

⑥ Superposition of plane-flow sols.

$\left\{ \begin{array}{l} \nabla^2 \phi = 0 \\ \nabla^2 \psi = 0 \end{array} \right\} \rightarrow$  linear eq.  $\rightarrow$  superposition is possible.

$$\phi_1, \phi_2 \rightarrow c_1 \phi_1 + c_2 \phi_2$$

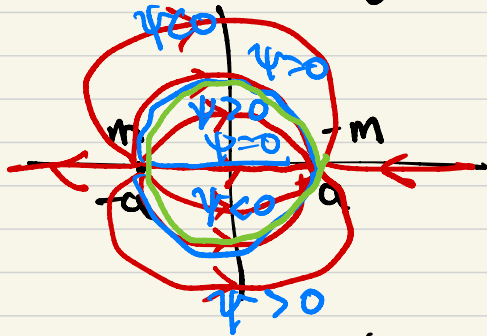
$$\psi_1, \psi_2 \rightarrow c_1 \psi_1 + c_2 \psi_2$$

$$\underline{P_1, P_2 \rightarrow c_1 P_1 + c_2 P_2 \quad X}$$

$$\left( \begin{array}{l} u = \frac{\partial \phi}{\partial x} \\ v = \frac{\partial \psi}{\partial y} \end{array} \right. \quad \left. \begin{array}{l} u_1 = \frac{\partial \phi_1}{\partial x} \\ u_2 = \frac{\partial \phi_2}{\partial x} \end{array} \right)$$

$$\frac{v_2}{2} + \frac{P}{\rho} = \text{const}$$

① Source + an equal sink



$$\psi = \psi_{\text{source}} + \psi_{\text{sink}} \quad \left( \begin{array}{l} \psi = m\theta = m \tan^{-1} \frac{y}{x} \\ \phi = m \ln r \end{array} \right)$$

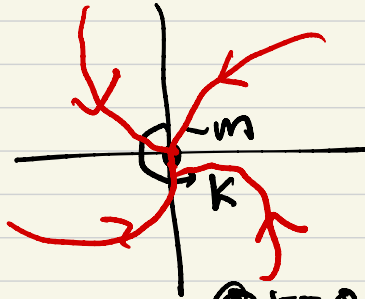
$$= m \tan^{-1} \frac{y}{x+a} - m \tan^{-1} \frac{y}{x-a}$$

$$= -m \tan^{-1} \frac{2ay}{x^2 + y^2 - a^2}$$

$$\phi = \phi_{\text{source}} + \phi_{\text{sink}} = \frac{1}{2} m \ln[(x+a)^2 + y^2] - \frac{1}{2} m \ln[(x-a)^2 + y^2]$$

$$u = \frac{\partial \phi}{\partial x}, \quad v = -\frac{\partial \psi}{\partial x} \rightarrow P \text{ from Bernoulli eq.}$$

② sink + vortex at the origin



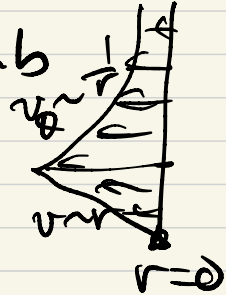
$$\psi = -m\theta - k \ln r$$

$$\phi = -m \ln r + k\theta$$

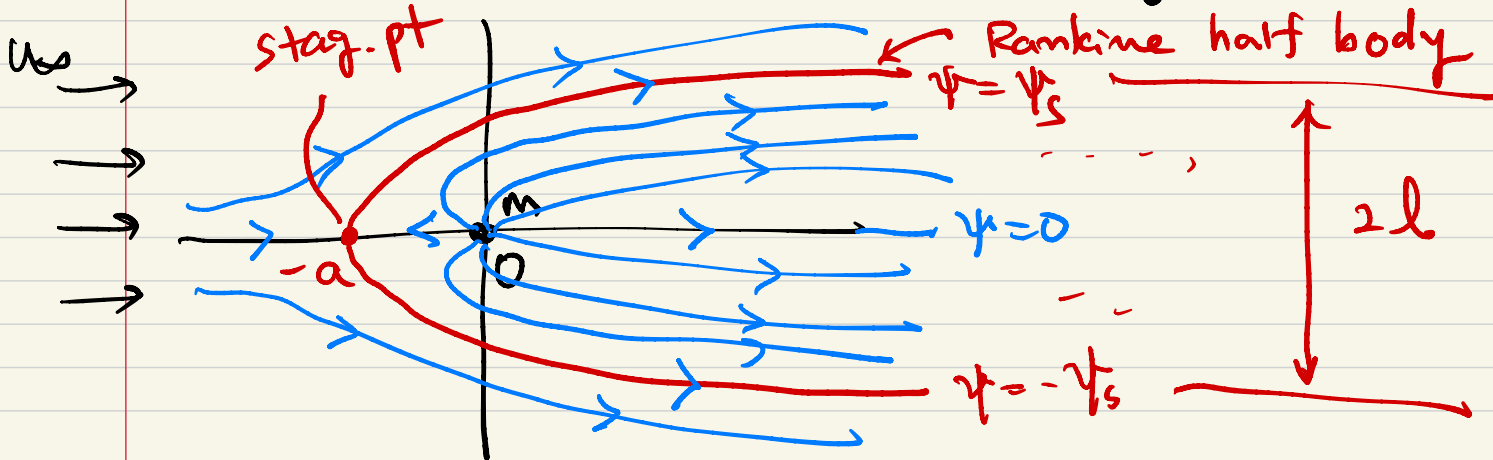
tornado, rapidly draining bathtub

@  $r=0$ ,  $v_\theta \rightarrow \infty$

In reality, near  $r=0$ , solid body rotation



③ uniform stream + a source at the origin



$$\psi = u_{\infty} r \sin \theta + m \theta \quad \rightarrow \quad u = \frac{\partial \psi}{\partial y} = u_{\infty} + \frac{m}{r} \cos \theta$$

$$\phi = u_{\infty} r \cos \theta + m \ln r \quad \rightarrow \quad v = -\frac{\partial \phi}{\partial z} = \frac{m}{r} \sin \theta$$

stag. pt:  $u = v = 0$

$$\theta = 0, \pi \rightarrow u = u_{\infty} - \frac{m}{r} \Rightarrow \boxed{a = \frac{m}{u_{\infty}}}$$

$$m = \frac{Q}{2\pi b} \rightarrow \frac{Q}{b} = 2\pi m : \text{flow rate from source}$$

$$= 2l u_{\infty} \rightarrow l = \frac{\pi m}{u_{\infty}} = \pi a$$

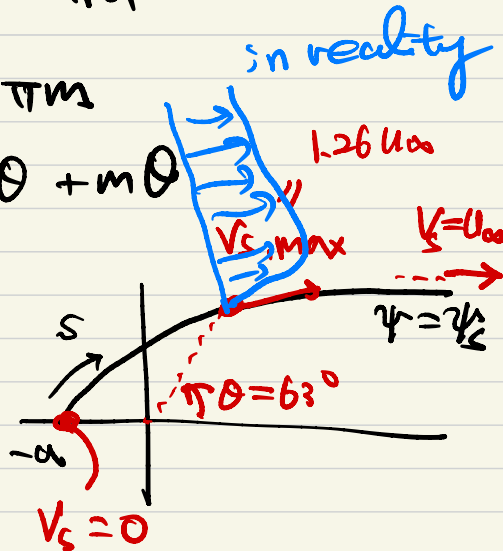
$$\frac{Q}{b} = \int d\psi = 2\psi_s = 2\pi m \rightarrow \psi_s = \pi m$$

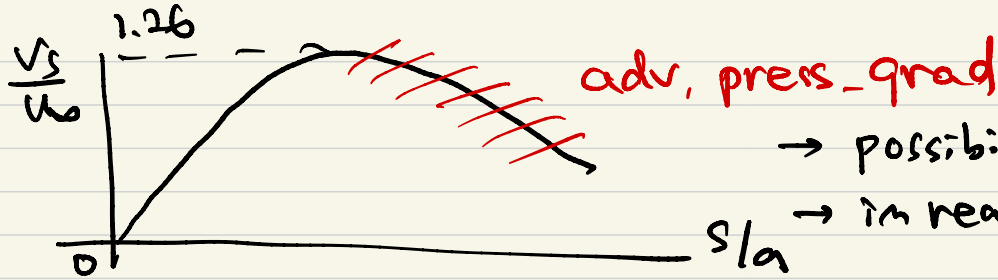
upper surface:  $\psi_s = \pi m = u_{\infty} r \sin \theta + m \theta$

$$\rightarrow r = \frac{m(\pi - \theta)}{u_{\infty} \sin \theta}$$

$$V_s^2 = u^2 + v^2 = u_{\infty}^2 \left( 1 + \frac{a^2}{r^2} + \frac{2a}{r} \cos \theta \right)$$

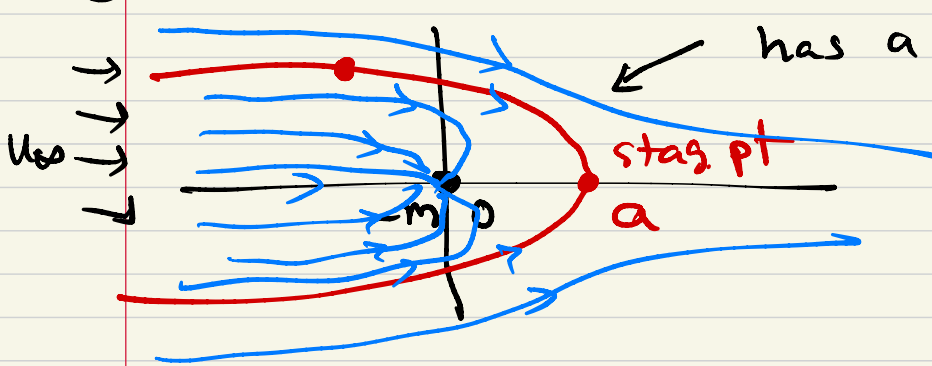
$$\frac{\partial V_s^2}{\partial \theta} = 0 : V_{s, \max} = 1.26 u_{\infty} \text{ @ } \theta = 63^\circ$$



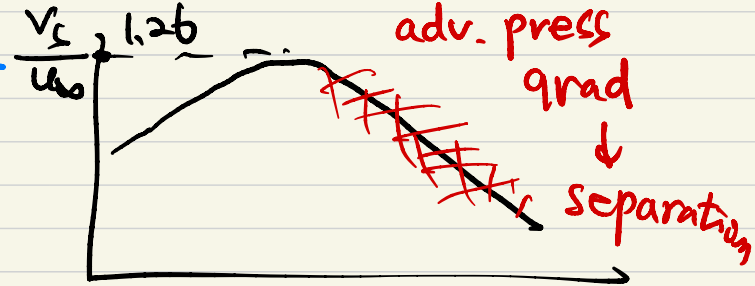


→ possibility of flow sep.  
 → in reality, no sep.

④ uni form stream + a sink

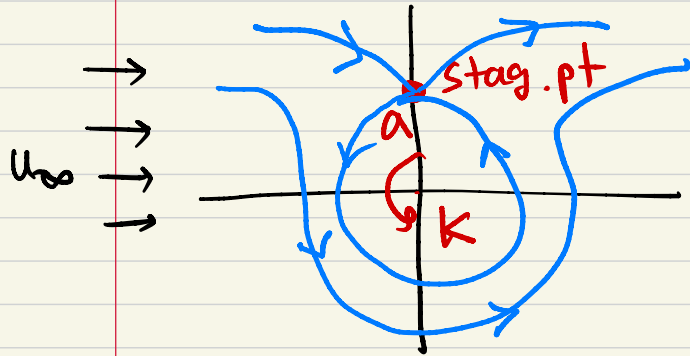


has a mirror image of  $u_0$  + source





⑤ Flow past a vortex = unif stream + vortex



$$\psi = u_{\infty} r \sin \theta - k \ln r$$

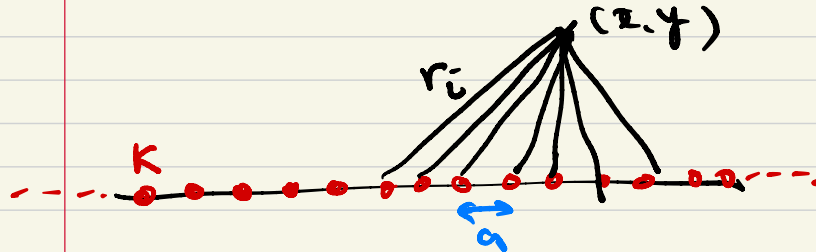
$$\phi = u_{\infty} r \cos \theta + k \theta$$

$$v_r = \frac{1}{r} \frac{\partial \psi}{\partial \theta} = u_{\infty} \cos \theta$$

$$v_{\theta} = -\frac{\partial \psi}{\partial r} = -u_{\infty} \sin \theta + \frac{k}{r}$$

stag. pt.  $v_r = v_{\theta} = 0 : \theta = \frac{\pi}{2}, r = a = \frac{k}{u_{\infty}}$

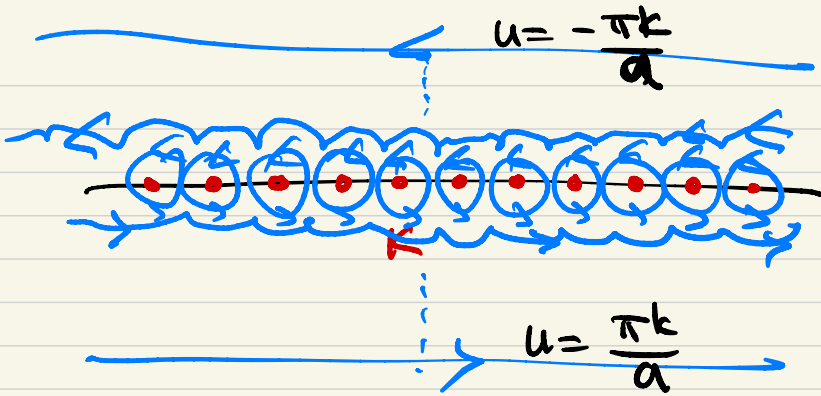
⑥ Infinite row of vortices



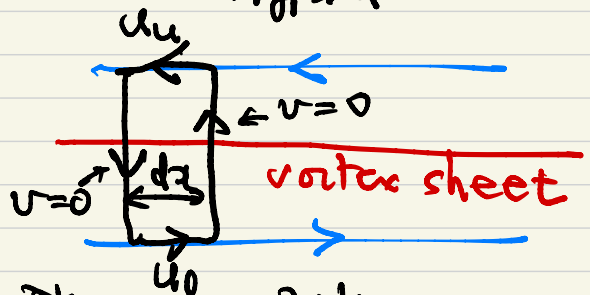
$$\psi = -k \sum_{i=1}^{\infty} \ln r_i$$

(complex variable)

$$= -\frac{1}{2} k \ln \left[ \frac{1}{2} \left( \cosh \frac{2\pi y}{a} - \cos \frac{2\pi x}{a} \right) \right]$$



$$u = \frac{\partial \psi}{\partial y} \Big|_{|y| \gg a} = \pm \frac{\pi k}{a}$$

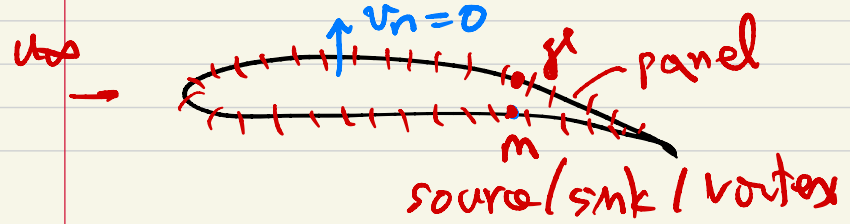


$$d\Gamma = \int \underline{u} \cdot d\underline{l} = -\frac{\pi k}{a} \cdot (-dx) + \frac{\pi k}{a} (dx) = \frac{2\pi k}{a} dx$$

$$\frac{d\Gamma}{dx} = \frac{2\pi k}{a} = \gamma$$

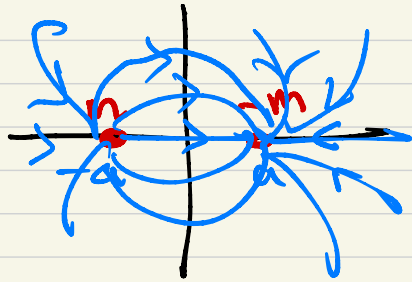
: strength of vortex sheet  
circulation per unit length of  
the vortex sheet

↓  
This is used to simulate a thin-body shape  
like air-foil, flat plate...



⇒ panel method

⑦ Doublet : source + sink pair w/ a vanishingly small distance  $a$



$$a \rightarrow 0$$

$2am = \text{const} = \lambda$  : strength of doublet

$$\psi = \lim_{\substack{a \rightarrow 0 \\ 2am = \lambda}} (-m \tan^{-1} \frac{2ay}{x^2 + y^2 - a^2})$$

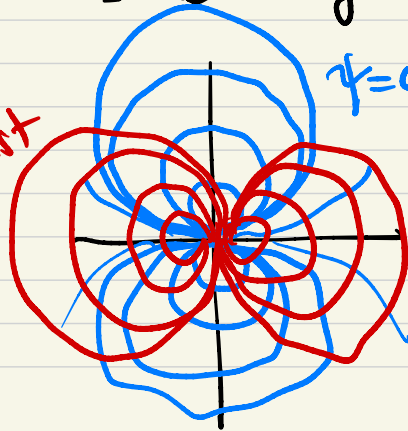
$$\left( \tan^{-1} x = x - \frac{1}{3}x^3 + \frac{1}{5}x^5 - \dots \right)$$

$$= -\frac{2amy}{x^2 + y^2} = -\frac{\lambda y}{x^2 + y^2} = -\frac{\lambda \sin \theta}{r}$$

$\rightarrow x^2 + (y + \frac{\lambda}{2\psi})^2 = (\frac{\lambda}{2\psi})^2$  : streamlines are circles.  
( $\psi = \text{const}$ )

$$(0, -\frac{\lambda}{2\psi})$$

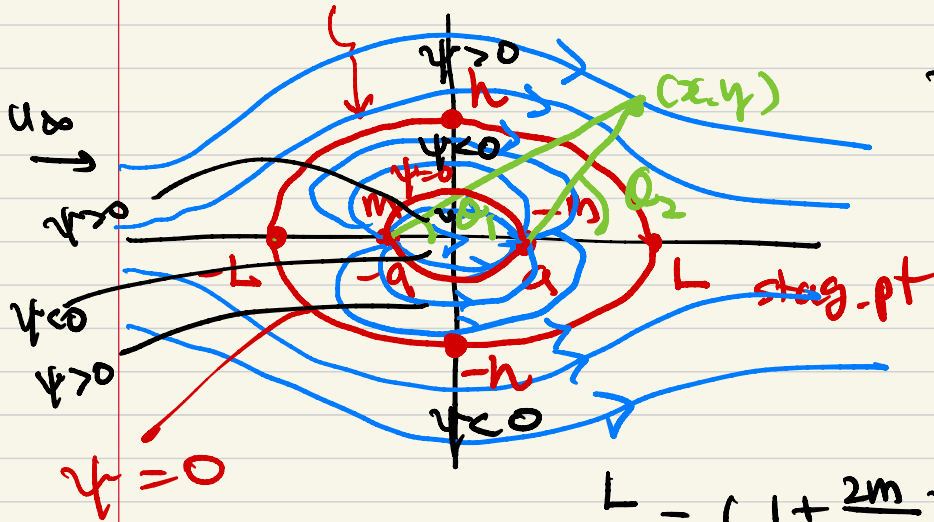
$\phi = \text{const}$



Similarly  $\phi = \frac{\lambda x}{x^2 + y^2}$

$$\rightarrow (x - \frac{\lambda}{2\phi})^2 + y^2 = (\frac{\lambda}{2\phi})^2$$

⑧ Rankine oval : uniform stream + source + sink



$$\psi = u_0 y - m \tan^{-1} \frac{2xy}{x^2 + y^2 - a^2}$$

$$= u_0 r \sin \theta + m(\theta_1 - \theta_2)$$

$$\left. \begin{aligned} u &= \frac{\partial \psi}{\partial y} = \dots \\ v &= -\frac{\partial \psi}{\partial x} = \dots \end{aligned} \right\} \begin{aligned} u &= v = 0 \\ &(\text{stag. pt.}) \end{aligned}$$

$$\frac{L}{a} = \left(1 + \frac{2m}{u_0 a}\right)^{\frac{1}{2}}$$

$$\textcircled{a} \quad y=h, x=0 : \quad \psi = u_0 h - m \tan^{-1} \frac{2ah}{h^2 - a^2} = 0 \rightarrow \frac{h}{a} = \cot \frac{h/a}{\frac{2m}{u_0 a}}$$

$$V_s^2 = u^2 + v^2 \text{ on } \psi=0 \rightarrow \frac{\partial V_s^2}{\partial x} = 0$$

$$\rightarrow \frac{u_{\max}}{u_0} = 1 + \frac{2m/u_0 a}{1 + u^2/a^2} \quad \textcircled{a} \quad \theta = 90^\circ$$

For  $m/u_0 a = 1$ ,

$$u_{\max} = 1.74 u_0$$