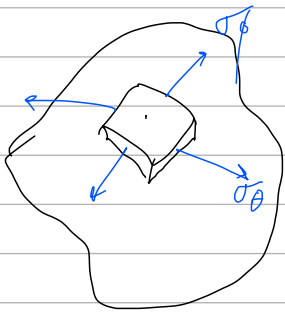


3.6. Thin-walled pressure vessels



$$\underline{\underline{\frac{p}{t} = \frac{\sigma_\theta}{r_\theta} + \frac{\sigma_\phi}{r_\phi}}}$$

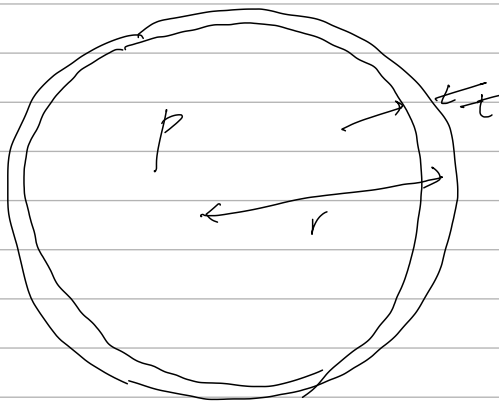
→ Membrane equation

$$N_\theta = \sigma_\theta \cdot t$$

$$N_\phi = \sigma_\phi \cdot t$$

$$p = \frac{N_\theta}{r_\theta} + \frac{N_\phi}{r_\phi}$$

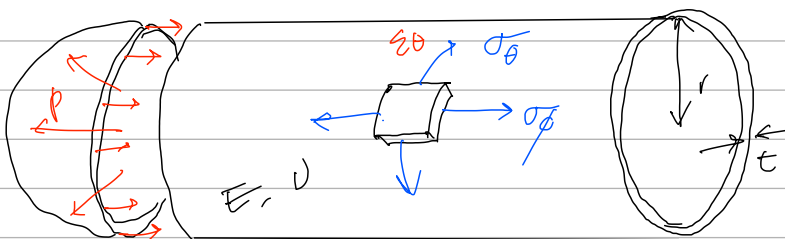
Spherical membrane



$$\frac{p}{t} = \frac{2\sigma}{r}$$

$$\rightarrow \sigma = \frac{pr}{2t}$$

Circular cylinder



σ_θ : hoop stress

σ_ϕ : axial stress

$$\frac{p}{t} = \frac{\sigma_\theta}{\cancel{r_\theta} r} + \frac{\sigma_\phi}{\cancel{r_\phi} \infty}$$

$$\sigma_\theta = \frac{pr}{t}$$

✓

$$2\pi r \cdot t \cdot \sigma_\phi = \pi r^2 p$$

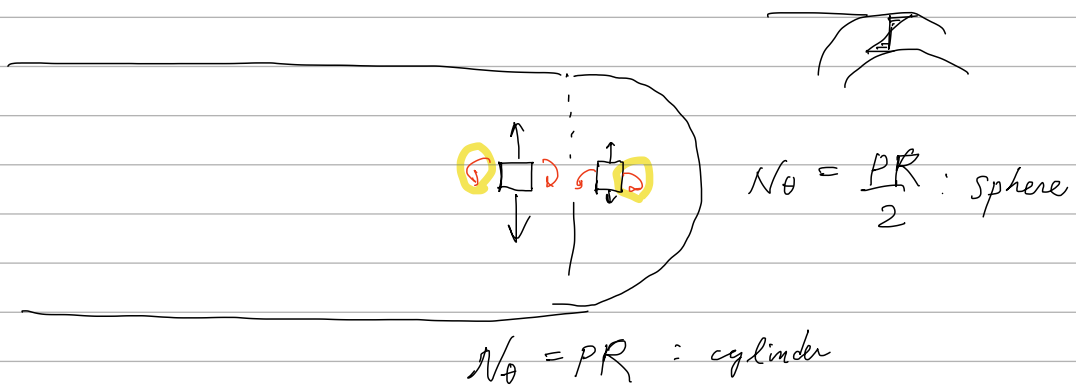
$$\sigma_\phi = \frac{pr}{2t}$$

What would be the change of radius when $p: 0 \rightarrow p$

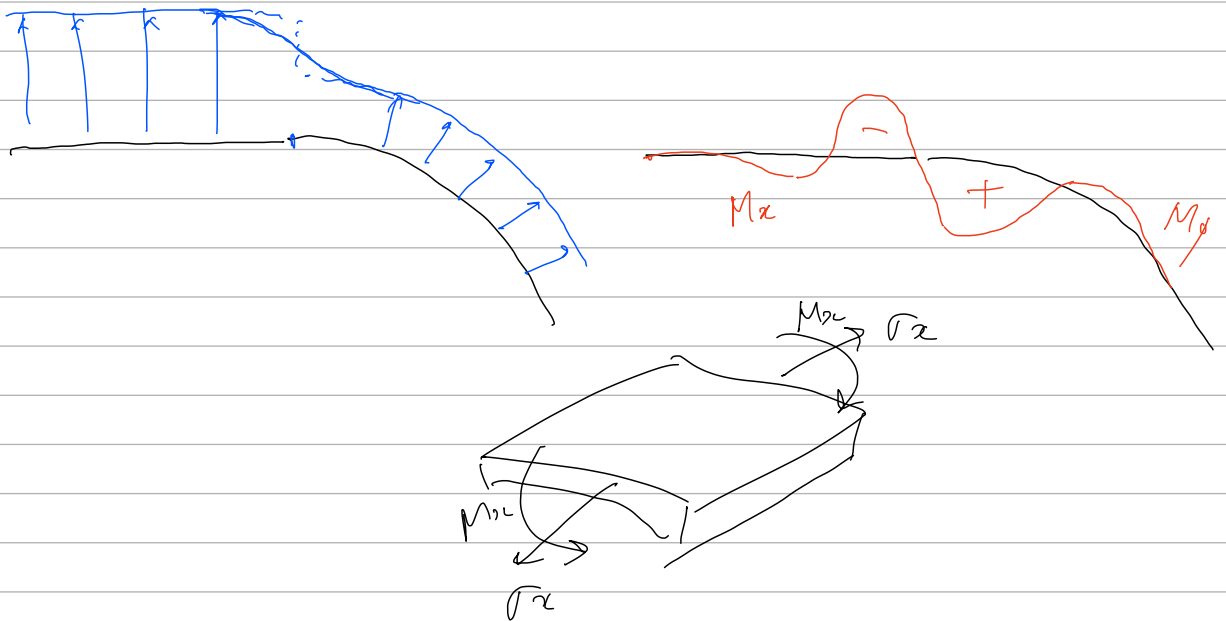
$$\begin{aligned} \epsilon_{\theta} &= \frac{\sigma_{\theta}}{E} - \nu \frac{\sigma_r}{E} \\ &= \frac{1}{E} \left(\frac{pr}{t} - \frac{\nu pr}{2t} \right) \\ &= \frac{pr}{Et} \left(1 - \frac{\nu}{2} \right) \end{aligned}$$

$$\frac{\Delta r}{r} = \epsilon_{\theta}$$

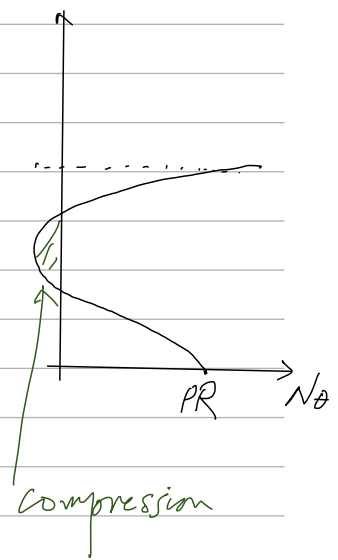
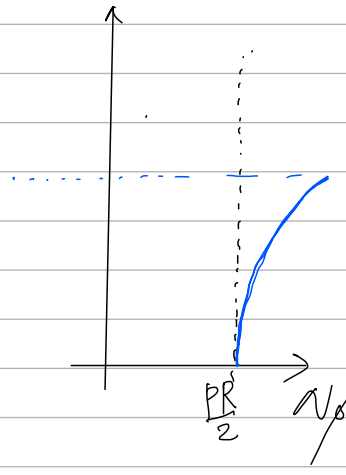
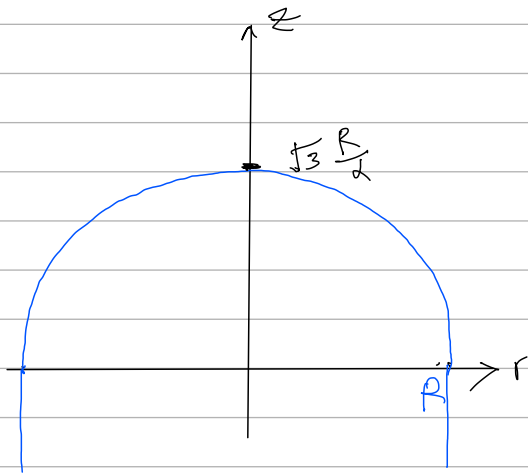
5-4.2. Boiler end



Sudden discontinuity in hoop stress
 \rightarrow induces an additional distribution of bending moments



Discontinuity in the hoop forces is caused by the sudden change in the curvature from $0 \rightarrow \frac{1}{R}$
 \rightarrow Second order continuous profile



$$(r^2 + \alpha^2 z^2)^2 + 2R^2(r^2 - \alpha^2 z^2) = 3R^4$$

$$r=0 : \alpha^4 z^4 - 2R^2 \alpha^2 z^2 - 3R^4 = 0$$

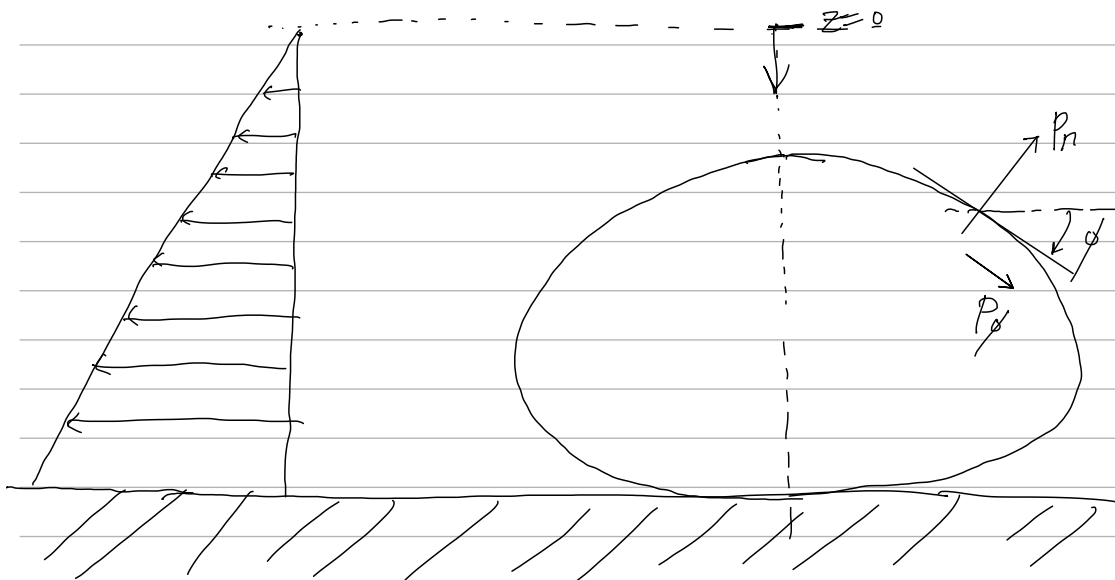
$$(\alpha^2 z^2 - 3R^2)(\alpha^2 z^2 + R^2) = 0$$

$$z = \sqrt{3} \frac{R}{\alpha}$$

Compression can be avoided by $\alpha < 1.9$



5.4.3. Drop-shaped tank



$$\underline{N_\phi} = \underline{N_\theta} = A$$

Fully stressed axisymmetric tank (echinodomes)

∴ shape identical to that of a drop of liquid resting on a plane surface

$$p_\phi = 0, \quad p_n = \rho g z$$

$$\frac{1}{R_\phi} + \frac{1}{R_\theta} = \frac{\rho g z}{A}$$

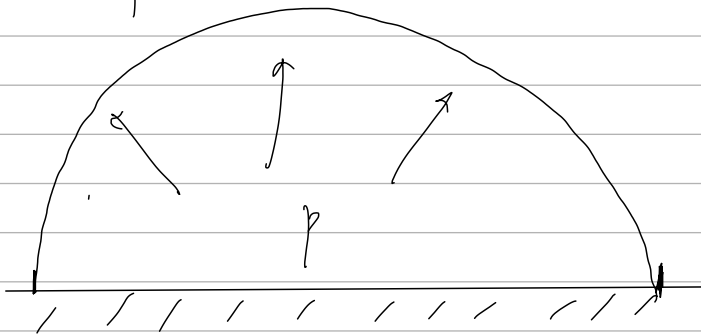
For axisymmetric profile $\frac{1}{R_\phi} = \frac{d \sin \phi}{dr}, \quad \frac{1}{R_\theta} = \frac{\sin \phi}{r}$
(From A-30, 39)

$$\frac{d \sin \phi}{dr} + \frac{\sin \phi}{r} = \frac{\rho g}{A} \cdot z, \quad \tan \phi = \frac{dz}{dr}$$

3.6. Pneumatic domes

∴ what is the best shape for an axisymmetric pneumatic dome with a vertical profile at the base?

- Hemisphere

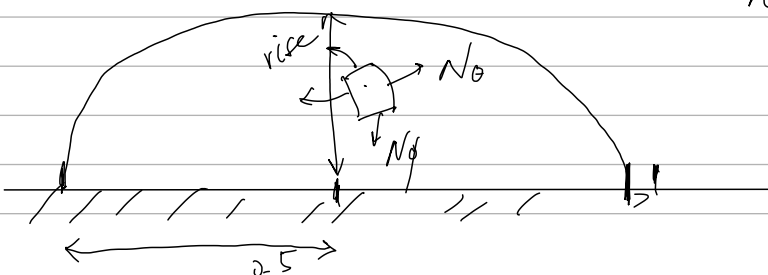


$$N_\phi = N_\theta = \frac{\rho R}{2}$$

→ Dead space at the top of the dome

Q. Shallowest possible shape that avoids the formation of wrinkles (compression)?

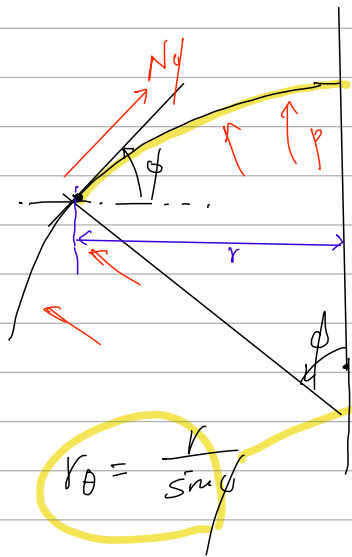
- Ellipsoid



$N_\phi > 0$ for all ellipsoid

N_θ can be negative if rise is less than

$$\frac{1}{2\sqrt{2}} = 0.35$$



$$\pi r^2 p = 2\pi r N_\phi \sin \phi$$

$$N_\phi = \frac{pr}{2 \sin \phi} = \frac{pr_0}{2}$$

$$\frac{N_\theta}{r_\theta} + \frac{N_\phi}{r_\phi} = p$$

$$\frac{N_\theta}{r_\theta} = p - \frac{N_\phi}{r_\phi} = p - \frac{pr_0}{2r_\phi}$$

$$= p \left(1 - \frac{r_\theta}{2r_\phi} \right)$$

$$= 0$$

$$r_\phi = \frac{r_\theta}{2}$$

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$$\frac{1}{r_\phi} = \frac{d \sin \phi}{dr} = \cos \phi \frac{d\phi}{dr}$$

$$\frac{1}{r_\theta} = \frac{\sin \phi}{r}$$

$$r_\phi = \frac{dr}{\cos \phi d\phi} = \frac{1}{2} \cdot \frac{r}{\sin \phi}$$

$$\frac{dr}{r} = \frac{d\phi}{2 \tan \phi}$$

Integration

$$\ln c \cdot r = \frac{1}{2} \ln \sin \phi$$

$$c \cdot r = \sqrt{\sin \phi} \rightarrow c^2 r^2 = \sin \phi$$

B.C. $r = r_0$ at $\phi = 90^\circ$

$$c^2 r_0^2 = \sin 90^\circ = 1, \quad c^2 = \frac{1}{r_0^2}$$

$$\frac{r^2}{r_s^2} = \sin^2 \theta$$

Parachutes

- tightly packed for transportation
- once deployed, form a stable structure with high drag

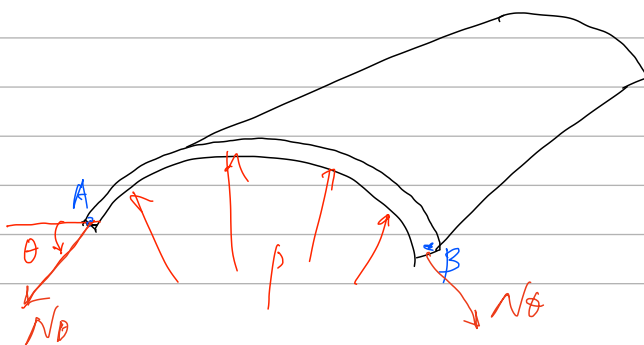
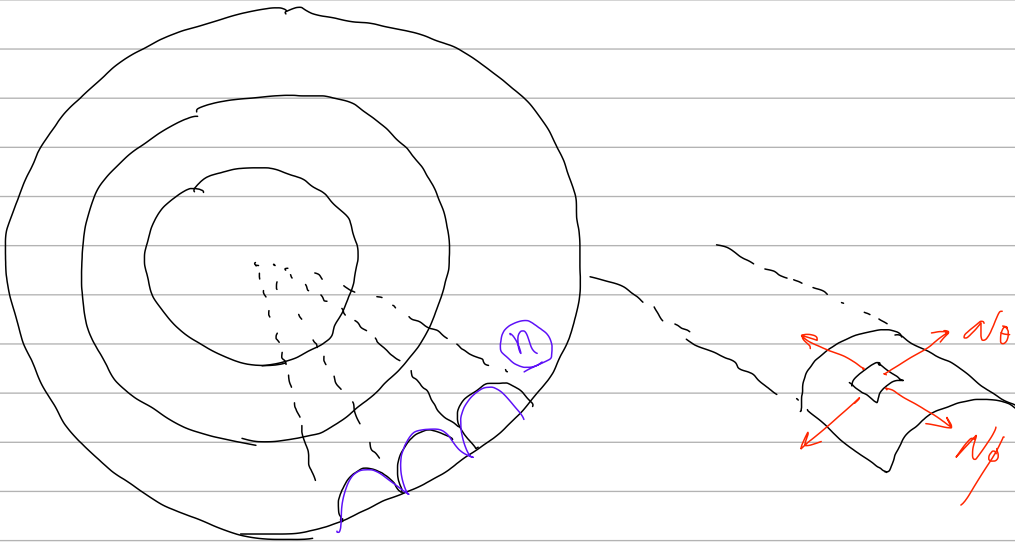
$$D = C_D \cdot \rho \cdot \frac{1}{2} v^2 \cdot A$$

\uparrow drag coefficient
 \uparrow projected area

A : increases structural mass

C_D : 0.5 ~ 1

→ compromise between A and C_D



$$p \overline{AB} = 2 N_\theta \sin \theta$$

$n \uparrow$
 $\theta \uparrow$) → reduce N_θ