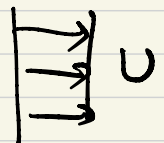


## 8.2 Elementary plane flow solutions

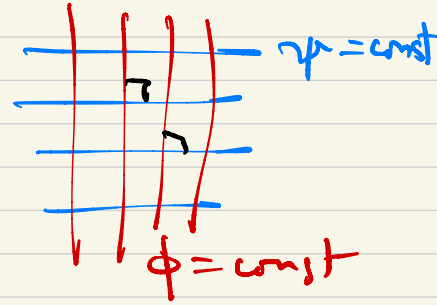
- Uniform stream:  $u=U, v=0$



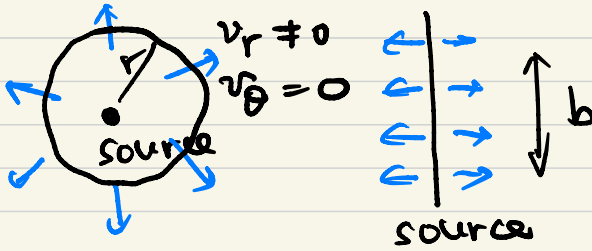
$$u=U = \frac{\partial \phi}{\partial x} = \frac{\partial \psi}{\partial y}$$

$$v=0 = \frac{\partial \phi}{\partial y} = -\frac{\partial \psi}{\partial x}$$

$$\left. \begin{array}{l} \phi = Ux \\ \psi = Uy \end{array} \right\}$$



- Line source / sink @ the origin



$$Q = v_r 2\pi r b$$

$$v_r = \frac{Q}{2\pi r b} = \frac{m}{r} = \frac{1}{r} \frac{\partial \psi}{\partial \theta} = \frac{\partial \phi}{\partial r}$$

$$m = Q/2\pi b : \text{source strength}$$

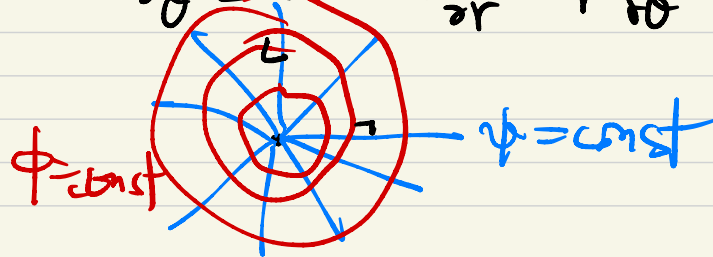
$$m > 0 \text{ source}$$

$$m < 0 \text{ sink}$$

$$\rightarrow \psi = m\theta$$

$$\phi = m \ln r$$

$$v_\theta = 0 = -\frac{\partial \psi}{\partial r} = \frac{1}{r} \frac{\partial \phi}{\partial \theta}$$



• Line irrotational vortex (free vortex)

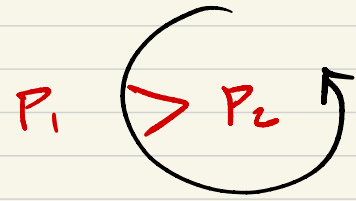
$\omega = 0$   
 $v_r = 0$   
 $v_\theta \neq 0$

$\rightarrow v_\theta = f(r)$   
 $\nabla \times \underline{v} = 0 \rightarrow \omega_z = \frac{1}{r} \frac{\partial}{\partial r} (r v_\theta) - \frac{1}{r} \frac{\partial v_r}{\partial \theta} = 0$

$r v_\theta = \text{const} = K$

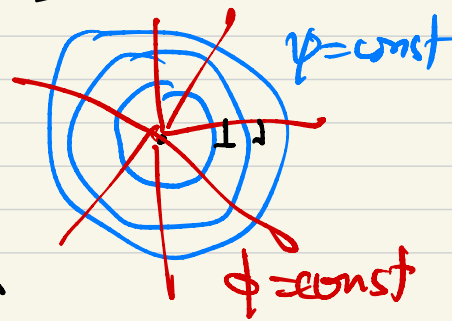
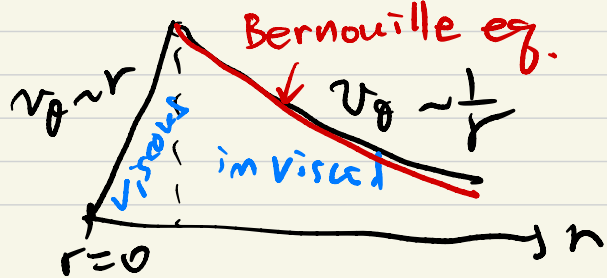
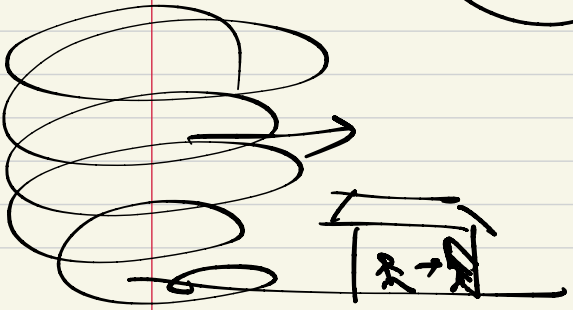
$v_\theta = \frac{K}{r}$      $K$ : vortex strength

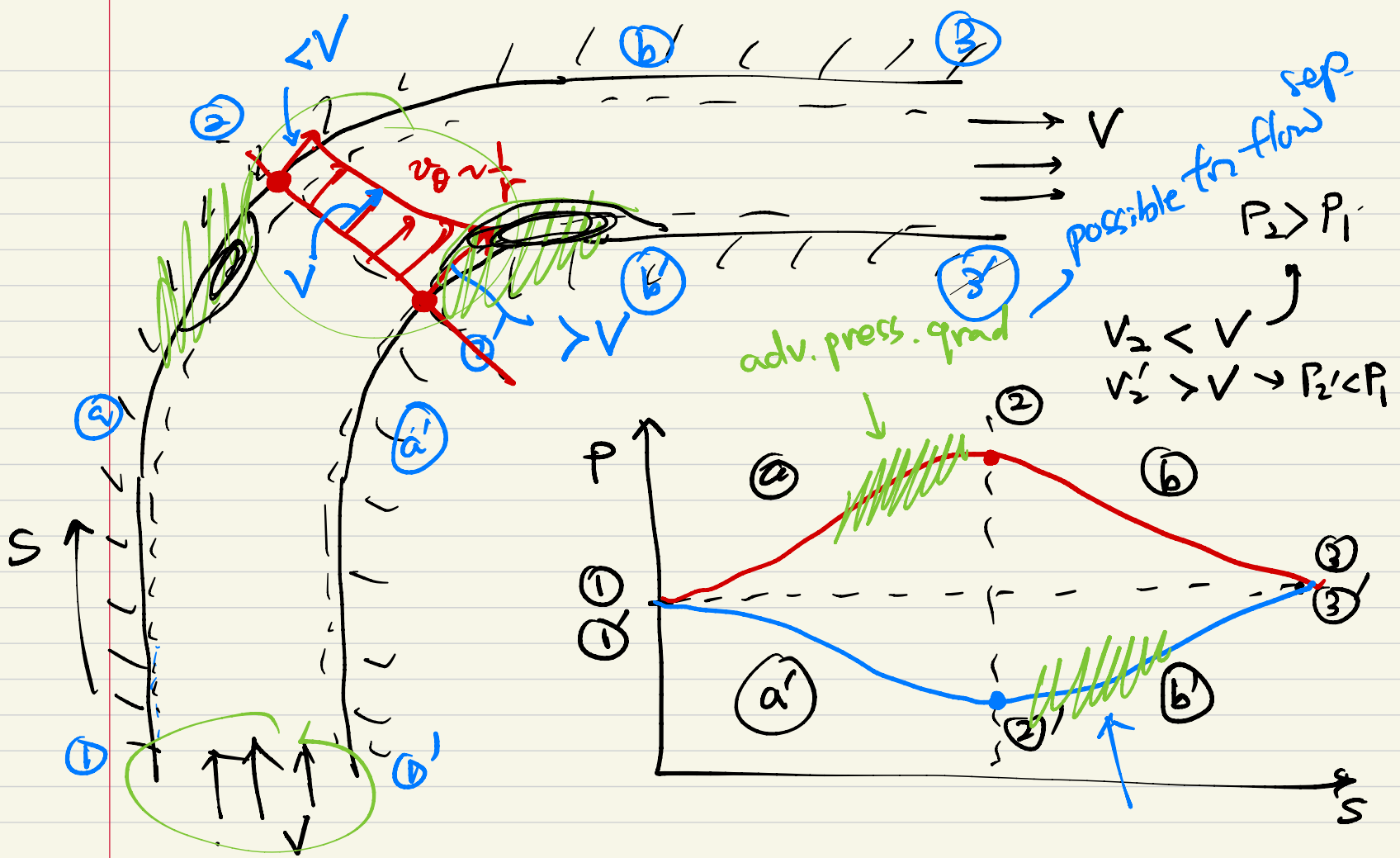
As  $r \rightarrow 0$ ,  $v_\theta \rightarrow \infty \Rightarrow$  pressure @ vortex center is lowest

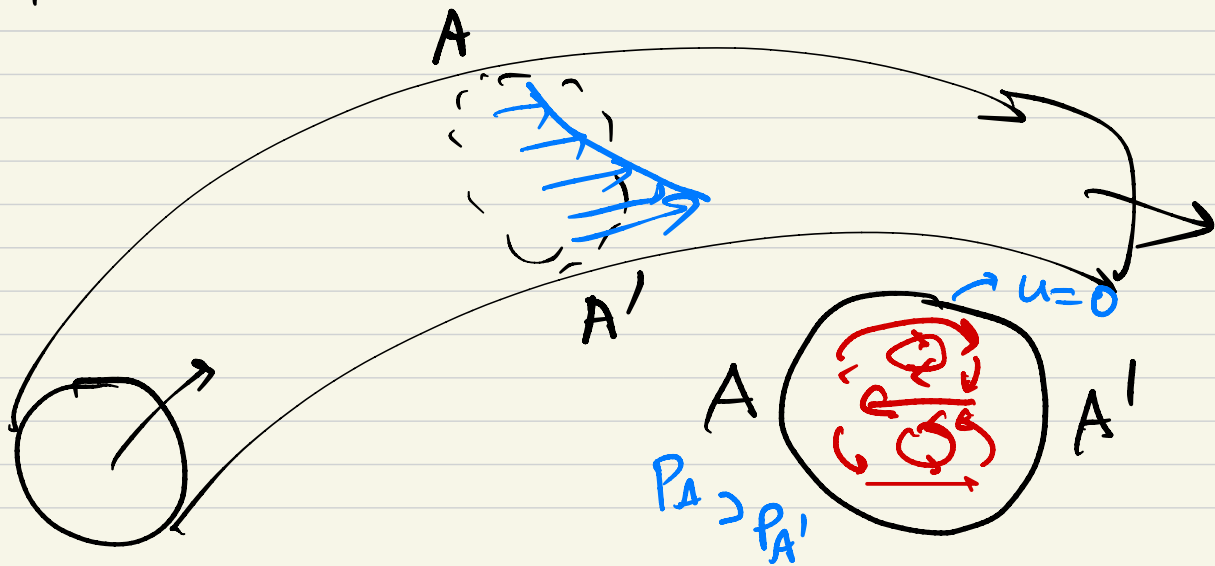
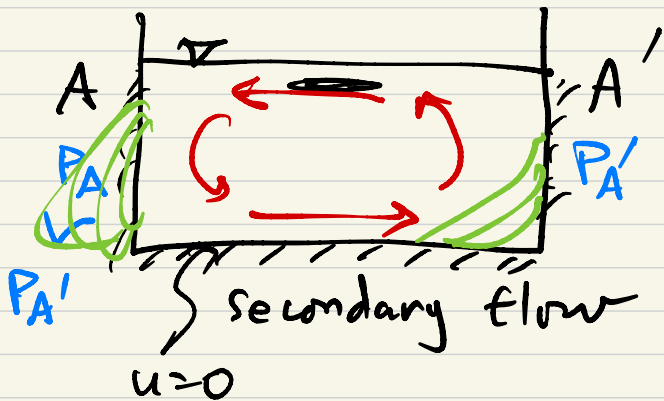
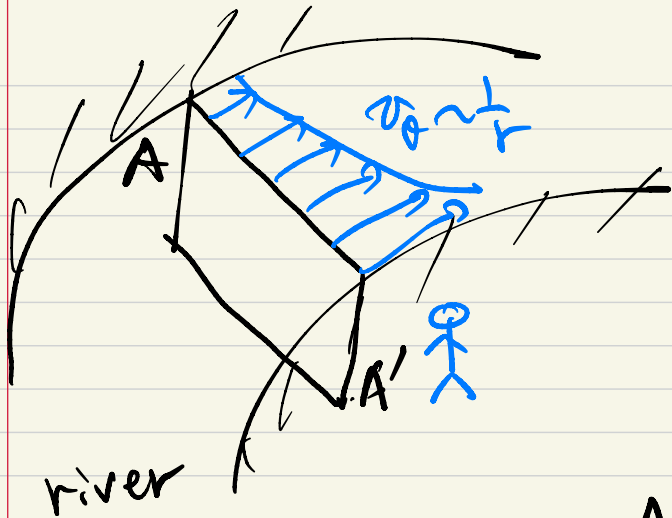


$v_r = \frac{1}{r} \frac{\partial \phi}{\partial \theta} = \frac{\partial \phi}{\partial r} = 0$   
 $v_\theta = -\frac{\partial \phi}{\partial r} = \frac{1}{r} \frac{\partial \phi}{\partial \theta} = \frac{K}{r}$

$\left. \begin{array}{l} \phi = -k \ln r \\ \phi = k \theta \end{array} \right\}$







r-momentum eq      irrotational vortex ( $v_r=0, v_\theta = \frac{c}{r}$ )

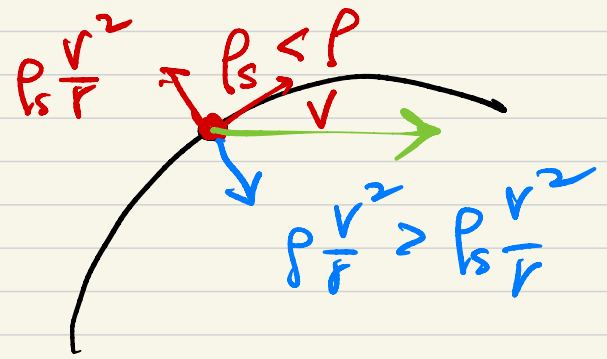
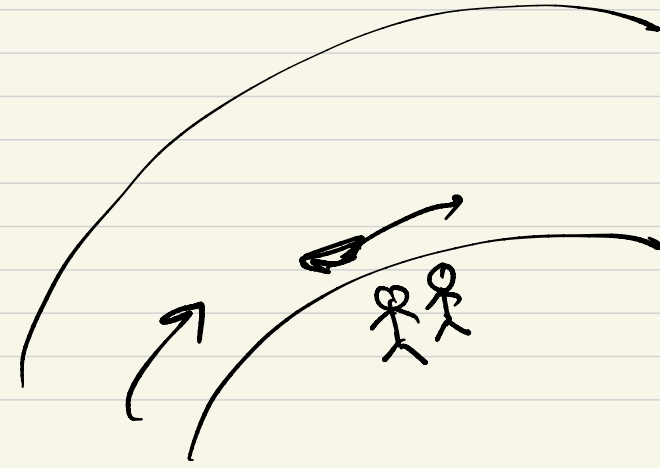
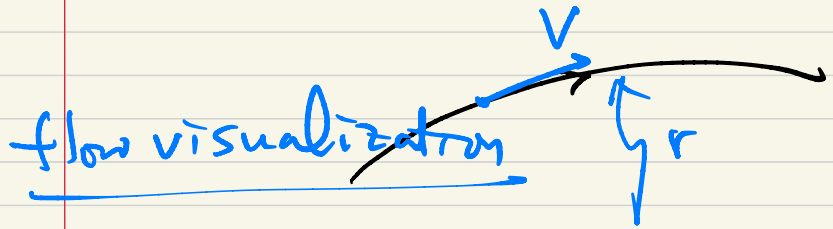
$$\frac{\partial v_r}{\partial t} + (v \cdot \nabla) v_r - \frac{1}{r} v_\theta^2 = -\frac{1}{\rho} \frac{\partial p}{\partial r} + \cancel{2} \left( \cancel{\nabla^2 v_r} - \cancel{\frac{v_r}{r^2}} - \cancel{\frac{2}{r^2} \frac{\partial v_\theta}{\partial \theta}} \right)$$

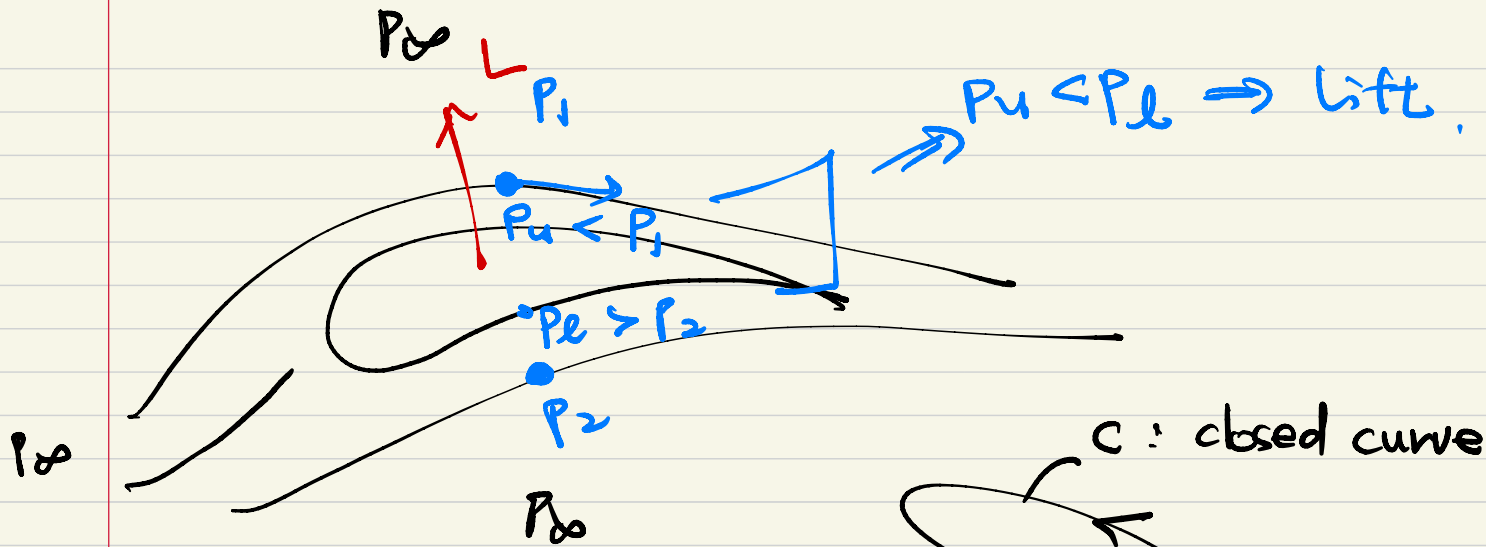
$$\rightarrow \frac{\partial p}{\partial r} = \rho \frac{v_\theta^2}{r}$$

$$\frac{\partial p}{\partial r} = \frac{\rho v_\theta^2}{r} = \frac{\rho v^2}{r} > 0$$

$$\frac{\partial p}{\partial n} = \frac{\rho v^2}{r} > 0 \quad \text{Euler-n eq.}$$

↑  
normal





• Circulation  $\Gamma$  ( $\hat{z}$  direction)

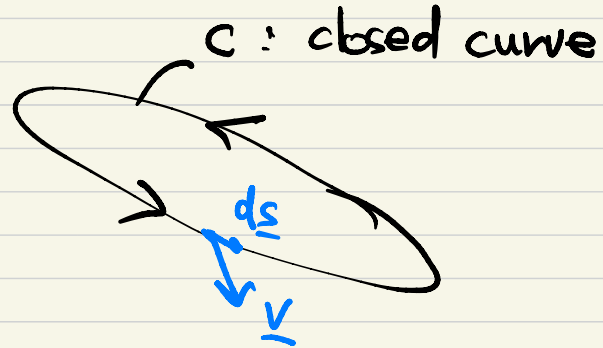
$$\Gamma \equiv \oint_c \underline{v} \cdot d\underline{s}$$

inviscid  
viscous

$$= \int_c (u dx + v dy)$$

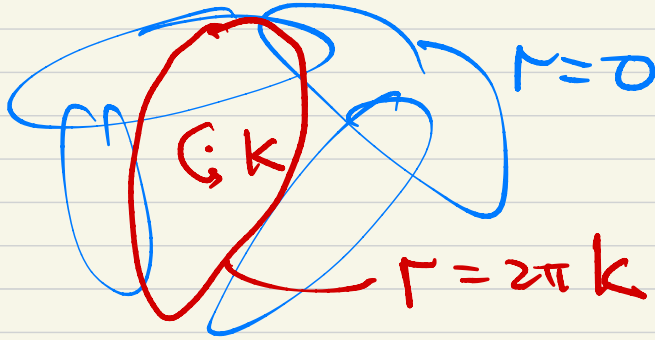
irrotational vortex ( $\underline{\omega} = 0$ )

$$\Gamma = \oint_c \nabla \phi \cdot d\underline{s} = \oint_c d\phi$$



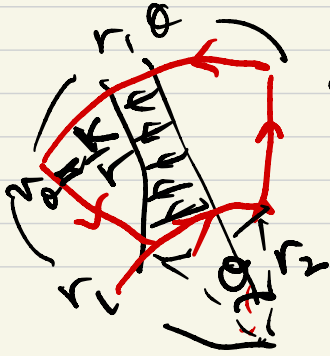
$\Gamma = 0$  for irrotational flow on the whole domain.

Free vortex  $\phi = k\theta \rightarrow \Gamma = \oint_C d\phi = 2\pi k \neq 0$



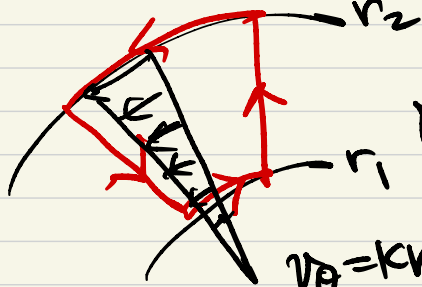
$\Gamma$  denotes the net algebraic strength of all the vortex filaments contained within the closed curve.

$\Gamma \neq 0 \quad \omega_z = \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \neq 0$  rotational flow  
 $= 0$  irrotational "



$$\begin{aligned} \oint \mathbf{v} \cdot d\mathbf{s} &= +\frac{1}{r_1} k \cdot r_1 \theta - \frac{k}{r_2} \cdot r_2 \theta \\ &= 0 \end{aligned} \quad \text{irrotational flow}$$

viscous flow



$$\int \underline{v} \cdot d\underline{s} = kr_2 \cdot r_2 \theta - kh_1 \cdot h_1 \theta$$

$$= k\theta (r_2^2 - h_1^2) \neq 0$$

$v_\theta \sim r$   
 $v_\theta = kr$   
 (solid-body rotation)

low  $p$

$$\nabla^2 \underline{v} = 0$$

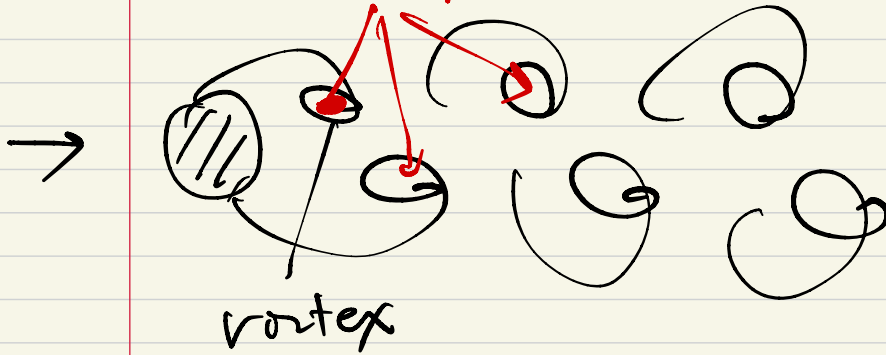
$$\nabla p = \rho(\underline{g} - \underline{a}) = -\rho \underline{a}$$

$$\underline{v} = \underline{\Omega} \times \underline{r}$$

$$\underline{a} = \underline{\Omega} \times (\underline{\Omega} \times \underline{r}) = -\rho r \Omega^2 \hat{r}$$

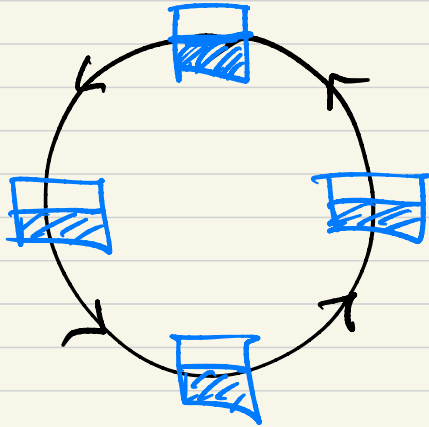
$$\frac{\partial p}{\partial r} = \rho r \Omega^2 = \rho \frac{v^2}{r} > 0$$

same as that of irrotational vortex

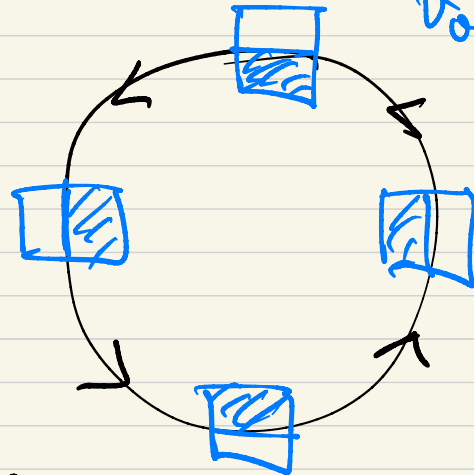




irrotational circular motion  $v_0 \sim \frac{1}{r}$



rotational circular motion  $v_0 \sim r$



vortex

vorticity →

$$N-S \Rightarrow u_i^{(x)} \cdot P(x)$$

$$\text{loop, } \underbrace{\left( \frac{v}{R} \right)}_R \underbrace{\left( \frac{\nabla v}{S} \right)}_S \quad \left( \lambda_2 \right)$$