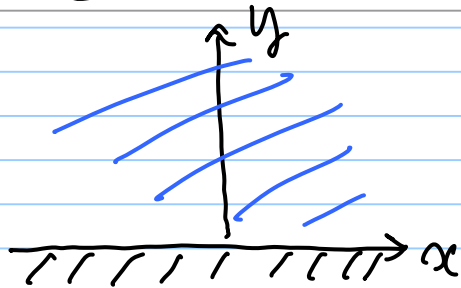


③ Stokes's First Problem (Rayleigh Prob.)

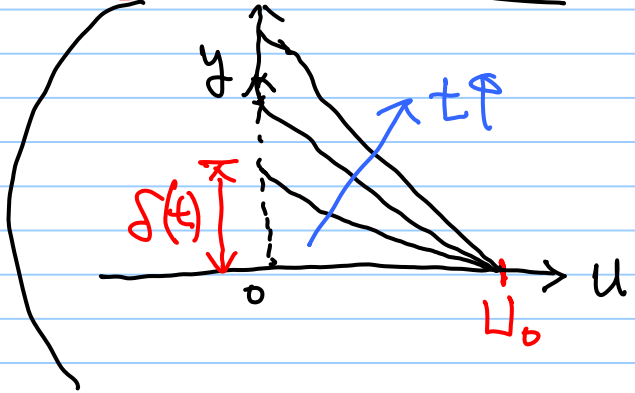


$u = 0$ @ $t = 0$
 impulsely moving plate : $u = U_0$ (for $t > 0$)
 $\hookrightarrow u = u(y), \frac{\partial \rho}{\partial x} = 0$

$\Rightarrow \frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2}$

w/ BC's $u(y, t) : \text{finite}$

$u(0, t) = \begin{cases} 0 & \text{for } t \leq 0 \\ U_0 & \text{for } t > 0 \end{cases}$



$\delta(t) : \text{penetration depth}$

↓ simple scaling analysis : $\frac{U_0}{t} \sim \nu \frac{U_0}{\delta^2} \Rightarrow \underline{\delta \sim \sqrt{\nu t}}$.

: this problem describes a viscous diffusion process

boundary-layer \rightarrow (transfer of x -moment into y -direction)

like behavior . \rightarrow as ν increases, the penetration depth becomes thicker.

• Wall-shear stress, $\tau_w = \mu \frac{\partial u}{\partial y} \sim \frac{\mu U_0}{\delta} \sim \frac{\mu U_0}{\sqrt{\nu t}}$.

$$\therefore \frac{\tau_w \delta}{\mu U_0} \sim 1.$$

* Similarity solution .

• shape of vel. profile is similar at all times.

(PDE \rightarrow ODE)

indep. var. : y, t . \rightarrow dep. var. $u(y, t)$.

$$\hookrightarrow \frac{u(y, t)}{U_0} = f(\eta)$$

\hookrightarrow dimensionless similarity variable.

$$\eta = \alpha \cdot \frac{y}{t^n}$$

put η into the gov. eq.

$$\textcircled{*} \rightarrow -U_0 \cdot n \cdot f' \cdot \eta \cdot t^{-n} = 2\nu \cdot U_0 \cdot f'' \cdot \alpha^2 \cdot t^{-2n} \rightarrow \text{we want this to be ODE.}$$

to make ' η ' dimensionless & for convenience, need $n = 1/2$.

$$\text{we choose } \alpha^2 = \frac{1}{4\nu} \Rightarrow \eta = \frac{y}{2\sqrt{\nu t}}$$

resulting ODE : $f'' + 2\eta \cdot f' = 0$.

$$\ln f' = -\eta^2 + \ln A, \quad f' = A e^{-\eta^2}$$

$$\therefore f(\eta) = A \int_0^{\eta} e^{-\xi^2} d\xi + B. \quad f \in u/W_0.$$

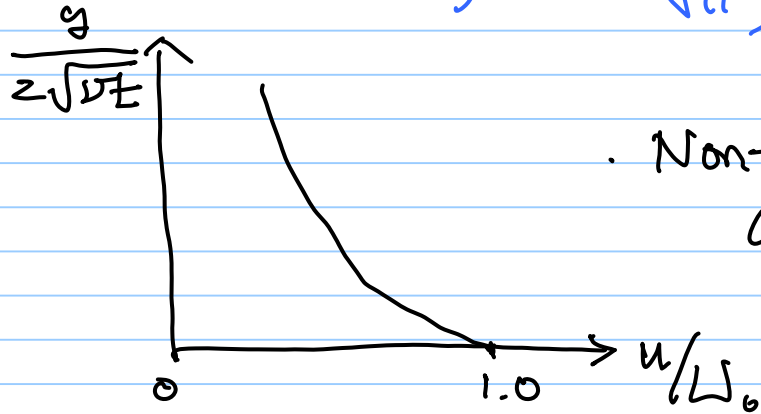
• BC : $u(0, t) = W_0 \quad (t > 0) \rightarrow f(0) = 1 \rightarrow B = 1$.

$$\left(\eta = \frac{y}{2\sqrt{\alpha t}} \right)$$

• IC : $u(y, 0) = 0 \rightarrow \lim_{\eta \rightarrow \infty} f = 0 \rightarrow A = -\frac{2}{\sqrt{\pi}}$.

$$\therefore \frac{u(y, t)}{W_0} = f(\eta) = 1 - \frac{2}{\sqrt{\pi}} \int_0^{\eta} e^{-\xi^2} d\xi = 1 - \operatorname{erf}(\eta) = 1 - \operatorname{erf}\left(\frac{y}{2\sqrt{\alpha t}}\right)$$
$$\underbrace{\hspace{10em}} = \operatorname{erfc}\left(\frac{y}{2\sqrt{\alpha t}}\right).$$

$$\operatorname{erf}(x) \equiv \frac{2}{\sqrt{\pi}} \int_0^x e^{-t^2} dt, \quad 1 - \operatorname{erf}(x) \equiv \operatorname{erfc}(x).$$

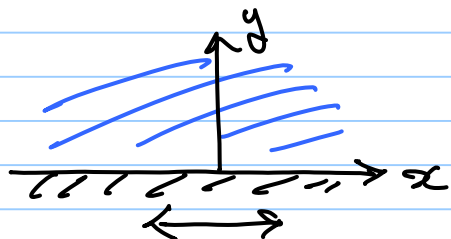


• Non-dimensionalized vel. profile
collapsed to one curve \leftrightarrow similarity.

* penetration depth behaves like a boundary layer

agree with the scaling analysis \leftarrow $\left(\begin{array}{l} \frac{u}{W_0} = 0.01 \rightarrow \delta = 3.6 \sqrt{\nu t} \\ \text{wall shear, } \tau_w = \frac{\mu W_0}{\sqrt{\pi \nu t}} \end{array} \right.$

⊕ Stokes' Second Prob : oscillating wall.



$$u(0,t) = U_0 \cdot \cos(nt)$$

$$= \text{Re}[W_0 e^{int}]$$

↳ real part,

$$e^{int} = \cos(nt) + i \sin(nt)$$

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2}$$

($u(y,t)$ has a finite value)

Assume, $u(y,t) = \text{Re}[w(y) \cdot e^{int}]$ → into mtrn eq.

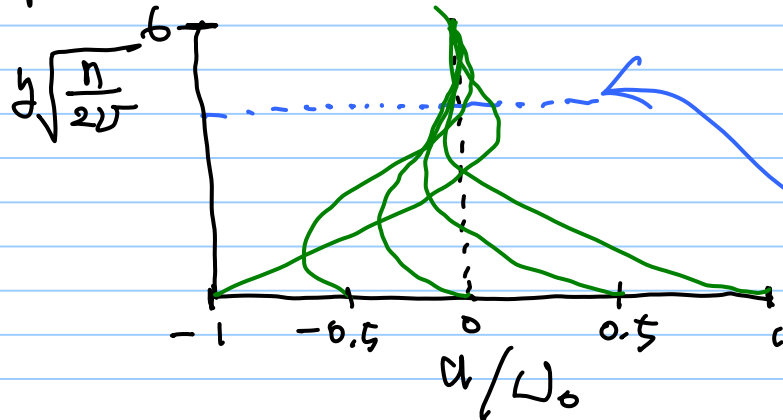
$$\Rightarrow w(y) = A \cdot \exp\left\{- (1+i) \sqrt{\frac{\nu}{2}} y\right\} + B \cdot \exp\left\{(1+i) \sqrt{\frac{\nu}{2}} y\right\}$$

then, $\frac{u(y,t)}{U_0} = \exp\left\{-\sqrt{\frac{\eta}{2\nu}} y\right\} \cdot \cos\left(\omega t - \sqrt{\frac{\eta}{2\nu}} y\right)$

i) exponentially \downarrow decay in y -direction

ii) phase shift between wall motion and fluid velocity by $\sqrt{\frac{\eta}{2\nu}} y$.

- e.g.) for instantaneous vel. profiles.



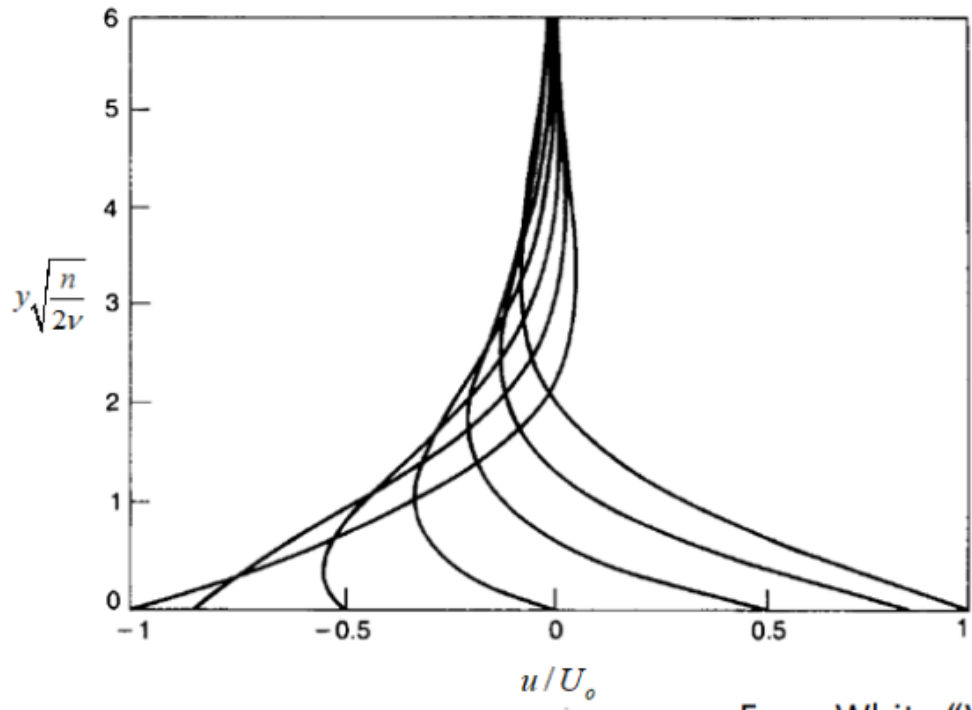
white "Viscous Fluid Flow"

· if $\exp\left\{-\sqrt{\frac{\nu}{2\nu}} y\right\} = 0.01$, or $\sqrt{\frac{\nu}{2\nu}} y = 4.6$

then, $\delta \doteq 6.4 \sqrt{\frac{\nu}{n}} \sim \sqrt{\frac{\nu}{n}}$.

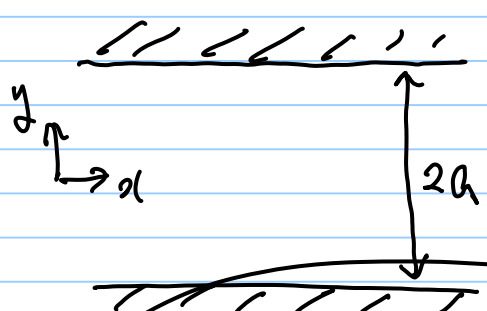
so, we can say

- i) low-frequency motion will be dominant in diffusing x -motion into y -direction.
- ii) small phase shift for low-freq. motion.



From White "Viscous Fluid Flow"

⑤ Pulsating flow between parallel surfaces.



1D, incompressible, $\sigma = \text{const}$.

$$\frac{\partial p}{\partial x} = P_x \cdot \cos(\omega t)$$

amp. of pressure-grad. oscillation

$$\frac{\partial v}{\partial t} = -\frac{\partial p}{\partial x} + \nu \frac{\partial^2 u}{\partial y^2}, \quad \text{BC's: } u(-a, t) = u(a, t) = 0.$$

$$\Rightarrow \frac{\partial p}{\partial x} = \text{Re} [P_x \cdot e^{i\omega t}] \longrightarrow u(y, t) = \text{Re} [w(y) e^{i\omega t}]$$

$$\therefore u(y, t) = \text{Re} \left[i \frac{P_x}{\rho \nu} \left\{ 1 - \frac{\cosh \left\{ (1+i) \sqrt{\frac{\nu}{2\omega}} y \right\}}{\cosh \left\{ (1+i) \sqrt{\frac{\nu}{2\omega}} a \right\}} \right\} e^{i\omega t} \right]$$

i) $u(y, t)$ oscillates @ frequency of n .

ii) \llcorner has a phase shift, function of y .

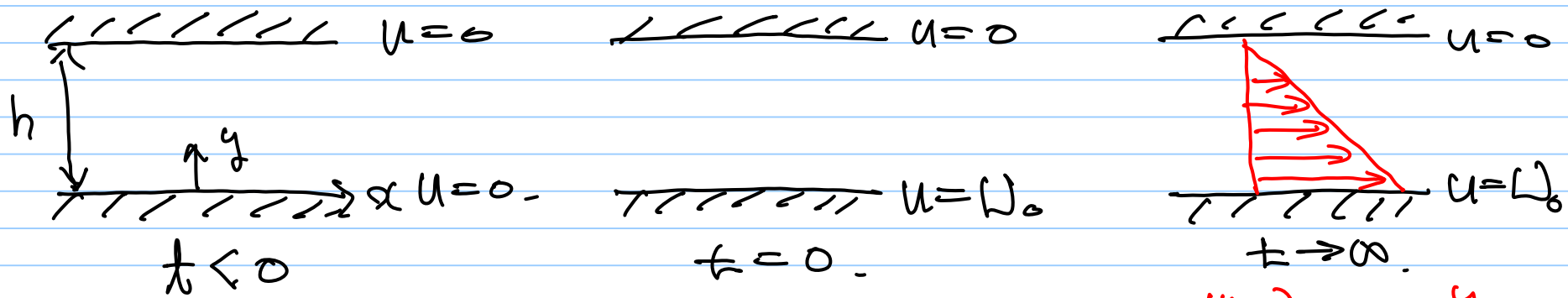
$\left. \begin{array}{l} \sqrt{\frac{n}{2\nu}} a \ll 1 \\ \sqrt{\frac{n}{2\nu}} a \gg 1 \end{array} \right\} : \text{quasi-steady.}$

$\sqrt{\frac{n}{2\nu}} a \gg 1 : \text{unsteady flow}$
① thin viscous region will ^{only} occur near the wall

② phase difference between pressure and velocity, pressure and shear.

③ phase difference between velocities at different y 's.

⑥ Transient (unsteady) flow for Couette flow formation.
 (lower wall moves, for convenience)



$$\frac{u(y, \infty)}{U_0} = 1 - \frac{y}{h}$$

Let's define, $w(y, t)$ as.

$$\frac{w(y, t)}{U_0} = \frac{u(y, t)}{U_0} - \left(1 - \frac{y}{h}\right) ; \text{ difference between } \hat{u} \text{ and final steady solution.}$$

$$t^* = \frac{25t}{h^2}, \quad y^* = \frac{y}{b}, \quad \frac{u}{u_0} = u^*, \quad \frac{w}{w_0} = w^*$$

$$\rightarrow w^* = u^* - (1 - y^*)$$

then, intm. conservation eq.

$$\frac{\partial w^*}{\partial t^*} = \frac{\partial^2 w^*}{\partial y^{*2}} \quad \left. \begin{array}{l} w^*(0, t^*) = 0 \\ w^*(1, t^*) = 0 \\ w^*(y^*, 0) = -1 + y^* \end{array} \right\}$$

sep. of variables

$$w^*(y^*, t^*) = f(y^*) \cdot g(t^*) \quad \therefore \frac{f''}{f} = \frac{g''}{g} = \text{const} = -\lambda^2$$

$$\rightarrow \left\{ \begin{array}{l} f = A \cdot \sin(\lambda y^*) + B \cdot \cos(\lambda y^*) \\ g = C \cdot \exp(-\lambda^2 t^*) \end{array} \right.$$

$$\begin{aligned}
 \omega^*(0, t^*) = 0 &\rightarrow B = 0. \\
 \omega^*(1, t^*) = 0 &\rightarrow \lambda = n\pi. \\
 \omega^*(y^*, 0) = \sum A_n \sin(n\pi y^*) = -1 + y^*. &\rightarrow f = A_n \sin(n\pi y^*)
 \end{aligned}$$

• Fourier orthogonality.

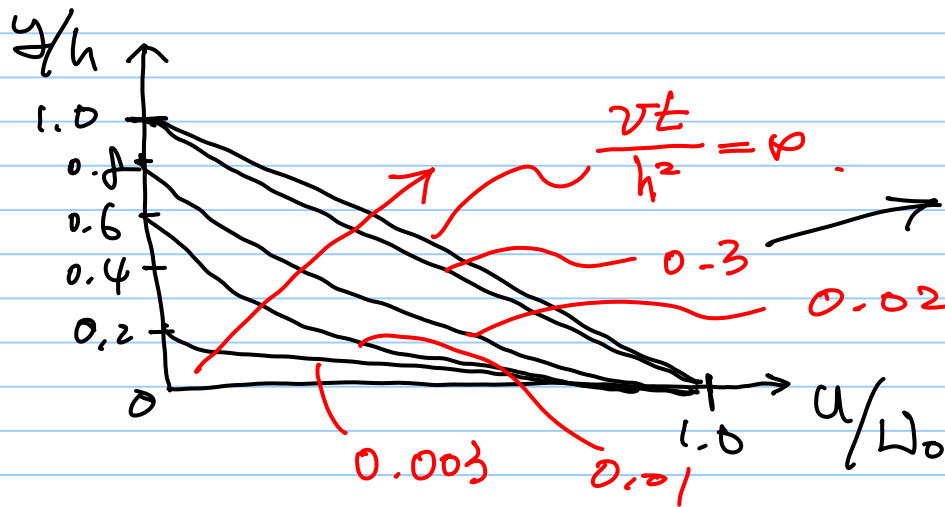
$$f(x) = a_0 + \sum_{n=1}^{\infty} \{ a_n \cos(n\pi x) + b_n \sin(n\pi x) \}$$

$$\rightarrow \int_0^{2\pi} \cos(n\pi x) \cdot \sin(m\pi x) dx = 0 \quad (m \neq n \neq 0)$$

$$\underline{\underline{A_n}} = \int_{-1}^1 (-1 + y^*) \cdot \sin(n\pi y^*) dy^* = \underline{\underline{-\frac{2}{n\pi}}}$$

$$\omega^* = u^* - (1 - y^*), \quad u^* = \omega^* + (1 - y^*)$$

$$\therefore \frac{u(y,t)}{U_0} = \left(1 - \frac{y}{h}\right) - \frac{2}{\pi} \sum_{n=1}^{\infty} \frac{1}{n} \exp\left(-n^2 \pi^2 \frac{\nu t}{h^2}\right) \cdot \sin \frac{n\pi y}{h}$$



if $h = 10 \text{ mm}$, air.
 $\rightarrow t \approx 2.0 \text{ sec}$

- upper plate prevents the return diffusion
 \rightarrow steady flow
- "n=1" component survives the longest.