# Frequency Domain Techniques and Loop Shaping

Francis Ch.1-7 Chen Ch. 9

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# **Course Info**

• Control Systems I

Introductory graduate-level control course, mainly focusing on linear systems, bridge between System Control Theory (undergrad. classical control) and Control System II (nonlinear/optimal control).

- Instructor: Dongjun Lee (djlee@snu.ac.kr, 301-1517, 880-1724)
- Teaching Assistants:
  - Nguyen Hai-Nguyen (Lead: hainguyen@snu.ac.kr, 301-211, 880-1690)
  - Hoyong Lee, Jeongseob Lee (hylee0428, overjs94@snu.ac.kr, same as above)
- Prerequisites
  - $-\,$  System Control Theory (ME2794.002100) or equivalent; or by the consent of instructor
- Grading
  - 1. HW 20% (score 0/1, 0.5/1, 1/1)
  - 2. Mid-term exam  $40\% \ 4/19/2017 \ W \ 7-9:30 pm$
  - 3. Final exam 40% 6/14/2017 W 7-10pm

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# **Course Topics**

- 1. Loop shaping and fundamental limitations
- 2. Control design using Youla parameterization, coprime factorization
- 3. Linear system state-space representation and solution
- 4. Stability
- 5. Controllability, observability, canonical decomposition
- 6. Minimal realization, balanced realization
- 7. State-space control design, state estimation, separation principle
- 8. Linear quadratic regulator (LQR)
- 9. Kalman-Bucy filter and linear quadratic Gaussian (LQG)
- 10. Kalman estimation and extended Kalman filtering (EKF)

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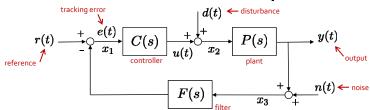


# **Course Info**

- Textbooks
  - Linear System Theory and Design, Chen, Oxford University Press
  - Feedback Control Theory, Doyle, Francis & Tannenbaum, Dover
  - Control System Design, Friedland, Dover
- References
  - Multivariable Feedback Control, Skogestad & Postlethwaite, Willey
  - Control System Design, Goodwin, Gradebe & Salgado, Prentice Hall
  - Linear System Theory, Rugh & Kailath, Prentice Hall
  - Principles of Robot Motion, Choset et al, MIT Press
  - Optimal Control: Linear Quadratic Methods, Anderson & Moore, Dover



# **Basic Feedback Loop**



- Plant P(s), controller C(s), filter F(s), all SISO.
- Exogeneous signals: reference r(t), disturbance d(t), noise n(t).
- Signal signals: tracking error e(t), output y(t), control u(t) (or  $x_1, x_2, x_3$ ).
- Closed-loop transfer functions:

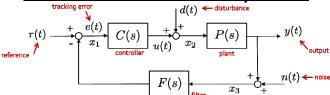
$$\left(egin{array}{c} x_1 \ x_2 \ x_3 \end{array}
ight) = rac{1}{1+PCF} \left[egin{array}{ccc} 1 & -PF & -F \ C & 1 & -CF \ PC & P & 1 \end{array}
ight] \left(egin{array}{c} r \ d \ n \end{array}
ight)$$

• Well-posedness (stronger than stability): all closed-loop TFs exist  $\Leftrightarrow$   $1 + P(s)C(s)F(s) \neq 0 \ \forall s \in \mathcal{C} \ (\text{e.g.}, \ P = C = 1, F = -1).$ 

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# Well-Posedness and Proper TFs



• Closed-loop transfer functions:

$$\left(\begin{array}{c} x_1 \\ x_2 \\ x_3 \end{array}\right) = \frac{1}{1+PCF} \left[\begin{array}{ccc} 1 & -PF & -F \\ C & 1 & -CF \\ PC & P & 1 \end{array}\right] \left(\begin{array}{c} r \\ d \\ n \end{array}\right)$$

- Ill-posedness arises due to the lack of time to response (cf. integrator).
- CL system well-posed if P(s) is strictly proper and C(s), F(s) are proper.
- G(s) is (strictly) stable if analytic in closed-RHP (i.e.,  $\text{Re}(s) \geq 0$ ).
- G(s) is proper if  $G(j\infty)$  is finite (i.e.,  $\deg(N) = \deg(D)$ ).
- G(s) is strictly proper if  $G(j\infty) = 0$  (i.e.,  $\deg(D) > \deg(N)$ ).
- CL system well-posed if 1 + P(s)C(s)F(s) is not strictly proper.

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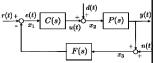
• Closed-loop transfer functions:

$$\left( \begin{array}{c} e \\ y \\ u \end{array} \right) = \frac{1}{1 + PCF} \left[ \begin{array}{ccc} 1 & -PF & -F \\ PC & P & -PCF \\ C & -PCF & -CF \end{array} \right] \overline{ \left( \begin{array}{c} r \\ d \\ n \end{array} \right) }$$

- Def: The CL system is internally stable if all nine TFs are stable.
  - Not enough just to ensure stability of input-output TF (i.e.,  $\frac{PC}{1+PCF}$ ).
  - Internal instability with unstable pole-zero cancelation (e.g., with  $P=\frac{1}{s^2-1},\,C=\frac{s-1}{s+1},\,F=1,\,r\to y$  stable, yet,  $d\to y$  unstable)
- Th. 1-1: CL system is internally stable iff no CL poles in  $Re(s) \ge 0$ .
  - Write P, C, F as coprime polynomials (i.e., with no common factors):  $P = \frac{N_P}{M_P}, C = \frac{N_C}{M_C}, F = \frac{N_F}{M_F} \Rightarrow \frac{1}{1 + PCF} = \frac{M_P M_C M_F}{N_P N_C N_F + M_P M_C M_F}.$   $- \text{ CL poles} = \text{zeros of } N_P N_C N_F + M_P M_C M_F = 0 \text{ (w/o PZ-cancelation)}.$

  - $-\text{ CL TF matrix: } \frac{1}{N_PN_C+M_PM_C} \left[ \begin{array}{ccc} M_PM_C & -N_PM_C & -M_PM_C \\ M_PN_C & M_PM_C & -M_PN_C \\ N_PN_C & -N_PM_C & M_PM_C \end{array} \right]$
  - $(\Leftarrow)$  obvious;  $(\Rightarrow)$  if not, that unstable pole should be canceled in all nine TFs  $\rightarrow$  impossible w/ coprimeness among  $M_{\star}, N_{\star}$ .

# Internal Stability $r(t) \rightarrow c(t) \rightarrow c($



• Feedback system transfer functions:

$$\left(\begin{array}{c} e \\ y \\ u \end{array}\right) = \frac{1}{1+PCF} \left[\begin{array}{ccc} 1 & -PF & -F \\ PC & P & -PCF \\ C & -PCF & -CF \end{array}\right] \left(\begin{array}{c} r \\ d \\ n \end{array}\right)$$

- Th. 1-2: The CL system is internally stable if and only if:
  - 1.  $1 + PCF = \frac{N_P N_C N_F + M_P M_C M_F}{M_P M_C M_F}$  has no zeros in Re(s)  $\geq$  0; and
  - 2. No unstable pole-zero cancelation when the loop-TF PCF is formed.
  - $(\Rightarrow)$  First condition obvious. Also, from Th. 1-1, no unstable zeros of  $N_P N_C N_F + M_P M_C M_F \rightarrow$  no unstable pole-zero cancelation possible.
  - (⇐) Will show that, under two conditions, CL poles necessarily stable. Suppose not. Then,  $\exists s_o$ , s.t.,  $[N_P N_C N_F + M_P M_C M_F](s_o) = 0$ ,  $\operatorname{Re}(s_o) \geq 0.$ 
    - 1. Suppose  $[M_PM_CM_F](s_o) \neq 0 \rightarrow [\frac{M_PM_CM_F + N_PN_CN_F}{M_PM_CM_F}](s_o) = [1 + PCF](s_o) = 0 \rightarrow \text{contradict to condition 1.}$
    - 2. Suppose  $[M_P M_C M_F](s_o) = 0 \rightarrow [N_P N_C N_F](s_o) = 0 \rightarrow \text{unstable}$ PZ-cancelation  $\rightarrow$  contradict to condition 2.

# **Unit Feedback System**



• Unit-feedback CL transfer functions with F(s) = 1:

$$\left(\begin{array}{c} e \\ y \\ u \end{array}\right) = \frac{1}{1+PC} \left[\begin{array}{ccc} 1 & -P & -1 \\ PC & P & -PC \\ C & -PC & -C \end{array}\right] \left(\begin{array}{c} r \\ d \\ n \end{array}\right) \xrightarrow[]{r(t)_{+}} e(t) \xrightarrow[]{e(t)} \left(\begin{array}{c} l(t) \\ l(t) \\ l(t) \end{array}\right) \xrightarrow[]{p(t)} \left(\begin{array}{c} l(t) \\ l(t) \\ l(t) \end{array}\right) \xrightarrow[]{e(t)} \left(\begin{array}{c} l(t) \\ l(t) \\ l(t)$$

- Nyquist stability theorem: Construct Nyquist plot of L = PC, indenting to RHP around poles on jw-axis. Let n denote unstable poles of P and C. Then, CL system is internally stable iff Nyquist plot doesn't pass through (-1,0) and encircles it exactly n-times counter-clockwise.
- Sensitivity function  $S := \frac{e}{r} = \frac{1}{1+L} = \frac{1}{1+PC} = \frac{M_P M_C}{N_P N_C + M_P M_C}$ .
  - Step tracking if S has at least one zero at s=0; ramp tracking if two zeros at s=0 (cf. system type, smaller  $\rightarrow$  better performance).
- Complementary sensitivity function  $T := \frac{y}{r} = \frac{L}{1+L} = \frac{PC}{1+PC} = \frac{N_P N_C}{N_P N_C + M_P M_C}$ .
  - Total transfer function from reference r to output y.
  - S is sensitivity of total TF T against  $\Delta P$ :  $S = \frac{\partial T/T}{\partial P/P}$ .
  - -S(s) + T(s) = 1 with T relevant to robust stability.

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# **Control Parameterization: Stable P**

• Unit-feedback transfer functions:

$$\begin{pmatrix} e \\ y \\ u \end{pmatrix} = \frac{1}{1 + PC} \begin{bmatrix} 1 & -P & -1 \\ PC & P & -PC \\ C & -PC & -C \end{bmatrix} \begin{pmatrix} r \\ d \\ n \end{pmatrix} \xrightarrow{r(t) + c(t)} C(s) \xrightarrow{t} C(s) \xrightarrow{t} P(s) \xrightarrow{t} P(s)$$

- Control synthesis: given P, design C s.t., feedback system (1) is internally stable and (2) acquires some desired properties (e.g., tracking).
- Parameterize stabilizing  $C \to \text{tweak } C$  to achieve desired properties.
- Th. 1-3: Suppose  $P \in \mathcal{S}$ , where  $\mathcal{S}$  is the set of all stable, proper, real-rational functions. Then, all the stabilizing controllers are given by

$$C = \frac{Q}{1 - PQ}, \quad \forall Q \in \mathcal{S} \qquad \stackrel{\frac{r(\theta : t_0)}{t_0} - \frac{y(t)}{t_0}}{\uparrow} \stackrel{y(t)}{\longleftarrow} \stackrel{\frac{r(\theta : t_0)}{t_0} - \frac{y(t)}{t_0}}{\uparrow} \stackrel{y(t)}{\longleftarrow} \stackrel{y(t)$$

- Affine relation: S = 1 PQ,  $T = PQ \rightarrow$  not work for unstable P.
- (Ex) Design ramping tracking control for  $P(s) = \frac{1}{(s+1)(s+2)}$ .
  - S with at least two zeros at  $s = 0 \rightarrow Q = \frac{as+b}{s+1} \rightarrow CL$  stable.
  - $-S = 1 PQ = \frac{s^3 + 4s^2 + (5 a)s + (2 b)}{(s + 1)^2(s + 2)} \rightarrow a = 5, b = 2 \text{ (increase order}[C]).$

# Coprime Factorization

• **Bezout identity:** given polynomials  $N(s), M(s), \exists$  polynomials X(s), Y(s)

N(s)X(s) + M(s)Y(s) = 1

iff N(s), M(s) are coprime  $(X = s + 1, Y = s^2 + s;$  also for integers).

• Coprimeness of TFs: we say  $N(s), M(s) \in \mathcal{S}$  are coprime if  $\exists X(s), Y(s) \in \mathcal{S}$  $\mathcal{S}$  s.t.,

N(s)X(s) + M(s)Y(s) = 1

This coprimeness of N, M holds is iff N(s), M(s) have no common zero in  $\operatorname{Re}(s) \geq 0$  and at  $s \to \infty$  (i.e., both N, M can't be strictly proper).

• Coprime factorization: for any real-rational TF G(s), we can always write its coprime factorization over S s.t.,

 $G(s) = rac{N(s)}{M(s)}, \quad NX + NY = 1, \ N, M, X, Y \in \mathcal{S}$ 

- (Ex)  $G(s) = \frac{1}{s-1} \to N(s) = \frac{1}{s+1}$ ,  $M(s) = \frac{s-1}{s+1}$  (all pass). What if  $(N, M) = (\frac{1}{(s+1)^2}, \frac{s-1}{(s+1)^2})$  or  $(N, M) = (\frac{(s-2)}{(s+1)^2}, \frac{(s-1)(s-2)}{(s+1)^2})$ ? (Ex: Euclid against  $G(s) = \frac{1}{(s-1)(s-2)} \to N = \frac{1}{(s+1)^2}$ ,  $M = \frac{(s-1)(s-2)}{(s+1)^2}$ ,
- $X = \frac{19s-11}{s+1}, Y = \frac{s+6}{s+1}.$

# Youla Parameterization

• Th. 1-4: Consider unit feedback system with possibly unstable  $P = \frac{N}{M}$ , where  $N, M \in \mathcal{S}$  are coprime with  $NX + MY = 1, X, Y \in \mathcal{S}$ . Then, all the stabilizing controllers are given by

$$C = rac{X + MQ}{Y - NQ}, \quad orall Q \in \mathcal{S}$$

- Main idea of proof:
  - (Lemma 1-1) Let  $C = \frac{N_C}{M_C}$  be a comprime factorization. Then, the feedback system is internally stable iff  $(NN_C + MM_C)^{-1} \in \mathcal{S}$ .
  - $(\Leftarrow)$  Given  $Q \in \mathcal{S}$  and C as above, define  $N'_C := X + MQ$ ,  $M'_C :=$ Y - NQ (not necessarily coprime)  $\rightarrow$  put these into NX + MY = 1, we have  $NN'_C + MM'_C = 1$ , implying  $N'_C, M'_C$  are coprime. Further, CL stable from Lem. 1-1 w/  $NN'_C + MM'_C \in \mathcal{S}$ .
  - ( $\Rightarrow$ ) Suppose internal stability with C and let  $C = \frac{N_C}{M_C}$  its coprime factorization. Define  $V := (NN_C + MM_C)^{-1} \in \mathcal{S}$  (Lem. 1-1). Define Q s.t.,  $M_CV =: Y - NQ \rightarrow \text{put this into } NN_CV + MM_CV = 1$  $\rightarrow NN_CV + M(Y - NQ) = 1$ . Also, from NX + NY = 1, N(X + NQ) = 1MQ) + M(Y - NQ) = 1. Thus,  $N_C V = X + MQ$ , and  $C = \frac{N_C V}{M_C V} = 1$  $\frac{X+MQ}{Y-NQ}$ . Further,  $(NX+MY)Q=YN_CV-XM_CV\to Q\in\mathcal{S}$ .

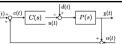
# Q-Parameterization: Example

• Th. 1-4: Consider unit feedback system with possibly unstable  $P = \frac{N}{M}$ where  $N, M \in \mathcal{S}$  are coprime with  $NX + MY = 1, X, Y \in \mathcal{S}$ . Then, all the stabilizing controllers are given by

$$C = rac{X + MQ}{Y - NQ}, \quad orall Q \in \mathcal{S}$$

- S = M(Y NQ) and T = N(X + MQ), still all affine in Q.
- (Ex) Design control C for  $P = \frac{1}{(s-1)(s-2)}$  s.t., (1) CL system internally stable; (2) perfect step tracking when d = 0; and (3) rejection of sinusoid disturbance d of 10 rad/s.
  - From Th. 1-4,  $C = \frac{X+MQ}{Y-NQ}$  with  $Q \in \mathcal{S}$ . With this,  $H_{r\to y} = T = N(X+MQ)$  and  $H_{d\to y} = N(Y-NQ)$ .
  - Step tracking/stabilization (dc-gain):  $H_{r\to y}(0) = 1 = N(0)[X(0) + y]$  $M(0)Q(0) = 1 \rightarrow Q(0) = 6.$
  - Disturbance rejection (zero at jw):  $H_{d\to u}(jw) = 0 = N(10j)[Y(10j) N(10j)Q(10j)] \rightarrow Q(10j) = -94 + 70j.$
  - Three constraints for  $Q: Q = a + \frac{b}{s+1} + \frac{c}{(s+1)^2} \to C(s)$ .

# Pole Placement Control



• Unit feedback system: 
$$P = \frac{N_P}{M_P}$$
,  $C = \frac{N_C}{M_C}$  coprime polynomials  $N_{\star}$ ,  $M_{\star}$ , 
$$\begin{pmatrix} e \\ y \\ u \end{pmatrix} = \frac{1}{N_P N_C + M_P M_C} \begin{bmatrix} M_P M_C & -N_P M_C & -M_P M_C \\ N_P N_C & N_P M_C & -N_P N_C \\ N_C M_P & -N_P N_C & -N_C M_P \end{bmatrix} \begin{pmatrix} r \\ d \\ n \end{pmatrix}$$

• Pole-placement control: given desired CL-CE F(s) and  $N_P(s)$ ,  $M_P(s)$ , design  $N_C(s), M_C(s)$  s.t.,

$$N_C(s)N_P(s) + M_C(s)M_P(s) = F(s)$$

- From Bezout's identity, there exist coprime polynomials X, Y s.t.,  $N_PX + M_PY = 1$  iff  $N_P, M_P$  are coprime.
- Thus, we can always find a solution  $\bar{N}_C(s) = F(s)X(s), \bar{M}_C(s) =$ F(s)Y(s) iff  $N_P, M_P$  are coprime (may not stabilizing though).
- General solution:  $N_C(s) = \bar{N}_C(s) + M_P(s)Q(s), M_C(s) = \bar{M}_C(s)$  $N_P(s)Q(s)$  for any polynomial Q(s) with

$$C(s) = rac{X(s)F(s) + M_P(s)Q(s)}{Y(s)F(s) - N_P(s)Q(s)}$$

implying that unit-feedback cannot alter open- $H_{r \to y} = \frac{N_C N_P}{N_P N_C + M_P M_C}$  implying that unit-feedback cannot alter open loop zeros  $\to$  two-DOF control for model (i.e., pole-zero) matching.

# Pole Placement via Linear Algebra

• Pole-placement control: given desired CL-CE F(s) and  $N_P(s), M_P(s)$ , design  $N_C(s), M_C(s)$  s.t.,

 $N_C(s)N_P(s) + M_C(s)M_P(s) = F(s)$ 

• Given  $\deg(M_P) = n$ , choose  $\deg(M_C) = m$  to match  $\deg(F) = n + m$ . The larger m is (i.e., more DOFs), the more likely can match F(s). Then,

$$N_P(s) = a_{p0} + a_{p1}s + ...a_{pn}s^n$$
,  $M_P(s) = b_{p0} + b_{p1}s + ...b_{pn}s^n$ ,  $N_C(s) = a_{c0} + a_{c1}s + ...a_{cm}s^m$ ,  $M_C(s) = b_{c0} + b_{c1}s + ...b_{cm}s^m$ ,  $F(s) = f_0 + f_1s + ...f_{n+m}s^{n+m}$ 

with  $a_{pn} = 0$ ,  $b_{pn} \neq 0$ ,  $b_{cm} \neq 0 \rightarrow f_{n+m} = b_{pn}b_{cm} + a_{pn}a_{cm} = b_{pn}b_{cm} \neq 0$ .

• The pole-placement can then be written by linear equation (Chern Ch.9):

$$[S_m]_{(n+m+1)\times 2(m+1)}[b_{co};a_{co};...;b_{cm};a_{cm}]=[f_o;...;f_{n+m}]$$

which has a solution if  $2(m+1) \ge n+m+1$  due to the structure of  $S_m$ .

• Th. 1-5: For the above problem, for any F(s) with  $\deg(F) = n + m$ , there always exist a proper  $C = \frac{N_C}{M_C}$  with  $\deg(C) = m$ , if  $m \ge n - 1$ .

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# Example: Chen 9.2

- For  $P = \frac{s-2}{s^2-1}$ , design a proper compensator C so that y can track any step response.
  - 1. Choose desired CL poles:  $F(s) = (s+2)(s^2+2s+2) = s^3+4s^2+6s+4$ .
  - 2. Set up linear pole-placement equation:

$$\begin{bmatrix} -1 & -2 & 0 & 0 \\ 0 & 1 \\ 1 & 0 & 0 \\ \hline 0 & 0 & 1 \\ 1 & 0 \end{bmatrix} \begin{pmatrix} b_o \\ a_o \\ b_1 \\ a_1 \end{pmatrix} = \begin{pmatrix} 4 \\ 6 \\ 4 \\ 1 \end{pmatrix}$$

with  $S_m$  full column rank (i.e., solution exists if  $m \ge n - 1$ ).

- 3. Compute the control law:  $C = -\frac{22s+23}{3s+34}$ .
- 4. Analyze dc-tracking performance:  $T(0) = \frac{P(0)C(0)}{1+P(0)C(0)} = \frac{23}{6}$ .
- 5. Design pre-compensator  $P(s) = \frac{6}{23}\bar{P}(s), \ \bar{P}(s) \in \mathcal{S}$  with P(0) = 1.

# **Internal Model Control**

• Suppose we want to track a class of reference signal r(t) while rejecting a class of disturbance d(t), given by

particle d(t), given by  $r(s) = \frac{\hat{N}_r(s)}{D_r(s)}, \quad d(s) = \frac{\hat{N}_d(s)}{D_d(s)}$ 

where  $D_r, D_d$  are **known**, yet,  $\hat{N}_r, \hat{N}_d$  unknown (e.g., any ramp  $r(s) = \frac{\hat{N}_r(s)}{s^2}$ , any step+sinusoid with known  $w d(s) = \frac{\hat{N}_d(s)}{s(s^2+w^2)}$ ).

- Incorporate this information of r, d into the controller to achieve robust tracking and disturbance rejection  $\rightarrow$  internal model control.
- Define  $\phi(s)$  to be least common denominator of unstable (i.e., non-vanishing) poles of r(s), d(s) (e.g.,  $\phi = s^2(s^2 + w^2)$ ) and incorporate into control.
- $H_{d \to y}(s) = \frac{N_P M_C \phi}{N_P N_C + M_P M_C \phi} \to y(s) = \frac{N_P M_C \phi}{N_P N_C + M_P M_C \phi(s)} \frac{\hat{N}_d(s)}{D_d(s)} = \frac{N_P M_C \phi(s)}{F(s)} \frac{\hat{N}_d(s)}{D_d(s)}$   $\to \text{ with CL-CE } F(s) \text{ Hurwitz}, \ y(\infty) = \lim_{s \to 0} s \frac{N_P M_C \phi(s)}{F(s)} \frac{\hat{N}_d(s)}{D_d(s)} = 0.$
- $H_{r \to e}(s) = \frac{M_P M_C \phi}{N_P N_C + M_P M_C \phi} \to e(s) = \frac{M_P M_C \phi}{N_P N_C + M_P M_C \phi(s)} \frac{\hat{N}_r(s)}{D_r(s)} = \frac{M_P M_C \phi(s)}{F(s)} \frac{\hat{N}_r(s)}{D_r(s)}$   $\to \text{ with CL-CE } F(s) \text{ Hurwitz, } e(\infty) = \lim_{s \to 0} s \frac{M_P M_C \phi(s)}{F(s)} \frac{\hat{N}_r(s)}{D_r(s)} = 0.$

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# **Pole Placement for IMC**

• Suppose we want to track a class of reference signal r(t) while rejecting a class of disturbance d(t), given by

$$r(s)=rac{\hat{N}_r(s)}{D_r(s)}, \ \ d(s)=rac{\hat{N}_d(s)}{D_d(s)}$$

• CL pole placement with IMC:

$$N_P(s)N_C(s) + M_P(s)M_C(s)\phi(s) = F(s)$$

where  $N_P, M_P$  are coprime. Thus, if no common zeros between  $\phi(s)$  and  $N_P(s), N_P$  and  $M_P\phi$  are coprime  $\to$  from **Th.1-5**,  $\exists$  a proper  $C(s) = \frac{N_C}{M_C}$  if  $m \ge n + n_d - 1$ ,  $n = \deg(P)$ ,  $n_d = \deg(\phi)$ ,  $m = \deg(C)$ .

- A simplest form of IMC is I-control to reject constant disturbance.
- (Chen Ex 9.3) Design control for  $P = \frac{s-2}{s^2-1}$  to track any step reference.
  - IMC  $\phi(s) = s$ . Then,  $m \ge 2 \to \deg[F(s)] = 5$ .
  - Choose  $F(s) = (s+2)(s^2+4s+5)(s^2+2s+5)$ .
  - Solve LA to obtain  $N_C, M_C$ :  $C(s) = \frac{N_C(s)}{M_C(s)\phi(s)} = -\frac{96.3s^2 + 118.7s + 25}{s(s^2 + 127.3)}$ .
  - $-T(0) = \frac{P(0)C(0)}{1+P(0)C(0)} = 1$  (de-tracking).  $H_{d\to y}(0) = \frac{P(0)}{1+P(0)C(0)} = 0$

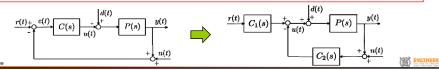
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 $\sum_{t=0}^{t} n(t)$ 

# Model Matching Control

- So far, we have been working on CL pole placement control (i.e., F(s)), yet, CL system behavior also (possibly severely) depends on zeros as well.
- Model matching problem: given P, design a proper C s.t., the CL behavior  $H_{r\to u}(s)$  matches with a desired TF  $H_o(s)$ .
- Unit feedback can do arbitrary pole-placement F(s), yet, not general model matching  $H_o(s) \to \text{FF}$  pre-compensator  $C_1 + \text{FB}$  control  $C_2$ .
- Even with two-DOF control, not arbitrary  $H_o(s)$  implementable due to OL-dynamics P(s) (e.g.,  $H_{r\to y} = \frac{N_P N_1}{N_P N_2 + M_P M_2}$  w/ unmovable zeros of P).
- Th. 1-6: Consider  $P = \frac{N_P}{M_P}$ . Then,  $H_o(s) = \frac{E(s)}{F(s)}$  is implementable iff:
  - 1. CL-CE F(s) is Hurwitz.
  - 2.  $\deg(F) \deg(E) \ge \deg(M_P) \deg(N_P)$  (relative degree can't decrease).
  - 3. All zeros of  $N_P(s)$  with zero/positive real parts are also zeros of E(s)(non-minimum phase zeros not removable).



# **Model Matching Procedure**

Given  $P = \frac{N_P}{M_P}$  and  $H_o = \frac{E}{F}$  (satisfying **Th. 1-6**), design proper  $C_1 = \frac{N_1}{M_C}$  and  $C_2 = \frac{N_2}{M_C}$  s.t.,  $H_{r \to y}(s) = H_o(s)$  while ensuring CL internal stability.

- 1. Define coprime  $\bar{E}, \bar{F}$  s.t.,  $\frac{H_o}{N_P} = \frac{\bar{E}}{FN_P} = \frac{\bar{E}}{\bar{F}}$ .

   May attempt  $H = \frac{PC_1}{1+PC_2} = \frac{N_PN_1}{M_PM_C+N_PN_2} = H_o = \frac{\bar{E}N_P}{\bar{F}} \rightarrow N_1 = \bar{E}$  and solve  $M_C, N_2$  from  $M_PM_C + N_PN_2 = \bar{F}$  with coprime  $N_P, M_P$  $\rightarrow$  typically, deg( $\bar{F}$ ) not enough (e.g., improper  $C_2$  w/ HO P).
- 2. Augment  $\bar{F}(s)$  w/ a Hurwitz  $\hat{F}(s)$  s.t.,  $\deg(\bar{F}\hat{F}) \geq 2n-1$ ,  $n = \deg(M_P)$ .
- 3. Rewrite  $H_o=\frac{\bar{E}\hat{F}N_P}{\bar{F}\hat{F}}=H=\frac{N_PN_1}{M_PM_C+N_PN_2},$  and choose/solve for  $N_2,M_C$ :

$$N_1(s) = ar{E}(s)\hat{F}(s), \quad M_P(s)M_C(s) + N_P(s)N_2(s) = ar{F}(s)\hat{F}(s)$$

which has a solution  $N_2, M_C$ , since  $N_P, M_P$  coprime and  $m = \deg(M_C) =$  $\deg(\bar{F}\hat{F}) - \deg(M_p) \ge n - 1$  (cf. **Th. 1-5**).

4.  $C_2 = \frac{N_2}{M_C}$  is proper (**Th. 1-5**) Also, for  $C_1 = \frac{\bar{E}\hat{F}}{M_C} = \frac{N_1}{M_C}$ , from item 2 of **Th. 1-6**,  $\deg(\bar{F}\hat{F}) - \deg(\bar{E}\hat{F}N_P) \ge \deg(M_P) - \deg(N_P) \to \deg(\bar{E}\hat{F}) = \deg(N_1) \le \deg(\bar{F}\hat{F}) - \deg(M_P) = n + m - n = m \to C_1$  also proper.

# Model Matching: Example 9.8 Chen

- (Ex 9.8) Given  $P = \frac{s-2}{s^2-1}$ , match  $H_o(s) = \frac{-(s-2)(4s+2)}{(s+2)(s^2+2s+2)}$ .
  - $H_o(s)$  ensures step and ramp tracking:  $H_o=rac{-4s^2+6s+4}{s^3+4s^2+6s+4}$
  - $-H_o(s)$  is implementable:
  - Compue  $\bar{E}(s), \bar{F}(s)$ :  $\frac{H_o}{N_P} = \frac{\bar{E}}{\bar{F}} = \frac{-(4s+2)}{s^3+4s^2+6s+4}$ .
  - Compute  $\hat{F}(s)$ :  $\deg(\bar{F}\hat{F}) \geq 2n-1=3 \rightarrow \hat{F}=1$ .
  - Compute  $C_1$  via direct substitution and Bezout:  $N_1=-(4s+2)$ ,  $M_C=s+34/3 \rightarrow C_1=\frac{-(4s+2)}{s+34/3}$ .
  - Compute  $C_2$  via Bezout:  $C_2 = \frac{-(22s+23)}{3s+34}$ .

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# Nominal Performance

• We have considered tracking of step, ramp or sinusoid sign freq. Yet, in practice, often, need to track signals with fr



- For tracking, ideally, we want  $||S||_{\infty} = \frac{e}{r} \equiv 0 \rightarrow \text{impossib}$
- Nominal performance:  $||W_pS||_{\infty} < 1$ , where  $W_p(s) \in S$  is frequency-dependent performance weight:
  - $-W_p(s) = \frac{100}{(s+1)^3}$  for tracking up to 1rad/s with error less than 1/100.
  - $-W_p(s) = \frac{s/M + w_B^*}{s + w_B^*A}$  for CL bandwidth  $w_B^*$  (w/  $|W_p(jw_B^*)| \approx 1$ ), LF track  $|e_{ss}| < A < 1$  and HF max peak of |S(jw)| < M (stability).
- Signal norm of  $u(t) \in \Re$ :
  - Defining properties: (1)  $||u|| \ge 0$ ; (2) ||u|| = 0 iff  $u(t) \equiv 0$ ; (3)  $||au|| = |a| \cdot ||u||$ ; (4)  $||u + v|| \le ||u|| + ||v||$ .
  - 2-norm:  $||u||_2 := \sqrt{\int_{-\infty}^{+\infty} |u(t)|^2 dt}; \infty$ -norm  $||u||_{\infty} := \sup_t |u(t)|,$
- System norm of H(s) (or h(t)):
  - -\[ \infty\]-norm:  $||H||_{\infty}:=\sup_{w}|H(jw)|$  with  $||y||_{2}=||H||_{\infty}||u||_{2}$  (via Parseval's identity); if  $u(t)=A\sin wt$ ,  $w\in\Re$ ,  $||y||_{\infty}=||H||_{\infty}A$ .

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# **Multiplicative Uncertainty**

- So far, we have rather neglected uncertainty in system, we only degrade performance but also destabilize CL system.
- Goal: maintain CL-system stability (i.e., robust stability)
- Multiplicative uncertainty: plant perturbed by uncertainty given by

$$\tilde{P} = (1 + W_u \Delta) P, \quad \frac{\tilde{P} - P}{P} = W_u \Delta \Rightarrow \left| \frac{\tilde{P}(jw) - P(jw)}{P(jw)} \right| \leq |W_u(jw)|$$

where  $W_u \in \mathcal{S}$  is uncertainty weight (e.g., large for HP),  $\Delta \in \mathcal{S}$  is (any) nominal uncertainty with  $||\Delta||_{\infty} < 1$  w/o unstable pole cancelation of P (i.e., allowable uncertainty).

- Given scattered gain/phase measurements  $(M_{ik}, \phi_{ik})$  at  $w_i$ :  $\left|\frac{M_{ik}e^{j\phi_{ik}}}{M_ie^{j\phi_i}} 1\right| < |W_u(jw_i)|$
- Nominal  $P(s)=1/s^2$  perturbed by delay up to  $\tau=0.1s$  with  $\tilde{P}=e^{-\tau s}\frac{1}{s^2}\Rightarrow |\frac{\tilde{P}}{P}-1|=|e^{-j\tau w}-1|<|W_u(jw)|$  for  $W_u=\frac{0.21s}{0.1s+1}$ .

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# **Robust Stability**

- Given P and uncertainty information  $W_u$ , design control law C to ensure robust stability for any plant  $\tilde{P} \in \mathcal{P} := \{(1 + W_u \Delta)P\}$ .
- Th. 1-7: Assume C stabilizes nominal plant P, with PC and  $\tilde{P}C$  having same number of RHP-poles. Then, C provides robust stability for any plant  $\tilde{P} \in \mathcal{P}$  iff  $||W_uT||_{\infty} < 1$ , where  $T = \frac{PC}{1+PC}$ .
  - (⇐) Convert feedback diagram and apply small-gain theorem.
  - ( $\Rightarrow$ ) Suppose not, i.e.,  $|W_u(jw)T(jw)| \ge \gamma \ge 1$ . Consider Nyquist plot of  $\tilde{L} = \tilde{P}C$  w.r.t. (-1,0)-point:

$$1 + \tilde{L} = 1 + \tilde{P}C = (1 + L)(1 + \Delta W_u T)$$

where  $1+L\neq 0$  with # of encirclement same as OP RHP-poles. Then, can find  $\Delta$  w/  $|\Delta(jw)|=1/\gamma\leq 1$ ,  $\angle\Delta(jw)=-\pi+\angle W_uT$  at  $w\to \tilde{L}$  touches  $(-1,0)\to \text{unstable}\to \text{contradiction}$ .

• Small gain theorem: Let  $H_1$  and  $H_2$  be (possibly nonlinear) stable systems with finite IO-gains  $||H_1||, ||H_2||$ . If  $||H_1|| \cdot ||H_2|| < 1$ , their feedback system is also stable.

# **Robust Performance**

• Given P and uncertainty  $W_u$ , design control C to ensure **robust stability** and **robust performance** for any plant  $\tilde{P} \in \mathcal{P} := \{(1 + W_u \Delta)P\}$ , i.e.,

$$||W_uT||_{\infty} < 1, \quad ||W_p\tilde{S}||_{\infty} < 1, \quad \forall \tilde{P} \in \mathcal{P}$$

where 
$$\tilde{S} = \frac{1}{1+\tilde{L}} = \frac{1}{1+\tilde{P}C}$$
. Then,  $||W_p \tilde{S}||_{\infty} = \left\|\frac{W_p S}{1+\Delta W_u T}\right\|_{\infty} < 1$ 

• Th. 1-8: A necessary and sufficient condition for RP (also RS) is

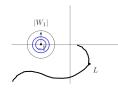
$$|| |W_p S| + |W_u T| ||_{\infty} < 1$$

- $\begin{array}{l} -\ (\Leftarrow)\ ||W_uT||_{\infty} < 1\ \text{obvious. Also,}\ |||W_pS| + |W_uT|||_{\infty} < 1 \to |W_pS| + \\ |W_uT| < 1 \to \frac{|W_pS|}{1 |W_uT|} < 1 \to \left\|\frac{W_pS}{1 + \Delta W_uT}\right\|_{\infty} \leq \left\|\frac{|W_pS|}{1 |W_uT|}\right\|_{\infty} < 1. \end{array}$
- $\ (\Rightarrow) \ \left\| \frac{W_p S}{1 + \Delta W_u T} \right\|_{\infty} < 1 \rightarrow \left\| \frac{W_p S}{1 |W_u T|} \right\|_{\infty} \le \left\| \frac{W_p S}{1 + \Delta W_u T} \right\|_{\infty} < 1.$
- Maximum tolerable uncertainty  $||\Delta||_{\infty} \leq \beta$ :
  - RS:  $1 + \tilde{L} = (1 + L)(1 + \Delta W_u T) \rightarrow |\beta W_u T| < 1 \rightarrow \beta < \frac{1}{||W_u T||_{\infty}}$
  - $\text{ RP: } \left\| \frac{W_p S}{1 + \Delta W_u T} \right\|_{\infty} < 1 \rightarrow \left| \frac{W_p S}{1 \beta |W_u T|} \right| < 1 \rightarrow \beta < \left\| \frac{1 |W_p S|}{W_u T} \right\|_{\infty}.$

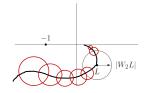
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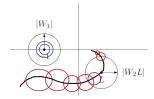
# **Graphical Representations**



• Nominal performance:  $||W_p S||_{\infty} < 1$  $|1 + L(jw)| > |W_p(jw)|, \forall w$ 



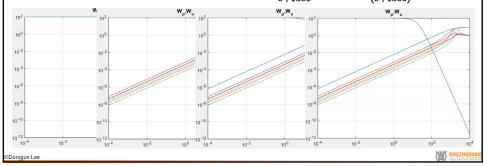
• Robust stability:  $||W_uT||_{\infty} < 1$  $|1 + L(jw)| > |W_u(jw)L(jw)|, \ \forall w$ 



• Robust performance:  $|||W_pS| + |W_uT|||_{\infty} < 1$  $|1 + L(jw)| > |W_p(jw)| + |W_u(jw)L(jw)|, \ \forall w$ 

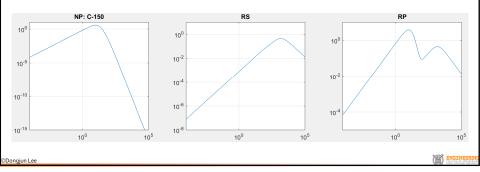
# **RP Design Example**

- Nominal plant (integrator):  $P = \frac{1}{s}$ .
- Perturbed plant:  $\tilde{P} = \frac{1}{s} \frac{k}{ms^2 + bs + k} \rightarrow \frac{\tilde{P} P}{P} = \frac{s^2 + 2\zeta w_n s}{s^2 + 2\zeta w_n s + w_n^2}$ .
- Performance specification:  $w_c = 20 \text{rad/s}$ , dc-tracking error  $\leq 1\%$
- Design performance weight function:  $W_p = \frac{100w_c^5}{(s+w_c)^5}$ .
- Uncertainty:  $w_n \in [200, 500]$ Hz,  $\zeta \in [0.1, 0.5]$ .
- $\bullet$  Design uncertainty weight function:  $W_u = \frac{9s}{s+1500} \to W_u = \frac{9s(s+200)}{(s+1500)^2}$



# **RP Design Example**

- Perturbed plant:  $\tilde{P} = \frac{1}{s} \frac{k}{ms^2 + bs + k}$ .  $W_u = \frac{9s(s + 200)}{(s + 1500)^2}$ ,  $W_p = \frac{100}{(\frac{9}{20} + 1)^5}$ .
- $w_n \in [200, 500]$ Hz,  $\zeta \in [0.1, 0.5]$ ;  $w_c = 20$ rad/s, dc-tracking error  $\leq 1\%$ .
- Design P-control C(s) = K s.t.,  $|||W_p S| + |W_u T|||_{\infty} < 1$ .
  - Design K for NP: K = 600 and check RS and RP.
  - Decrease K improves RS, yet, degrades RP  $\rightarrow$  P-control can't satisfy both (or reduce performance  $W_p$  or improve system-ID  $W_u$ --).
  - More complicated control  $\rightarrow$  Loop shaping.



# **Loop Shaping**

ullet Loop shaping: graphical technique to shape loop-transfer function L(s)=P(s)C(s) to satisfy RP and internal stability:

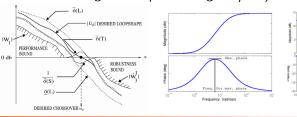
$$|||W_{p}S| + |W_{u}T|||_{\infty} < 1$$

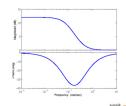
if such control law exists, where  $S = \frac{1}{1+L}$  and  $T = \frac{L}{1+L}$ .

 $\bullet\,$  Typically, L should have large gain in LF (for performance) and low gain in HF (for robust stability). More precisely, basic loop shaping condition:

$$|L(jw)| > \frac{|W_p|}{1-|W_u|} \text{ (LF)}, \quad |L(jw)| < \frac{1-|W_p|}{|W_u|} \text{ (HF)}$$

• Simplest loop shaping: lead (to increase  $w_c$  while improving PM) and lag (to increase tracking in LF w/o affecting GM/PM).





# **Loop Shaping: Derivation**

• Stability-performance trade-off: A necessary condition for L(s) to satisfy RP is

$$\min\{|W_p(jw)|, |W_u(jw)|\} < 1, \quad \forall w \ge 0$$

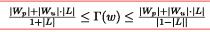
can't allow high performance 8 high uncertainty at same band

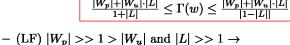
i.e., both  $|W_n|$  and  $|W_n|$  can't be larger than 1 at the same time.

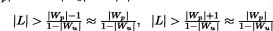
- Suppose not, i.e., at w',  $|W_p| \ge 1$  and  $|W_u| \ge 1$ . WLG, assume  $|W_p| \geq |W_u|$ . Then, at w',

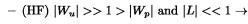
$$|W_p S| + |W_u T| \ge |W_u S| + |W_u T| = |W_u S| + |W_u (1 - S)| \ge |W_u| \ge 1$$

• Now, define  $\Gamma(w):=\frac{|W_p|}{|1+L|}+\frac{|W_uL|}{|1+L|}$ . Then, RP iff  $\Gamma(w)<1\ \forall w$ . Also,









$$|L| < \tfrac{1 - |W_p|}{|W_u| - 1} \approx \tfrac{1 - |W_p|}{|W_u|}, \quad |L| < \tfrac{1 - |W_p|}{1 + |W_u|} \approx \tfrac{1 - |W_p|}{|W_u|}$$

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# **Loop Shaping: Conditions**

- We assume stable/minimum-phase P(s) (i.e., no RHP pole/zero) to avoid unstable/non-minimum-phase L = PC (general case  $\rightarrow$  Doyle Ch. 8).
- A necessay condition for  $W_p, W_u$ :  $\min\{|W_p(jw)|, |W_u(jw)|\} < 1$ .
- HF roll-off of L should be at least as fast as that of P (proper C).
- Slop of |L| at crossover frequency  $w_c$  (i.e.,  $|L(jw_c)|$ ) should be as gentle as possible (-20[dB/dec] to -40[dB/dec]).
- Loop shaping is based on approximation: NS, RP shoul be checked a posterior (RP assumes RS, RS assumes NS).
- Bode's gain formula: For a non-minimum phase stable L with all positive coefficients, its phase  $\angle L(jw_o)$  is uniquely given by: with  $\nu := \ln(w/w_o)$ ,

$$\angle L(jw_o) = rac{1}{\pi} \int_{\infty}^{+\infty} rac{d \ln |L|}{d 
u} \ln rac{|w+w_o|}{|w-w_o|} d 
u$$

where, if constant slop  $\frac{d \ln |L|}{d \nu} = c$  at  $w_o$ ,  $\angle L(j w_o) = -\frac{c \pi}{2}$ .

- The stiffer the slop of  $|L(jw_c)|$  is, the less the PM is.
- For system w/ RHP-zeros, phase angle larger than minimum angle above.

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# **Loop Shaping: Procedure**

1. For stable/minimum-phase nominal plant P, design  $W_p$ ,  $W_u$  s.t.,

$$\min\{|W_p|,|W_u|\}<1,\ \forall w$$

2. On  $(\log w, 20 \log |L|)$  plane, plot LF and HF bounds:

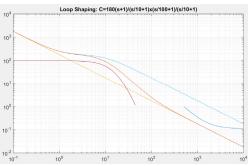
$$\frac{\left|W_{p}\right|}{1-\left|W_{u}\right|}, \quad \left|W_{p}\right| >> 1 > \left|W_{u}\right| \left(\text{LF}\right) \quad \frac{1-\left|W_{p}\right|}{\left|W_{u}\right|}, \quad \left|W_{u}\right| >> 1 > \left|W_{p}\right| \left(\text{HF}\right)$$

- 3. Construct a desired loop-TF candidate L = PC s.t.,
  - |L| is above (or below) the LF (or HF) bounds.
  - Roll-off of L at HF at least as fast as P.
  - Slop of L at crossover frequency as gentle as possible (<-40 dB/dec).
- 4. Check RP by observing if  $|W_p S| + |W_u T| < 1$ .
- 5. Check NS by ensuring roots of 1 + L(s) = 0 in LHP.
- 6. Determine the controller C(s) = L(s)/P(s).

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# **Loop Shaping: Example**

- Nominal plant;  $P = \frac{1}{s}$ . perturbed plant:  $\tilde{P} = \frac{1}{s} \frac{k}{ms^2 + bs + k}$ .
- $w_n \in [200, 500]$ Hz,  $\zeta \in [0.1, 0.5]; w_c = 20 \mathrm{rad/s},$  dc-tracking error  $\leq 1\%$
- Do loop shaping to design C(s) s.t.,  $|||W_pS| + |W_uT|||_{\infty} < 1$ .
  - Check L(s) with C(s) = 180, i.e.,  $L(s) = 180\frac{1}{s} \rightarrow \text{violate LF bound.}$
  - Shape L(s) w/ Lead  $\to$   $L(s) = 180\frac{1}{s}\frac{s+1}{\frac{s}{10}+1}$   $\to$  violate HF bound.
  - Shape L(s) w/ Lag  $\rightarrow L(s) = 180\frac{1}{s}\frac{s+1}{\frac{s}{10}+1}\frac{\frac{s}{100}+1}{\frac{s}{10+1}} \rightarrow \text{LF/HF}$  bounds satisfied

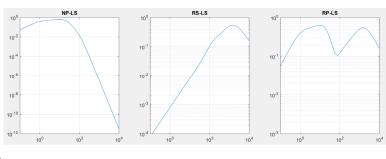


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# **Loop Shaping: Example**

- Nominal plant;  $P = \frac{1}{s}$ . perturbed plant:  $\tilde{P} = \frac{1}{s} \frac{k}{ms^2 + bs + k}$ .
- $W_u = \frac{9s(s+200)}{(s+1500)^2}$ ,  $W_p = \frac{100}{(\frac{s}{20}+1)^5}$ .
- Do loop shaping to design C(s) s.t.,  $|||W_pS| + |W_uT|||_{\infty} < 1$ .
  - With  $C(s)=180\frac{s+1}{\frac{8}{10}+1}\frac{\frac{8}{100}+1}{\frac{8}{10+1}}\to \text{LF/HF}$  bounds satisfied.
  - Check RP via Bode plot of  $|W_pS| + |W_uT|$ .
  - Check NS via CL CE: CL poles  $-0.9925, -99.5023 \pm 90.45j$ .



# Waterbed Effects

• Bode's integral theorem for sensitivity function: Suppose L(s) has relative-degree  $\geq 1$ , has M RHP poles  $p_i$  (with  $Re(p_i) > 0$ ), and k = $\lim_{s\to\infty} sL(s)$  (e.g., k=0 if relative-degree  $\geq 2$ ). Then,

$$\int_0^\infty \ln |S(jw)| dw = -k rac{\pi}{2} + \pi \cdot \sum_{i=1}^M \mathrm{Re}(p_i)$$

- If you "push down" |S(jw)| at some frequency-band, it is lead into "swelling-up" at another frequency-band.
- Overall level of |S(jw)| will increase if open-loop system is unstable with fast poles (difficult to stabilize).
- Bode's integral theorem for complementary sensitivity function: Suppose L(s) has at least 1 pole at 0, M RHP zeros  $z_i$ , with  $k_v =$  $\lim_{s\to 0} sL(s)$  (e.g.,  $k_v = \infty$  if type  $\geq 2$ ). Then,

$$\int_0^\infty rac{1}{w^2} \ln |T(jw)| dw = -rac{\pi}{2k_v} + \pi \cdot \sum_{i=1}^M rac{1}{z_i}$$

- Waterbed effect for |T(jw)|.
- Overall level of |T(jw)| increase w/RHP-zeros (bad for performance).



# **RHP Poles and RHP Zeros**

• Effect of combined RHP zeros/poles: Suppose L(s) has  $N_z$  RHPzeros  $z_j$  and  $N_p$  RHP-poles  $p_i$ . Then,  $\forall j=1,..,N_z, i=1,..,N_p$ ,

$$||W_p S||_{\infty} \geq \prod_{i=1}^{N_p} \frac{|z_j + \bar{p}_i|}{|z_j - p_i|} |W_p(z_j)|, \quad ||W_u T||_{\infty} \geq \prod_{i=1}^{N_p} \frac{|\bar{z}_j + p_i|}{|z_j - p_i|} |W_p(z_j)|$$

- It would be extremely difficult to control if RHP-pole and RHP-zero are close with each other (i.e., unstable mode nearly uncontrollable/unobservable).
  - 1. Suppose L(s) has RHP-pole p and RHP-zero z. Then, S(p) =0, T(p) = 1 and S(z) = 1, T(z) = 0.
  - 2. Then, we can write  $S = S_{ap}S_{mp} = \frac{s-p}{s+p}S_{mp}$  where  $S_{ap}$  is all-pass with  $|S_{ap}(jw)| = 1$  and  $|S_{mp}(jw)| = |S(jw)|$ . Further,  $|S_{mp}(z)| =$  $|S(z)|/|S_{zp}(z)| = |\frac{z+p}{z-p}|.$
  - 3. Moreover, from maximum modulus theorem,

$$||W_pS||_{\infty} = \sup_w |W_p(jw)S_{mp}(jw)| \geq \sup_{\mathrm{Re}(s) \geq 0} |W_p(s)S_{mp}(s)|$$

$$|| \geq W_p(z) || S_{mp}(z) | = |W_p(z)| \frac{|z+p|}{|z-p|}$$

 $\geq W_p(z)||S_{mp}(z)| = |W_p(z)|\frac{|z+p|}{|z-p|}$  • If  $W_p = W_u = 1$ ,  $||S||_{\infty} \geq \frac{|z+p|}{|z-p|}$  and  $||T||_{\infty} \geq \frac{|z+p|}{|z-p|}$ , again, shows difficulty of control.

# **Bandwidth Limitation with RHP Poles/Zeros**

- RHP-poles typically require aggressive/fast control to stabilize. RHP-zeros typically require non-aggressive/slow control due to inverse response.
- Effect of RHP zeros: approximate bound for the open-loop bandwidth  $w_B$  of L(s) is given by

$$w_B pprox w_C \le egin{cases} |z|/4 & ext{if } \operatorname{Re}(z) >> \operatorname{Im}(z) \ |z|/2.8 & ext{if } \operatorname{Re}(z) = \operatorname{Im}(z) \ |z| & ext{if } \operatorname{Re}(z) << \operatorname{Im}(z) \end{cases}$$



- CL-BW  $w_c$  is limited by z and should be slower w.r.t. z.
- RHP-zeros close to origin is bad.
- Effect of RHP-poles:  $w_C \approx w_B > 2p$ , i.e. should be fast enough to stabilize RHP-pole.
- It would be extremely difficult to control system w/ RHP-poles and RHP-zeros close with each other; w/ slow RHP-zeros and fast RHP-poles.
- Ex) inverted pendulum:  $G_1(s) = \frac{-g}{s^2(Mls^2 (M+m)g)}$  and  $G_2(s) = \frac{ls^2 g}{s^2(Mls^2 (M+m)g)}$  (e.g., short/light rod, small m/M, large m).