

Last lecture

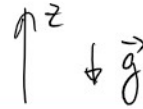
Flow lines \rightarrow streamlines, pathlines, streaklines

Forces on fluids $\left\{ \begin{array}{l} \text{surface} \rightarrow p, \tau \\ \text{body} \rightarrow \text{gravity} \end{array} \right.$

pressure \rightarrow absolute, gage, vacuum

Pascal's Principle

Hydrostatic formula \rightarrow incompressible $\rightarrow p(z)$
 water $\vec{v} = 0$



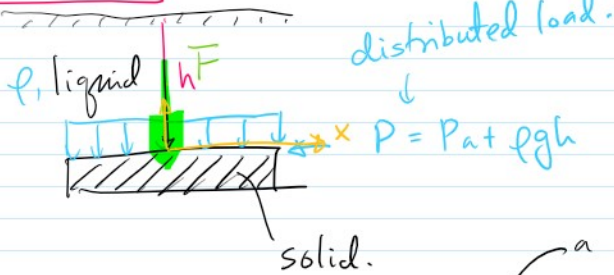
$p(z)$ for compressible fluid (atmosphere)

HW#2 2-5, 35, 68, 85 Due 9/29

No class on 9/21

FORCES ON PLANE SURFACES.

HORIZONTAL $P_a = P_{atm}$



distributed load.

$P = P_a + \rho gh$

Weight of the liquid above the plate

Force due to pressure on plate

Vertical force acting on solid surface = $P A = (P_a + \rho gh) A = F$

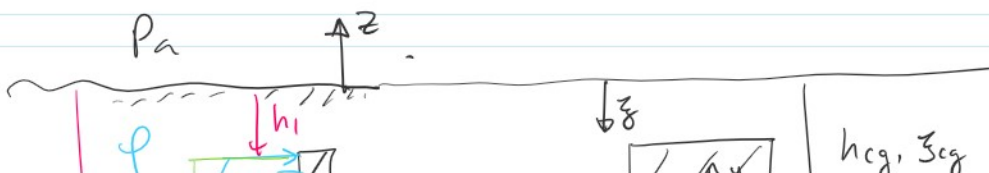
uniform pressure acting
 x_{cp} center of pressure = center of gravity.
 location where F acts.

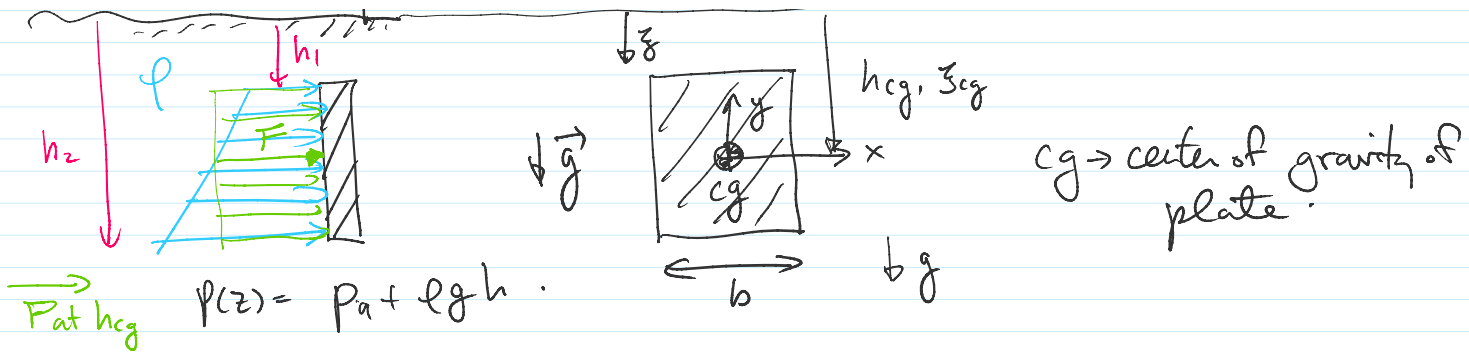
$x, y \rightarrow$ distance from center of gravity in x, y dir

$F x_{cp} = \int x p dA = 0 \rightarrow \underline{x_{cp} = 0}$

$F y_{cp} = \int y p dA = 0 \rightarrow \underline{y_{cp} = 0}$

VERTICAL SURFACE.





Assume $p_a = 0$

$$F = \left(\frac{\rho g h_1 + \rho g h_2}{2} \right) b (h_2 - h_1) = \rho g h_{cg} b (h_2 - h_1)$$

$$h_{cg} = \frac{h_2 + h_1}{2}$$

$F_{x_{cp}} = 0$ p uniform along x . \rightarrow $x_{cp} = 0$.

$$F_{y_{cp}} = \int y p dA = \int y \rho g (z_{cg} - y) dA.$$

$$z_{cg} = \frac{1}{A} \int z dA \quad \& \quad z = z_{cg} - y.$$

$$F_{y_{cp}} = -\rho g \int y^2 dA = -\rho g I_{xx}$$

$$I_{xx} = \frac{b(h_2 - h_1)^3}{12} \quad \text{for a rectangular plate.}$$

$$y_{cp} = -\frac{\rho g I_{xx}}{\rho_{cg} A}$$

Force on an inclined surface.

$$\delta = \frac{h}{\sin \theta} \quad z_{cg} = \frac{1}{A} \int z dA \quad \rightarrow \quad z_{cg} A = \int z dA$$

$$F = p_a A + \rho g z_{cg} \sin \theta A \rightarrow \int h dA = \int z \sin \theta dA = z_{cg} \sin \theta A.$$

$$F = p_a A + \rho g h_{cg} A \rightarrow \text{independent of shape of } A.$$

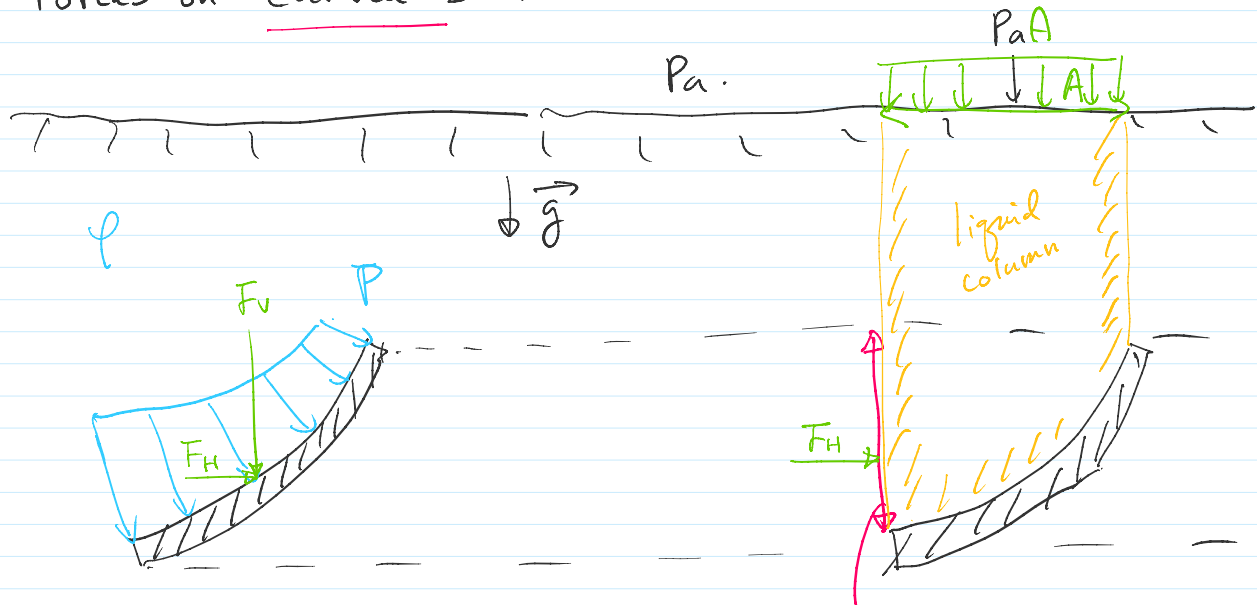
$$F_{y_{cp}} = \int y p dA$$

$$F_{x_{cp}} = \int x p dA$$

$$y_{cp} = \frac{-\rho g \sin \theta I_{xx}}{\rho g A}$$

$$x_{cp} = \frac{-\rho g \sin \theta I_{xy}}{\rho g A}$$

Forces on curved surfaces.



F_H is equivalent to the force on the vertical plane area formed by the projection of the curved surface onto the vertical plane.

vertical surface of projection of original surface onto the vertical axis

////// \rightarrow liquid column above the plate.

F_v equals the weight of the entire column of fluid above the curved plate.

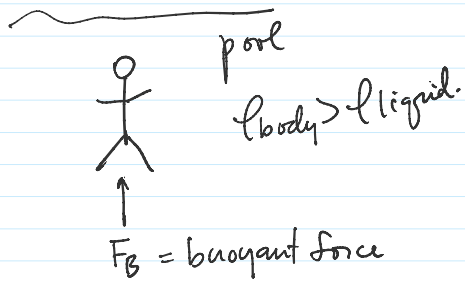
(in case of air-water combination, $\rho_{\text{water}} \sim 1000 \times \rho_{\text{air}}$)

Buoyancy \rightarrow Archimedes.

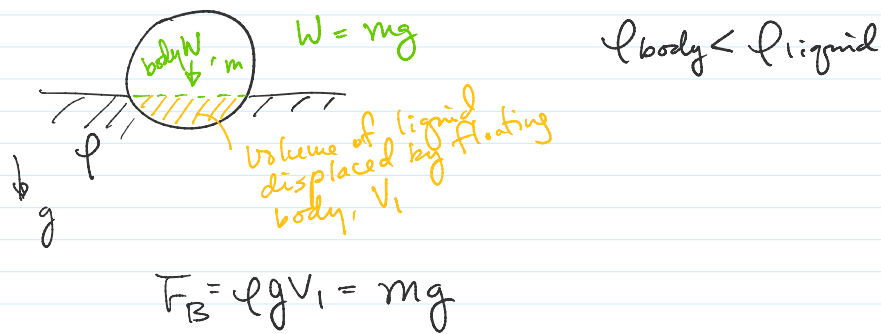
(e.g. person submerged in a pool)

1. A submerged body is subject to a buoyant force equivalent to the weight of the fluid displaced by the body

pool



2. A floating body displaces exactly the volume of fluid whose weight is equivalent to the body's weight.



$$F_B = \int (p_2 - p_1) dA$$

$$= - \int \rho g (z_2 - z_1) dA_H \quad \text{horizontal projected area.}$$

$$= \rho g \underbrace{(z_1 - z_2)}_{\text{Volume of body.}} dA_H$$

$$F_B = \rho g \text{Volume}_{\text{body.}}$$

↑
for a submerged body.

On 9/23

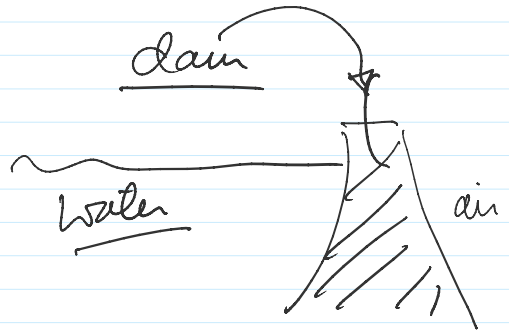
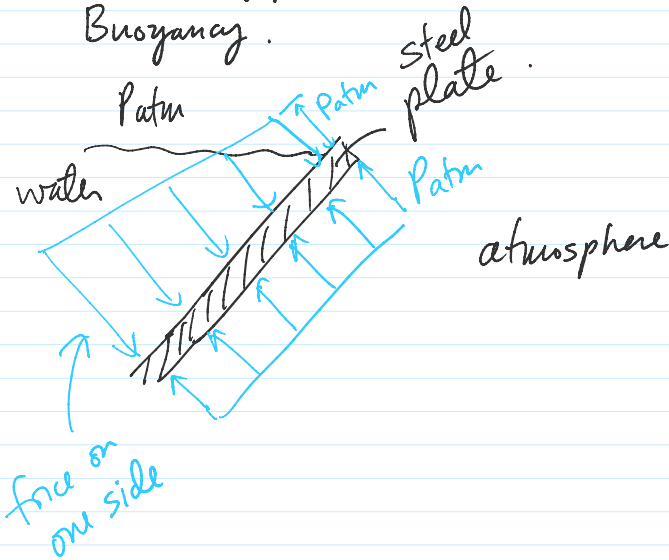
↓
 ρ in moving fluids

Last lecture

forces on $\left\{ \begin{array}{l} \text{plane surfaces - vertical} \\ \text{due to } p \quad \quad \quad \text{horizontal} \\ \quad \quad \quad \quad \quad \quad \quad \text{inclined} \\ \text{curved surfaces} \end{array} \right.$

$F, x_{cp}, y_{cp}, I_{xx}, \text{etc.}$

Buoyancy.



Pressure distribution in Rigid Body Motion

$\zeta = 0$
 $\mu \nabla^2 \vec{V} = 0$
 no deformation \rightarrow no shear force.

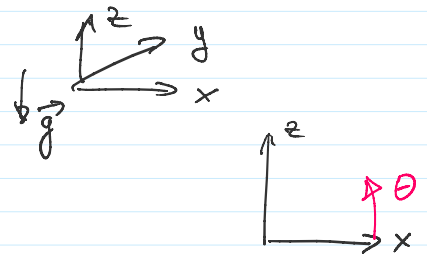
1. translation (linear)
2. rotation

Translation

Uniform linear \vec{a}

$\rho \vec{a} = -\nabla p + \rho \vec{g} + \mu \nabla^2 \vec{V}$ $\zeta = 0$
 $\nabla p = \rho(\vec{g} - \vec{a})$

Rigid body motion



x: $\rho dy dz - (p + \frac{\partial p}{\partial x} dx) dy dz = dm \cdot a_x \rightarrow -\frac{\partial p}{\partial x} = \rho a_x$

y: $-\frac{\partial p}{\partial y} = \rho a_y$

z: $\rho dx dy - (p + \frac{\partial p}{\partial z} dz) dx dy - g dm = dm \cdot a_z \rightarrow -\frac{\partial p}{\partial z} = \rho(g + a_z)$

$dp = \frac{\partial p}{\partial x} dx + \frac{\partial p}{\partial y} dy + \frac{\partial p}{\partial z} dz$

$$dp = \frac{\partial p}{\partial x} dx + \frac{\partial p}{\partial y} dy + \frac{\partial p}{\partial z} dz$$

1. Assume $a_y = 0$

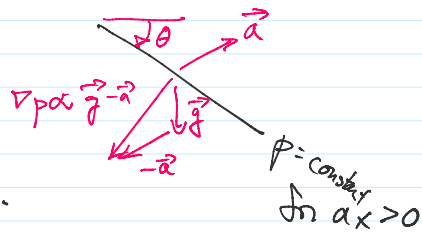
$$dp = -\rho a_x dx - \rho(g + a_z) dz.$$

isobar → surface of equivalent pressure → $p = \text{constant}$.
 ↑ ↑
 equal pressure line

$$dp = 0 = -\rho a_x dx - \rho(g + a_z) dz.$$

$p = \text{const.}$
 $p = \text{const.}$ $a_x < 0$
 $a_x = 0$

slope of the isobar $\frac{dz}{dx} \Big|_{p=\text{constant}} = \tan \theta = \frac{-a_x}{g + a_z}.$



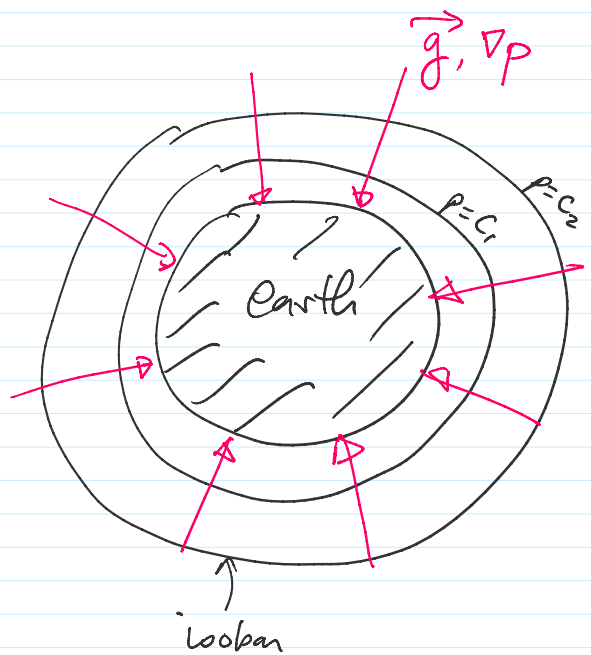
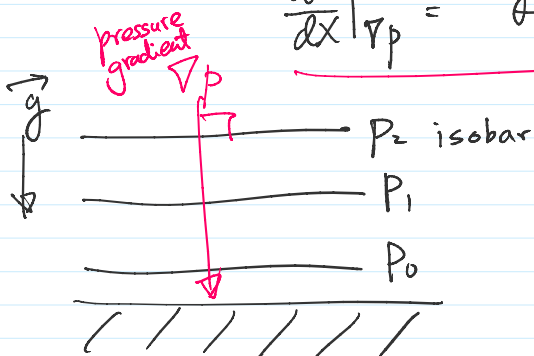
$$p = -\rho a_x x - \rho(g + a_z)z + C_0$$

B.C. at $x=0, z=0, p=p_0$

$$p = p_0 - \rho a_x x - \rho(g + a_z)z.$$

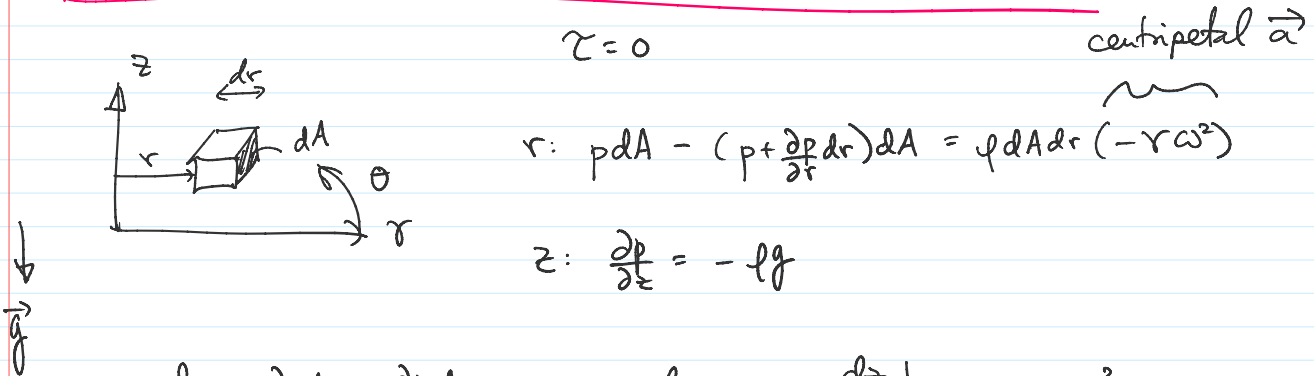
$\frac{dz}{dx} \Big|_{\nabla p}$ is perpendicular to $\frac{dz}{dx} \Big|_{p=\text{constant}}$.

$$\frac{dz}{dx} \Big|_{\nabla p} = \frac{g + a_z}{a_x}$$



Rigid Body Rotation (at constant $\omega = \text{angular velocity}$).

Rigid Body Rotation (at constant $\omega = \text{angular velocity}$).



$\tau = 0$ centrifetal \vec{a}

$$r: p dA - (p + \frac{\partial p}{\partial r} dr) dA = \rho dA dr (-r\omega^2)$$

$$z: \frac{\partial p}{\partial z} = -\rho g$$

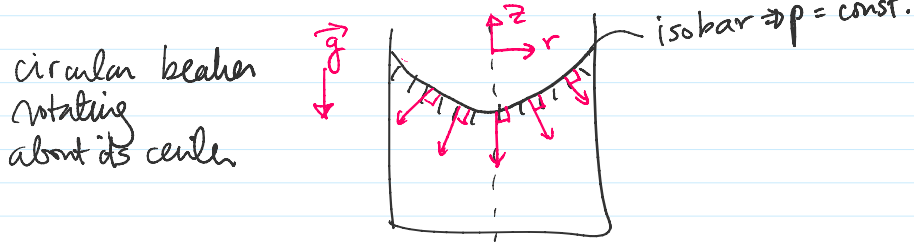
$$dp = -\frac{\partial p}{\partial r} dr + \frac{\partial p}{\partial z} dz \rightarrow dp = 0 \rightarrow \frac{dz}{dr} \Big|_{\text{isobar}} = \frac{\omega^2 r}{g}$$

$$p = \frac{\rho \omega^2 r^2}{2} - \rho g z + C$$

$$\frac{dz}{dr} \Big|_{\nabla p} = \frac{-g}{\omega^2 r}$$

B.C. @ $r=0$ & $z=0$, $p = p_0$.

$$p(r, z) = \frac{\rho \omega^2 r^2}{2} - \rho g z + p_0 \quad \text{isobar}$$



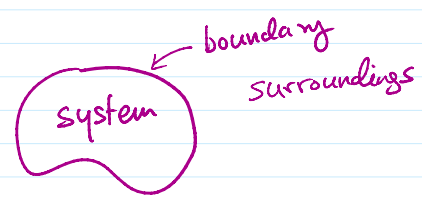
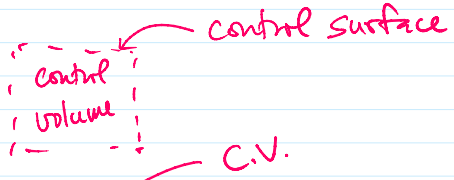
Ch. 3. Integral Control Volume Analysis. MOST IMPORTANT CHAPTER.

large scale
e.g. car compressor etc.

Space (not mass)
selected for analysis.

"system" (Lagrangian view)

"Control volume" (Eulerian view)

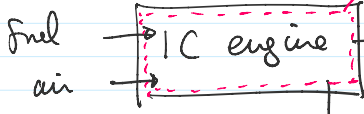


control volume = region or space selected for analysis.
C.S.
control surface separates the control volume from the rest.

system \equiv mass [selected for analysis] of interest
surroundings \equiv mass not included in the system
boundary separates system from surroundings.

Control volume from the rest.

interested in space



interested in the engine and NOT particular fluid particles

↑ apply conservation laws to C.V.

Reynolds Transport Theorem

Conservation Laws. (Lagrangian)

1. Conservation of mass
 $m_{sys} = \text{constant.}$
 $\frac{dm_{sys}}{dt} = \frac{Dm}{Dt} = 0$
2. Momentum conservation.
 $\frac{D(m\vec{v})}{Dt} = m\vec{a} = \sum \vec{F}$
3. Energy conservation (1st Law of Thermo)

$$\frac{DE}{Dt} - \dot{E} = \dot{Q} - \dot{W}$$

rate of change in system energy heat transferred from the surroundings to the system work done by the system

Reynolds Transport Theorem

