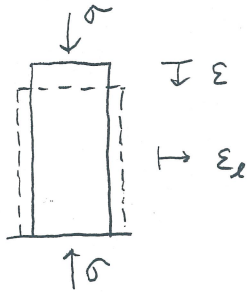


Chapter 4. Stiffness Analysis of Frames - I

4.1 Stress - strain relationships

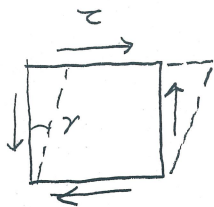
under uniaxial stress



$$\epsilon = \frac{\sigma}{E} \quad E = \text{elastic modulus}$$

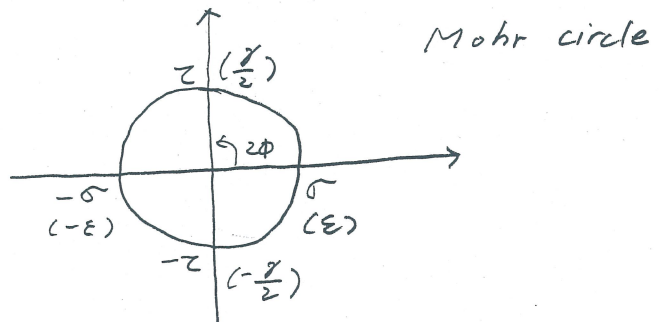
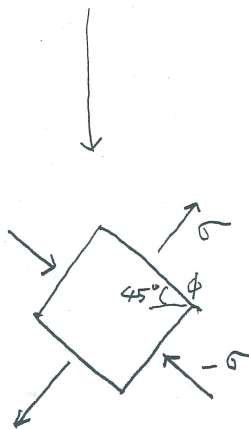
$$\begin{aligned} \epsilon_x &= -\nu \epsilon \\ &= -\nu \frac{\sigma}{E} \end{aligned} \quad \nu = \text{poisson's ratio}$$

under pure shear



$$\gamma = \frac{\tau}{G} \quad \gamma = 2 \epsilon_{xy} \quad (\text{engineering shear strain})$$

$$G = \frac{E}{2(1+\nu)}$$



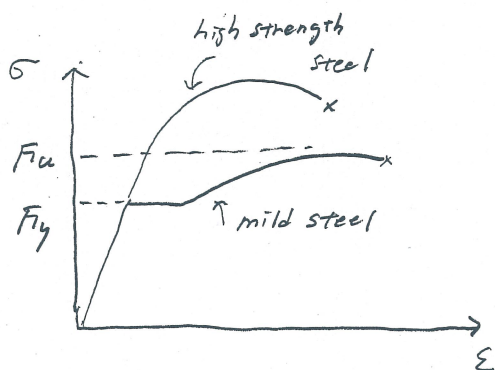
$$\epsilon = \frac{\sigma}{E} - \nu \frac{(-\sigma)}{E} = \frac{\sigma}{E} (1+\nu) = \frac{\gamma}{2}$$

under pure shear $\tau = \sigma$

$$\tau = \frac{E}{2(1+\nu)} \gamma = G \gamma$$

Material properties

steel



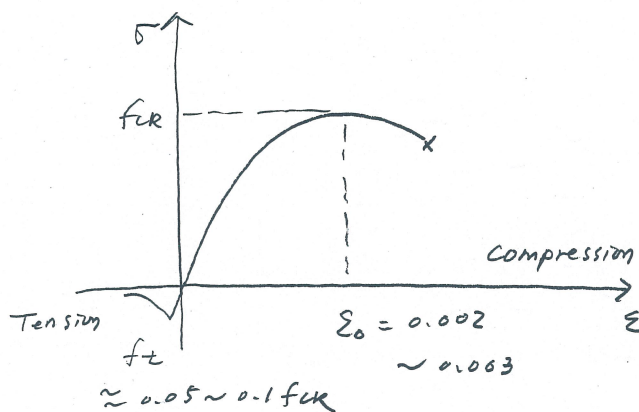
$$F_y = 250 \sim 700 \text{ MPa}$$

$$\nu = 0.3$$

$$E = 200,000 \text{ MPa}$$

Relatively expensive
 high strength and stiffness
 in both tension and compression
 low fire resistance

concrete



$$f_{ck} = 21 \sim 80 \text{ MPa}$$

(ordinary cement)

$$\nu = 0.15$$

$$E = 4700 \sqrt{f_{ck}} \text{ MPa}$$

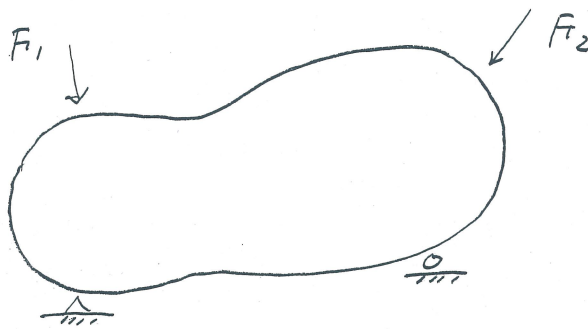
economical
 high strength only in compression
 high fire resistance
 high durability

4.2 Work and energy

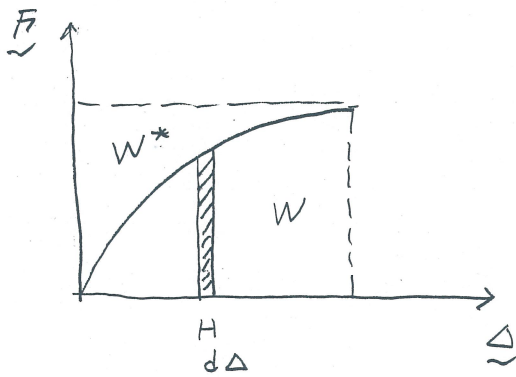
Force \Rightarrow vector with magnitude and orientation

Work \Rightarrow scalar with only magnitude ($= \sum \text{force} \times \text{displacement}$)

Work principle can be conveniently used to derive \underline{K} matrix.



External Work



$$W = \text{work} = \int F d\Delta$$

$$W^* = \text{complimentary work}$$

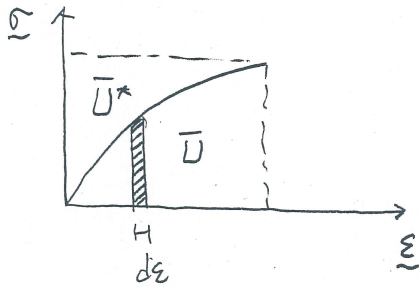
$$= \int \Delta dF$$

For linear elastic material

$$W = \frac{1}{2} \underline{F} \cdot \underline{\Delta} = \frac{1}{2} \underline{\Delta}^T \underline{K} \underline{\Delta}$$

$$W^* = \frac{1}{2} \underline{F} \cdot \underline{\Delta} = \frac{1}{2} \underline{F}_f^T \underline{d} \underline{F}_f$$

Internal Work



\bar{U} = strain energy density

$$= \int \underline{\sigma} d\underline{\varepsilon}$$

U = strain energy (internal energy)

$$= \int \bar{U} dV$$

\bar{U}^* = complimentary strain energy density

$$= \int \underline{\varepsilon} d\underline{\sigma}$$

$$U^* = \int \bar{U}^* dV$$

For linear elastic material,

$$\bar{U} = \frac{1}{2} \underline{\sigma} \cdot \underline{\varepsilon} = \bar{U}^*$$

Energy Conservation

External Work = Internal Work

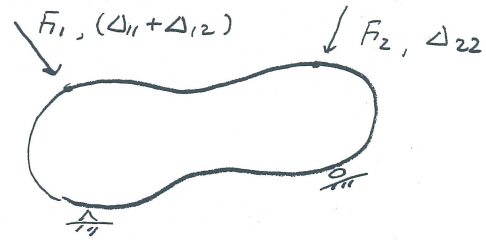
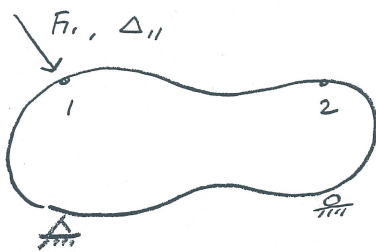
$$W = U \quad \text{and} \quad W^* = U^*$$

Example 4.1

4.3 Reciprocity

$$\left. \begin{array}{l} \underline{k} : k_{ij} = k_{ji} \\ \underline{d} : d_{ij} = d_{ji} \end{array} \right\} \Rightarrow \text{symmetry} \Rightarrow \text{computational efficiency}$$

Maxwell's Reciprocal Theorem



Apply F_1 and then $F_2 \Rightarrow$ work W_I

$$W_{I1} \text{ (By } F_1) = \frac{1}{2} \Delta_{11} F_1 = \frac{1}{2} d_{11} F_1 F_1$$

$$\begin{aligned} W_{I2} \text{ (By } F_2) &= \frac{1}{2} \Delta_{22} F_2 + \Delta_{12} F_1 \\ &= \frac{1}{2} d_{22} F_2 F_2 + d_{12} F_2 F_1 \end{aligned}$$

$$W_I = W_{I1} + W_{I2} = \frac{1}{2} d_{11} F_1^2 + d_{12} F_1 F_2 + \frac{1}{2} d_{22} F_2^2$$

Apply F_2 and then $F_1 \Rightarrow$ work W_{II}

$$W_{II} = W_{II1} + W_{II2} = \frac{1}{2} d_{11} F_1^2 + d_{21} F_1 F_2 + \frac{1}{2} d_{22} F_2^2$$

For linear elastic system,

$$W_I \equiv W_{II}$$

$$\Rightarrow d_{12} \equiv d_{21} \Rightarrow k_{12} \equiv k_{21}$$

4.4 Flexibility - Stiffness Transformations

4.4.1 Stiffness - to - flexibility transformation

$$\begin{bmatrix} \underline{F}_f \\ \underline{F}_s \end{bmatrix} = \begin{bmatrix} \underline{k}_{ff} & \underline{k}_{fs} \\ \underline{k}_{sf} & \underline{k}_{ss} \end{bmatrix} \begin{bmatrix} \underline{\Delta}_f \\ \underline{\Delta}_s \end{bmatrix}$$

if $\underline{\Delta}_s = 0$

$$\begin{bmatrix} \underline{F}_f \\ \underline{F}_s \end{bmatrix} = \begin{bmatrix} \underline{k}_{ff} \\ \underline{k}_{sf} \end{bmatrix} \underline{\Delta}_f$$

$$\underline{F}_f = \underline{k}_{ff} \underline{\Delta}_f \Rightarrow \underline{\Delta}_f = \underline{d} \underline{F}_f$$

$$\underline{d} = \underline{k}_{ff}^{-1}$$

4.4.2 flexibility - to - stiffness transformation

$$\begin{aligned} \underline{F}_f &= \underline{d}^{-1} \underline{\Delta}_f \\ &= \underline{k}_{ff} \underline{\Delta}_f \end{aligned}$$

$$\underline{k}_{ff} = \underline{d}^{-1}$$

other terms? \underline{k}_{fs} , \underline{k}_{sf} , \underline{k}_{ss} ?

$$\underline{F}_s = \underline{\Phi} \underline{F}_f \quad \text{by force-equilibrium}$$

$\underline{\Phi}$ = equilibrium matrix

$$= \underline{\Phi} \underline{d}^{-1} \underline{\Delta}_f = \underline{k}_{sf} \underline{\Delta}_f$$

$$\underline{k}_{sf} = \underline{\Phi} \underline{d}^{-1}$$

From reciprocal theorem,

$$\underline{k}_{fs} = \underline{k}_{sf}^T = (\underline{d}^{-1})^T \underline{\Phi}^T = \underline{d}^{-1} \underline{\Phi}^T$$

\underline{d}^{-1} = symmetric

$$\underline{F}_s = \underline{\Phi} \underline{F}_f = \underline{\Phi} [\underline{k}_{ff} \underline{\Delta}_f + \underline{k}_{fs} \underline{\Delta}_s]$$

$$= \underline{\Phi} [\underline{d}^{-1} \underline{\Delta}_f + \underline{d}^{-1} \underline{\Phi}^T \underline{\Delta}_s]$$

$$\underline{F}_s = \underline{k}_{sf} \underline{\Delta}_f + \underline{k}_{ss} \underline{\Delta}_s$$

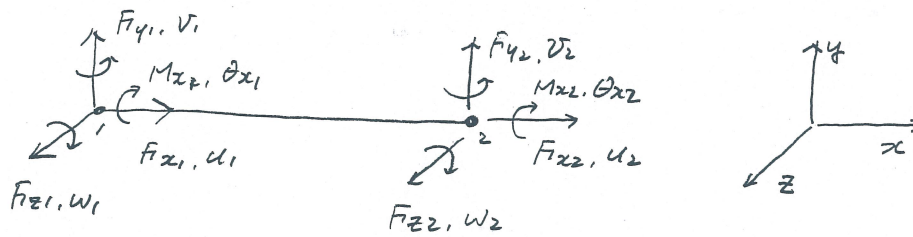
$$\underline{k}_{ss} = \underline{\Phi} \underline{d}^{-1} \underline{\Phi}^T$$

$$\underline{k} = \left[\begin{array}{c|c} \underline{k}_{ff} & \underline{k}_{fs} \\ \hline \underline{k}_{sf} & \underline{k}_{ss} \end{array} \right]$$

$$= \left[\begin{array}{c|c} \underline{d}^{-1} & \underline{d}^{-1} \underline{\Phi}^T \\ \hline \underline{\Phi} \underline{d}^{-1} & \underline{\Phi} \underline{d}^{-1} \underline{\Phi}^T \end{array} \right]$$

Example 4.2

4.5 Framework Element Stiffness Matrix



b Degree of freedom per each node

For 3 dimensional frame element, 12 dof

$$\underline{u} = \langle u_1, v_1, w_1, \theta_{x1}, \theta_{y1}, \theta_{z1}, u_2, v_2, w_2, \theta_{x2}, \theta_{y2}, \theta_{z2} \rangle$$

$$\underline{F} = \langle F_{x1}, F_{y1}, F_{z1}, M_{x1}, M_{y1}, M_{z1}, F_{x2}, F_{y2}, F_{z2}, M_{x2}, M_{y2}, M_{z2} \rangle$$

Assuming

- 1) small deformation
- 2) cross-section is bisymmetric
- 3) no warping (distortion of cross-section)

The behavior of a 3-D frame element can be uncoupled into four actions, which is independent

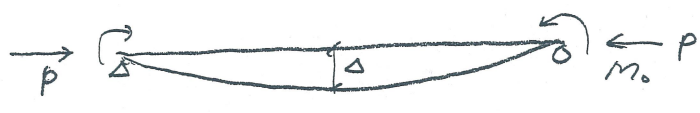
1) axial force $\underline{u} = \langle u_1, u_2 \rangle$

2) pure torsion $\underline{u} = \langle \theta_{x1}, \theta_{x2} \rangle$

3) major axis bending $\underline{u} = \langle v_1, \theta_{z1}, v_2, \theta_{z2} \rangle$

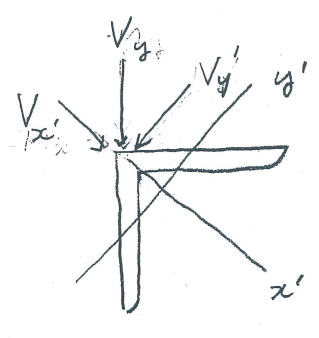
4) minor axis bending $\underline{u} = \langle w_1, \theta_{y1}, w_2, \theta_{y2} \rangle$

large deformation - coupled actions of flexure and axial compression



M at center = Mo + P * delta

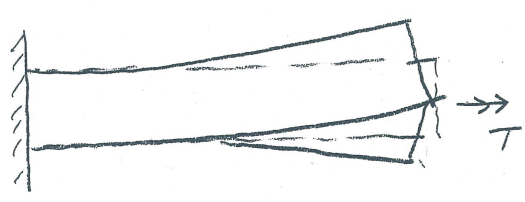
Asymmetric cross-section



Vy induces Mx' on My' in local axis.

coupled action of major axis bending and minor axis bending

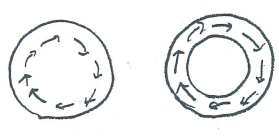
Warping (warping torsion versus pure torsion)



coupled action of torsion and minor axis bending.

when a torsional moment is applied to a asymmetric section or open section, the cross-section is distorted.

As a result, normal stresses are developed by the torsion



=> pure torsion causing shear stress only no warping

4.5.1 Axial Force member



$$u_2 = \int_0^L \epsilon dx = \int_0^L \frac{\sigma}{E} dx = \int_0^L \frac{F_{x2}}{EA} dx = \frac{F_{x2} L}{EA}$$

$$\underline{d} = \frac{L}{EA} \quad \underline{k}_{ff} = \underline{d}^{-1} = \frac{EA}{L}$$

$$F_{x1} = \underline{\phi} F_{x2} \quad \underline{\phi} = -1$$

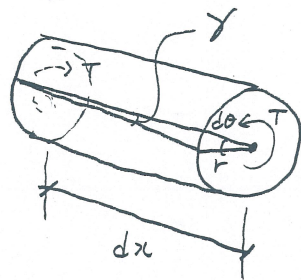
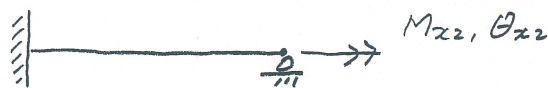
$$\underline{k}_{sf} = \underline{\phi} \underline{d}^{-1} = -\frac{EA}{L}$$

$$\underline{k}_{fs} = \underline{d}^{-1} \underline{\phi}^T = -\frac{EA}{L}$$

$$\underline{k}_{ss} = \underline{\phi} \underline{d}^{-1} \underline{\phi}^T = \frac{EA}{L}$$

$$\underline{k} = \frac{EA}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

4.5.2 Pure Torsional Member



$$\gamma \cdot dx = r d\theta$$

$$\gamma = r \frac{d\theta}{dx} = r \beta$$

$$\tau = G \gamma = G r \beta$$

$$dA = r d\phi dr$$

$$T = \int \tau \cdot r dA = \iint G r^2 \beta d\phi dr$$

$$= G J \beta$$

$$J = \text{torsional constant}$$

$$= \iint r^2 dr d\phi$$

$$\text{rate of twist } \beta \left(= \frac{d\theta_x}{dx} \right) = \frac{M_{x2}}{GJ}$$

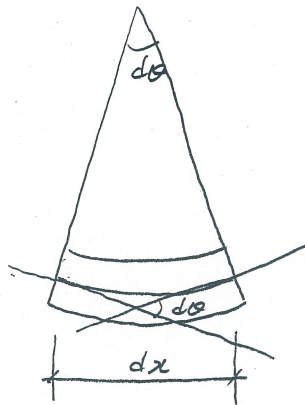
$$\theta_{x2} = \int_0^L \beta dx = \frac{L}{GJ} M_{x2}$$

$$\underline{d} = \frac{L}{GJ} \quad \underline{k}_{\text{eff}} = \underline{d}^{-1} = \frac{GJ}{L}$$

$$M_{x1} = -M_{x2} \Rightarrow \underline{\phi} = -1$$

$$\underline{k} = \frac{GJ}{L} \begin{bmatrix} 1 & -1 \\ -1 & 1 \end{bmatrix}$$

4.5.3 Beam bent about its z axis



A diagram of a beam element showing the relationship between strain ϵ and the angle of rotation $\frac{d\theta}{dx} = \phi$. The strain is shown as a linear distribution across the height y .

$$\epsilon = -y \frac{d\theta}{dx}$$

$$= -y \frac{d^2v}{dx^2}$$

$$\epsilon_x = -y \frac{d^2v}{dx^2}$$

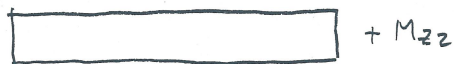
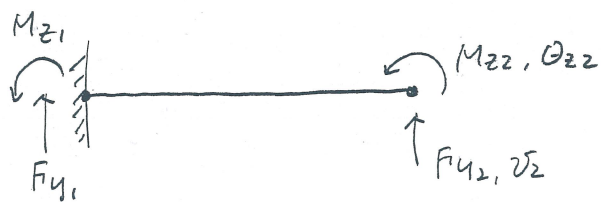
$$\sigma_x = -Ey \frac{d^2v}{dx^2}$$

$$M_x = -\int \sigma_x y dA = EI_z \frac{d^2v}{dx^2} = EI_z \phi$$

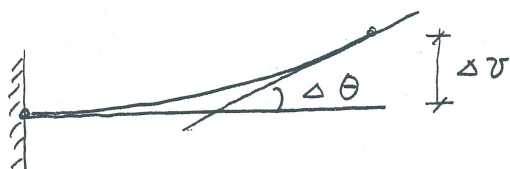
$M_x =$ Generalized stress

$\frac{d^2v}{dx^2} =$ Generalized strain

$EI_z =$ Generalized stiffness



Moment - area method (curvature area method)



$$\Delta \theta = \int \phi dx = \int \frac{M}{EI} dx (= \int d\theta)$$

$$\Delta v = \int d\theta \cdot x = \int \frac{M}{EI} \cdot x dx$$

$$\theta_{z2} = \frac{M_{z2}}{EI} \cdot L + \frac{F_{y2}L}{EI} \cdot \frac{L}{2}$$

$$v_2 = \frac{M_{z2}}{EI} L \cdot \frac{L}{2} + \frac{F_{y2}L}{EI} \cdot \frac{L}{2} \cdot \frac{2}{3} L$$

$$\begin{bmatrix} v_2 \\ \theta_{z2} \end{bmatrix} = \frac{L}{EI_2} \underbrace{\begin{bmatrix} \frac{L^2}{3} & \frac{L}{2} \\ \frac{L}{2} & 1 \end{bmatrix}}_{\underline{d}} \begin{bmatrix} F_{y2} \\ M_{z2} \end{bmatrix}$$

$$\underline{\Delta f} = \underline{d} \underline{F_f}$$

By force - equilibrium

$$F_{y1} = -F_{y2}$$

$$M_{z1} = -F_{y2} \cdot L - M_{z2}$$

$$\begin{bmatrix} F_{y1} \\ M_{z1} \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ -L & -1 \end{bmatrix} \begin{bmatrix} F_{y2} \\ M_{z2} \end{bmatrix}$$

$$\underline{\underline{F}}_s = \underline{\underline{\Phi}} \underline{\underline{F}}_f$$

$$\underline{\underline{k}} = \begin{bmatrix} \underline{\underline{d}}^{-1} & \underline{\underline{d}}^{-1} \underline{\underline{\Phi}}^T \\ \underline{\underline{\Phi}} \underline{\underline{d}}^{-1} & \underline{\underline{\Phi}} \underline{\underline{d}}^{-1} \underline{\underline{\Phi}}^T \end{bmatrix}$$

$$\text{kearrage } \underline{\underline{k}} = \frac{EI_z}{L} \begin{array}{c} \begin{array}{cccc} v_1 & \theta_1 & v_2 & \theta_2 \end{array} \\ \begin{array}{|c|c|c|c|} \hline 12/L^2 & 6/L & -12/L^2 & 6/L \\ \hline 6/L & 4 & -6/L & 2 \\ \hline -12/L^2 & -6/L & 12/L^2 & -6/L \\ \hline 6/L & 2 & -6/L & 4 \\ \hline \end{array} \end{array}$$

Eq. (4.32)

see examples 4.3 and 4.4

4.5.4 Beam bent about its y axis

$\underline{\underline{k}}$ for this action can be independently derived

in the same manner

see eq. (4.33)

Example 4.5

conjugate beam method

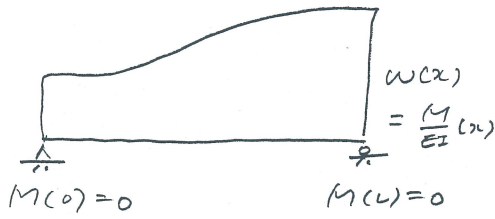
$$\frac{dM}{dx} = V$$

$$\frac{dV}{dx} = w$$

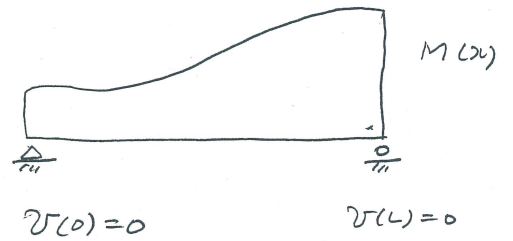


$$\frac{dV}{dx} = 0$$

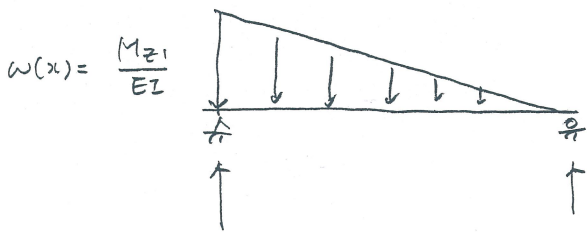
$$\frac{d\theta}{dx} = \phi \left(= \frac{M}{EI} \right)$$



conjugate beam



actual beam



$$V_1 = \frac{M}{EI} \cdot \frac{L}{3}$$

↓

$$\theta_{z1}$$

$$V_2 = \frac{M}{EI} \cdot \frac{L}{6}$$

↓

$$\theta_{z2}$$

4.5.5 Complete Element stiffness matrix

$$\underline{k} = \underline{k}_{\text{axial}} + \underline{k}_{\text{torsion}} + \underline{k}_{z\text{-bending}} + \underline{k}_{y\text{-bending}}$$

All actions can be superimposed because they are assumed to be independent \Rightarrow Eq. (4.34)

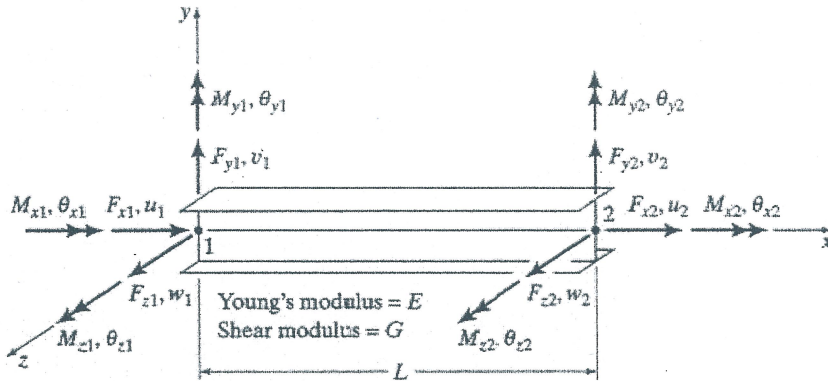


Figure 4.6 Bisymmetrical framework element.

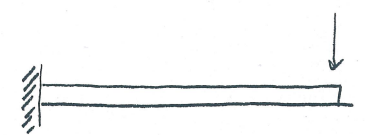
$\begin{Bmatrix} F_{x1} \\ F_{y1} \\ F_{z1} \\ M_{x1} \\ M_{y1} \\ M_{z1} \\ F_{x2} \\ F_{y2} \\ F_{z2} \\ M_{x2} \\ M_{y2} \\ M_{z2} \end{Bmatrix} = E$	$\frac{A}{L}$	0	0	0	0	0	0	$-\frac{A}{L}$	0	0	0	0	0	0	u_1
	0	$\frac{12I_z}{L^3}$	0	0	0	$\frac{6I_z}{L^2}$	0	$-\frac{12I_z}{L^3}$	0	0	0	0	$\frac{6I_z}{L^2}$	0	v_1
	0	0	$\frac{12I_y}{L^3}$	0	$-\frac{6I_y}{L^2}$	0	0	0	$-\frac{12I_y}{L^3}$	0	$\frac{6I_y}{L^2}$	0	0	0	w_1
	0	0	0	$\frac{J}{2(1+\nu)L}$	0	0	0	0	0	0	$-\frac{J}{2(1+\nu)L}$	0	0	0	θ_{x1}
	0	0	$-\frac{6I_y}{L^2}$	0	$\frac{4I_y}{L}$	0	0	0	$\frac{6I_y}{L^2}$	0	$\frac{2I_y}{L}$	0	0	0	θ_{y1}
	0	$\frac{6I_z}{L^2}$	0	0	0	$\frac{4I_z}{L}$	0	$-\frac{6I_z}{L^2}$	0	0	0	0	$\frac{2I_z}{L}$	0	θ_{z1}
	$-\frac{A}{L}$	0	0	0	0	0	$\frac{A}{L}$	0	0	0	0	0	0	0	u_2
	0	$-\frac{12I_z}{L^3}$	0	0	0	$-\frac{6I_z}{L^2}$	0	$\frac{12I_z}{L^3}$	0	0	0	0	0	$-\frac{6I_z}{L^2}$	v_2
	0	0	$-\frac{12I_y}{L^3}$	0	$\frac{6I_y}{L^2}$	0	0	0	$\frac{12I_y}{L^3}$	0	$\frac{6I_y}{L^2}$	0	0	0	w_2
	0	0	0	$-\frac{J}{2(1+\nu)L}$	0	0	0	0	0	0	$\frac{J}{2(1+\nu)L}$	0	0	0	θ_{x2}
	0	0	$-\frac{6I_y}{L^2}$	0	$\frac{2I_y}{L}$	0	0	0	$\frac{6I_y}{L^2}$	0	$\frac{4I_y}{L}$	0	0	0	θ_{y2}
	0	$\frac{6I_z}{L^2}$	0	0	0	$\frac{2I_z}{L}$	0	$-\frac{6I_z}{L^2}$	0	0	0	0	0	$\frac{4I_z}{L}$	θ_{z2}

4.6 A commentary on deformations and displacement variables

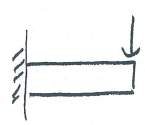
4.6.1 Neglected deformations

□ Transverse shear

$$\Delta = \Delta_{flexure} + \Delta_{shear}$$

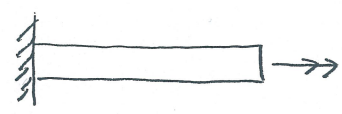


$$\Delta \approx \Delta_{flexure}$$



$$\Delta = \Delta_{flexure} + \Delta_{shear}$$

□ Warping torsion



pure torsion

$$T \rightarrow \tau$$



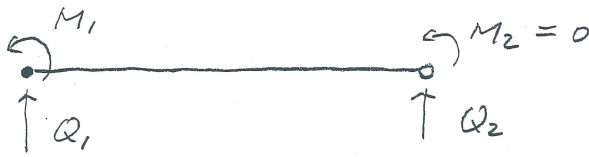
pure torsion
+ warping torsion

$$T \rightarrow \tau \neq \sigma$$

Torsion + minor axis bending
(\neq warping torsion)

Example 4.14

Stiffness of element with a hinge

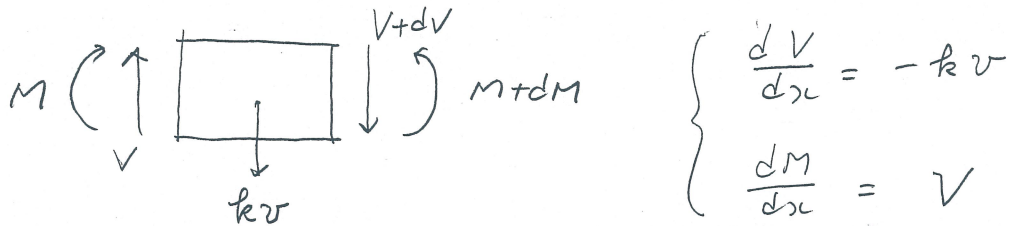
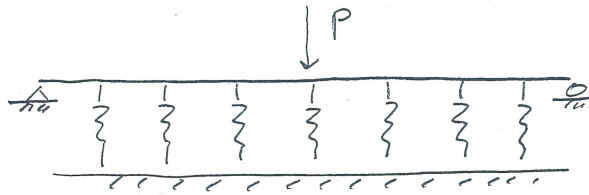


$$EI \begin{bmatrix} \frac{12}{L^3} & \frac{6}{L^2} & -\frac{12}{L^3} & \frac{6}{L^2} \\ \frac{6}{L^2} & \frac{4}{L} & -\frac{6}{L^2} & \frac{2}{L} \\ -\frac{12}{L^3} & -\frac{6}{L^2} & \frac{12}{L^3} & -\frac{6}{L^2} \\ \frac{6}{L^2} & \frac{2}{L} & -\frac{6}{L^2} & \frac{4}{L} \end{bmatrix} \begin{bmatrix} v_1 \\ \theta_1 \\ v_2 \\ \theta_2 \end{bmatrix} = \begin{bmatrix} Q_1 \\ M_1 \\ Q_2 \\ \textcircled{M_2} = 0 \end{bmatrix}$$

Eliminate Eq. (4) $\theta_2 = -\frac{3}{2L} v_1 - \frac{1}{2} \theta_1 + \frac{3}{2L} v_2$

$$EI \begin{bmatrix} \frac{3}{L^3} & \frac{3}{L^2} & -\frac{3}{L^3} \\ \frac{3}{L^2} & \frac{3}{L} & -\frac{3}{L^2} \\ -\frac{3}{L^3} & -\frac{3}{L^2} & \frac{3}{L^3} \end{bmatrix} \begin{bmatrix} v_1 \\ \theta_1 \\ v_2 \end{bmatrix} = \begin{bmatrix} Q_1 \\ M_1 \\ Q_2 \end{bmatrix}$$

Example 4.15



$$\phi \Rightarrow \frac{d^2 V}{dx^2} = \frac{M}{EI}, \quad \frac{d^2 M}{dx^2} = -kv$$

$$\Rightarrow \underline{EI \frac{d^4 V}{dx^4} = -kv}$$

$$B/c, \quad x=0 \quad v(0) = 0$$

$$M(0) = \frac{d^2 v}{dx^2}(0) = 0$$

$$x=l \quad v(l) = 0$$

$$M(l) = \frac{d^2 v}{dx^2}(l) = 0$$