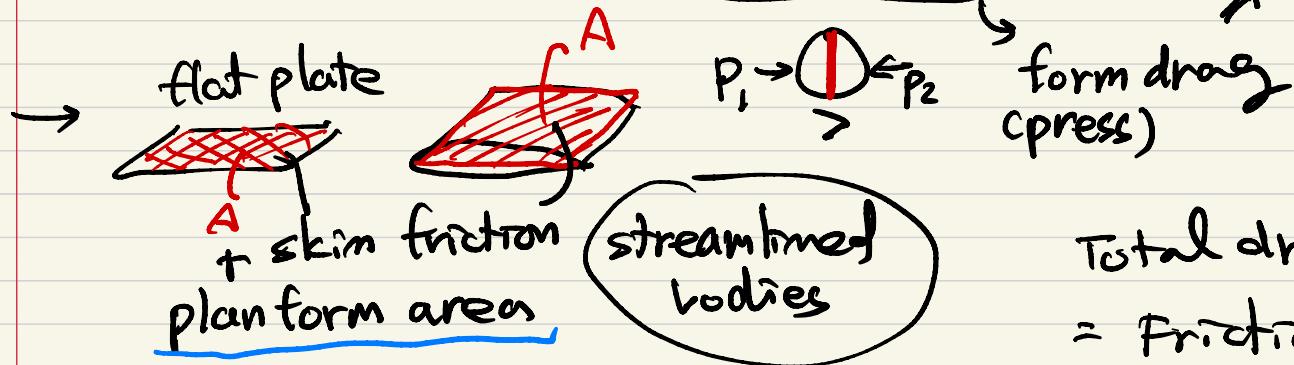
dimensional analysis

$$\rightarrow C_D = f(Re) \quad Re = \frac{Ud}{\nu}$$

" $\frac{D}{\frac{1}{2} \rho U^2 A}$ " : drag coeff.

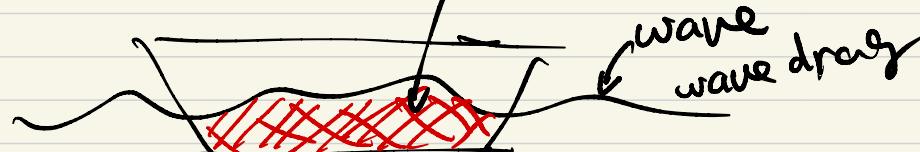


bluff bodies: $A \leftarrow \underline{\text{frontal area}}$

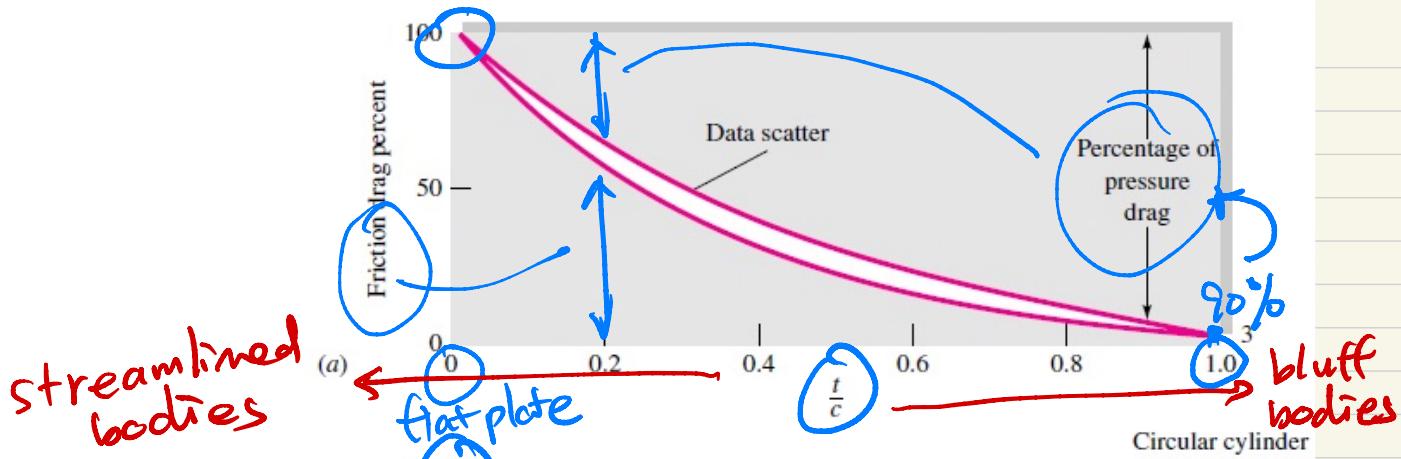


$$\frac{D_{\text{total}}}{2 \rho U^2 A}$$

Surface ship \leftarrow Skin friction



$$\begin{aligned} \text{Total drag} &\leftarrow \text{Friction drag} \\ &+ \text{Form drag} \\ &+ (\text{wave drag}) \end{aligned}$$



streamlined bodies

$$\frac{D}{\frac{1}{2} \rho U^2 A} = C_D$$

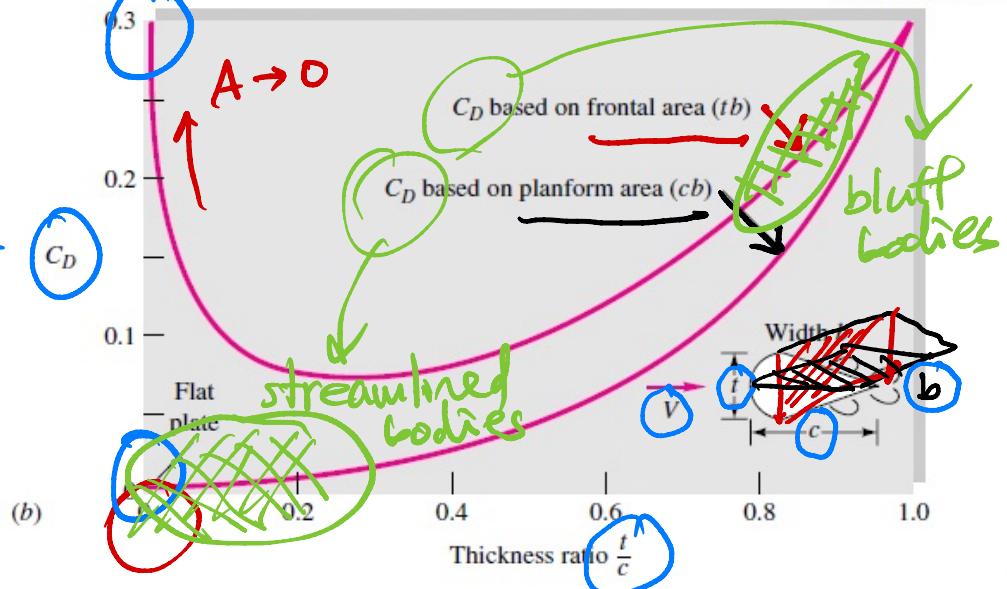
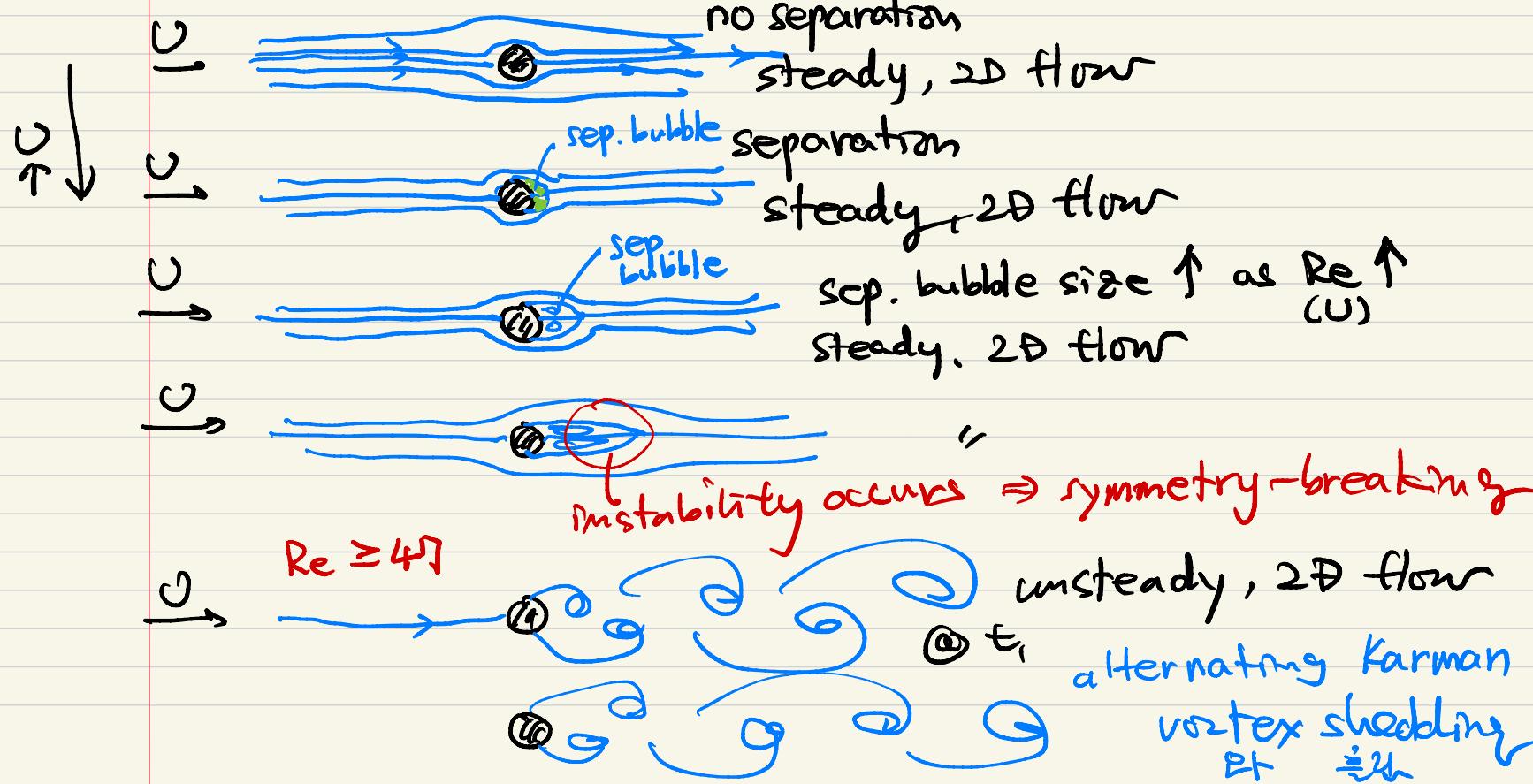


Fig. 7.12 Drag of a streamlined two-dimensional cylinder at $Re_c = 10^6$: (a) effect of thickness ratio on percentage of friction drag; (b) total drag versus thickness when based on two different areas.

- Flow past a circular cylinder  
- ↳ a representative 2D bluff body

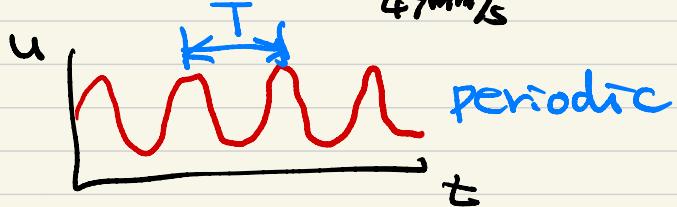


$$Re = 4\pi = \frac{Ud}{D} = \frac{U \times 1.5 \times 10^{-2}}{1.5 \times 10^{-5}} \rightarrow U = 4\pi \text{ mm/s}$$

~~4π~~ 1.5cm
47mm/s



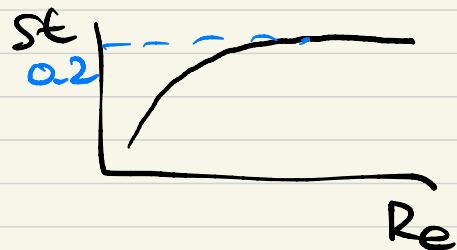
② $Re = 100$



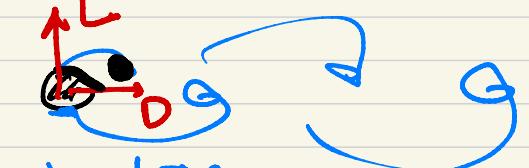
unsteady 2-D laminar flow

$$St = \frac{fd}{U} = \frac{d}{UT} \quad (f = \frac{1}{T})$$

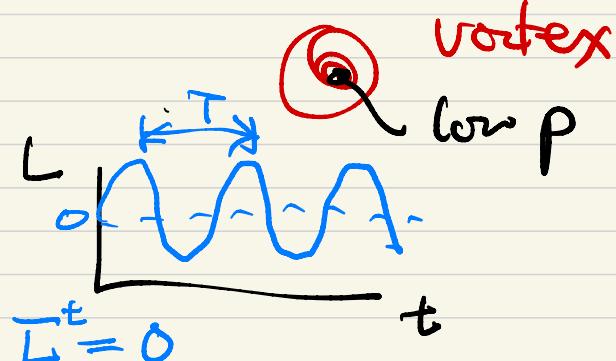
strouhal number (스트로날 수)

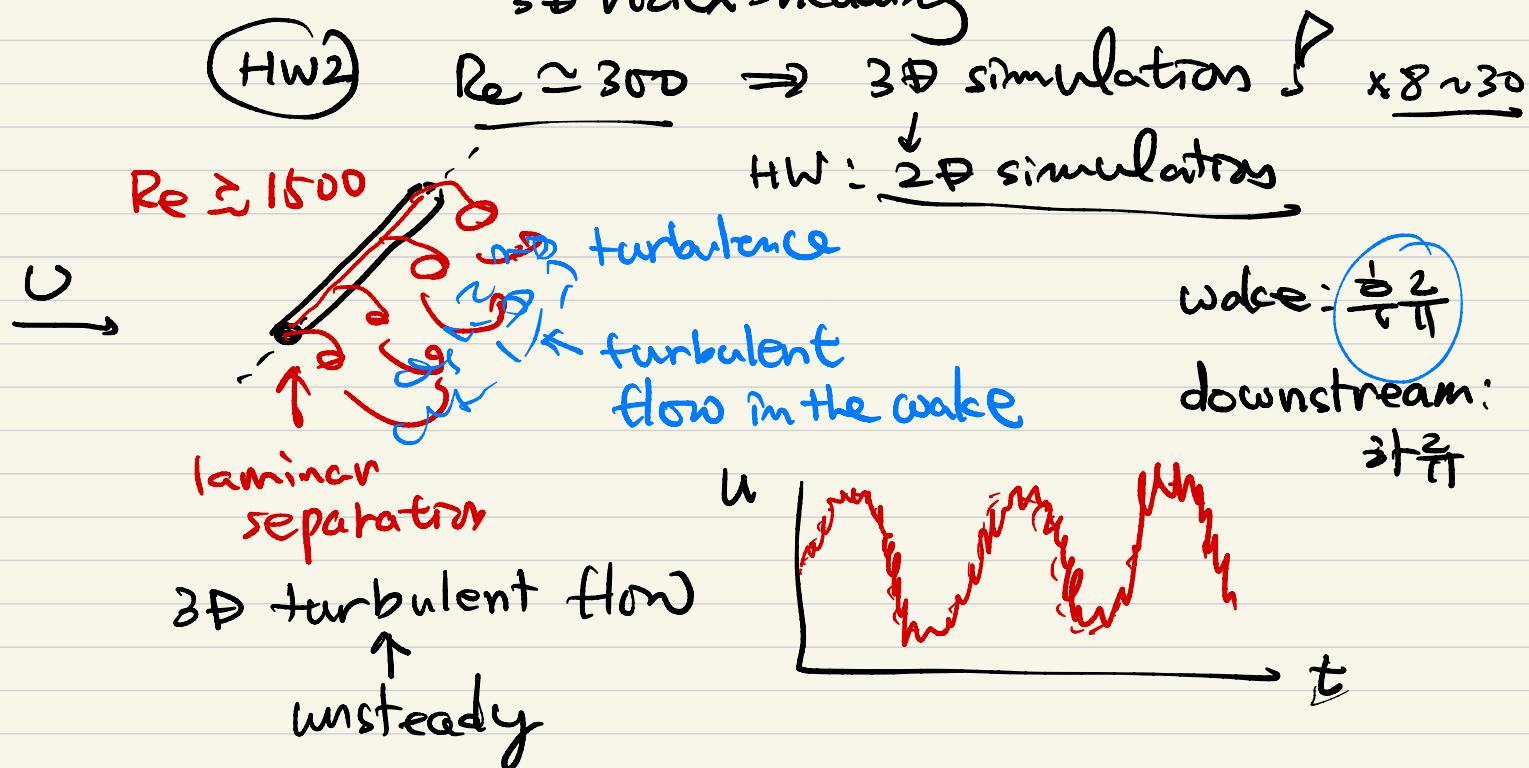
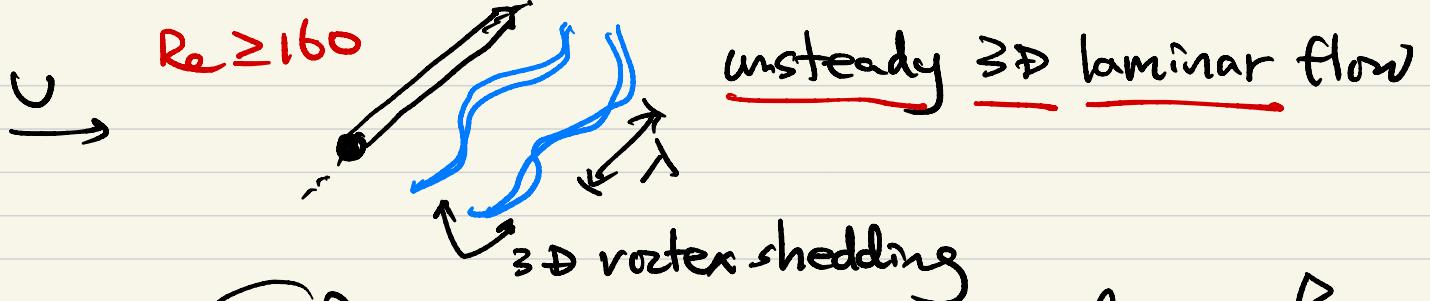


③ t_1



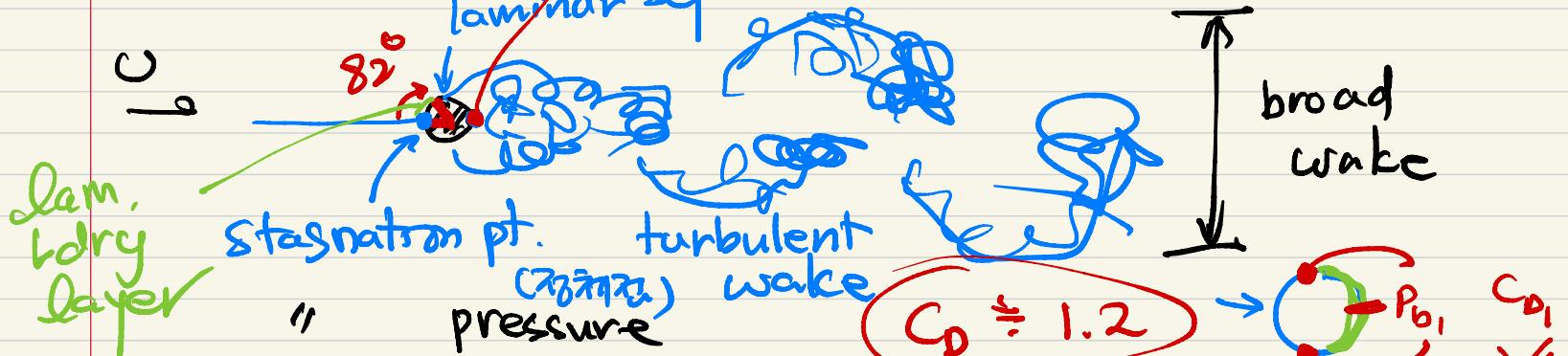
④ t_2



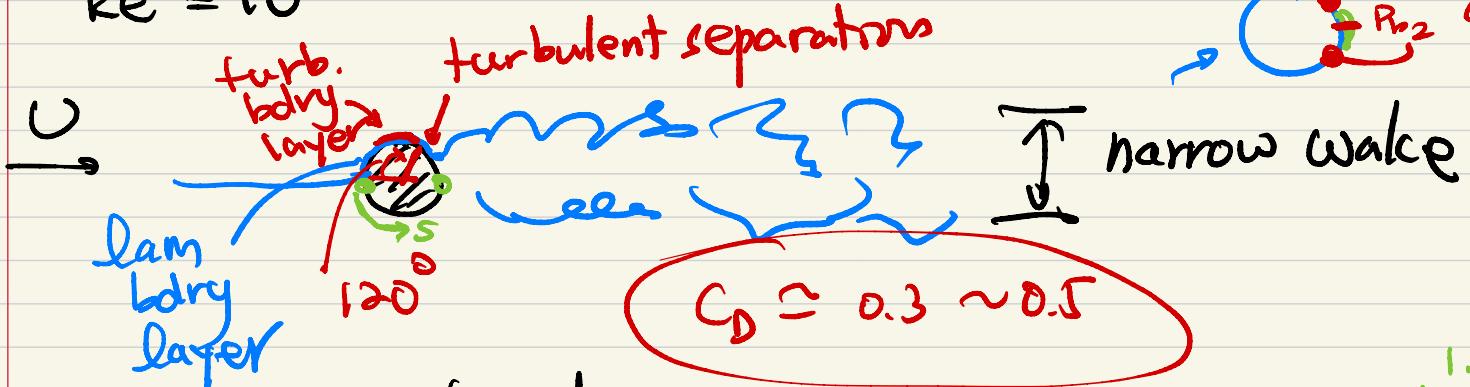


$$Re = 10^4 \sim 10^5$$

base point, base pressure ↓



$$Re = 10^6$$



$$Re = 10^6 = \frac{Ud}{\nu}$$

$$Re_s = \frac{Us}{\nu} = \frac{U \cdot \pi d / 2}{\nu} = \frac{Ud}{\nu} \cdot \frac{\pi}{2} \approx 1.5 \times 10^6$$

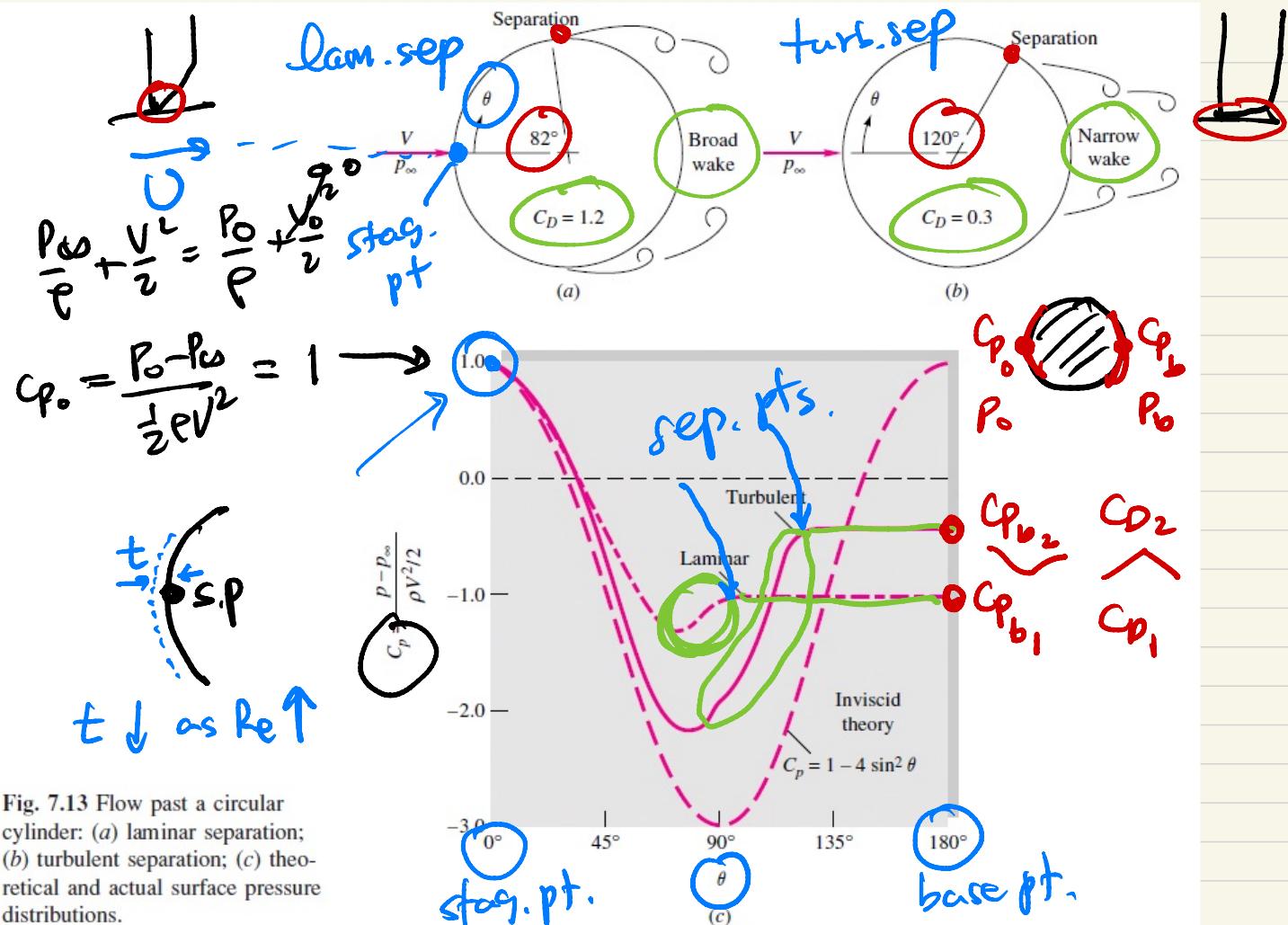


Fig. 7.13 Flow past a circular cylinder: (a) laminar separation; (b) turbulent separation; (c) theoretical and actual surface pressure distributions.