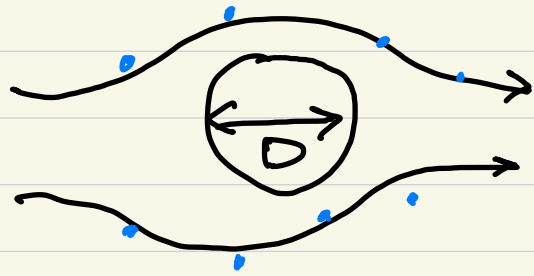


- Other forces on the particles in the non-uniform flow

① Effect of rarefied flow.



Knudsen number,

$$Kn = \frac{\lambda}{D} = \frac{\text{mean free path}}{\text{particle size}}$$

$$= \mu / c \rho_c$$

↳ speed of sound.

$$Kn = \frac{\mu}{c \rho_c D} = \frac{1}{c D / \nu_c} = \frac{1}{\underbrace{c D / \nu_c}_{\text{kinematic viscosity}} \underbrace{c}_{\text{charac. vel. scale}}} = \frac{Ma}{Re}$$

charac. vel. scale

- } $Kn < 10^{-3}$: continuum.
- } $10^{-3} < Kn < 0.25$: slip flow.
- } $0.25 < Kn < 10$: transitional flow

$Kn > 10$: free molecular flow

$$F_D = F_{D0} \cdot \frac{1}{1 + Kn \left\{ 2.49 + 0.84 \exp(-1.74/Kn) \right\}} \quad \text{: empirical Millikan (1923)}$$

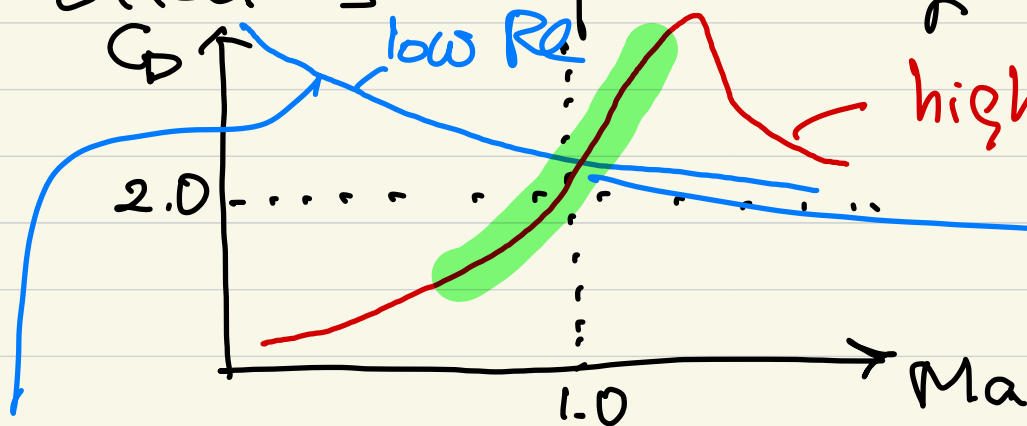
$$F_D = F_{D0} \cdot \left\{ 1 - 1.01619 \frac{k_g}{\sqrt{\pi} Kn} + \left(0.5 + 0.2349 \frac{k_g/p_s}{Kn+1} \right) \frac{k_g^2}{k_s^2} \right\} \quad \text{: theoretical Sone & Aoki (1977)}$$

$\frac{\sqrt{\pi}}{2} Kn$

k_g/p_s

(thermal conductivity of gas and solid)

* Effect of compressibility



due to the shock wave.
(form drag)

No peaks due to the rarefied flow effect.

$$C_D = 2 + (C_{D0} - 2) \exp\left(-\frac{3.07\sqrt{R} \zeta(Re) \cdot Ma}{Re}\right) + \frac{h(Ma)}{\sqrt{R} Ma} \exp\left(-\frac{Re}{2Ma}\right).$$

Crowe et al (1972)

$$\zeta(Re) = \frac{1 + Re(12.278 + 0.54Re)}{1 + 11.278Re} \text{ Hermansen (1979)}$$

$$h(Ma) = \frac{5.6}{1+Ma} + 1.7 \sqrt{\frac{T_d}{T_c}}$$

temperature of particle and gas.

Used for solid propellant rocket nozzle.

② Blowing effect. (phase change)

- surface of a burning (combustion) or evaporating droplet tends to reduce the C_D .

$$C_D = \frac{C_{D0} \leftarrow}{1+B} \quad (\text{Eisenklam et al. 1967})$$

B - transfer number.

- for an evaporating droplet

$$B = \frac{q \Delta T}{h_L}$$

h_L latent heat for evaporation.

- for a burning droplet (combustion)

$$B = \frac{q \Delta T + \alpha_{O_2} H / s}{h_L}$$

α_{O_2} oxygen concentration

H/s heat of combustion
stochastic rate for oxygen.

Yuen & Chen (1976) suggested that the blowing effects can be accounted for adjusting the Re based on the viscosity change and film temperature ($1/3$ rule).

$$T_f = T_d + \frac{1}{3} (T_c - T_d) \Rightarrow \text{determine } \mu.$$

$$\rightarrow \underline{Re} = \rho D U_r / \mu \rightarrow C_D = f(Re).$$

③ Thermophoresis (due to temperature gradient) in particular, for aerosol.

temperature is a measure of fluctuating KE of fluid molecule. \rightarrow more momentum exchange

→ force along the decreasing T. mean free path

- for a small particle ($D \ll \lambda$)

$$F_{th} = - \frac{\rho_c \lambda D^2 \Delta T_c}{T_d} \rightarrow V_{th} = - \frac{0.55 \mu_c}{\rho_c T_d} \Delta T_c$$

- for a large particle ($D > \lambda$)

temperature gradient is established within the particle depending on the conductivity of the particle.

$$F_{th} = - \frac{9 \pi D^2 \mu_c^2 H \Delta T_c}{2 \rho_c T_d} \rightarrow H = \left(\frac{1}{1 + 6.75 D} \right) \left(\frac{\frac{\rho_c}{\rho_d} + 4.4 D}{1 + 2 \frac{\rho_c}{\rho_d} + 0.8 D} \right)$$

$\sim \frac{\rho_c / \rho_d}{1}$ (thermal conductivity ratio)

$$\approx - \frac{9 \pi D^2 \mu_c^2 \Delta T_c}{2 \rho_c T_d} \cdot \frac{\rho_c}{\rho_d}$$

$$V_{th} = - \frac{3C_c \mu H \Delta T_c}{2C_c T_d} \text{ gas slip correction factor.}$$

④ Photophoresis (due to the gradient in the incident radiation)

- similar to the thermophoresis
- particle motion under the influence of asymmetric light (photons) absorption (momentum exchange) in a radiation beam.

⑤ Bjerknes force (due to the pressure gradient
↳ sound wave in a liquid)

- for a gas bubble in an acoustic pressure field, it can undergo volume pulsation.

· for non-zero ΔP , it can couple with the bubble oscillations to produce a translational force.

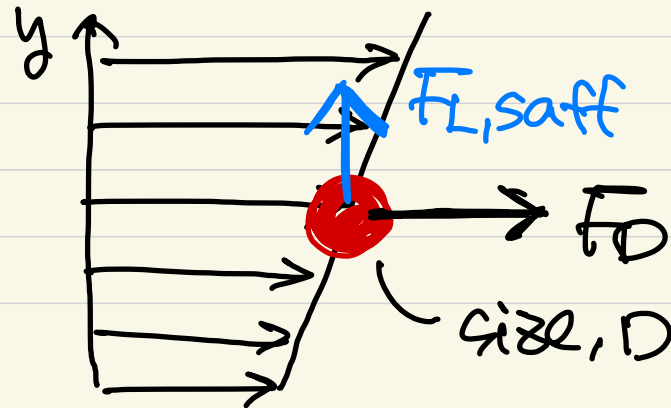
→ Primary Bjerknes force (1909)

$$F_{Bj} = -V_d \cdot \nabla P(r, t)$$

⑥ Lift force. (perpendicular to the major movement direction)

- shear-induced lift force (Saffman, 1965)

(uniform shear flow of an unbounded viscous fluid)



$$u = u(y)$$

$$F_{L, \text{saff}} = 1.61 \mu_c D |\bar{u}_r| \sqrt{Re_G}$$

$\bar{u}_r = \bar{u} - \bar{v}$: relative velocity

$$Re_G \equiv \frac{D^2}{2\nu_c} \left[\frac{du}{dy} \right]$$

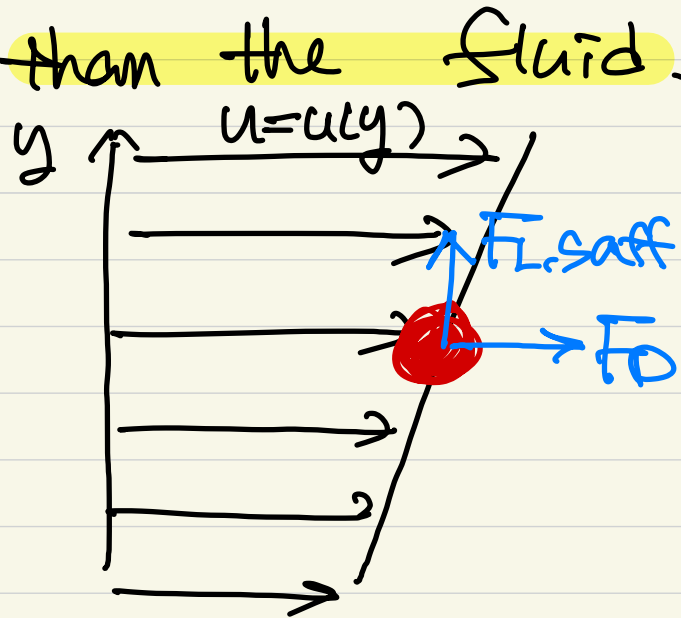
$$F_{L, \text{saff}} = 1.61 D^2 \sqrt{\mu_c \rho_c} |\bar{u}_r| \left| \frac{du}{dy} \right|^{0.5} \quad (\text{Shear Reynolds number})$$

$$= 1.61 D^2 \sqrt{\mu_c \rho_c} \cdot \frac{\bar{u}_r \times \bar{\omega}_c}{\sqrt{|\bar{\omega}_c|}}, \quad \bar{\omega}_c = \nabla \times \bar{u}$$

$u_r > 0$: $F_{L, \text{saff}}$ acts to the higher velocity region
 $u_r < 0$: " " " " lower " "

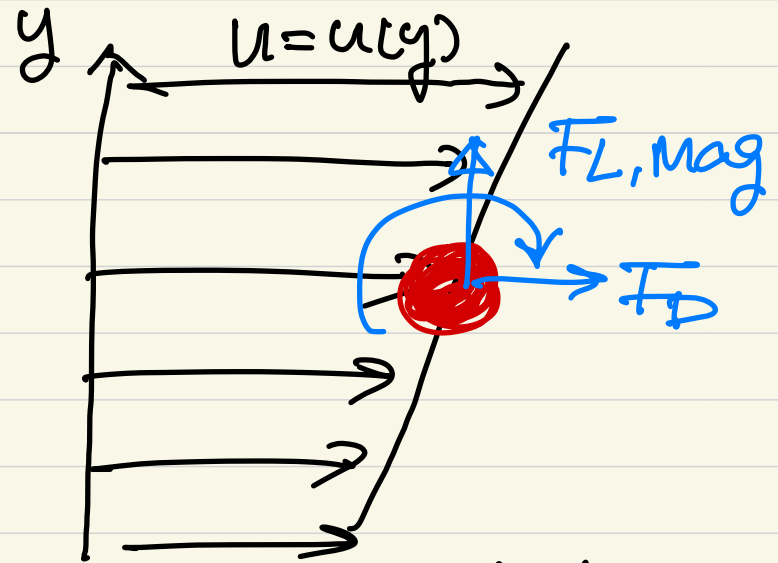
Saffman lift is valid when,
 $Re_r = u_r D / \nu \ll 1$, $Re_G \ll 1$, and $Re_r \ll Re_G$.

- Aerodynamic lift : the magnus effect appears when a particle is moving in a fluid w/ a rotation, caused by external sources other than the fluid.



particle rotation = fluid rotation.

$$\bar{\omega}_d = \bar{\omega}_c = \frac{1}{2}(\nabla \times \bar{u})$$



particle rotation \neq fluid rotation

$$\bar{\omega}_d \neq \bar{\omega}_c \left(= \frac{1}{2} \nabla \times \bar{u} \right)$$

relative vorticity

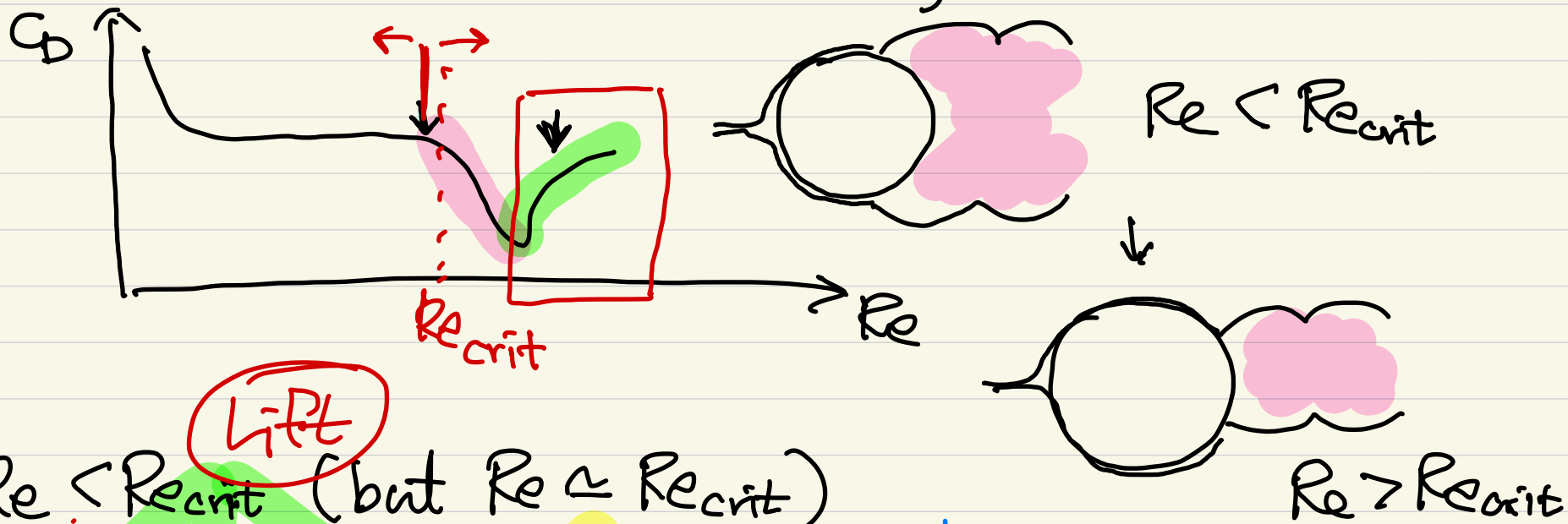
$$\bar{\omega}_r = \bar{\omega}_d - \bar{\omega}_c$$

- For low Reynolds number.

$$F_{L, \text{Mag}} = \frac{\pi}{8} D^3 \rho_c (\bar{u}_r \times \bar{\omega}_r),$$

$$\bar{\omega}_r = \bar{\omega}_d - \frac{1}{2} (\nabla \times \bar{u})$$

- For higher Re's, the asymmetric separation of the boundary layer ^(0.15-1.5) is responsible for the magnus effect. \rightarrow depends on the critical Reynolds number, Re_{crit} (drag crisis)



if $Re < Re_{crit}$ (but $Re \approx Re_{crit}$) Lift

$U_{\infty} - \Omega R$ separates earlier

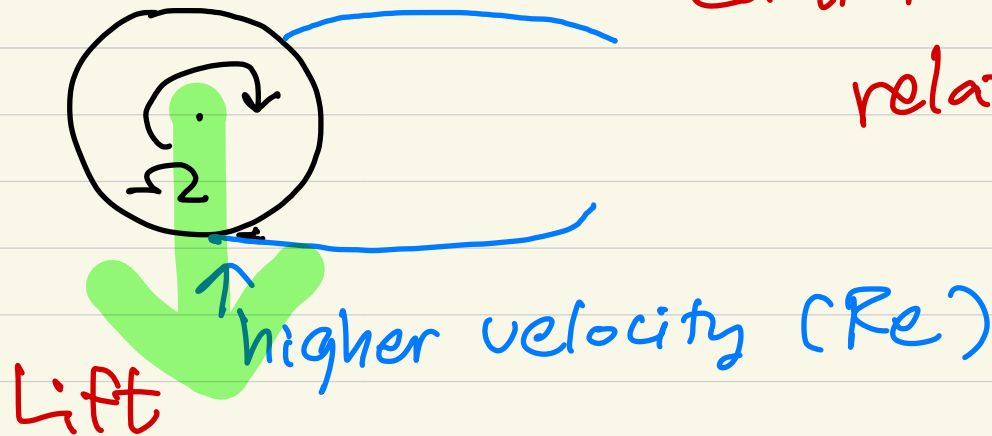
$U_{\infty} + \Omega R$ separation delayed.

U_{∞}

The diagram shows a sphere in a flow field with velocity U_{∞} indicated by vertical arrows on the left. The sphere rotates with angular velocity Ω , shown by a green arrow. The velocity at the top of the sphere is $U_{\infty} - \Omega R$, and at the bottom it is $U_{\infty} + \Omega R$. Blue arrows indicate that the flow separates earlier at the top and is delayed at the bottom.

if $Re > Re_{crit}$ (supercritical Re) \rightarrow sep. points move upstream as $Re \uparrow$.

∴ both upper and lower sides have a turb. separation. \rightarrow lower side separates earlier w/ higher relative velocity



$$u = \bar{u} + u'$$

① Turbulence effect

· relative turbulence intensity. $I_r \equiv \frac{u_{rms}}{|\bar{u} - \bar{v}|}$

· turb. length scale - particle size ratio

: drag coefficient increases w/ increasing the ratio at a given Re .

* To apply the eq. of particle motion in this chapter to the particle in a turb. flow, different scales in turbulence should be considered.

- Large scale : L_I (integral length scale)
 u' (velocity fluctuation)

- Small scale : η_k and v_k (Kolmogorov vel.)
(Kolmogorov length)

We will be back later on this!

$$v_k / u' \sim O(Re_\eta^{-1/2}), \quad \eta_k / L_I \sim O(Re_\eta^{-3/2})$$

Re_g : Re based on the Taylor (microscale)
particle size (inertia subrange)

- To apply the eq. we need $D/\eta_k \ll 1$, since η_k is the smallest scale at which the velocity is not uniform.
- Turbulence diffusion is dominated by large-scale "stirring", so the velocities of the order of u' over a long time lead to the statistically asymptotic behavior.
- For the velocity distribution, small-scale motion has to be considered to understand the